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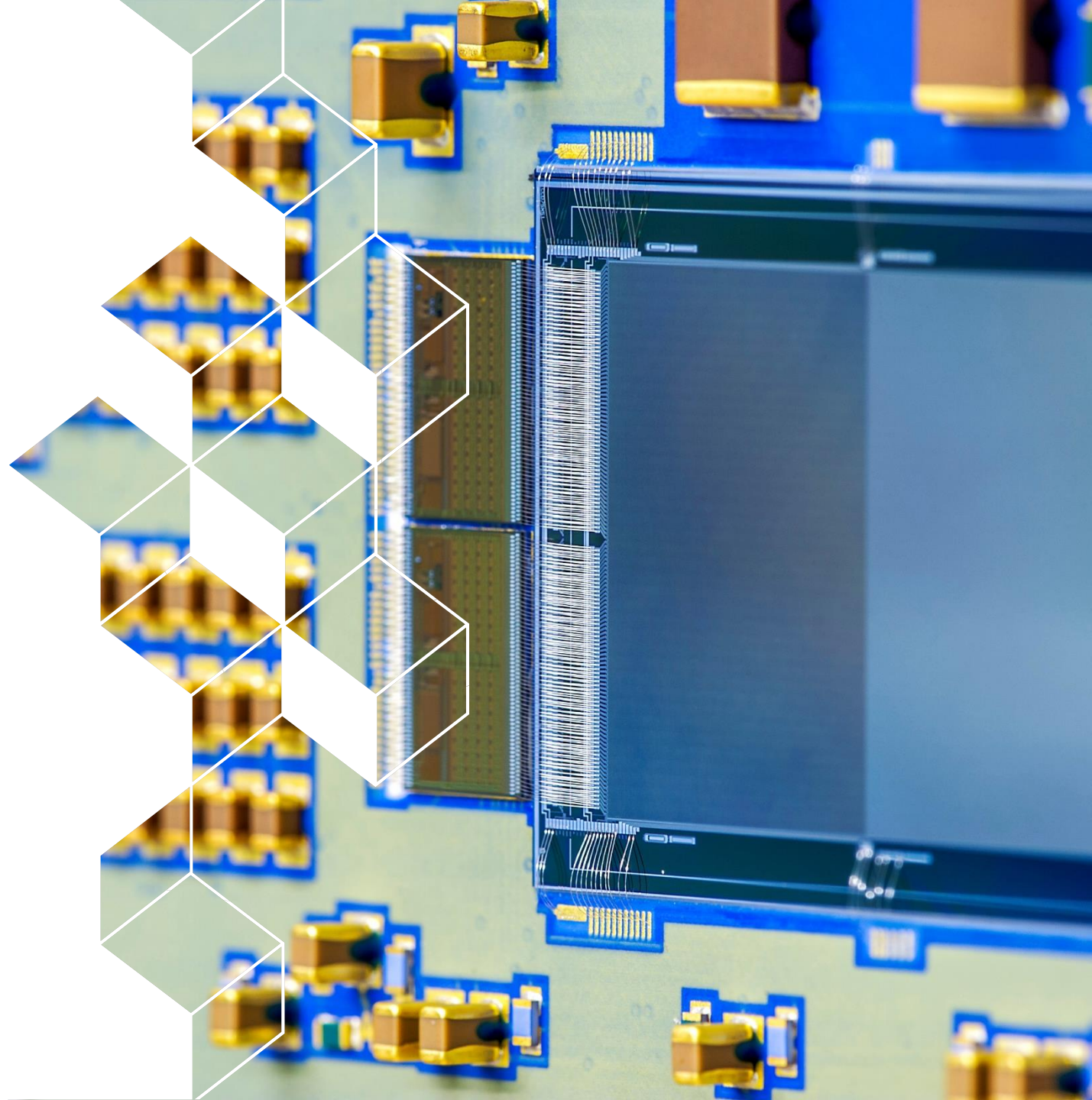


The development of the MXT camera of the SVOM space mission

From the first detector tests to the flight
instrument calibration

Aline Meuris

Journées thématiques IN2P3, 4 July 2023



Sommaire

1. SVOM, a gamma-ray burst mission

2. Design of the MXT camera, challenges and realizations

3. Spectral performance characterization of MXT





1 ■ The SVOM Mission

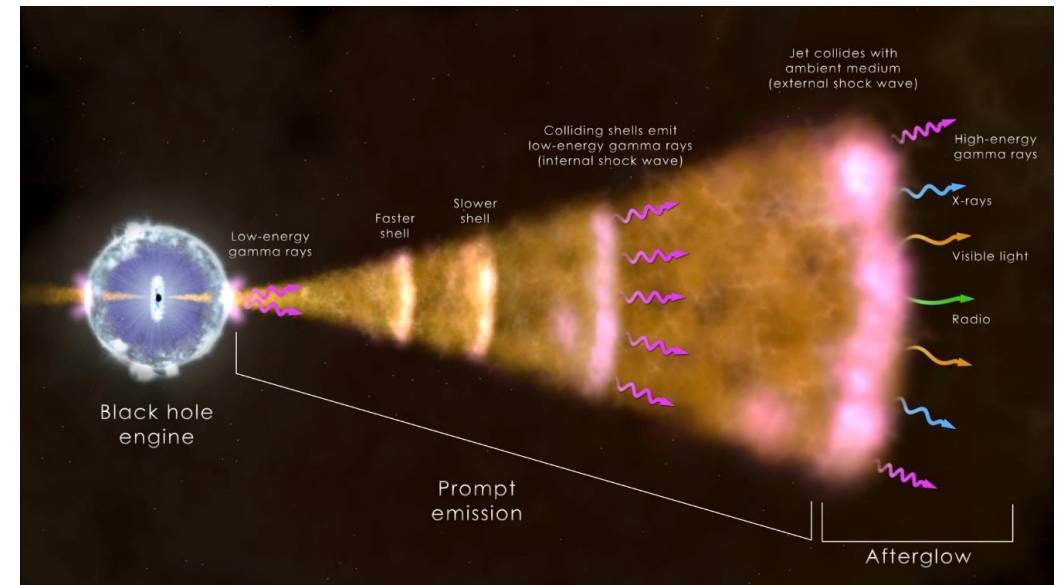
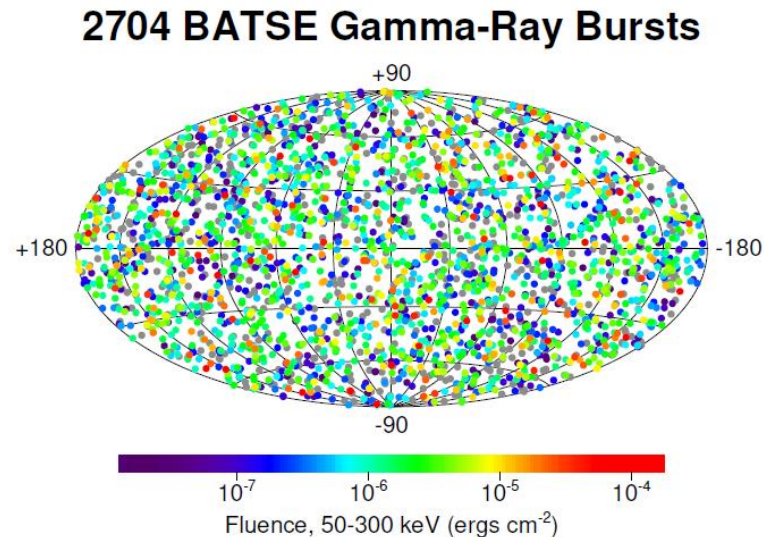
Gamma-ray bursts

