

LISA

The Instrument

H. Halloin

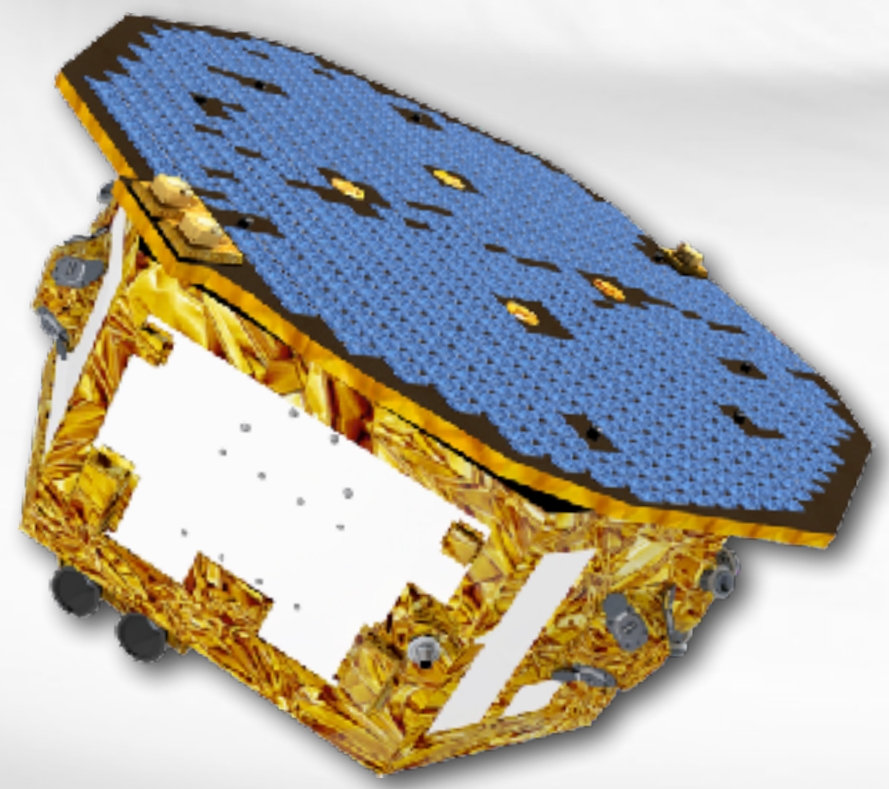
Université de Paris, CNRS,
Astroparticule et Cosmologie







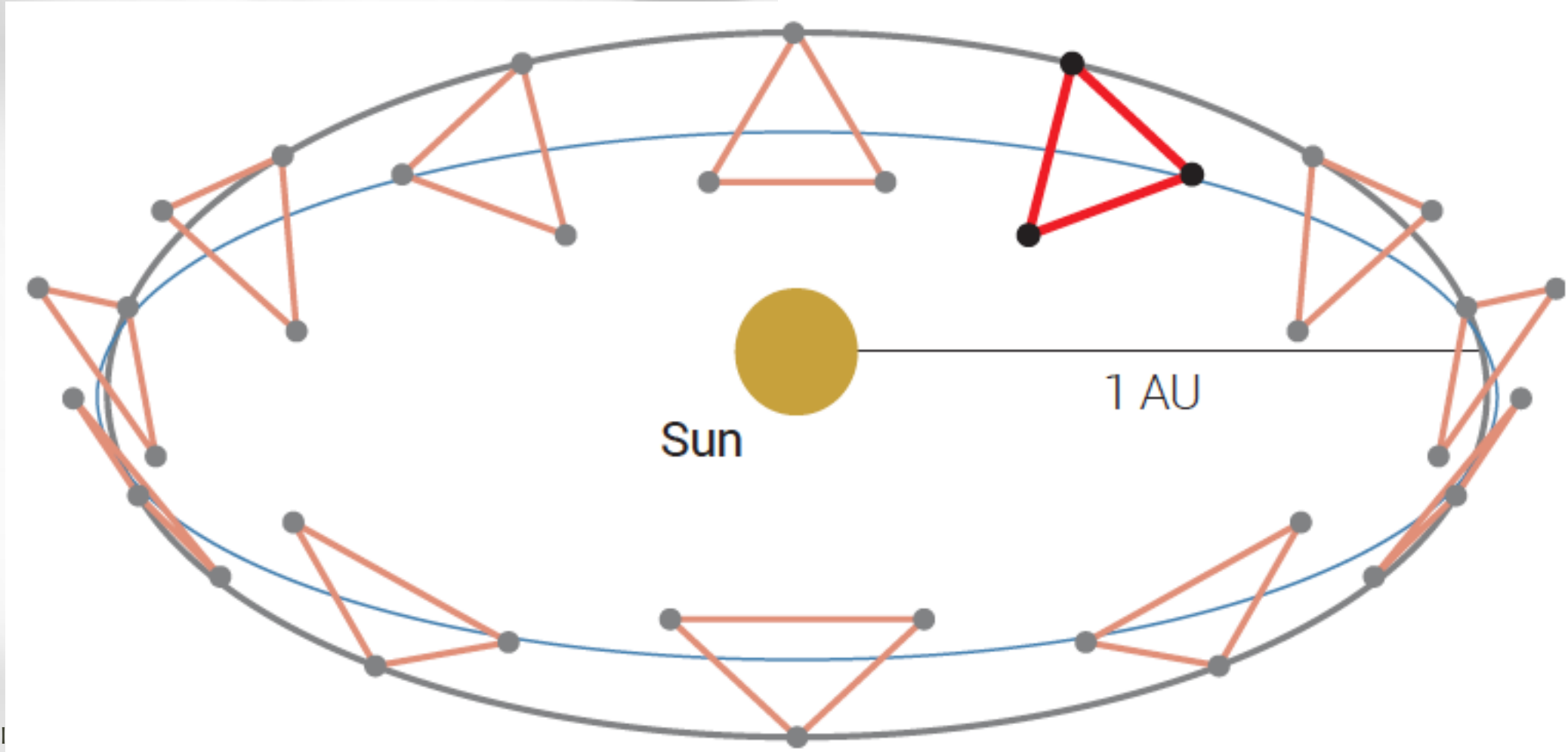
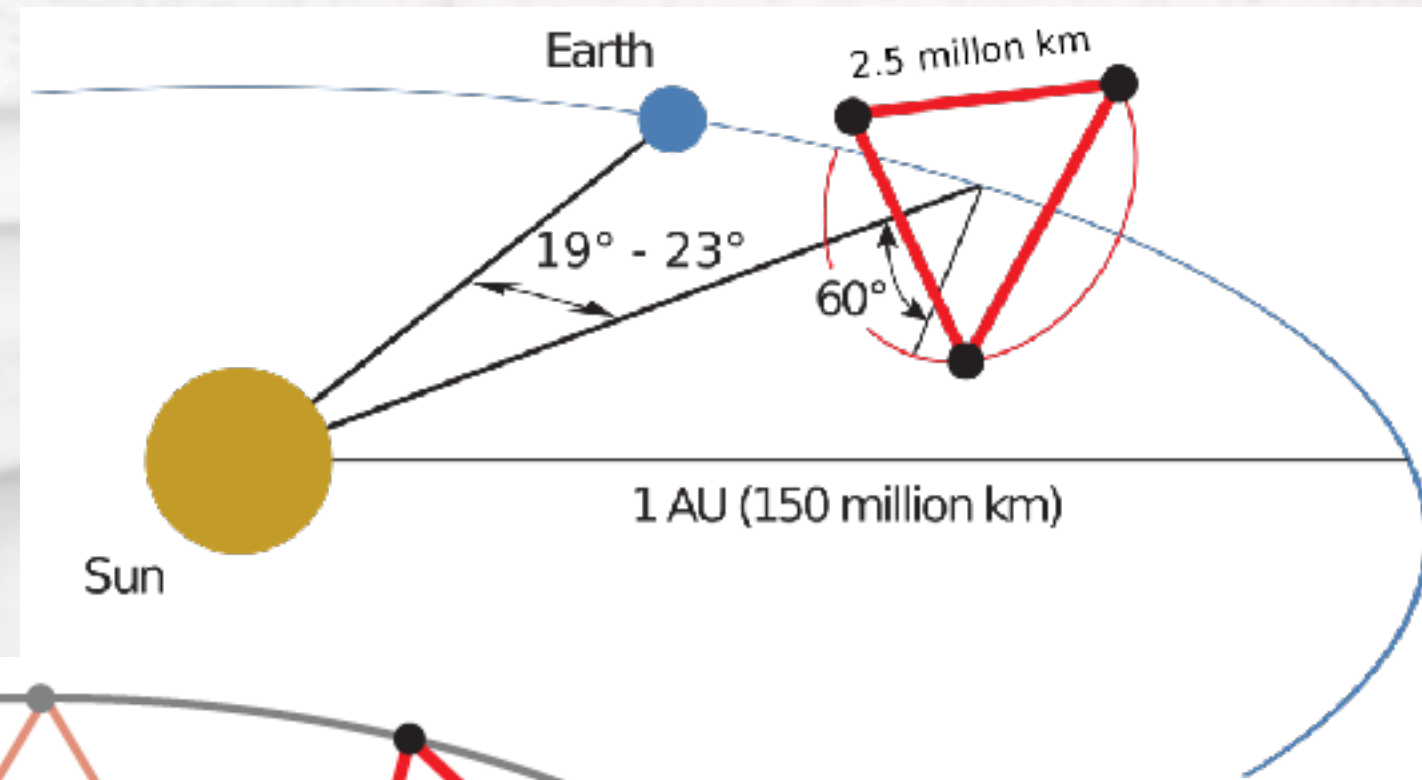


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Mission Profile

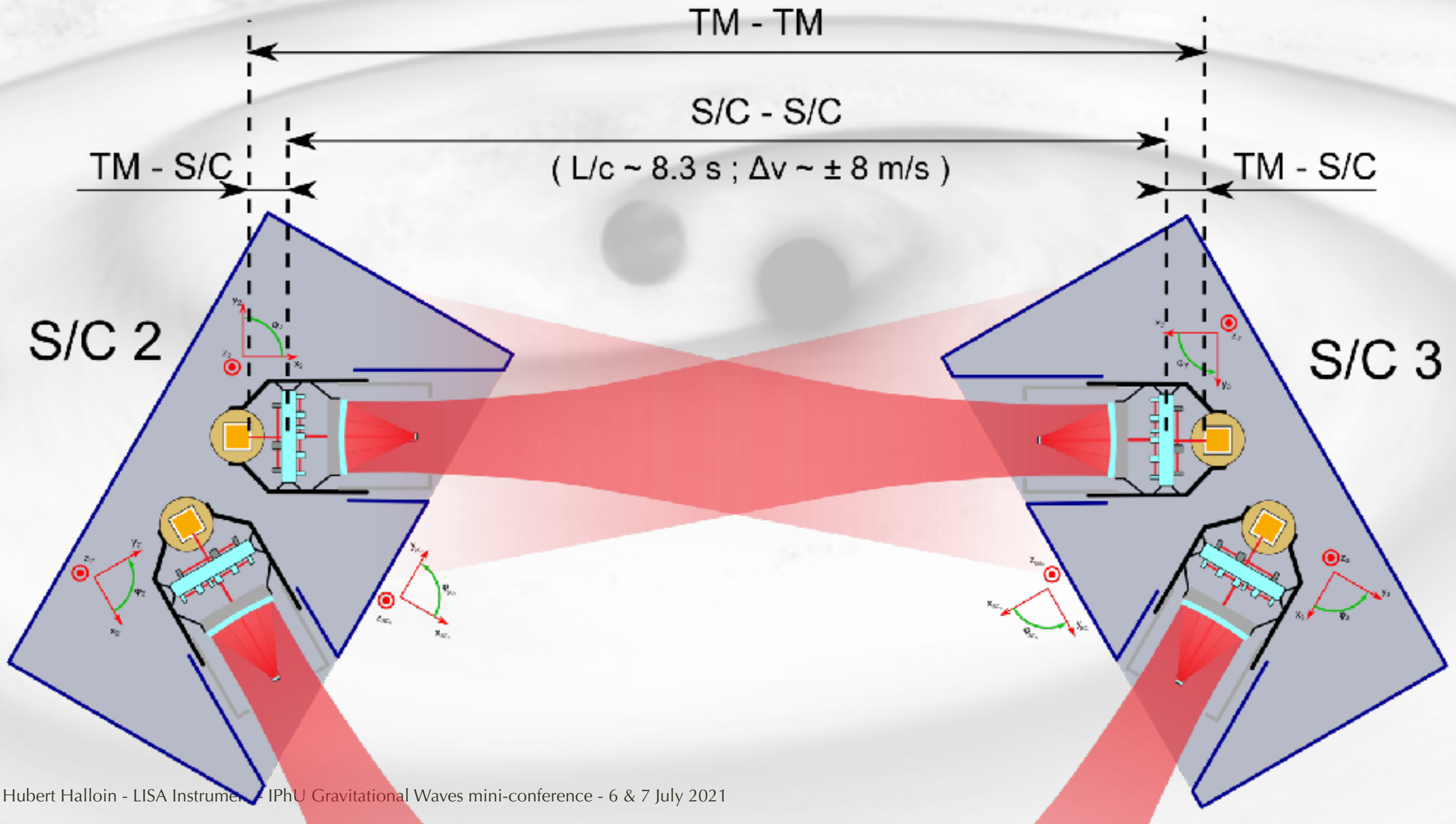


-  Long arms interferometer
-  Earth-like orbit, 19° to 23° trailing
-  Mission duration : minimum 6 years
(consumables and orbit stability for 10 years)
-  3 arms / 6 links ; 2.5 Mkm



