

A typical literature example of what not to do

First issue: correlations

Second issue: perihelion advances are not measured

Third issue: it is more complicated than just the dynamics

ARTEMIS (OCA)



Testing theories of gravity with planetary ephemerides

Olivier Minazzoli & Agnès Fienga

ARTEMIS
GéoAzur

Septembre 2025



OBSERVATOIRE
DE LA CÔTE D'AZUR



UNIVERSITÉ CÔTE D'AZUR

More in our review: Fienga and Minazzoli [2024]

SPRINGER NATURE Link

[Account](#)
[Find a journal](#)
[Publish with us](#)
[Track your research](#)
[Search](#)
[Cart](#)
[Home](#) > [Living Reviews in Relativity](#) > Article

Testing theories of gravity with planetary ephemerides

 Review Article | [Open access](#) | Published: 29 January 2024

 Volume 27, article number 1, (2024) | [Cite this article](#)
[Download PDF](#)

 You have full access to this [open access](#) article

[Living Reviews in Relativity](#)
[Aims and scope](#)

Agnès Fienga & Olivier Minazzoli

 8574 Accesses 29 Citations 6 Altmetric [Explore all metrics](#)

Abstract

We describe here how planetary ephemerides are built in the framework of General Relativity and how they can be used to test alternative theories. We focus on the definition of the reference frame (space and time) in which the planetary ephemeris is described, the equations of motion that govern the orbits of solar system bodies and electromagnetic waves. After a review on the existing planetary and lunar ephemerides, we summarize the results obtained considering full modifications of the ephemeris framework with direct comparisons with the observations of planetary systems, with a specific attention for the PPN formalism. We then discuss other formalisms such as Einstein–dilatation theories, the massless graviton and MOND. The paper finally concludes on some comments and recommendations regarding misinterpreted measurements of the advance of perihelia.

[Use our pre-submission checklist](#)

Avoid common mistakes on your manuscript.


 Part of a collection:
[Experimental Foundations of Gravitation](#)
[Sections](#)
[Figures](#)
[References](#)
[Abstract](#)
[Introduction](#)
[Basic concepts behind planetary ephemerides](#)
[Planetary and lunar ephemerides in general relat...](#)
[Tests of alternative theoretical frameworks with ...](#)
[Inconsistent tests with ephemeris outputs](#)
[Future directions](#)

Table of Contents

1. A typical literature example
of what not to do

2. First issue: correlations

3. Second issue: perihelion
advances are not measured

4. Third issue: it is more
complicated than just the
dynamics

Common issue in the literature

Example with a Yukawa suppression of gravity

Modified Newtonian potential:

$$\begin{aligned} U &= \sum_a \frac{Gm_a}{|\mathbf{x} - \mathbf{x}_a|} e^{-|\mathbf{x} - \mathbf{x}_a|/\lambda_g} \\ &= \sum_a \frac{Gm_a}{|\mathbf{x} - \mathbf{x}_a|} + \frac{1}{2} \sum_a \frac{Gm_a}{\lambda_g^2} |\mathbf{x} - \mathbf{x}_a| + O(Gmr^2/\lambda_g^3), \end{aligned}$$

Modified equation of motion:

$$\frac{d\mathbf{v}}{dt} = -\frac{Gm}{r^2} \left(1 - \frac{1}{2} \frac{r^2}{\lambda_g^2} \right),$$

Perihelion advance per orbit (assuming 2 bodies only):

$$\Delta\varpi = \pi \left(\frac{a}{\lambda_g} \right)^2 (1 - e^2)^{-1/2},$$

Common issue in the literature

Example with a Yukawa suppression of gravity

Perihelion advance per orbit (assuming 2 bodies only):

$$\Delta\varpi = \pi \left(\frac{a}{\lambda_g} \right)^2 (1 - e^2)^{-1/2},$$

Deduction of λ_g from uncertainties:

Table 1. Bounds on λ_g .