GRB are **short and bright flashes of gamma-ray radiation** that appear **randomly** over the entire sky

Prompt emission: Gamma-rays (ms to min); internal of the central engine and jets

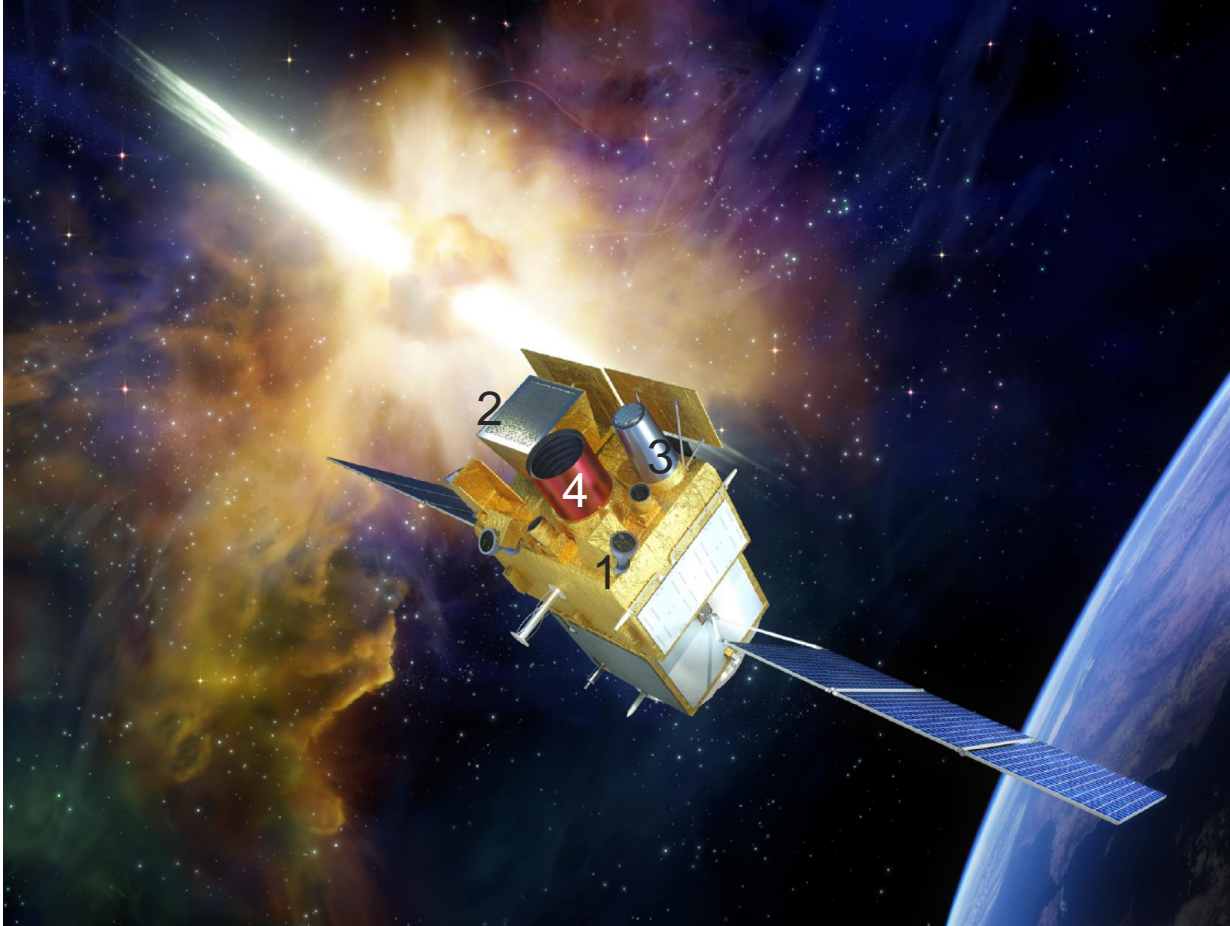
Afterglow emission: From X-rays to radio (days to months); external shocks with the environment

Motivations: physics in extreme conditions, multi-messenger astrophysics, early Universe probing



→ **Need for a new multi-wavelength observatory with real time data management**

The SVOM mission



LEO orbit, 930 kg - Slew in 3 min

Detection

1. **GRM**, prompt γ -ray emission (15 – 5000 keV)
2. **ECLAIRs**, GRB trigger (4-150 keV)

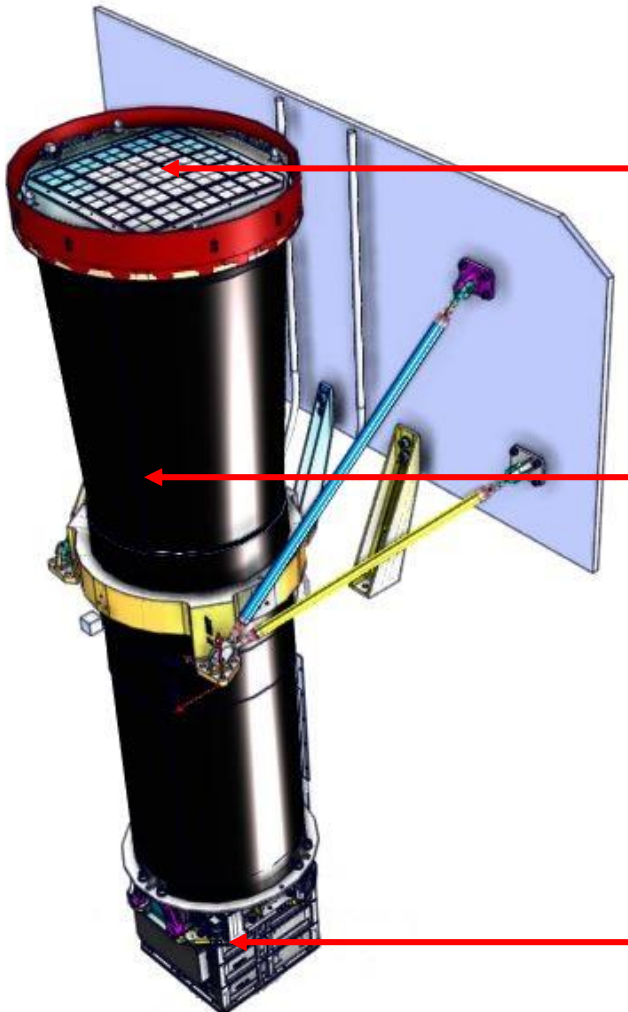
Localisation

3. **MXT**, X-ray afterglow (0,2-10 keV)
4. **VT**, Visible afterglow (400-1000 nm)

