### Colloque national CMB-France #6 **December 18, 2024**



### **Developing a Closed-Cycle Dilution Refrigerator for future CMB space missions** Focus on the Structural & Thermal Model

Valentin SAUVAGE Institut d'Astrophysique Spatiale, ORSAY, FRANCE



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### Developing a Closed-Cycle Dilution Refrigerator for future CMB space missions Focus on the Structural & Thermal Model



Valentin SAUVAGE

Bruno MAFFEI

Anaïs BESNARD

### Valentin SAUVAGE Institut d'Astrophysique Spatiale, ORSAY, FRANCE

Clémence DE JABRUN

Mehdi BOUZIT



# Context



### **REJECTED ONGOING ENDED**





# State of the art : the existing solutions



Planck Space Telescope [1]

### 100 mK **Open-Cycle Dilution Refrigerator**

[1] Triqueneaux et al. [2006] [2] Kelley *et al.* [2006] [3] Shirron *et al.* [2016] [4] Ezoe *et al.* [2019]





Suzaku Space Telescope [2], Hitomi Space Telescope [3], XRISM Space Telescope [4]

### 50 mK

### Adiabatic Demagnetization Refrigerator



# The Open-Cycle Dilution Refrigerator



Planck Space Telescope [1]

### 100 mK Open-Cycle Dilution Refrigerator

[1] Triqueneaux et al. [2006]





# **The Adiabatic Demagnetization Refrigerator**



Suzaku Space Telescope [1], Hitomi Space Telescope [2], XRISM Space Telescope [3]

### 50 mK

### Adiabatic Demagnetization Refrigerator

[1] Kelley *et al.* [2006] [2] Shirron *et al.* [2016] [3] Ezoe *et al.* [2019]



### A magnetic field is applied to a paramagnetic material



The magnetic field is slowly reduced

# State of the art : the existing solutions

Open Cycle Dilution Refrigerator

Provides 100 mK

Continuous temperature

Operates indefinitely

TRL 5 or more







IA Institut d'Astrophysiqu Orsay

Adiabatic Demagnetization Refrigerator







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# The limitations of the OCDR



Planck Space Telescope

Operation time: **2.5 years** Cooling power at 100 mK: 0.2 µW



LiteBIRD Space Telescope

Operation time: **3 years** Cooling power at 100 mK: 2 µW





<sup>3</sup>He: 12 000 liters STP

<sup>4</sup>He: 36 000 liters STP

<sup>3</sup>He: 63 000 liters STP

<sup>4</sup>He: 234 000 liters STP

Necessity of a closed-cycle that requires much less heliums

# **State of the art : the future solutions**









Adiabatic Demagnetization Refrigerator

Continuous Adiabatic Demagnetization Refrigerator









Duval et al. [2020]

LiteBIRD baseline





### What about the space CCDR?



### TRL 4

Component and/or breadboard functional verification in <u>a laboratory environment</u>

[1] Martin thesis [2009] [2] Chaudhry et al. [2012] [3] Volpe thesis [2014] [4] Sauvage et al. [2022], Sauvage thesis [2023]



Structural and Thermal Model [4]

. . . .

Development of an **Engineering Model in progress** 

### TRL 5

Component and/or breadboard critical function verification in <u>a relevant environment</u>

# **The Structural and Thermal Model**

Thermal aspects:

- Hosts the <sup>3</sup>He-<sup>4</sup>He dilution providing 2 μW of cooling power at 100 mK
- A heat sink at 1.7 K

Mechanical aspects:

- Supports a focal plane of 750 g on top of it
- Supports the vibrations of the launch (under 100 g), pushing the first mode above 140 Hz
- Limited size and mass (35 cm diameter, 25 cm height, 6 kg without the <sup>3</sup>He circulator)
- Holds the various sub-systems (capillaries, still, ...)

100 mK -





1.7 K

Support of the Planck HFI dilution

DM of Athena X-iFU (Institut Néel)



First design by IAS

Last design by IAS



# The struts

Purpose:

- Thermal insulation of the 100 mK stage from the 1.7 K stage
- Strong enough to withstand launch vibrations

Mechanical requirements:

- First vibration mode > 140 Hz (good stiffness)
- Choice of an isostatic structure

Thermal requirements:

$$\dot{Q} = \frac{S}{L} \int_{T_1}^{T_2} \kappa(T) dT$$

Strut sizing:

- Fixed length (limited by the requirements)
- Maximise IgZ/A (moment of inertia by surface area)



Carbon Fiber Reinforced Polymer

- Low thermal conductivity
- High resistance on tension/compression
- Lightweight



### **TOTAL: 7.8 μW from 1.7 K to 100 mK**

# The end fittings

- Avoid mounting stresses (no bending)
- Once tightened, it behaves like a fixed connection