Planet	$\sigma(\dot{\varpi})$ (mas yr ⁻¹)	λ_g bound (10 ¹⁴ km)
<i>Data from table 4 of [16]</i>		
Mercury	0.03	0.18
Venus	0.016	0.28
Earth	0.0019	0.88
Mars	0.00037	2.21
Jupiter	0.28	0.11
Saturn	0.0047	0.98
<i>Data from [17]</i>		
Mercury	0.02	0.22
Saturn	0.026	0.42

Table of Contents

1. A typical literature example of what not to do
2. First issue: correlations
3. Second issue: perihelion advances are not measured
4. Third issue: it is more complicated than just the dynamics

Common issue in the literature

Example with a Yukawa suppression of gravity

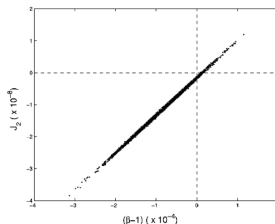
Correlations in INPOP17b:

	λ_g	a Mercury	a Mars	a Saturn	a Venus	a EMB	GM_\odot
λ_g	1	0.50	0.49	0.04	0.39	0.05	0.66
a Mercure	...	1	0.21	0.001	0.97	0.82	0.96
a Mars	1	0.03	0.29	0.53	0.06
a Saturn	1	0.003	0.02	0.01
a Venus	1	0.86	0.94
a EMB	1	0.73
GM_\odot	1

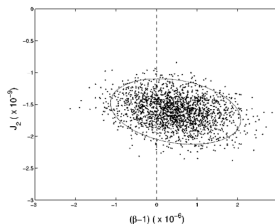
Bernus et al. [2019]

~ 400 parameters to adjust!

Another example of correlation: between J_2 and β



Model A



Model B

Milani et al. [2002]

A typical literature example of what not to do

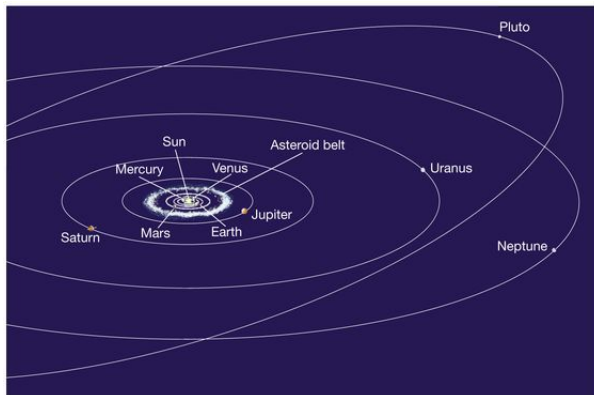
First issue: correlations

Second issue: perihelion advances are not measured

Third issue: it is more complicated than just the dynamics

Correlations: why?

All motions $\in \sim$ ecliptic plane & \sim small eccentricities



Correlations: why?

Spread in methods, data types and accuracies

Planet/type [unit]	s/c or method	Period	Averaged accuracy
<i>Mercury</i>			
Direct range [m]	Surface	1971.29: 1997.60	900
Radio science range [m]	Messenger	2011.23: 2014.26	5
Navigation range [m]	Mariner	1974.24: 1976.21	100
<i>Mercury</i>			
VLBI [mas]	Magellan, Venus Express	1990.70: 2013.14	2.0
Direct range [m]	Surface	1965.96: 1990.07	1400
Navigation range [m]	Venus Express	2006.32: 2011.45	7.0
<i>Mars</i>			
VLBI [mas]	MGs, MRO	1989.13: 2013.86	0.3
Navigation range [m]	Mars Express	2005.17: 2019.37	2.0
Radio Science range [m]	MGs	1999.31: 2006.70	2.0
Radio Science range [m]	MRO/MO	2002.14: 2014.00	1.2
Navigation range [m]	Viking	1976.55: 1982.87	20.0
<i>Jupiter</i>			
VLBI [mas]	Galileo	1996.54: 1997.94	11
	Juno	2016:2020	0.5
Optical RA/Dec [arcsec]	Transit+CCD	1924.34: 2008.49	0.3
Flyby RA/Dec [mas]	U,C, P, V	1974.92: 2001.00	8.0
Flyby range [m]	U,C, P, V	1974.92: 2001.00	2000
Radio science range [m]	Juno	2016.65: 2020.56	20
<i>Saturn</i>			
Optical RA/Dec [arcsec]	Transit+CCD	1924.22: 2008.34	0.3
VLBI RA/Dec [mas]	Cassini	2004.69: 2017.9	0.6
JPL H14 [m]	Cassini	2004.41: 2014.38	25.0
Navigation [m]	Cassini	2006.01: 2009.83	6.0
Radio science: Titan flybys [m]	Cassini	2006.01: 2016.61	15.0
Radio science: grand finale) [m]	Cassini	2017.35: 2017.55	1.0