Ground follow-up

GWAC (500-850 nm)
C-GFT (400-950 nm)
F-GFT (400-1700 nm)

The Microchannel X-ray Telescope



Optics based on microchannel pore optics in lobster-eye configuration



Telescope structure and radiator



Camera



42 kg, 60 W

Data processing unit



Key performance features

Localization within 2 arcmin in less than 10 min for 80 % GRB

38 cm² effective area at 1.5 keV

Short term sensitivity (10s)

2×10^{-10} erg cm⁻² s⁻¹ (~10 mcrab)

Long term sensibility (10 ks)

4×10^{-12} erg cm⁻² s⁻¹ (150 μ Crab)

The camera at a glance



Technical specifications

9 kg, 7 W, 20 cm x 20 cm x H27 cm
Fits in the \varnothing 240 mm telescope tube

Z

Functions:

- X-ray detection and event pre-processing
- Detector active cooling
- Energy calibration
- Proton shielding

Performance requirements

0.2-10 keV dynamic range
80 eV at 1.5 keV beginning of life
100 ms event time resolution

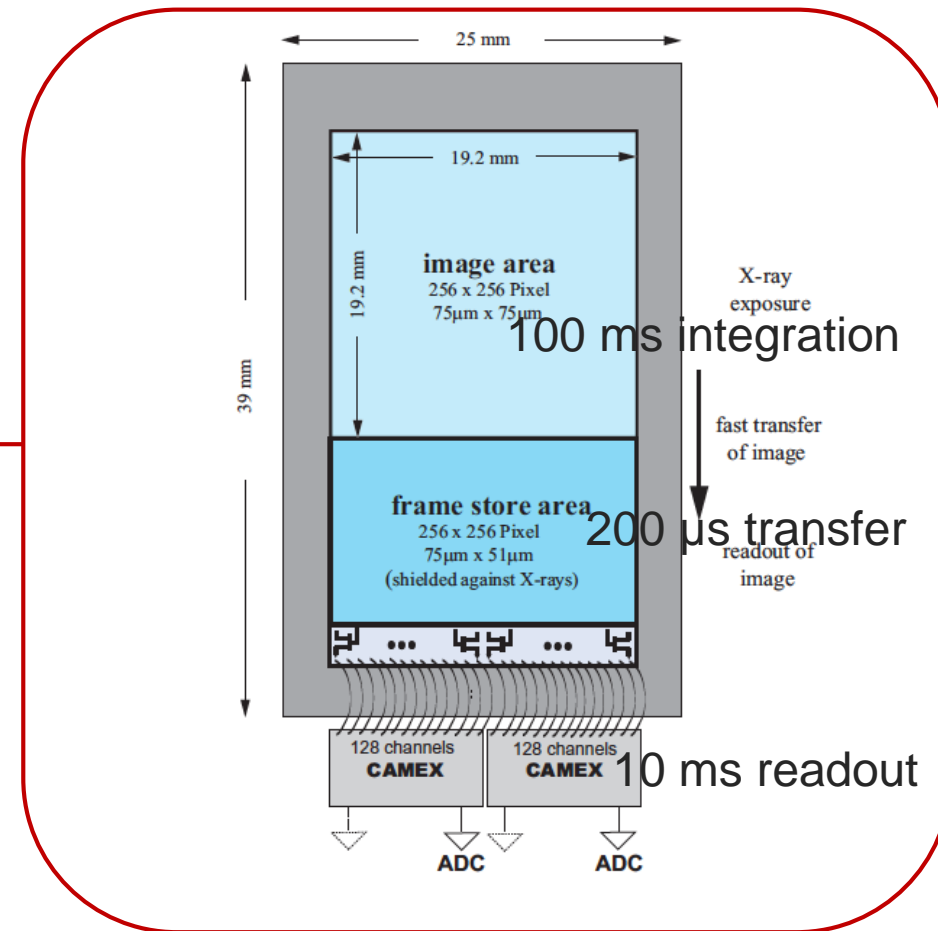
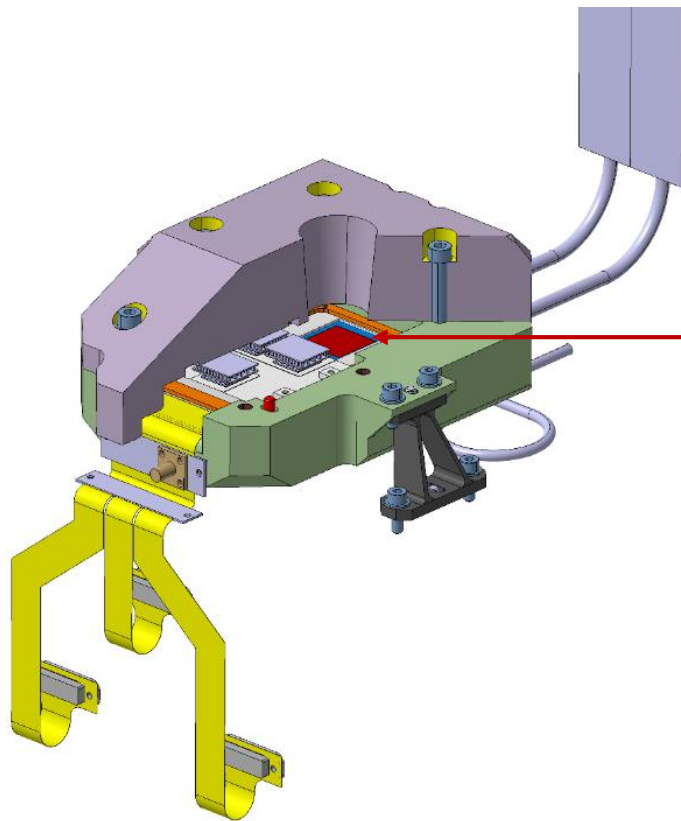
A. Meuris, NIM-A