End fittings have to be glued to CFRP (no data of the glue characteristics available at low temperature)

Inheritance of Planck: the glue have to work on compression to avoid breakage

Differential contraction tested a 77 K:

- CFRP contracts more than aluminium
- The end fittings are glued inside the CFRP tubes (Hysol 9395)







Hysol 9395 is pressure-injected to avoid air bubbles



# The thermal interfaces

Same thermal contraction to avoid differential deformations

Choice of Al6061-T6:

- Light and machinable
- Thermal isolation of the 100 mK cold plate (4 x 10<sup>-6</sup> W.m<sup>-1</sup>.K<sup>-1</sup>)
- Good thermal coupling at 1.7 K (4 W.m<sup>-1</sup>.K<sup>-1</sup>)







1.7 K stage











# **The Structural and Thermal Model**



 $\dot{Q}_{injected} = \dot{Q}_{struts} - \dot{Q}_{wires} \pm \dot{Q}_{heat switch}$ 

![](_page_16_Picture_3.jpeg)

![](_page_16_Picture_5.jpeg)

![](_page_16_Picture_6.jpeg)

![](_page_16_Figure_7.jpeg)

# **The Structural and Thermal Model**

![](_page_17_Figure_1.jpeg)

From 1.7 K to 100 mK • Predicted: 7.8 µW • Measured: 2.7 µW

**Thank you Kapitza resistance!** 

![](_page_17_Picture_5.jpeg)

We can do better! Addition of intermediate stages to intercept heat

![](_page_17_Picture_8.jpeg)

![](_page_17_Picture_9.jpeg)

![](_page_17_Figure_10.jpeg)

![](_page_17_Picture_11.jpeg)

# The Heat Exchanger Crowns

![](_page_18_Figure_1.jpeg)

![](_page_18_Picture_2.jpeg)

![](_page_18_Picture_3.jpeg)

Heat interceptor collars Designed to be repositioned HECs will be used to thermalize the electronic harnesses

![](_page_18_Picture_7.jpeg)

![](_page_18_Figure_8.jpeg)

### The small struts

- No structural function
- Flexible blades used as end fittings to have isostatic hexapodes
- Fewer parts than in Planck's design (used for main struts) —> easier mounting

![](_page_19_Picture_4.jpeg)

![](_page_19_Picture_5.jpeg)

### The Structural and Thermal Model

Reduced STM, from 1.7 K to 100 mK

- Predicted: 7.8 µW
- Measured: 2.7  $\mu$ W

### Full STM, from 1.7 K to 100 mK

- Predicted: 0.63 µW
- Measured: ..  $\mu W$

![](_page_20_Figure_7.jpeg)

### What's next?

### Next step (next January)

![](_page_21_Figure_2.jpeg)

![](_page_21_Picture_3.jpeg)

### **Design and integration**

Finalise the design to hold the free capillaries

Integration of the sub-systems (mixing chamber, capillaries, still, fountain pump, ...)

### Planning

May 2025: Validated STM **December 2025:** First still prototype June 2026: Final still version End of 2026: First EM of the CCDR

# Take home messages

- Accommodate future CMB missions requirements (e.g. LiteBIRD) but not only. CCDR could provides:
  - A continuous 100 mK (or 50 mK to be demonstrated)
  - A large cooling power (8  $\mu$ W at 100 mK and 1  $\mu$ W at 50 mK)
  - A compact and light system (<sup>3</sup>He circulator excluded)
  - A support for the focal plane
  - A thermalization for electronic harnesses
  - Compatible with any detector technology (e.g. no magnetic field)
- IAS is pushing the CCDR to TRL 5.
- We provide also properties on various materials

# See you next time

- Cryogenics in April 25'
- Low Temperature Detectors in June 25'
- Whenever you want at IAS

![](_page_22_Picture_14.jpeg)

![](_page_22_Picture_17.jpeg)

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![](_page_23_Picture_1.jpeg)

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![](_page_23_Picture_5.jpeg)

### The state of the art

![](_page_24_Figure_3.jpeg)

![](_page_24_Figure_4.jpeg)

![](_page_24_Picture_6.jpeg)

### **Principe of the dilution**

![](_page_25_Figure_1.jpeg)

![](_page_25_Picture_2.jpeg)

![](_page_25_Figure_3.jpeg)

# Principe of the fountain pump

![](_page_26_Figure_1.jpeg)

![](_page_26_Picture_2.jpeg)

# Principe of the fountain pump

![](_page_27_Figure_1.jpeg)

![](_page_27_Picture_2.jpeg)

![](_page_27_Picture_3.jpeg)