Correlations: why?

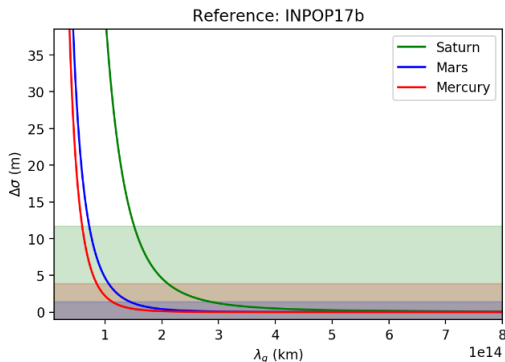
Temporal spread: accurate observations are rarely simultaneous

Planet/type [unit]	s/c or method	Period	Averaged accuracy
<i>Mercury</i>			
Direct range [m]	Surface	1971.29: 1997.60	900
Radio science range [m]	Messenger	2011.23: 2014.26	5
Navigation range [m]	Mariner	1974.24: 1976.21	100
<i>Mercury</i>			
VLBI [mas]	Magellan, Venus Express	1990.70: 2013.14	2.0
Direct range [m]	Surface	1965.96: 1990.07	1400
Navigation range [m]	Venus Express	2006.32: 2011.45	7.0
<i>Mars</i>			
VLBI [mas]	MGs, MRO	1989.13: 2013.86	0.3
Navigation range [m]	Mars Express	2005.17: 2019.37	2.0
Radio Science range [m]	MGs	1999.31: 2006.70	2.0
Radio Science range [m]	MRO/MO	2002.14: 2014.00	1.2
Navigation range [m]	Viking	1976.55: 1982.87	20.0
<i>Jupiter</i>			
VLBI [mas]	Galileo	1996.54: 1997.94	11
	Juno	2016:2020	0.5
Optical RA/Dec [arcsec]	Transit+CCD	1924.34: 2008.49	0.3
Flyby RA/Dec [mas]	U,C, P, V	1974.92: 2001.00	8.0
Flyby range [m]	U,C, P, V	1974.92: 2001.00	2000
Radio science range [m]	Juno	2016.65: 2020.56	20
<i>Saturn</i>			
Optical RA/Dec [arcsec]	Transit+CCD	1924.22: 2008.34	0.3
VLBI RA/Dec [mas]	Cassini	2004.69: 2017.9	0.6
JPL H14 [m]	Cassini	2004.41: 2014.38	25.0
Navigation [m]	Cassini	2006.01: 2009.83	6.0
Radio science: Titan flybys [m]	Cassini	2006.01: 2016.61	15.0
Radio science: grand finale) [m]	Cassini	2017.35: 2017.55	1.0

Effect of correlations

Example with a Yukawa suppression of gravity

Standard deviation without readjusting model parameters:

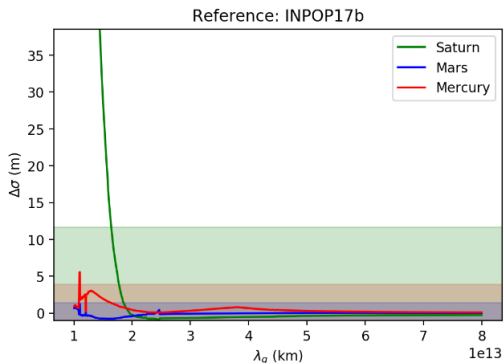


Bernus et al. [2019]