2 ■ The Camera design and integration

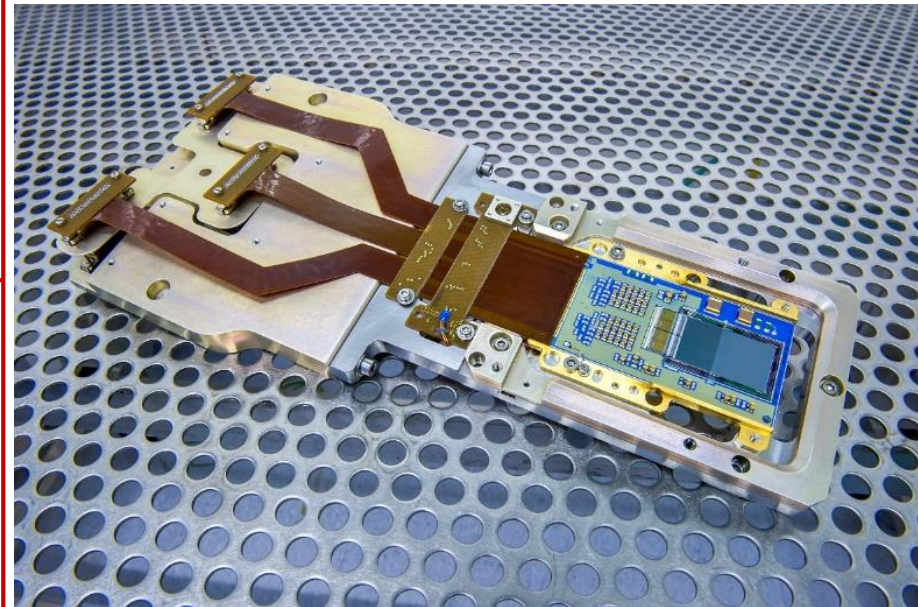
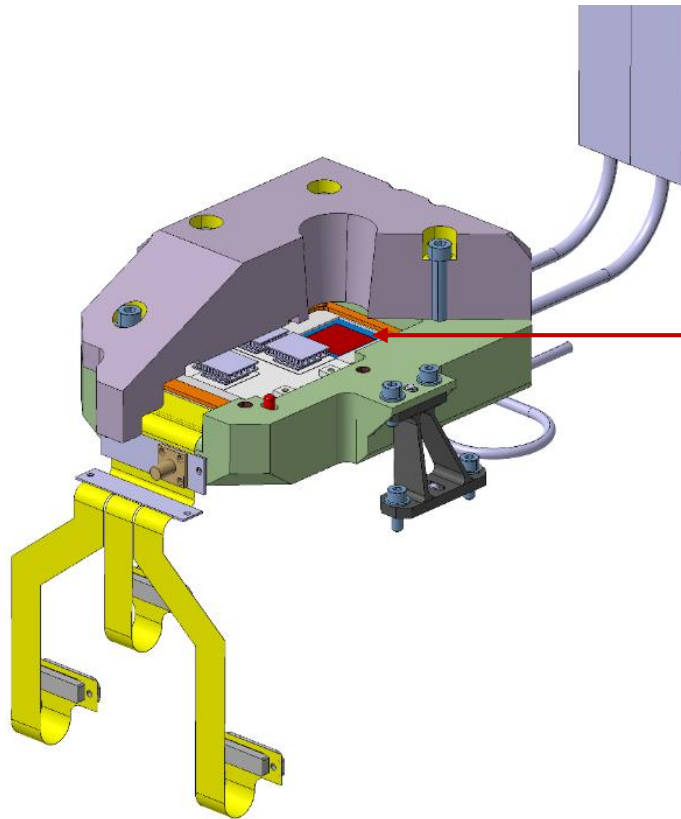
Focal plane assembly: detector assembly

- **Goal:** imaging spectroscopy from 0.2 to 10 keV with pnCCD and CAMEX ASIC provided by MPE.
- **Challenges:** space qualification

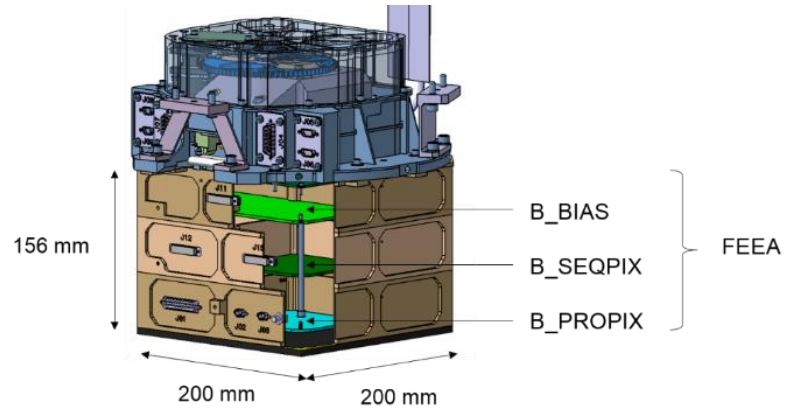


Focal plane assembly: detector assembly

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Front-end electronics: hardware



Challenges of the US free design:

- Charge transfer with high voltages
- Atmel FPGA performance

BIAS Board

Acquisition channels (ADC)
CCD/CAMEX bias voltages



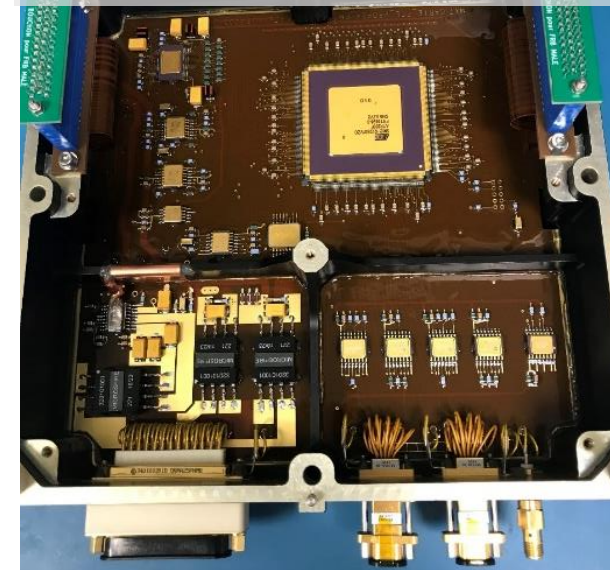
SEQPIX Board

Readout sequencing
Clocks for fast transfer

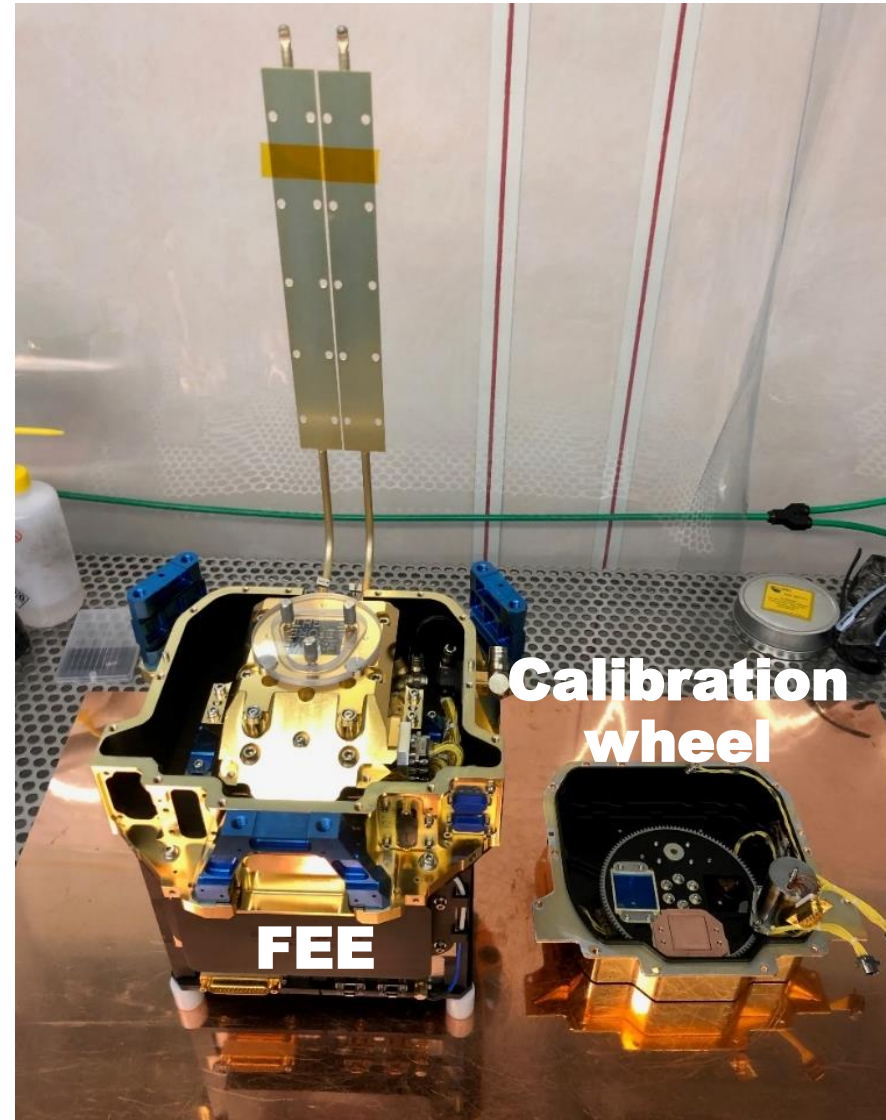
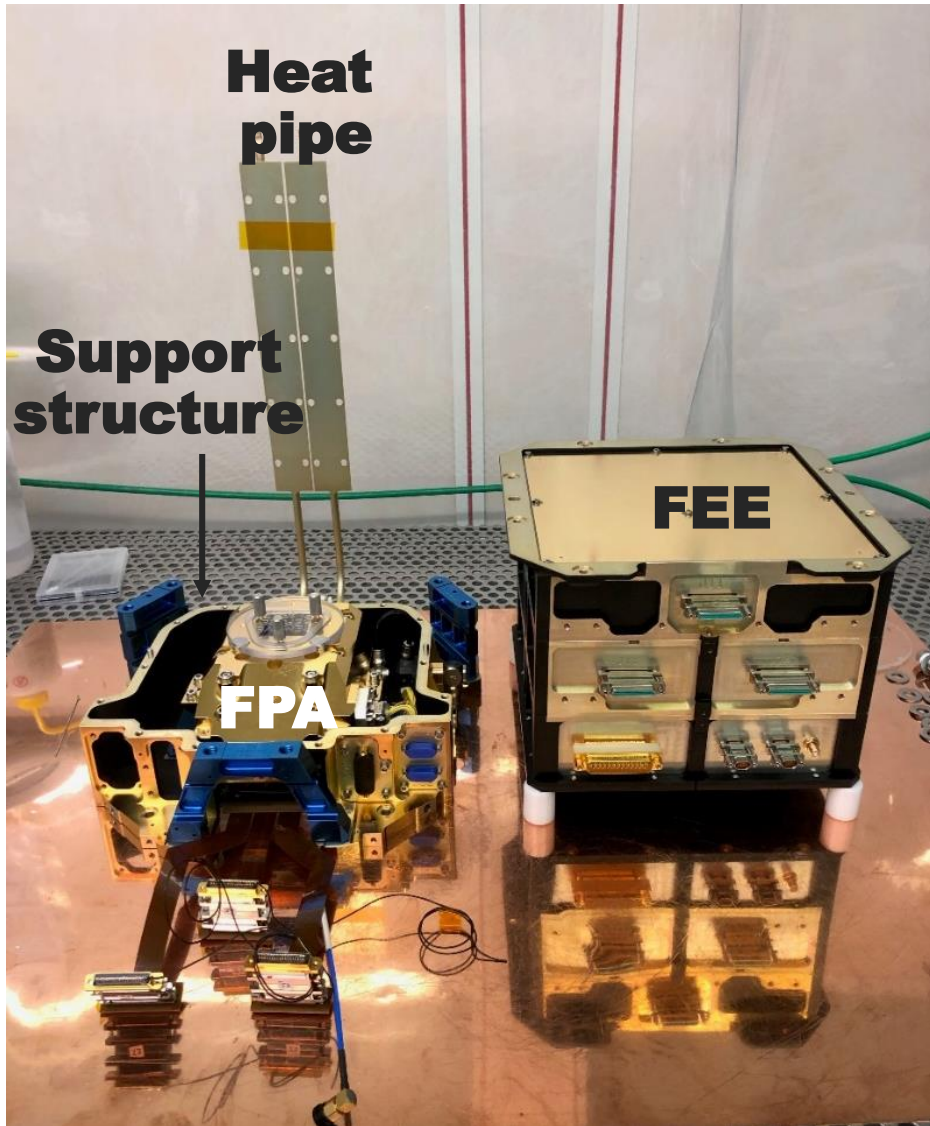