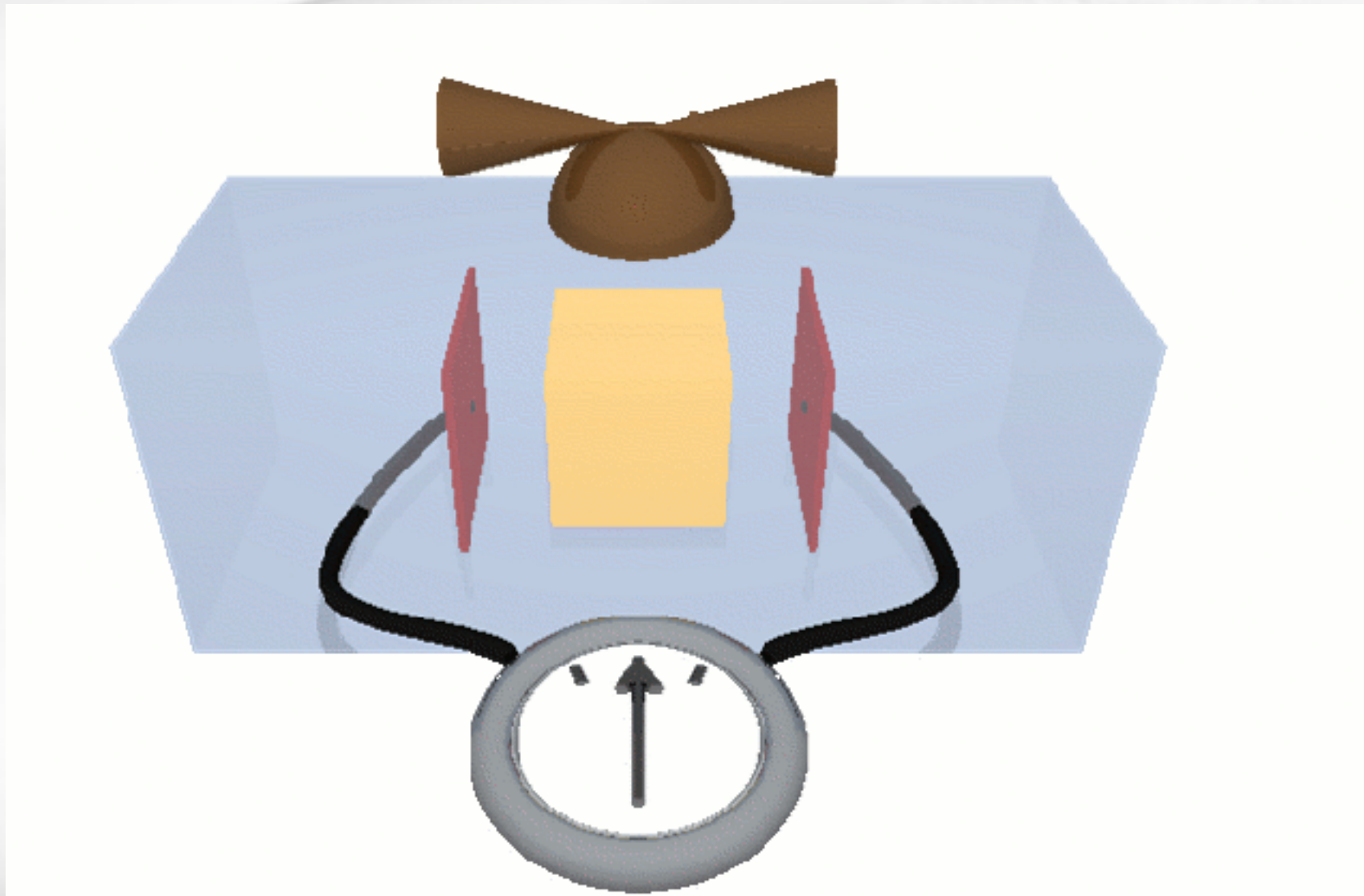
Measurement principle










- 2 TM / satellites (direct heritage from LISA Pathfinder)
- 2 steerable optical benches / satellite



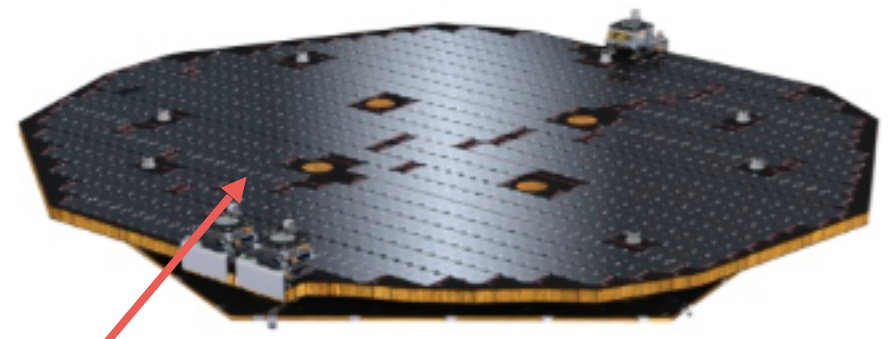
Drag-free flying ?

- Test masses must be protected from external perturbations (mainly solar wind)
- Technology demonstrator : LISA Pathfinder (2015-2017)

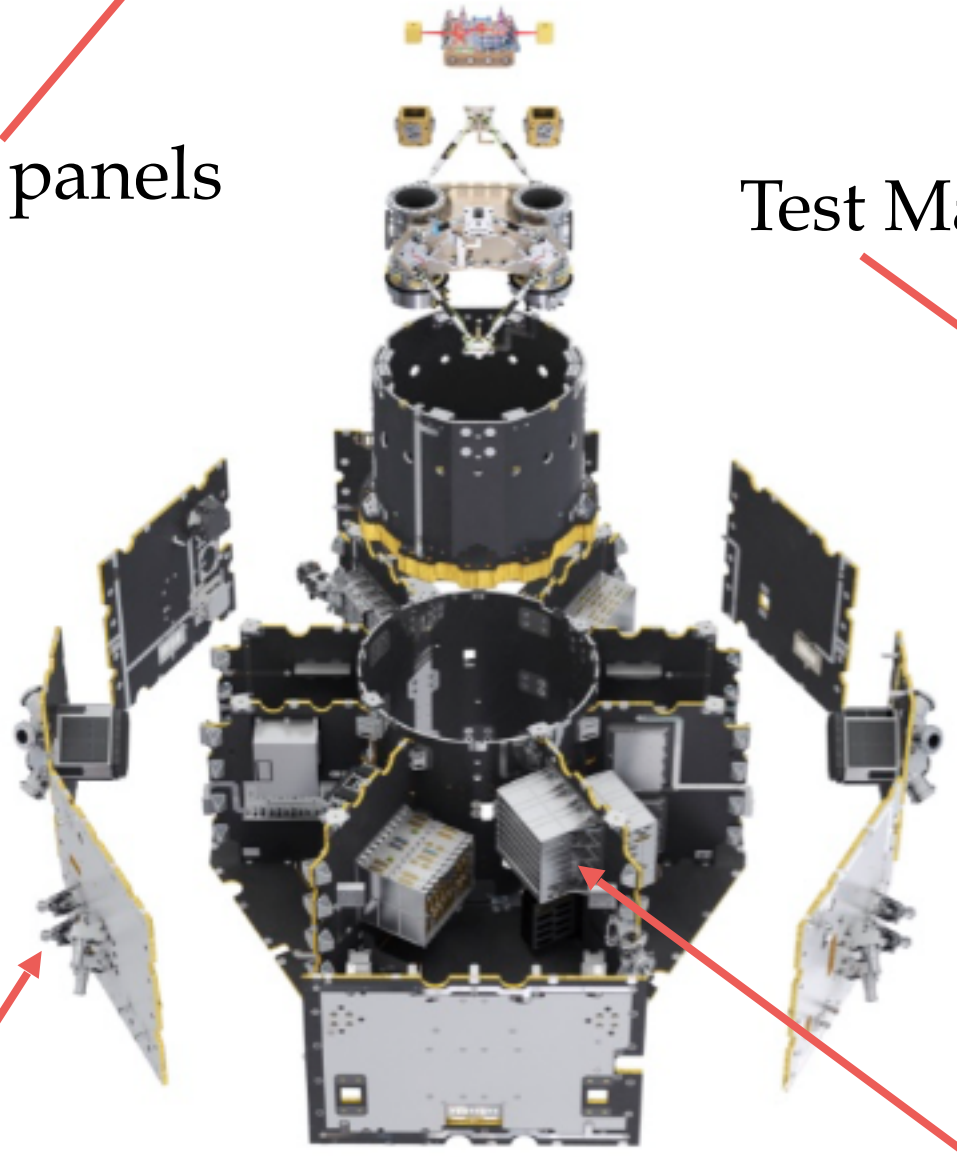


-  **Main goal:** demonstrate the possibility of "Free Fall" in space at the level of $\approx 10^{-14} \text{ m}\cdot\text{s}^{-2}/\sqrt{\text{Hz}}$, around 1 mHz
-  A number of effects have to be minimized:
 -  The static gravitational potential between the TMs and the SC,
 -  Residual links of the TMs w.r.t the SC via the residual vacuum,
 -  Cross talk between various electrostatic actuators,
 -  TM charging by cosmic rays that is eliminated by UV illumination,
 -  Temperature fluctuations ,
 -  Magnetic field fluctuations,
 -  ...

Lisa Pathfinder : A technology demonstrator



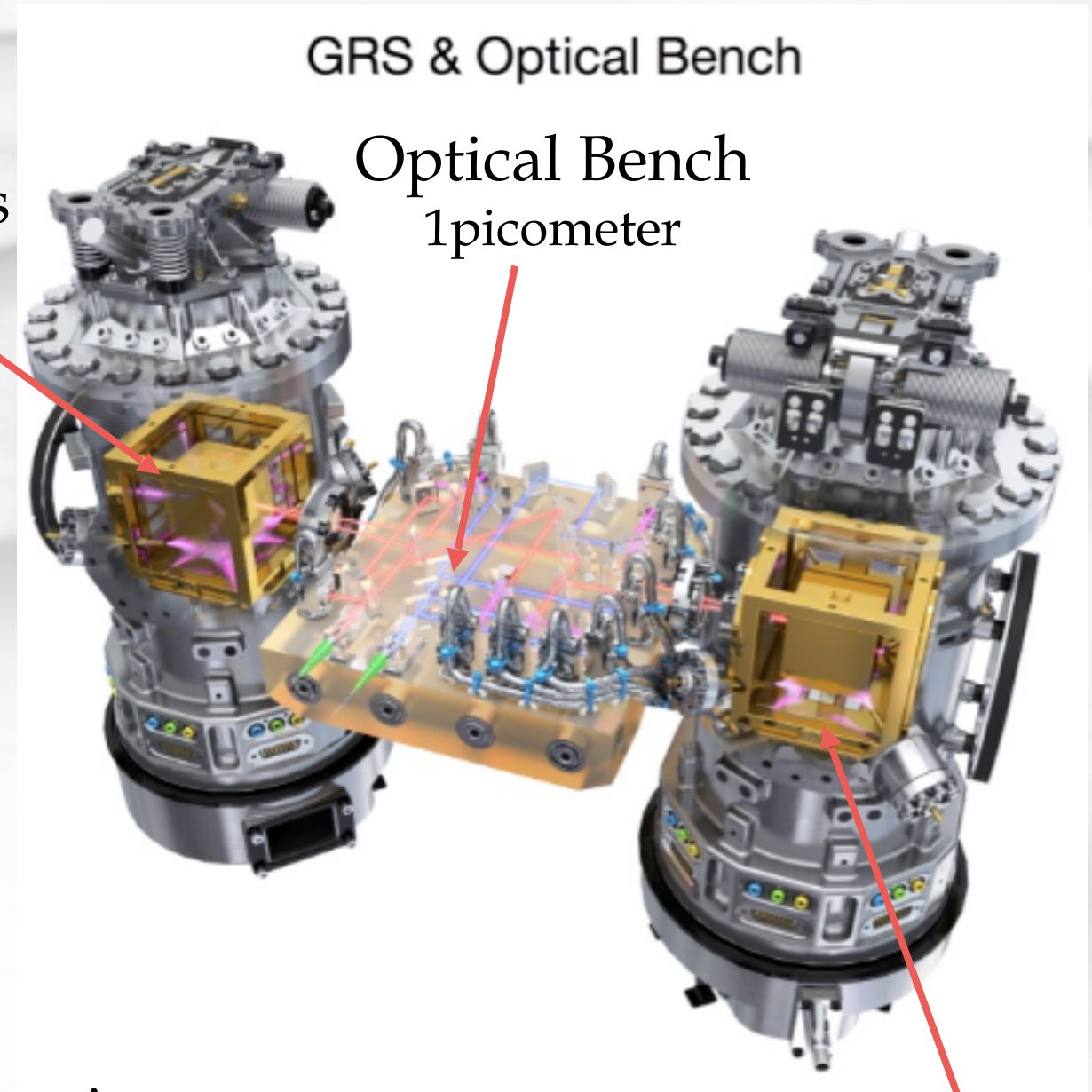
Solar panels



The micro-thrusters
Cold Gas (μ -Newton)

