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including in particular development from INREP group, in particular N. Quang Dam, Jean-Baptiste Bayle & Olaf Hartwig

> Cours LISA France Remote - 8 janvier 2021









- Introduction on LISA data processing
- Basic principle of TDI
- TDI with the current design
- TDI generators
- Propagation of noises and signal through TDI
- Laser locking
- Key ingredients of TDI
- New convention





'Survey' type observatory





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Phasemeters (carrier, sidebands, armlengths) + Gravitational Refe--rence Sensor + Auxiliary measurements

'Survey' type observatory





Phasemeters (carrier, sidebands, armlengths) + Gravitational Refe--rence Sensor

- Auxiliary measurements

'Survey' type observatory





Phasemeters (carrier, sidebands, armlengths) + Gravitational Refe--rence Sensor **Auxiliary measurements** 'Survey' type observatory



Calibrations

Resynchronisation (clocks)

Time-Delay Interferometry reduce laser noise

3 TDI time series



emitting between 0.02mHz and 1 Hz

Gravitational wave sources



Phasemeters (carrier, sidebands, armlengths) + Gravitational Refe--rence Sensor + Auxiliary measurements Survey type observatory Calibration (clocks) Resynchronisation (clocks) Time-Delay Interferometry reduce laser noise

Gravitational wave sources emitting between 0.02mHz and 1 Hz

3 TDI time series

GWs analysis

Catalogs of GW sources (posteriors & correlations), waveforms, etc

















• Generation rates:

System	Downlink rate [bit/s]
Phasemeter	5120,00
LA	64,00
DFACS	1088,00
GRS FEE	-
CGT	-
CMS	544,00
SciDiag	307,09
Housekeeping	4000,00
Total	11123,09

Generation volumes:

Volume	GB
Telemetered volume generated per day per S/C	0,20
Telemetered volume generated per day for the constellation	0,81
Onboard volume generated per day per S/C	2,79
Onboard Storage Required per S/C	39,12





LISA Ground Segment





LISA Ground Segment

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LISA Ground Segment

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LISA : Measurements

- Problem with 2.5x10⁹ m : A laser beam cannot make a round trip because too much intensity is lost.
 - 100pW received for 1 Watt emitted.
- Measurement with one arm
 and interference between
 two incoherent lasers in phase :
 - Distant laser
 - Local laser.

▶ 6 measurements ... at least!







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- Phase shift between the two beams measured by phasemeter.
- Beams from an external spacecraft, are delayed :
- delay operator D_i^{real} : • The measurement : $D_i^{real} x(t) = x \left(t - \frac{L_i^{real}}{c} \right)$.

 $s_1 = s_1^{GW} + s_1^{ShotNoise} + D_3^{real} p'_2^{lasernoise} - p_1^{lasernoise} - 2\delta^{Acc.Noise}$



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 $s_1 S_7 GW$

 $-s_1^{ShotNoise} + D_3^{real} p'_2^{lasernol}$





 $\mathbf{z} x \mid t$

 D_i^{rea}

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Pre-processing of the science data,

Tinto & Durandhar, Revue *Living Rev. Rel. 8 p 4* (2005) Durandhar, Nayak & Vinet, *PRD 65 102002* (2002)









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$$-D_{3}^{TDI} s'_{2}(t) = -D_{3}^{TDI} s'_{2} = p'_{2} \left(t - \frac{L_{3}^{TDI}}{c} \right)$$

$$s_{1}(t) = D_{3}^{real} p'_{2}(t) = p'_{2} \left(t - \frac{L_{3}^{real}}{c} \right)$$

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$$p'_{2}^{laser noise}$$

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$$s'_{2}$$

$$s'_{2}$$

$$s'_{2}$$





Co
 knowledge of delays : L^{TDI}_i = L^{real}_i
 interpolation due to the sampling of phasemeter signal

$$-D_{3}^{TDI} s'_{2}(t) = -D_{3}^{TDI} s'_{2} = p'_{2} \left(t - \frac{L_{3}^{TDI}}{c}\right)$$

$$s_{1}(t) + D_{3}^{TDI} s'_{2}(t)$$

$$= D_{3}^{real} p'_{2} - D_{3}^{TDI} p'_{2}$$

$$\simeq 0$$

$$s_{1}(t) + D_{3}^{TDI} s'_{2}(t)$$

$$= D_{3}^{real} p'_{2} - D_{3}^{TDI} p'_{2}$$

$$\simeq 0$$

g

► Pr





Time (s)



Time Delay Interferometry:

Tinto & Durandhar, Revue *Living Rev. Rel. 8 p 4* (2005) Durandhar, Nayak & Vinet, *PRD 65 102002* (2002) Vallisneri, *gr-qc/0504145* (2005)

- Combine delayed measurements to reduce laser noises, optical bench noises, ... ?
- Algebraic development: many combinations (generators)

 $X = -s_1 - D_3 s'_2 - D_3 D_{3'} s'_1 - D_3 D_{3'} D_{2'} s_3$ $+ s'_1 + D_{2'} s_3 - D_{2'} D_2 s_1 - D_{2'} D_2 D_3 s_{2'}$ $\simeq 0$