Effect of correlations

Example with a Yukawa suppression of gravity

Standard deviation after readjusting model parameters:



Bernus et al. [2019]

Effect of correlations

Example with a Yukawa suppression of gravity

Correlations in INPOP17b:

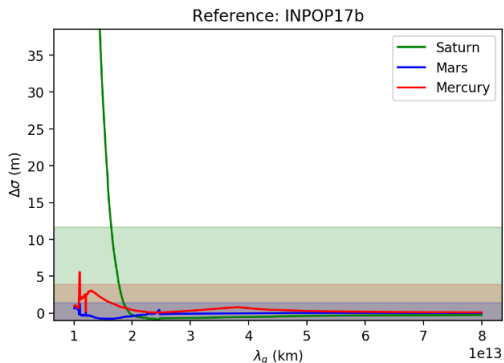
	λ_g	a Mercury	a Mars	a Saturn	a Venus	a EMB	GM_\odot
λ_g	1	0.50	0.49	0.04	0.39	0.05	0.66
a Mercure	...	1	0.21	0.001	0.97	0.82	0.96
a Mars	1	0.03	0.29	0.53	0.06
a Saturn	1	0.003	0.02	0.01
a Venus	1	0.86	0.94
a EMB	1	0.73
GM_\odot	1

Bernus et al. [2019]
 ~ 400 parameters to adjust!

Effect of correlations

Example with a Yukawa suppression of gravity

Standard deviation after readjusting model parameters:

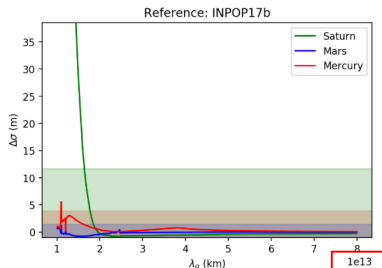
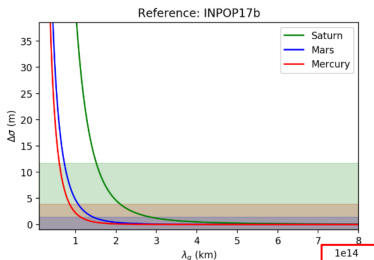


Bernus et al. [2019]

Effect of correlations

Example with a Yukawa suppression of gravity

Absorption of the effect of λ_g during the fit:

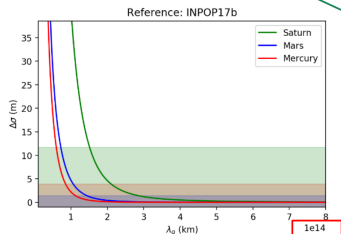


Bernus et al. [2019]

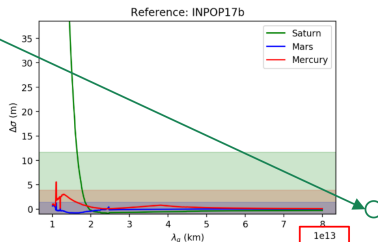
Effect of correlations

Example with a Yukawa suppression of gravity

Absorption of the effect of λ_g during the fit:



Without fit



With fit

Bernus et al. [2019]

Table of Contents

1. A typical literature example of what not to do
2. First issue: correlations
3. Second issue: perihelion advances are not measured
4. Third issue: it is more complicated than just the dynamics

Perihelion advances are not measured

Table 1. Bounds on λ_g .