Test Mass

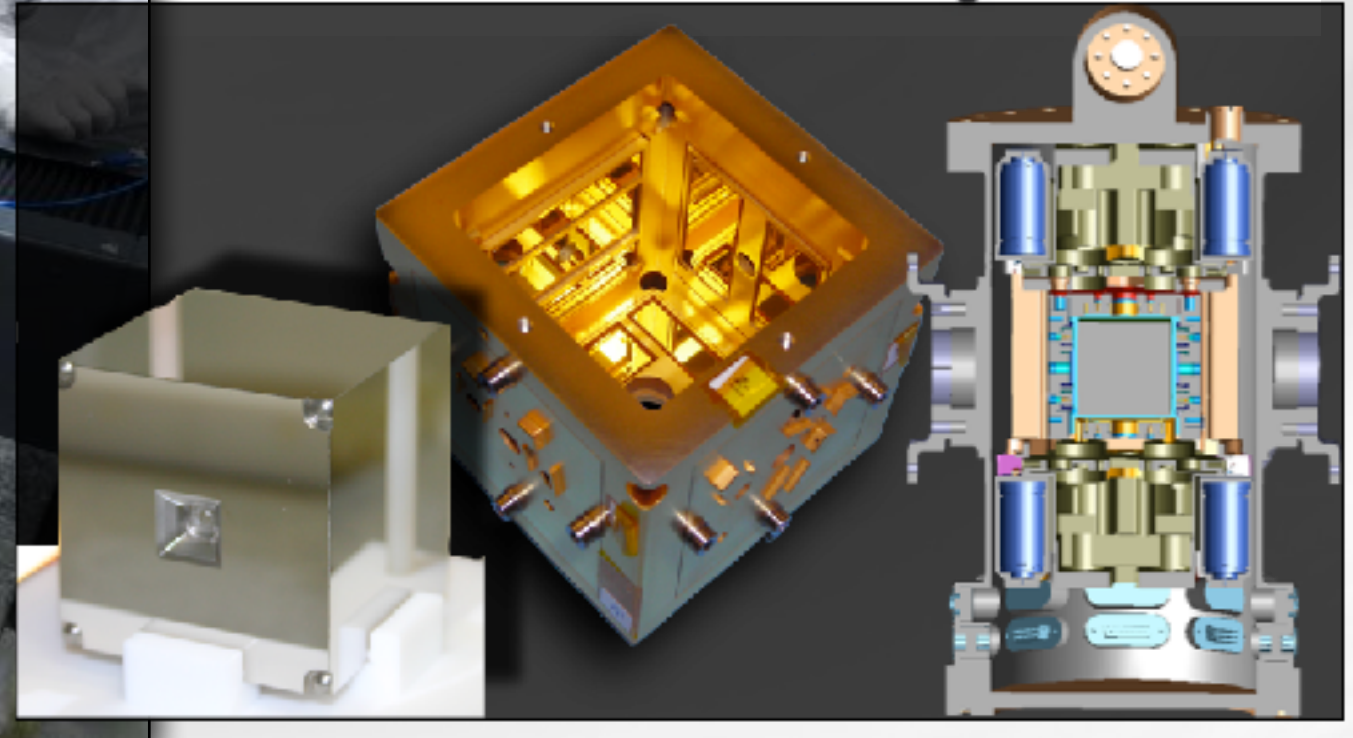
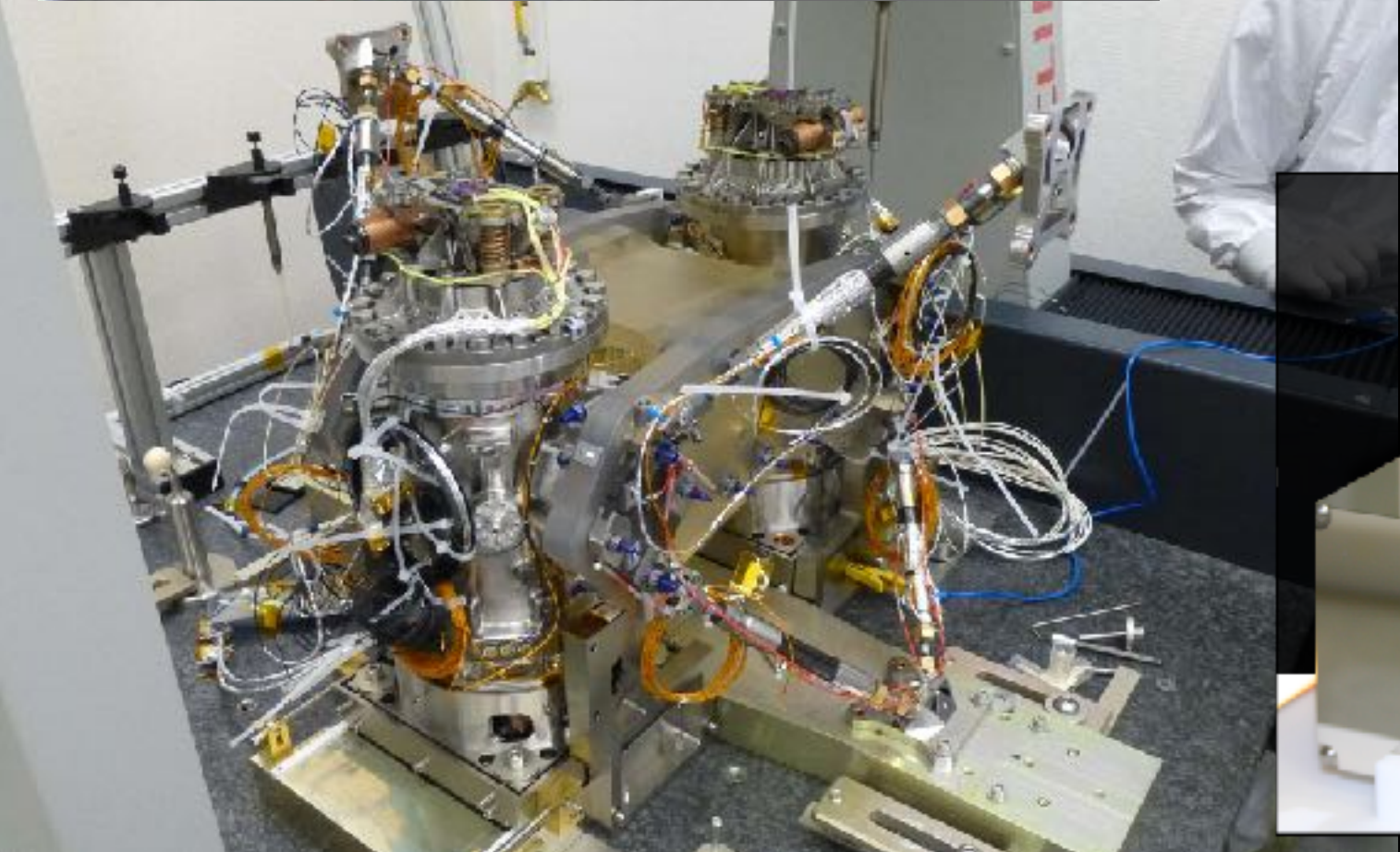
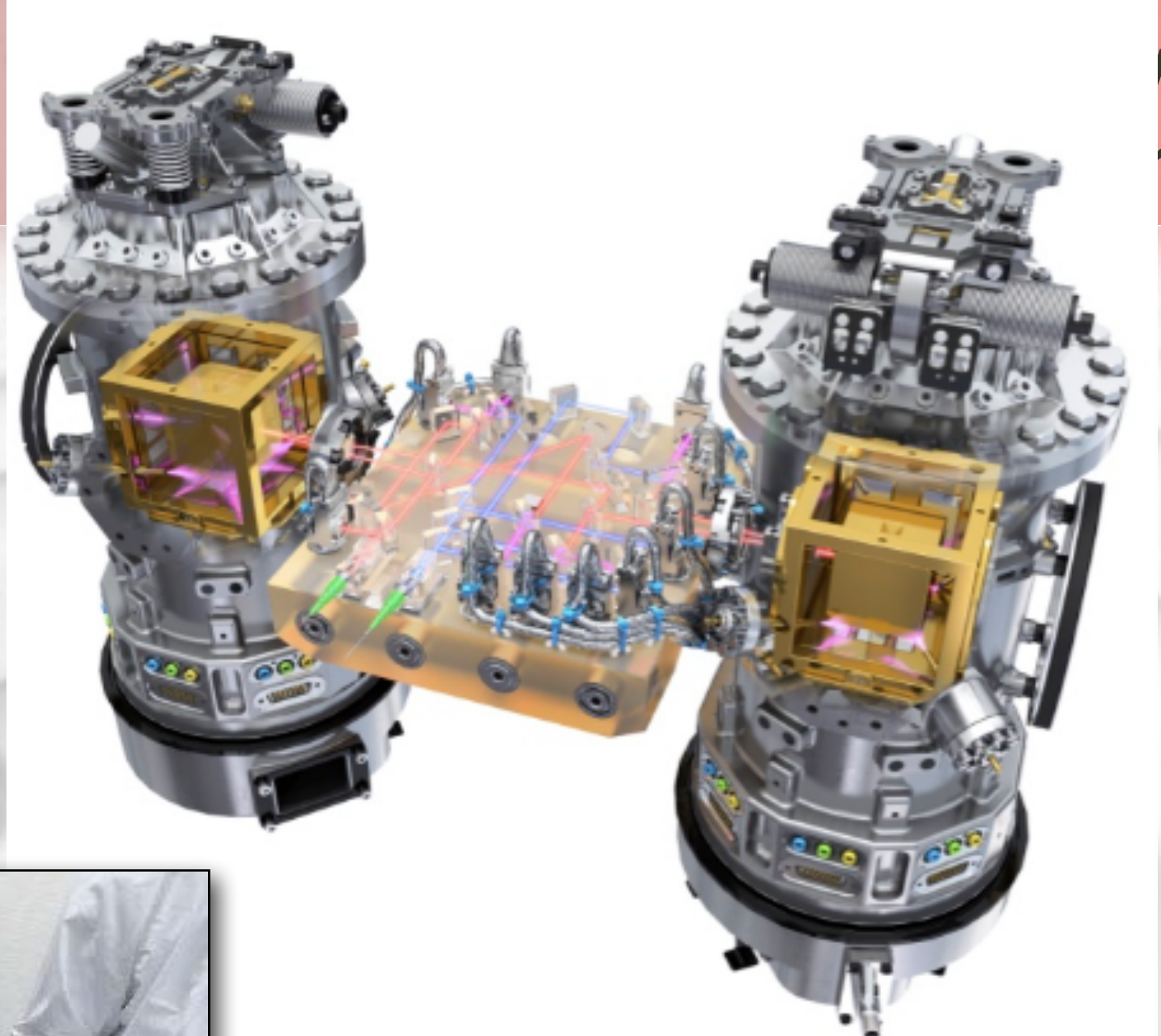
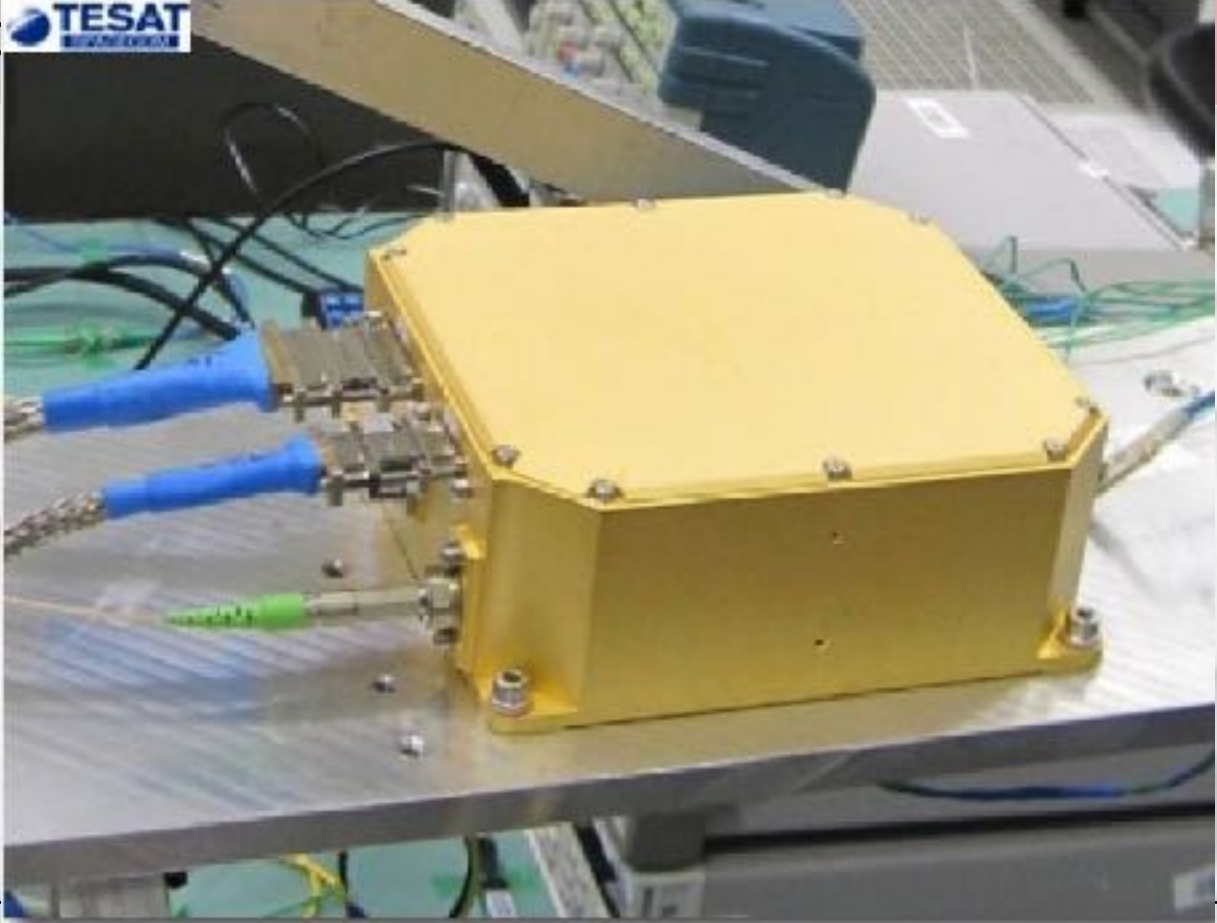
Electronics + computers