PROPIX Board

Interface with MDPU,
Event pre-processing

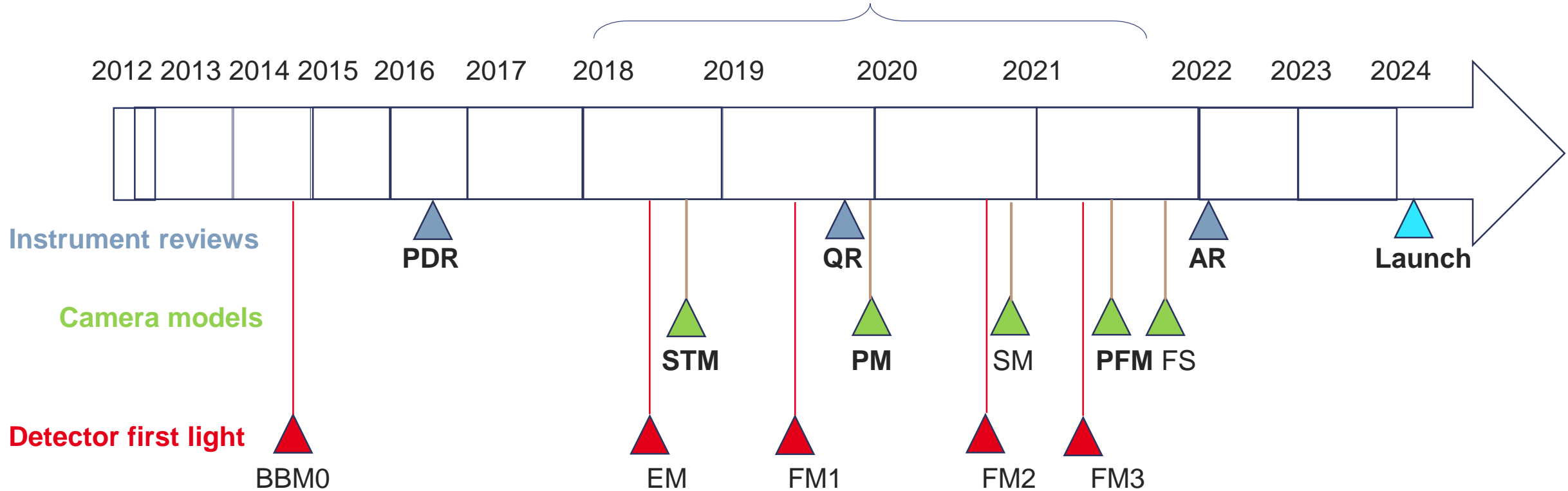


Camera integration



Project development

Acceleration phase of the camera development



PDR: Preliminary Design Review; QR: Qualification Review; AR: Acceptance Review

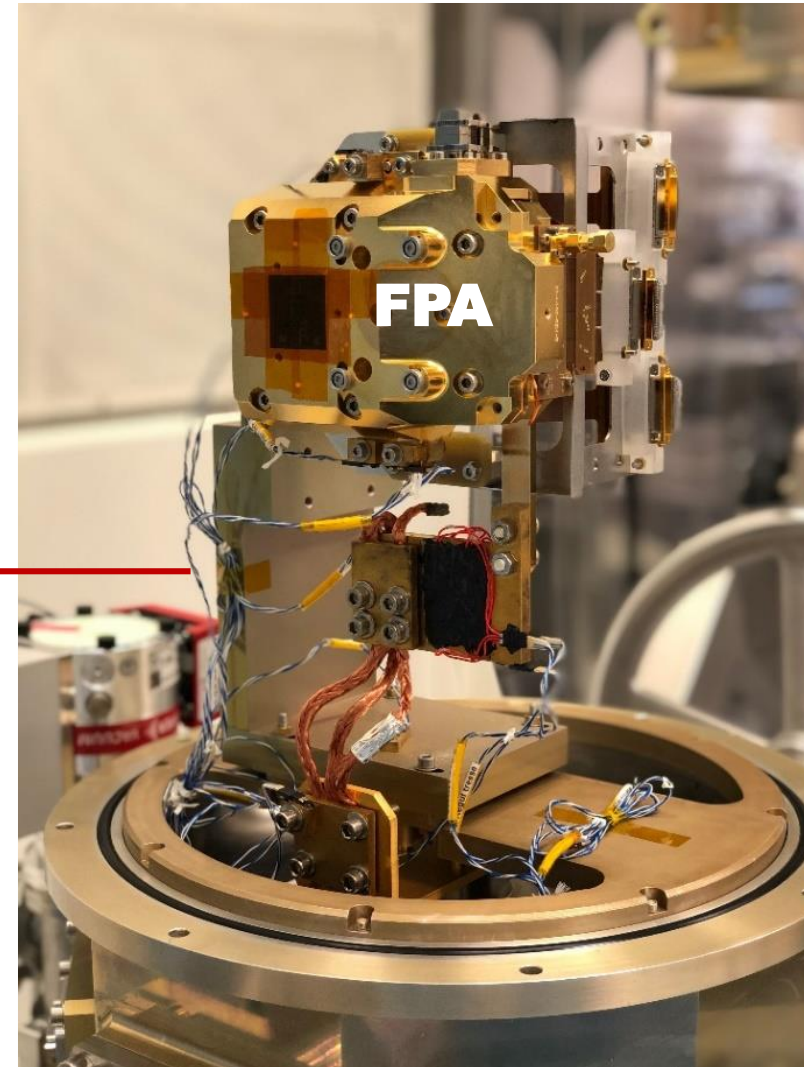
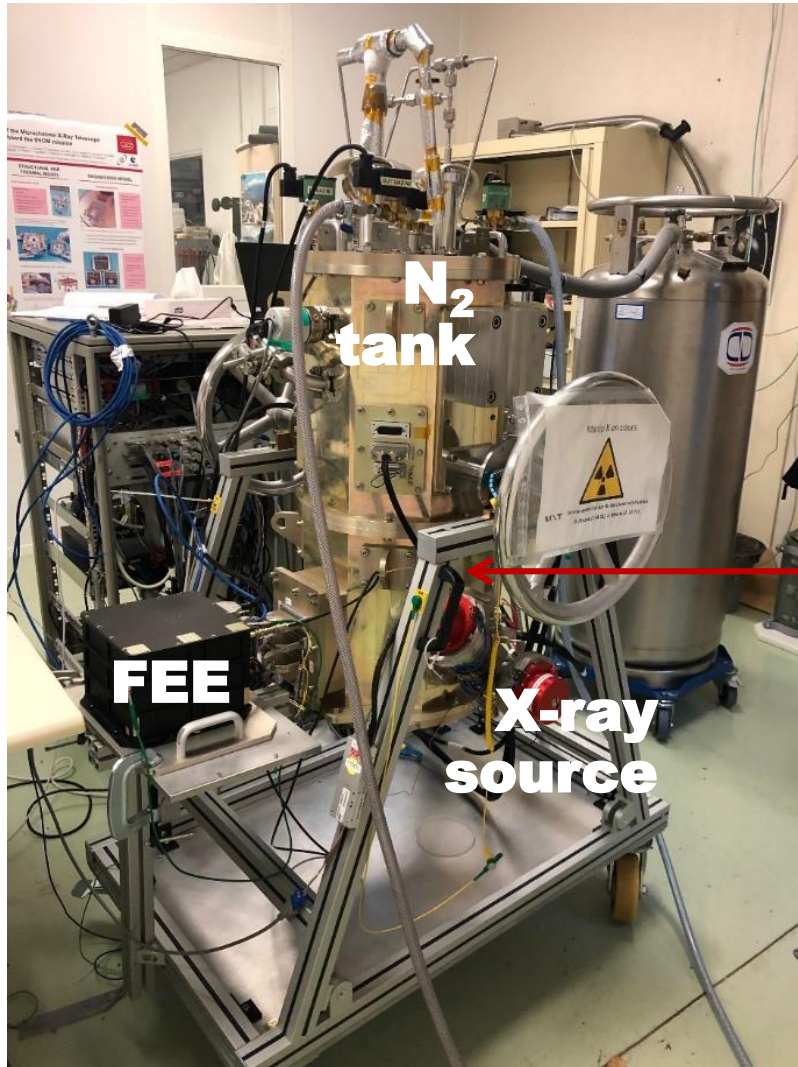
STM: Structural and Thermal Model; PM: Performance Model; SM: Schock Model; PFM: Proto-flight model; FS: Flight Spare