Planet	$\sigma(\dot{\varpi})$ (mas yr ⁻¹)	λ_g bound (10 ¹⁴ km)
<i>Data from table 4 of [16]</i>		
Mercury	0.03	0.18
Venus	0.016	0.28
Earth	0.0019	0.88
Mars	0.000 37	2.21
Jupiter	0.28	0.11
Saturn	0.0047	0.98
<i>Data from [17]</i>		
Mercury	0.02	0.22
Saturn	0.026	0.42

A typical literature example of what not to do

First issue: correlations

Second issue: perihelion advances are not measured

Third issue: it is more complicated than just the dynamics

Table of Contents

1. A typical literature example of what not to do
2. First issue: correlations
3. Second issue: perihelion advances are not measured
4. Third issue: it is more complicated than just the dynamics

The whole picture

The post-Newtonian framework is not only about the equations of motion!

- Dynamics of massive bodies
- Conserved quantities from Lagrangian or Hamiltonian of motion
- Shapiro effect on electromagnetic wave propagation
- Production of coordinate time for synchronisation between all clocks

Conclusion: The whole picture

Beware of correlations between parameters!

- Parameters must be adjusted in a novel theoretical framework
- Perihelion advances are deduced from the ephemerides (i.e., they are model-dependent).

The post-Newtonian framework is not only about the equations of motion!

- Dynamics of massive bodies
- Conserved quantities from Hamiltonian of motion
- Shapiro effect on electromagnetic wave propagation
- Production of coordinate time for synchronisation between all clocks

References I

- L. Bernus, O. Minazzoli, A. Fienga, M. Gastineau, J. Laskar, and P. Deram. Constraining the Mass of the Graviton with the Planetary Ephemeris INPOP. *Phys. Rev. Lett.*, 123(16):161103, October 2019. doi: 10.1103/PhysRevLett.123.161103.
- Agnès Fienga and Olivier Minazzoli. Testing theories of gravity with planetary ephemerides. *Living Reviews in Relativity*, 27(1):1, January 2024. doi: 10.1007/s41114-023-00047-0.
- Andrea Milani, David Vokrouhlický, Daniela Villani, Claudio Bonanno, and Alessandro Rossi. Testing general relativity with the BepiColombo radio science experiment. *Phys. Rev. D*, 66(8):082001, October 2002. doi: 10.1103/PhysRevD.66.082001.

Accuracy of ephemerides

Ephemerides	Gaillot (Gaillot and Le Verrier 1913)		DE102 (Standish 1983)		DE440/INPOP19a (Park et al. 2021; Fienga et al. 2019)		GR
Data span	1800–1913		1913–1983		1924–2021		$\Delta\dot{\varpi}$ "/yr ⁻¹
	Angle "	Distance Earth- km	Angle "	Distance Earth- km	Angle "	Distance Earth- km	
Mercury	1	450	0.050	5	0.002	0.004	0.43
Venus	0.5	100	0.050	2	0.002	0.006	0.14
Mars	0.5	150	0.050	0.050	0.001	0.0015	0.065
Jupiter	0.5	1400	0.1	10	0.010	0.020	0.019
Saturn	0.5	3000	0.1	600	0.001	0.020	0.010
Uranus	1	12,700	0.2	2540	0.050	10	0.005
Neptune	1	22,000	0.2	4400	0.050	50	0.0033
Pluto	1	24,000	0.2	4800	0.050	2400	0.0027

The whole picture

Dynamics: coordinate acceleration of massive bodies

$$\begin{aligned}
 \mathbf{a}_T = & - \sum_{A \neq T} \frac{\mu_A}{r_{AT}^3} \mathbf{r}_{AT} \\
 & - \sum_{A \neq T} \frac{\mu_A}{r_{AT}^3 c^2} \mathbf{r}_{AT} \left\{ v_T^2 + 2v_A^2 - 4\mathbf{v}_A \cdot \mathbf{v}_T - \frac{3}{2} \left(\frac{\mathbf{r}_{AT} \cdot \mathbf{v}_A}{r_{AT}} \right)^2 \right. \\
 & \left. - \frac{1}{2} \mathbf{r}_{AT} \cdot \mathbf{a}_A - 4 \sum_{B \neq T} \frac{\mu_B}{r_{TB}} - \sum_{B \neq A} \frac{\mu_B}{r_{AB}} \right\} \\
 & + \sum_{A \neq T} \frac{\mu_A}{c^2 r_{AT}^3} [4\mathbf{r}_{AT} \cdot \mathbf{v}_T - 3\mathbf{r}_{AT} \cdot \mathbf{v}_A] (\mathbf{v}_T - \mathbf{v}_A) \\
 & + \frac{7}{2} \sum_{A \neq T} \frac{\mu_A}{c^2 r_{AT}} \mathbf{a}_A,
 \end{aligned}$$