GRS & Optical Bench

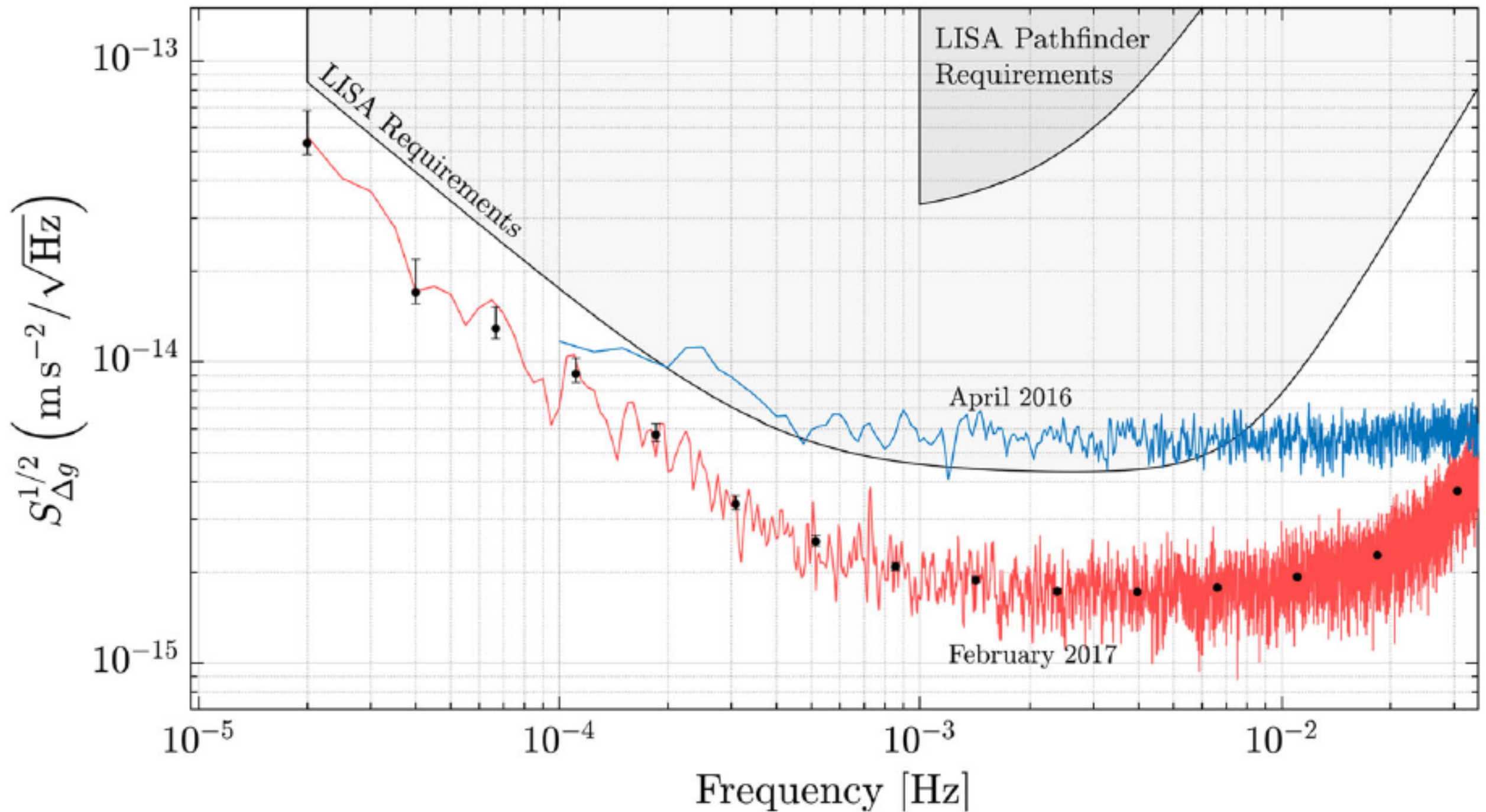
Optical Bench
1picometer

UV illumination

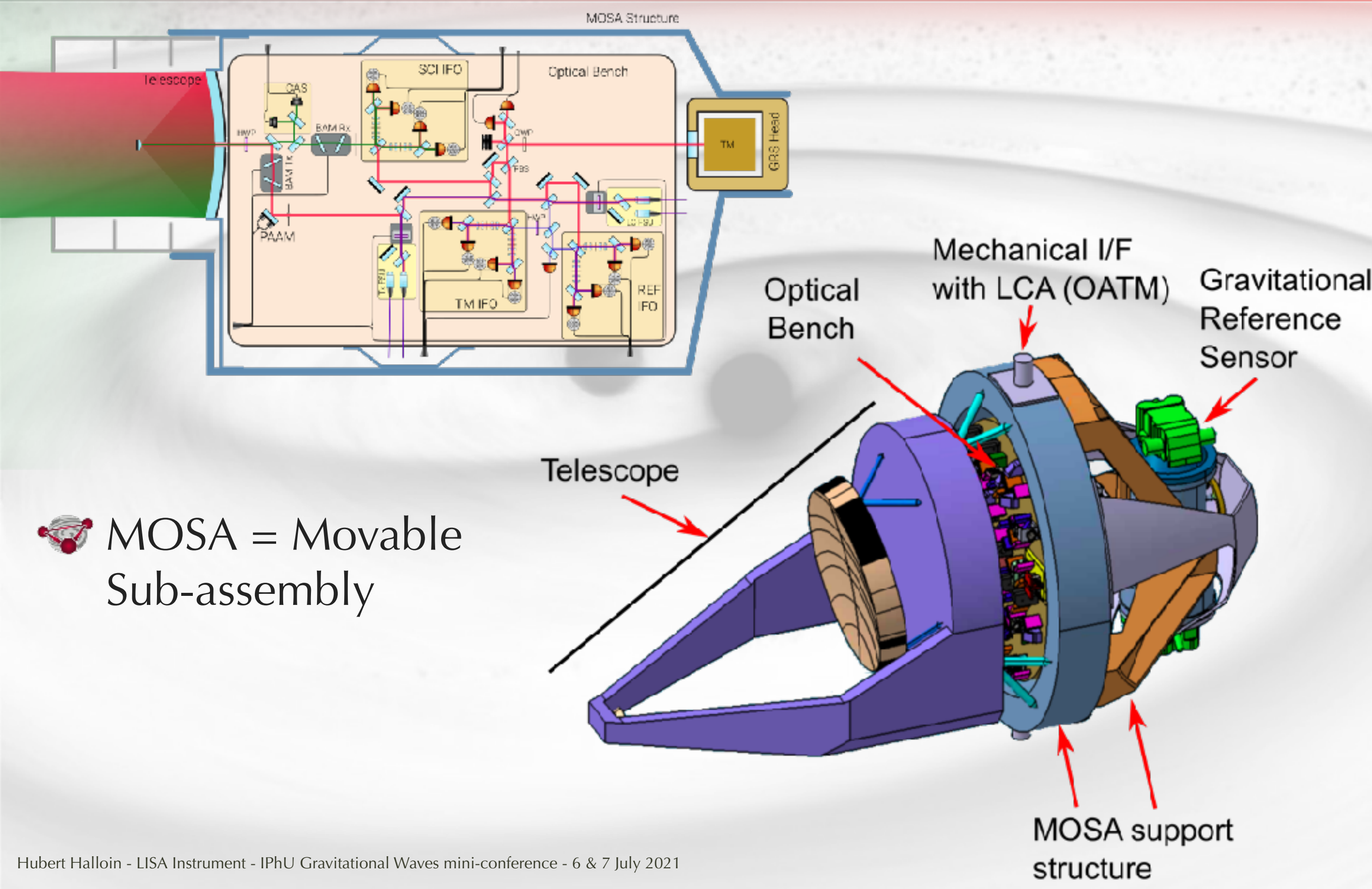





Beyond the Required LISA Free-Fall Performance: New LISA Pathfinder Results down to $20 \mu\text{Hz}$