3 ■ The performance characterization

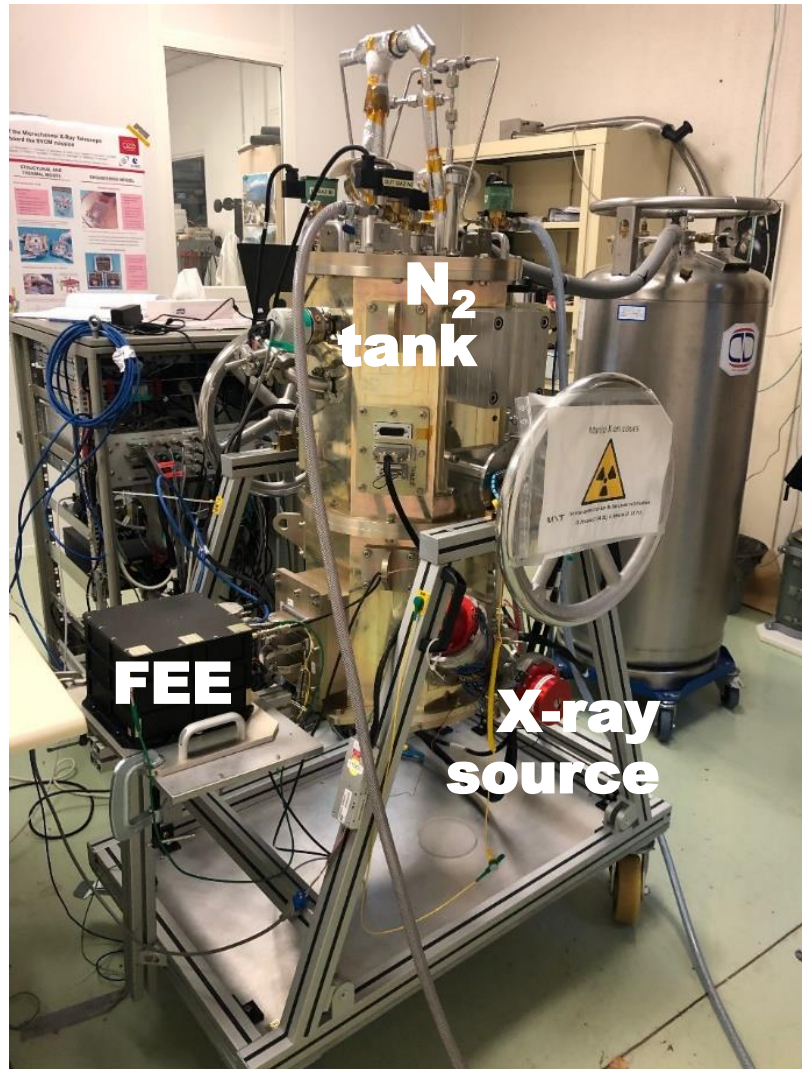
Spectral performance of the detection chain

Full-custom cryostat for design validation and initial detector characterizations

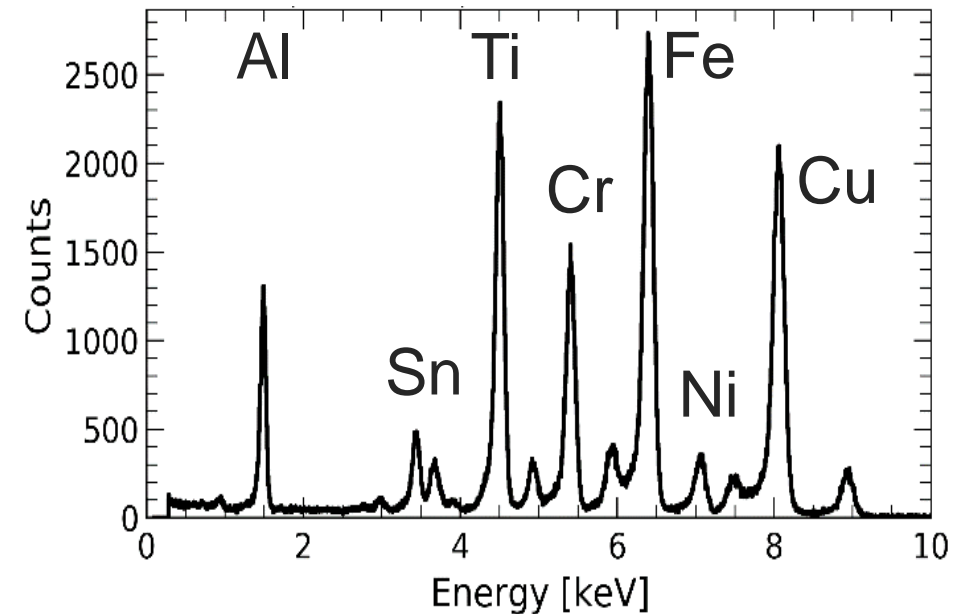


Spectral performance of the FPA

Full-custom cryostat for design validation and initial detector characterizations



Calibration source based on composite fluorescence targets.



+ Complementary spectral data with SOLEIL synchrotron (see Poster P02-13)

→ 3 defect-free flight model detector assemblies

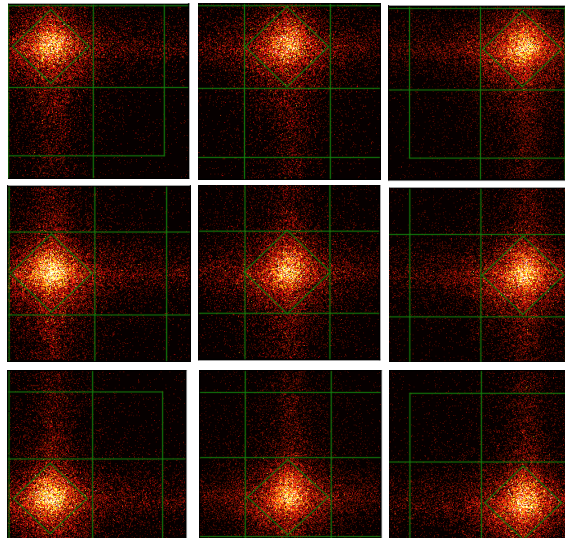
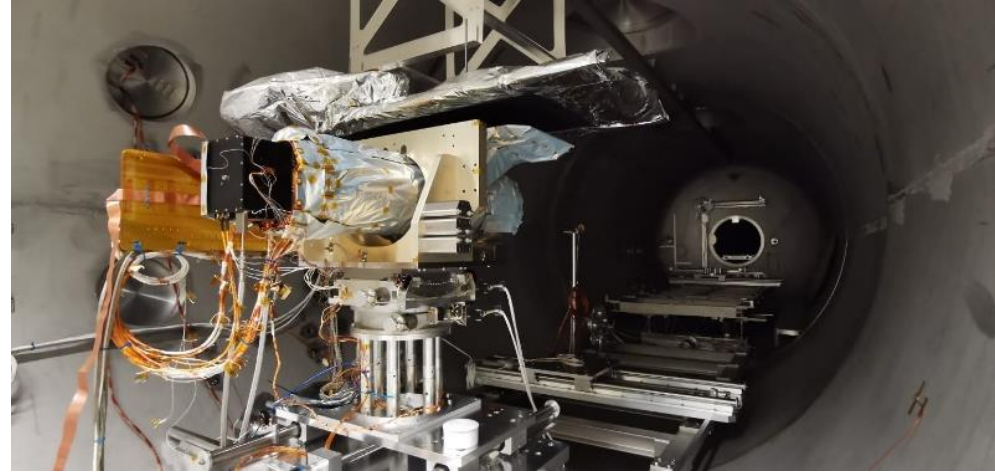
Instrument calibration in panter test facility