The whole picture

Dynamics: Lagrangian, Hamiltonian and first integrals

$$\begin{aligned}
 L_B = & -m_B c^2 + m_B \frac{v_B^2}{2} + \sum_{A \neq B} \frac{G m_A m_B}{r_{AB}} + \frac{m_B v_B^4}{8c^2} \\
 & + \frac{1}{c^2} \sum_{A \neq B} \frac{G m_A m_B}{r_{AB}} \left[-4 \mathbf{v}_A \cdot \mathbf{v}_B + \frac{3}{2} v_B^2 + 2 v_A^2 - \frac{(\mathbf{v}_A \cdot \mathbf{r}_{AB})^2}{2 r_{AB}^2} - \frac{\mathbf{r}_{AB} \cdot \mathbf{a}_A}{2} \right] \\
 & - \frac{1}{c^2} \sum_{A \neq B} \frac{G m_A m_B}{r_{AB}} \left[\sum_{D \neq A} \frac{G m_D}{r_{AD}} + \sum_{D \neq B} \frac{1}{2} \frac{G m_D}{r_{DB}} \right] + \mathcal{O}(c^{-4}).
 \end{aligned}$$

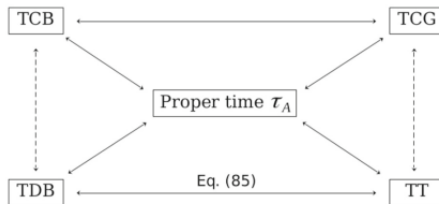
The whole picture

Dynamics: light rays and the Shapiro delay

$$c(t_r - t_e)_{GRT} = R + \sum_A 2 \frac{\mu_A}{c^2} \ln \frac{\mathbf{n} \cdot \mathbf{r}_{rA} + r_{rA} + \frac{4\mu_A}{c^2}}{\mathbf{n} \cdot \mathbf{r}_{eA} + r_{eA} + \frac{4\mu_A}{c^2}},$$

The whole picture

Time



The whole picture

Time

The difference between TT and TDB is produced by planetary ephemerides, by integrating the following equation together with the equations of motion (Klioner 2008; Fienga et al. 2009)

$$\frac{d(TT - TDB)}{d(TDB)} = \left(L_B + \frac{1}{c^2} a \right) (1 + L_B - L_G) - L_G + \frac{1}{c^4} b, \quad (85)$$

with L_B and L_G are defining constants for TDB relatively to TCB and TT relatively to TCG, respectively (see e.g. Klioner 2008, Petit and Luzum 2010 for the full definition) and where

$$\begin{aligned} a &= -\frac{1}{2} v_T^2 - \sum_{A \neq T} \frac{\mu_A}{r_{AT}} \\ b &= -\frac{1}{8} v_T^4 + \frac{1}{2} \left[\sum_{A \neq T} \frac{\mu_A}{r_{AT}} \right]^2 + \sum_{A \neq T} \frac{\mu_A}{r_{AT}} \left\{ 4 \mathbf{v}_T \cdot \mathbf{v}_A - \frac{3}{2} v_T^2 - 2 v_A^2 \right. \\ &\quad \left. + \frac{1}{2} \mathbf{a}_A \cdot \mathbf{r}_{AT} + \frac{1}{2} \left(\frac{\mathbf{v}_A \cdot \mathbf{r}_{AT}}{r_{AT}} \right)^2 + \sum_{B \neq A} \frac{\mu_B}{r_{BA}} \right\}, \end{aligned} \quad (86)$$