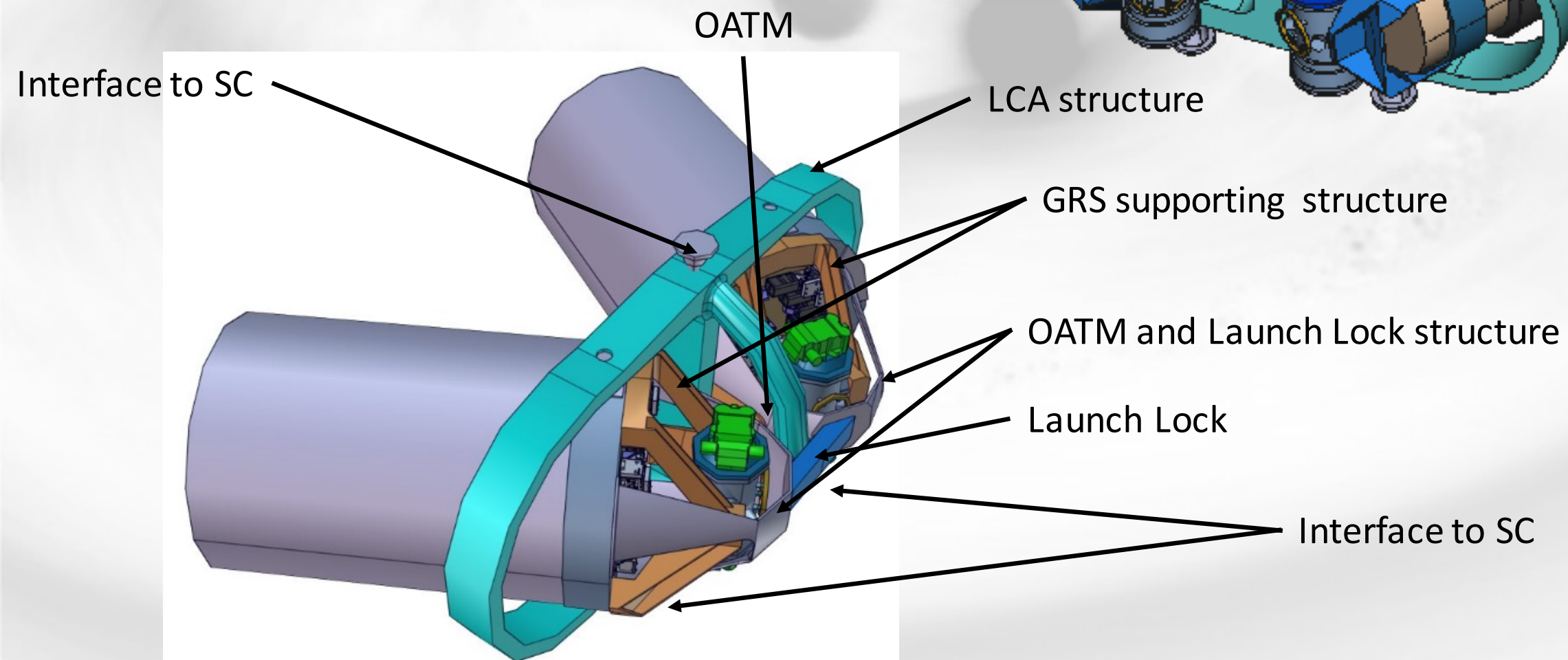
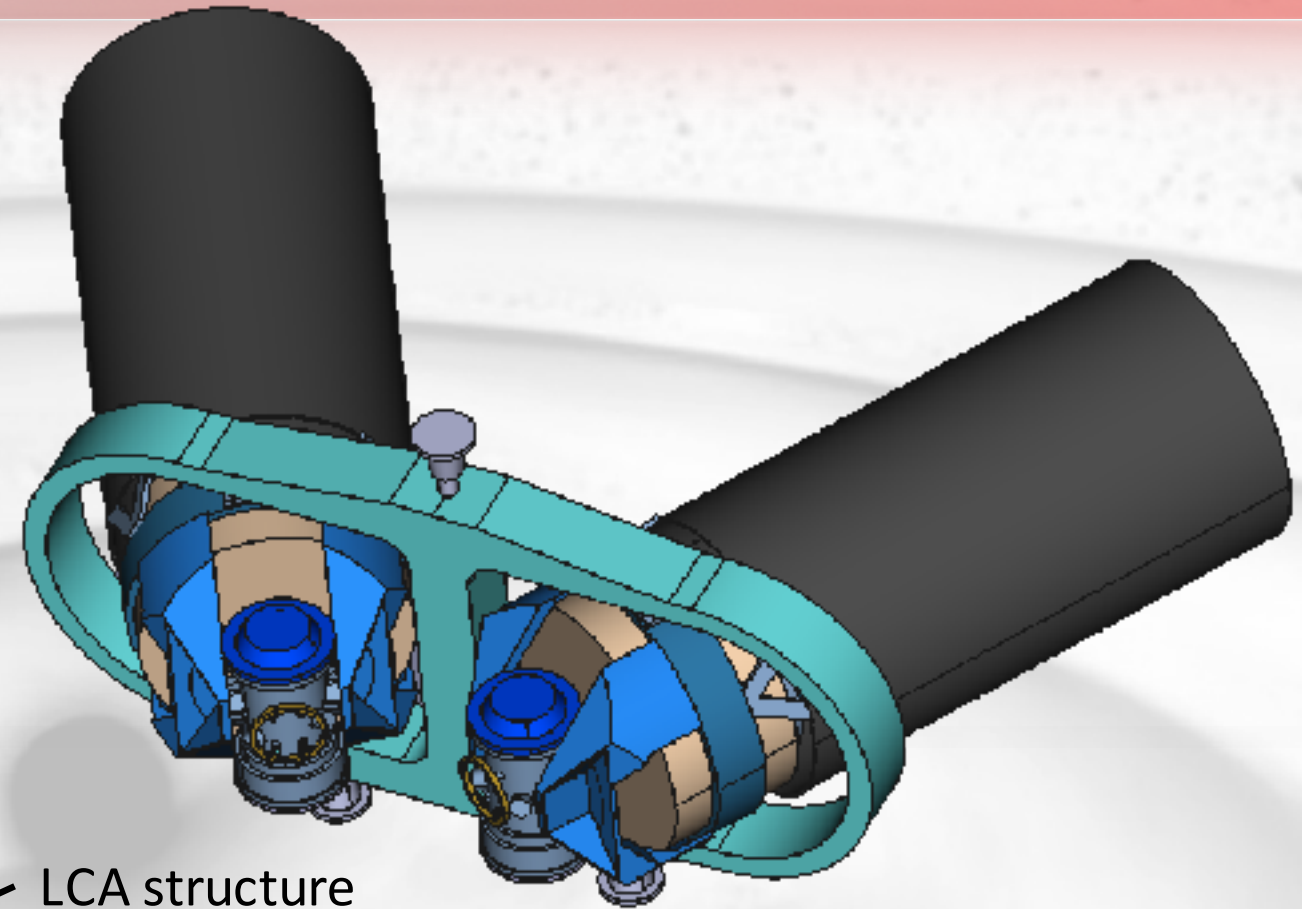
Scheme of a instrument



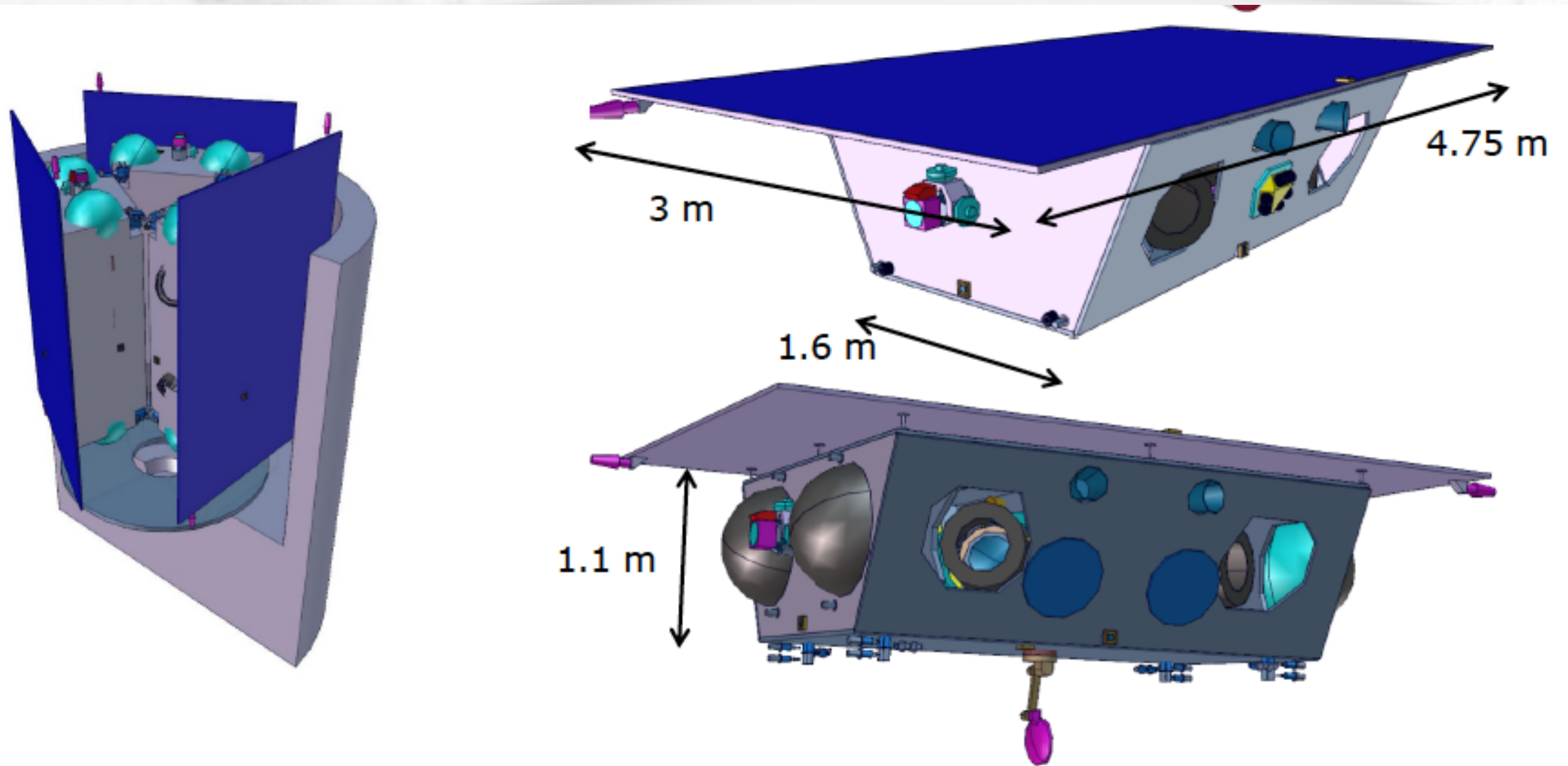

MOSA = Movable Sub-assembly

Two MOSAs form an LCA ...