X-ray facility in
Neuried (Germany)



Calibration campaign with the flight
instrument in October-November
2021

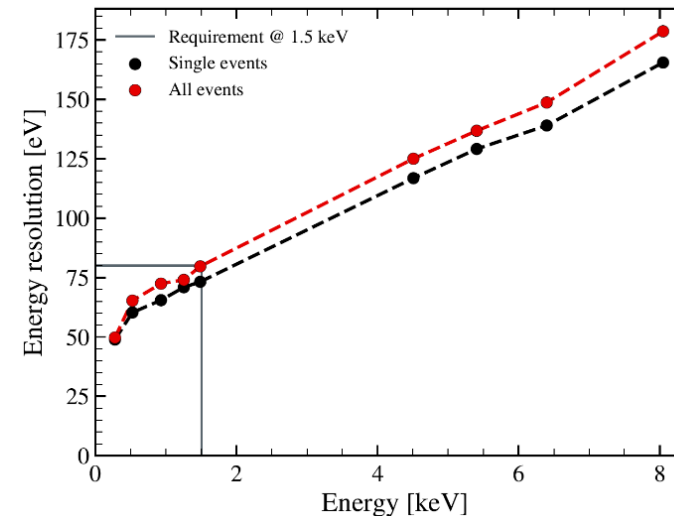
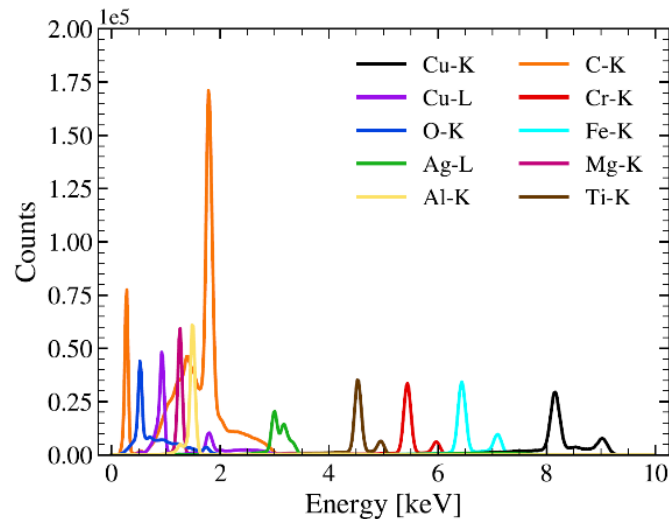
PSF, Effective area, vignetting
characterisation



Spectral calibration of the camera

Performance at -65°C (flight operating temperature)

- Gain uniformity within 1 %
- Uniform dark noise map, 45 eV low energy threshold
- 80 eV FWHM @ 1.5 keV with all events



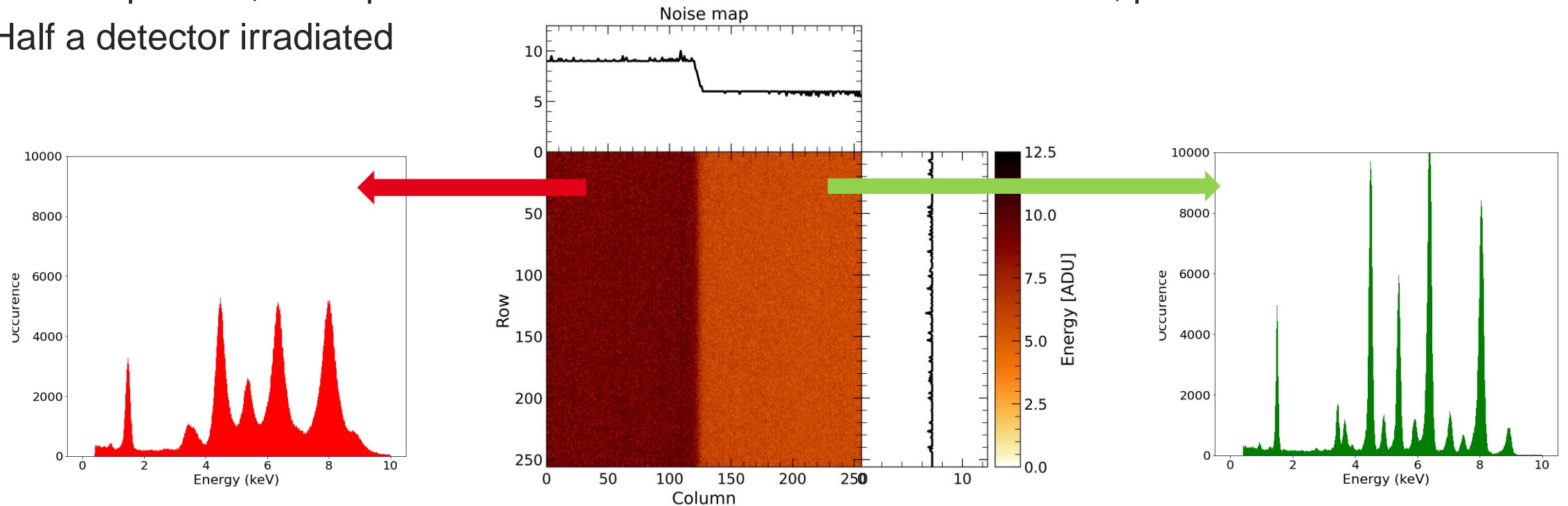
Calibration

- Model of multiple events statistics and spectral calibration method

End-of-life performance

Proton irradiation at Arronax cyclotron in June 2022 to simulate 3 years in orbit with margins

- 50 MeV protons, $6 \cdot 10^9$ protons/cm² fluence, $2 \cdot 10^5$ protons/cm²/s flux, passive irradiation @RT
- Half a detector irradiated



- On-going characterization of FWHM, CTI on-going: See Poster C. Plasse

Conclusions

The MXT camera is a compact system that includes all functions for the optimal performance of the detector.

Its development and validation represent:

- 7 detector assemblies (2 with dummy detector for qualification)
- 3 complete camera models (STM, PM, PFM)
- One full custom cryostat and the adaptation of 3 vacuum chambers
- 3 calibration test campaigns (2 in PANTER, 1 in SOLEIL)

The MXT camera fulfills all mission requirements, with spectral performance at the state of the art.

The MXT instrument is ready to fly: right now integrated in the satellite platform for final acceptance tests. Launch planned for beginning 2024.

THANK YOU FOR YOUR ATTENTION