 LCA = LISA Core Assembly



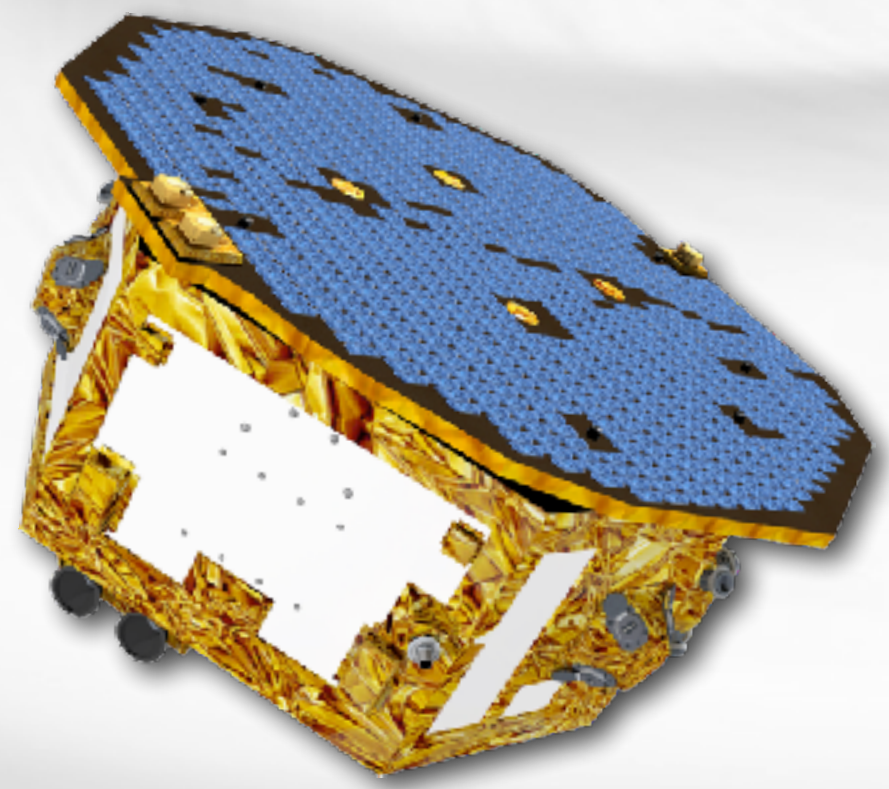
Three S/C fit into an Ariane 6.4



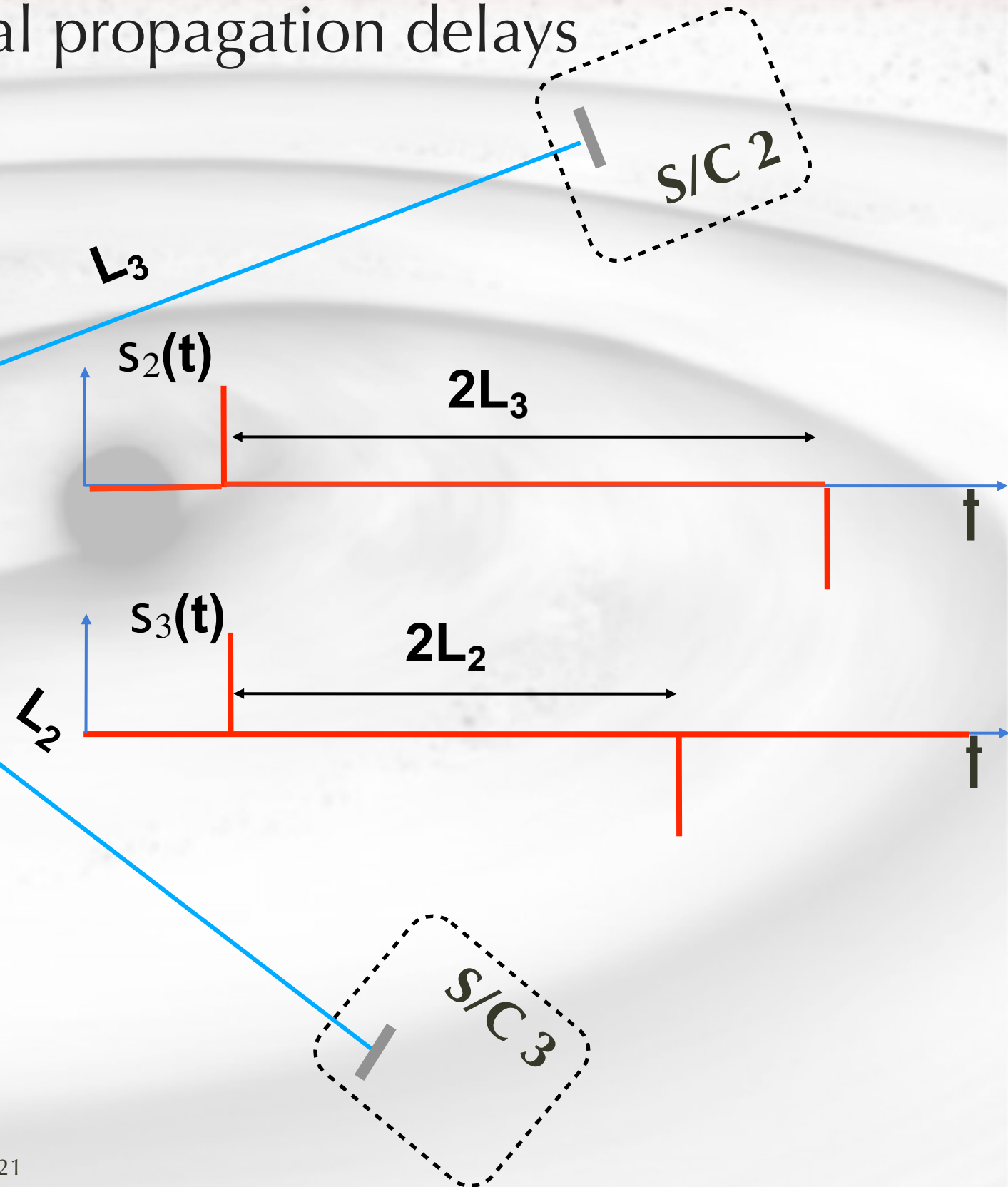
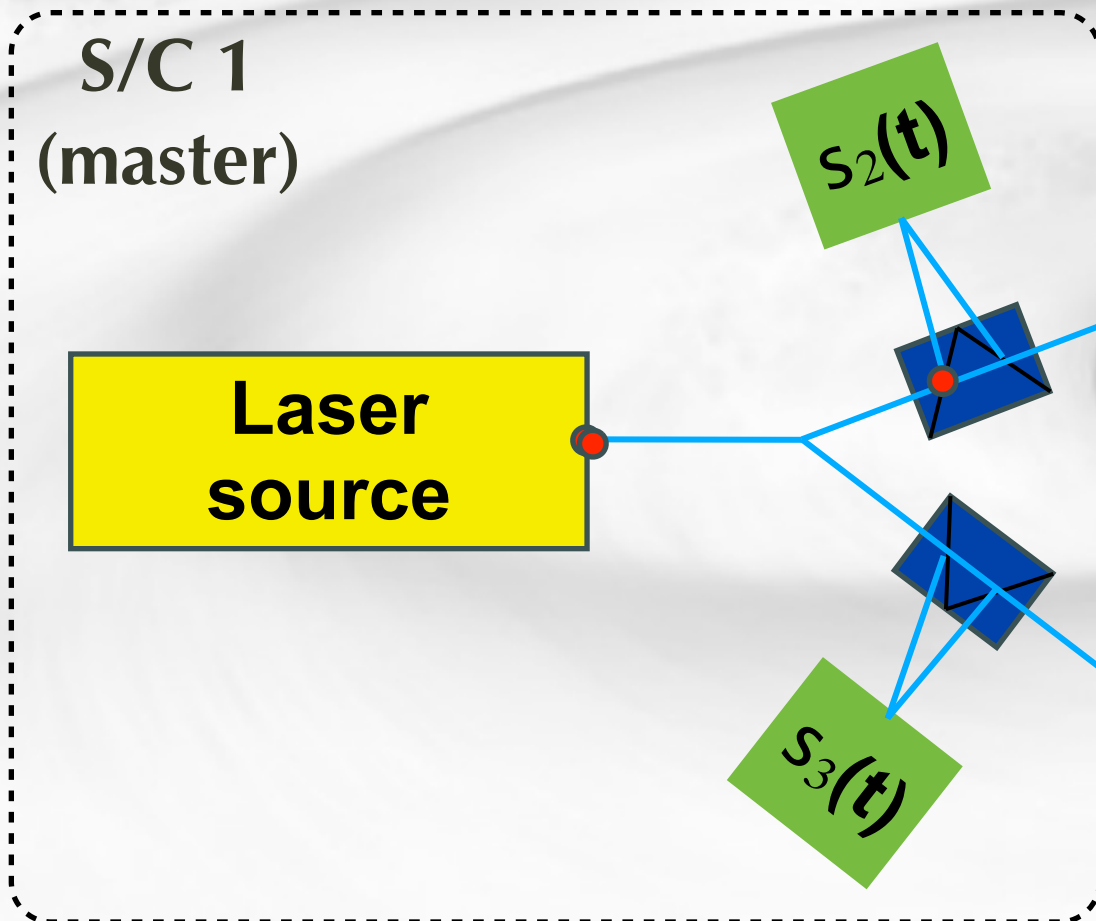


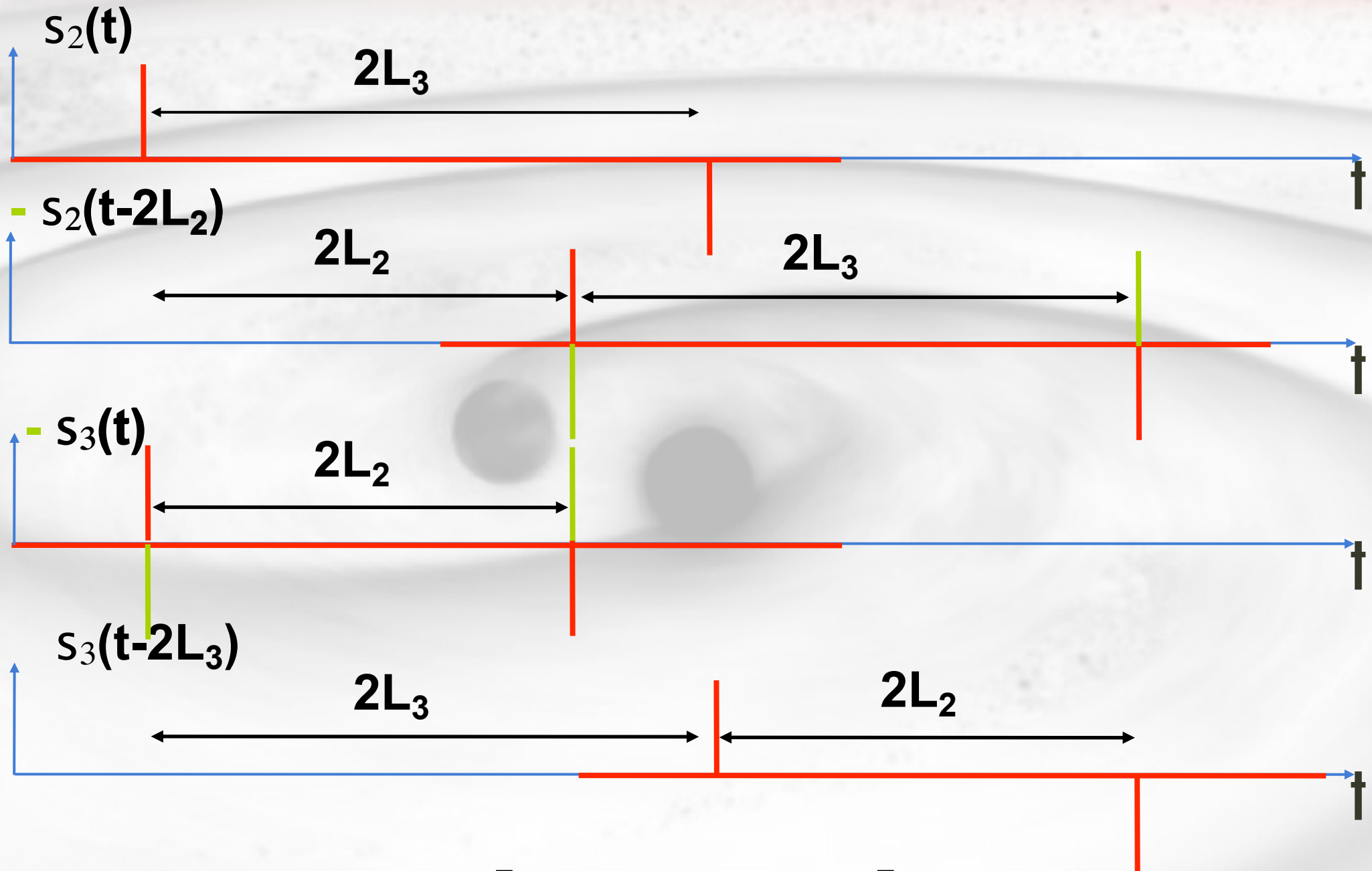
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Measurement Scheme



Not a Michelson : unequal propagation delays



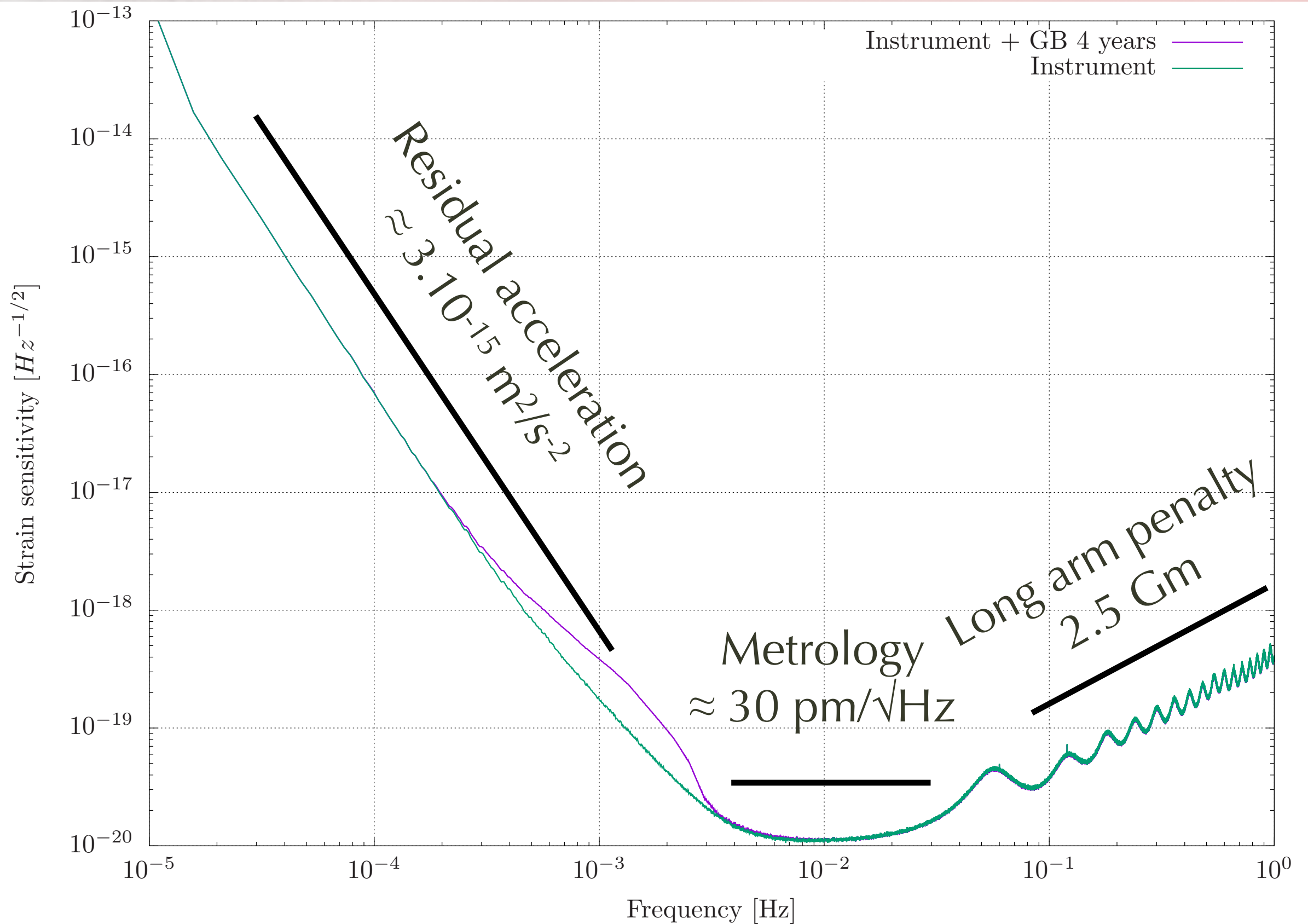


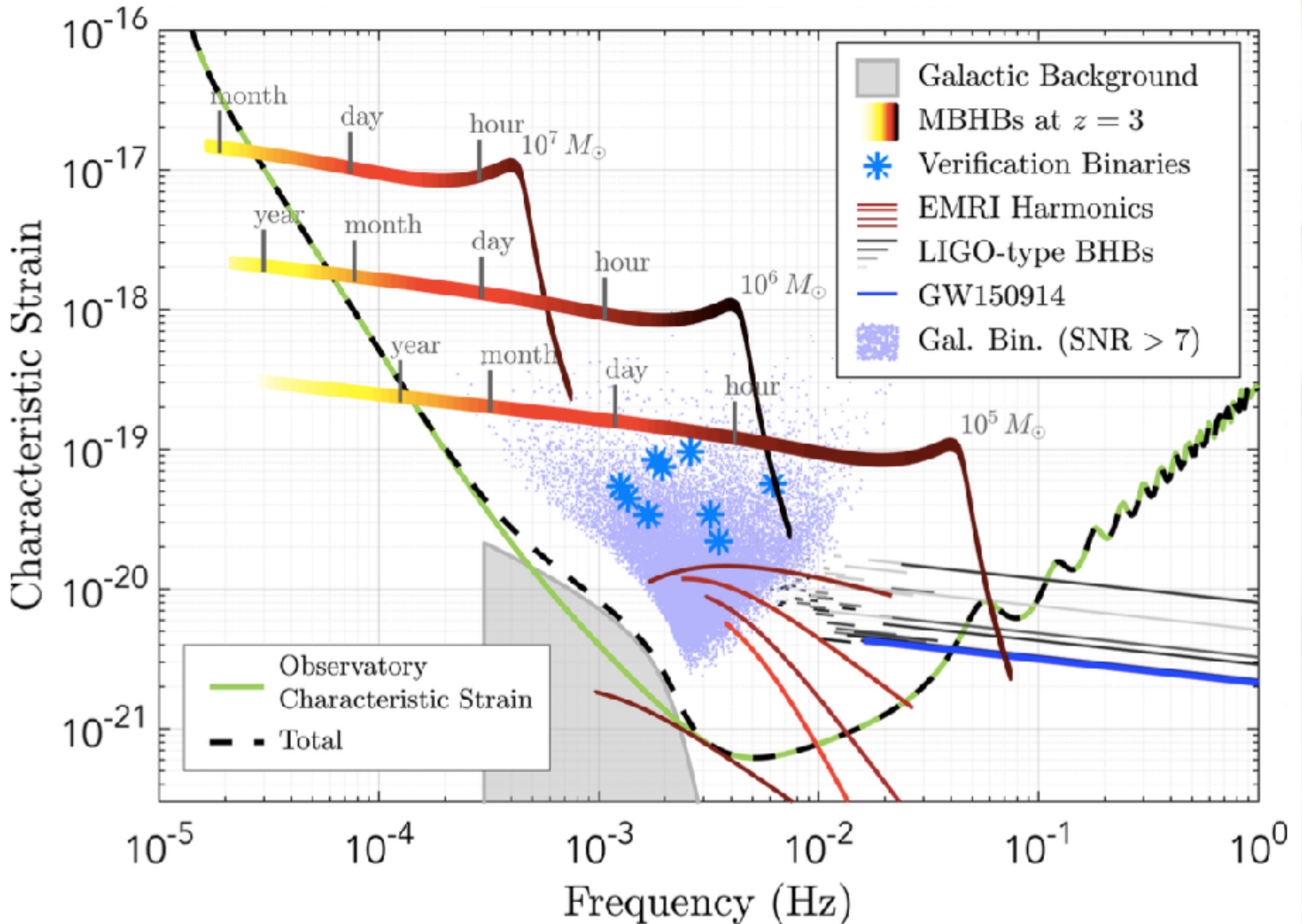
$$s_2(t) + s_3(t-2L_3) - [s_3(t) + s_2(t-2L_2)] = 0 \dots$$

$$s_2(t) + s_3(t-2L_3) - [s_3(t) + s_2(t-2L_2)] = 0 \dots$$

- ⇒ Cancellation of propagated noises (mostly laser phase noise)
 - required noise reduction >8 orders of magnitude
- ⇒ Transfer function shaping (no signal at f multiple of 1/(2L) ~60 mHz)
- ⇒ Requires the knowledge of :
 - Armlength at a **few meters accuracy**
 - Relative clock jitters at a **few ns**
- ⇒ (Much) more sophisticated TDI combinations compensate for non commutativity of delay operators (fluctuating arm length), non reciprocity of propagation (Sagnac effect), etc.
- ⇒ Residues of TDI combinations put constraints on :
 - relative clock drifts
 - antialiasing and interpolation filters
 - data sampling
 -

- 🌐 Drag free performance : $3 \times 10^{-15} \text{ m.s}^{-2}/\sqrt{\text{Hz}}$
 - 🌐 DC value: $\sim 10^{-11} \text{ m.s}^{-2}$
 - 🌐 Validated with LISA Pathfinder !
- 🌐 6 laser links, 2.5 Mkm
 - 🌐 Measurement bandpass : [0.1 mHz : 1 Hz]
- 🌐 Telescopes:
 - 🌐 $\sim 30 \text{ cm}$ diameter,
 - 🌐 Pathlength stability: $\sim 1 \text{ pm}/\sqrt{\text{Hz}}$
- 🌐 Laser
 - 🌐 Nd:YAG (1064 nm), 1.8 W emitted (received $\sim 400 \text{ pW}$)
 - 🌐 RIN : $< 10^{-8} /\sqrt{\text{Hz}}$ above 5 MHz
 - 🌐 $\sim 100 \text{ Hz}/\sqrt{\text{Hz}}$
- 🌐 Timing jitter in clock distribution: $\sim 4 \times 10^{-14} \text{ s}/\sqrt{\text{Hz}}$
- 🌐 Absolute ranging accuracy: $\sim 1 \text{ m}$
- 🌐 Thermal stability (optical bench): $< 10 \text{ } \mu\text{K}/\sqrt{\text{Hz}}$ at 1 mHz
- 🌐 Laser beam pointing jitter: $\sim 10 \text{ nrad}/\sqrt{\text{Hz}}$

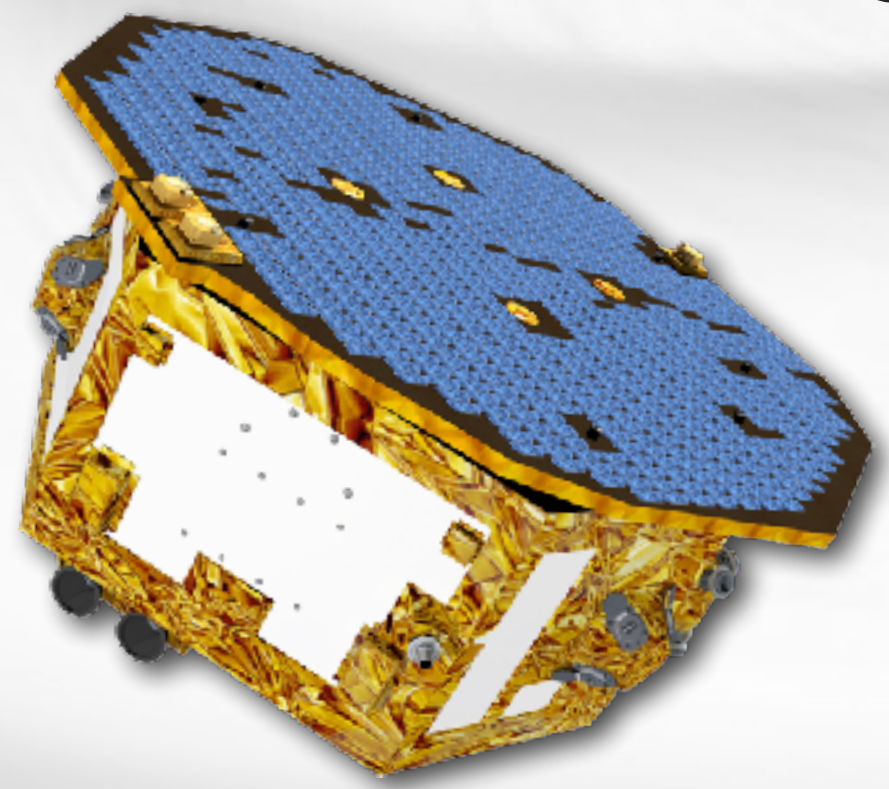






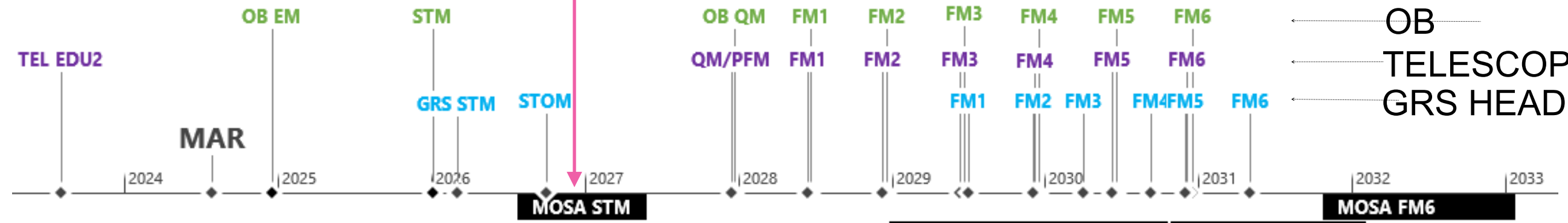
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Testing the instrument

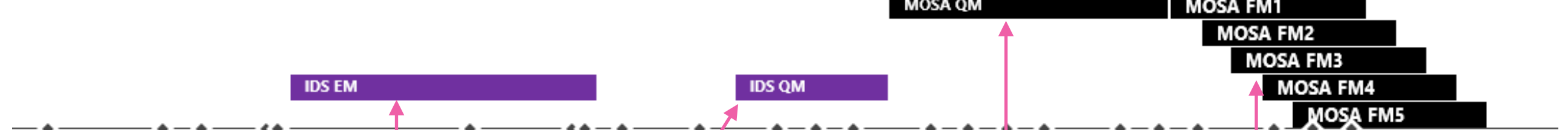


Full picture of MOSA alignment tests level (last update - 22/06/2021)

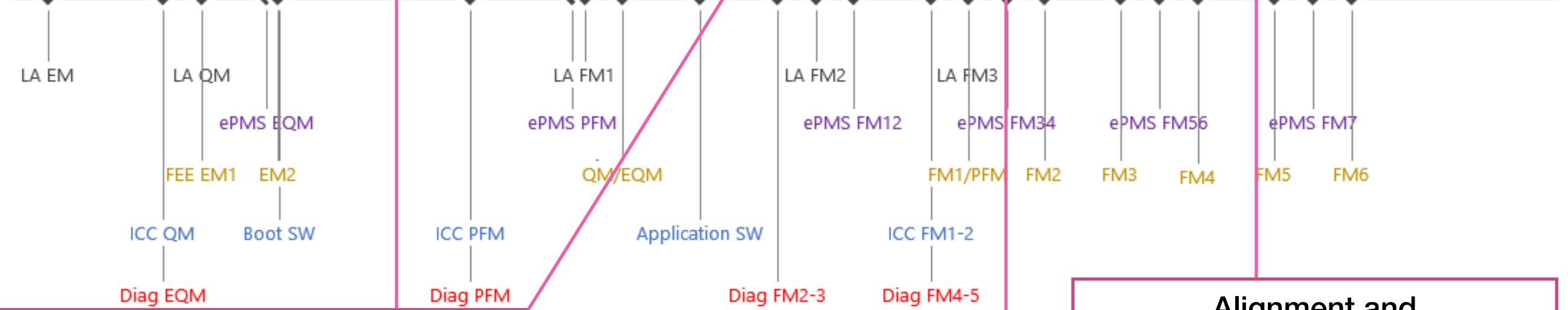
MOSA



AIVT



COMOSA

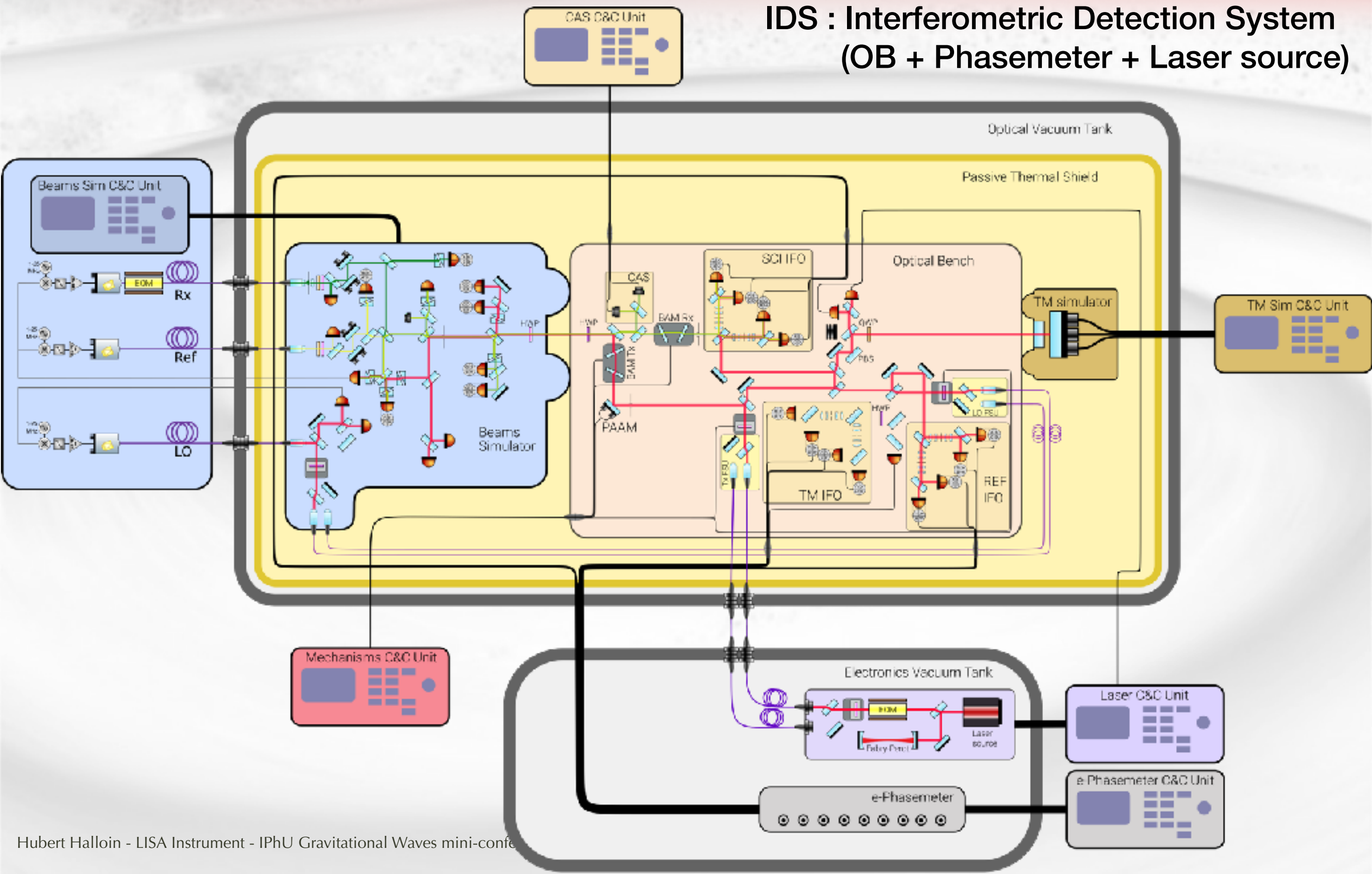


Metrology design validation (EM) and performance tests (QM) (OB, laser, Phasemeter)

Alignment and extended performance tests, before and after qualification campaigns (Tel, OB, GRS STOM)

Alignment and performance verification tests, before and after flight acceptance campaigns (Tel, OB, GRS STOM)

IDS : Interferometric Detection System
(OB + Phasemeter + Laser source)

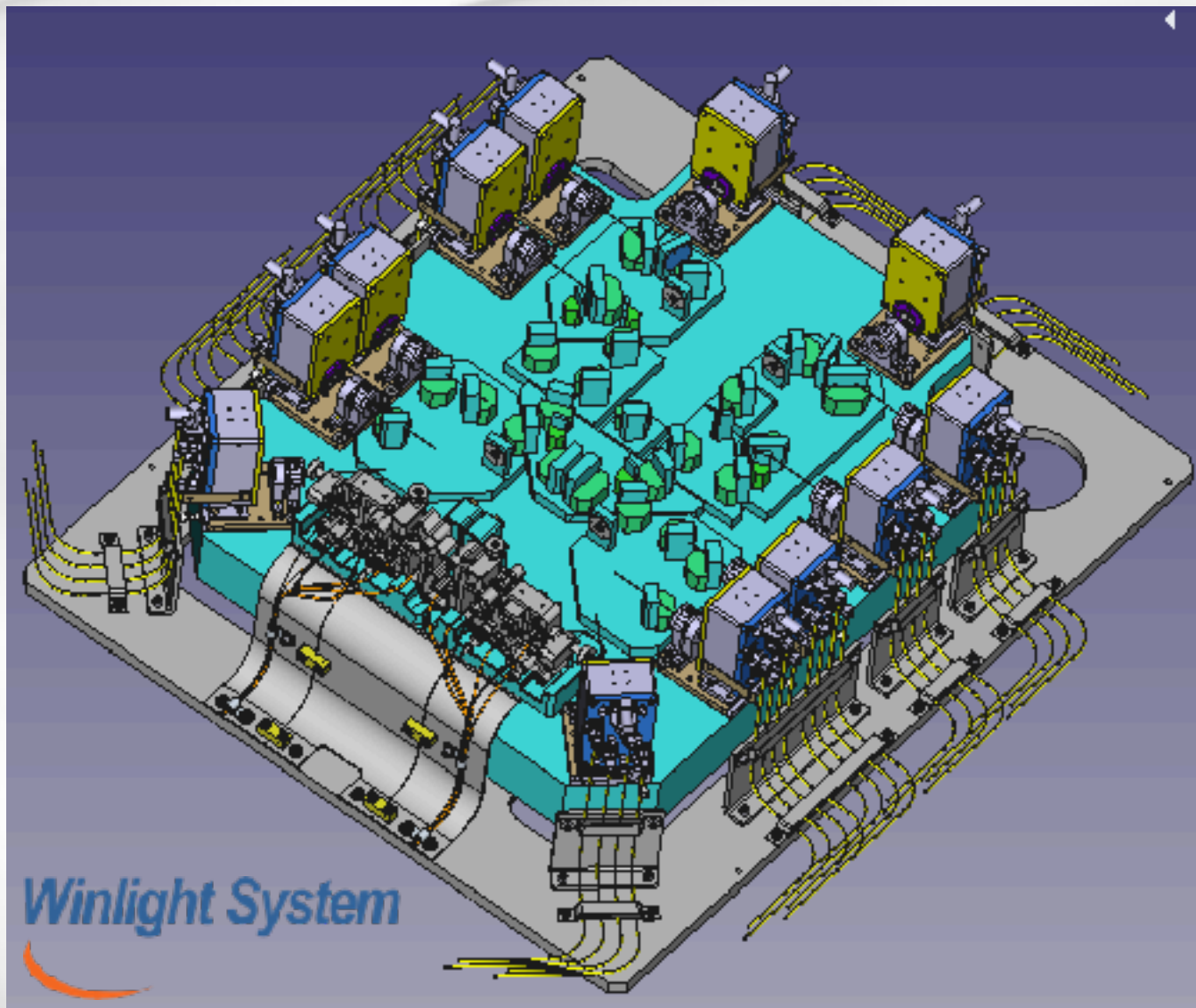




ZIFO



Zerodur base plate + optically contacted elements



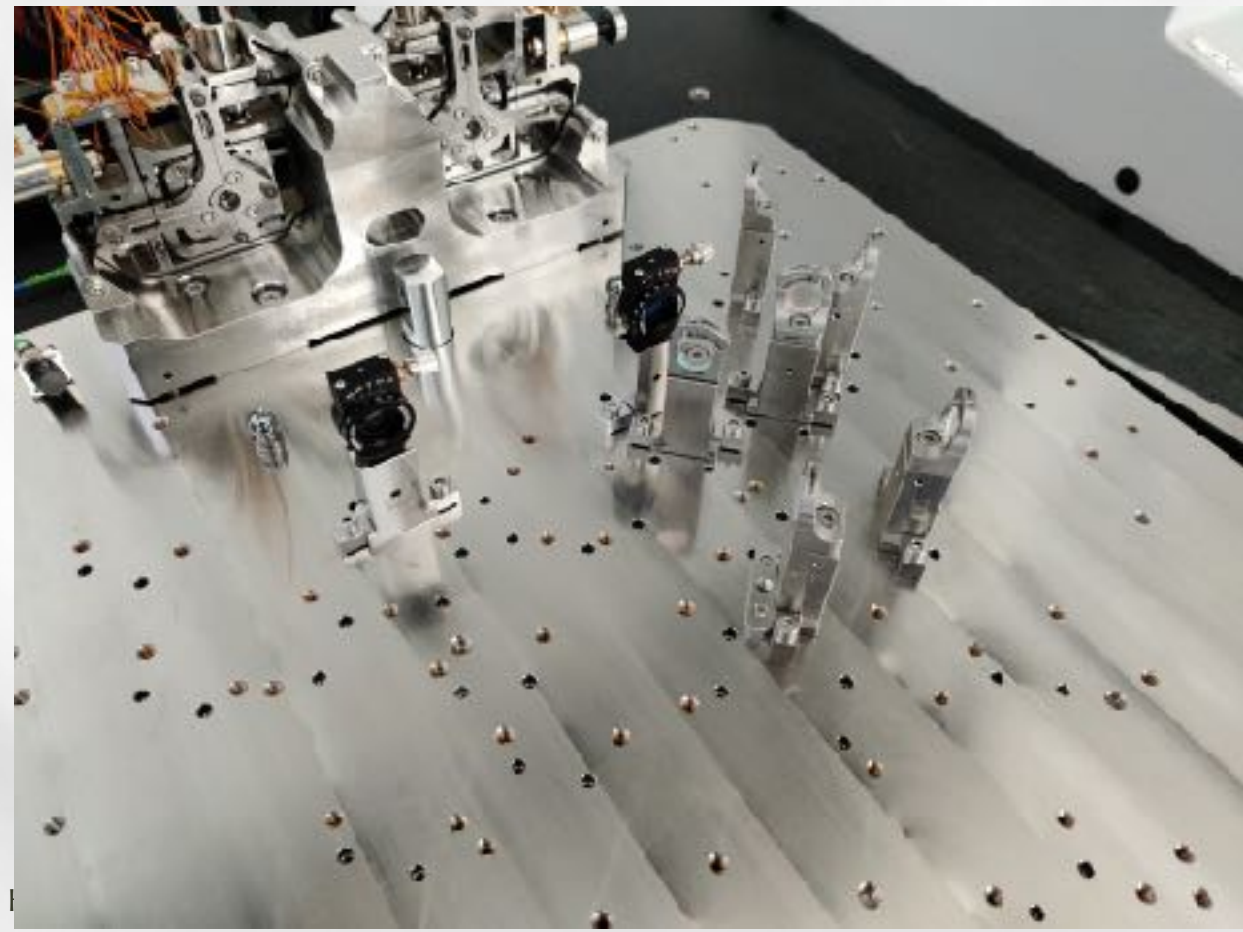
MIFO



Invar base plate + COTS optical mounts



Prototype Invar bench





Thank you !