- Different precisions level
 - 1st generation: rigid formation of LISA : $D_{i^{\prime}}\,s=D_{i}\,s,$
 - generation 1.5: Sagnac effect : $D_{i'} s \neq D_i s$ but $D_j D_i s = D_i D_j s$,
 - 2nd generation: flexing and Sagnac effect : $D_j D_i s \neq D_i D_j s$







► TDI generation 1

 $X_{1st} = \left(1 - D_2^2, 0, -D_2 + D_2 D_3^2, -1 + D_3^2, D_3 - D_2^2 D_3, 0\right)$

► **TDI** generation 1.5 $X_{1.5} = (1 - D_2 D'_2, 0, -D'_2 + D'_2 D'_3 D_3, -1 + D'_3 D_3, D_3 - D_2 D'_2 D_3, 0)$

► **TDI 2nd generation: until 7 delay operators combined** $X_{2nd} = (1 + D_3D'_3D'_2D_2D'_2D_2 - D'_2D_2 - D'_2D_2D_3D'_3,$

0,

 $D_{3}D'_{3}D'_{2} + D_{3}D'_{3}D'_{2}D_{2}D'_{2} - D'_{2} - D'_{2}D_{2}D_{3}D'_{3}D_{3}D'_{3}D'_{2},$ $D_{3}D'_{3} + D_{3}D'_{3}D'_{2}D_{2} - 1 - D'_{2}D_{2}D_{3}D'_{3}D_{3}D'_{3},$ $D_{3} + D_{3}D'_{3}D'_{2}D_{2}D'_{2}D_{2}D_{3} - D'_{2}D_{2}D_{3} - D'_{2}D_{2}D_{3}D'_{3}D_{3},$ 0)





Reduction of laser noises by 8 orders of magnitude !

A GW is hidden here !



Phasemeter (cut off due to the filter required for digitalization of signal)

Petiteau & al, Phys. Rev. D (2008)



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Phasemeter (cut off due to the filter required for digitalization of signal)

TDI Michelson

Petiteau & al, Phys. Rev. D (2008)


- Exchange of laser beam to form several interferometers
- Phasemeter measurements on each of the 6 Optical Benches:
 - Distant OB vs local OB
 - Test-mass vs OB
 - Reference using adjacent OB
 - Transmission using sidebands
 - Distance between spacecrafts
- Noises sources:
 - Laser noise : 10⁻¹³ (vs 10⁻²¹)
 - Clock noise (3 clocks)
 - Acceleration noise (see LPF)
 - Read-out noises
 - Optical path noises





 Configuration and conventions (from Jean-Baptiste Bayle's PhD)

(WARNING: not following the new LISA Consortium conventions)









Long arm interferometer (IFO)





Test-Mass Interferometer (IFO)





Reference Interferometer (IFO)





► With clock noises: 3 Ultra-Stable Oscillator (USO)

- Pilot tone sampled by the ADC and subtracted from measurement to cancel ADC timing jitter
- USO used to generate pilot tone
- => USO noises in measurements as scaled by the relevant beat note frequency

$$\begin{split} s_{1} &= \theta_{1}^{s} \Big[H_{1} + \mathbf{D}_{3} p_{2'} - p_{1} - \left(\hat{\mathbf{n}}_{3} \cdot \mathbf{D}_{3} \frac{\mathbf{v}_{\Delta_{2'}}}{c} + \hat{\mathbf{n}}_{3'} \cdot \frac{\mathbf{v}_{\Delta_{1}}}{c} \right) - a_{1} q_{1} \Big] + N_{1}^{s} , \\ s_{1'} &= \theta_{1'}^{s} \Big[H_{1'} + \mathbf{D}_{2'} p_{3} - p_{1'} - \left(\hat{\mathbf{n}}_{2'} \cdot \mathbf{D}_{2'} \frac{\mathbf{v}_{\Delta_{3}}}{c} + \hat{\mathbf{n}}_{2} \cdot \frac{\mathbf{v}_{\Delta_{1'}}}{c} \right) - a_{1'} q_{1} \Big] + N_{1'}^{s} \\ \epsilon_{1} &= \theta_{1}^{\tau} \Big[p_{1'} - p_{1} + 2\hat{\mathbf{n}}_{3'} \cdot \left(\frac{\mathbf{v}_{\Delta_{1}}}{c} - \frac{\mathbf{v}_{\delta_{1}}}{c} \right) + \mu_{1'} - b_{1} q_{1} \Big] + N_{1}^{\epsilon} , \\ \epsilon_{1'} &= \theta_{1'}^{\tau} \Big[p_{1} - p_{1'} + 2\hat{\mathbf{n}}_{2} \cdot \left(\frac{\mathbf{v}_{\Delta_{1'}}}{c} - \frac{\mathbf{v}_{\delta_{1'}}}{c} \right) + \mu_{1} - b_{1'} q_{1} \Big] + N_{1'}^{\epsilon} , \\ \tau_{1} &= \theta_{1}^{\tau} \Big[p_{1'} - p_{1} + \mu_{1'} - b_{1} q_{1} \Big] + N_{1}^{\tau} , \\ \tau_{1'} &= \theta_{1'}^{\tau} \Big[p_{1} - p_{1'} + \mu_{1} - b_{1'} q_{1} \Big] + N_{1'}^{\tau} . \end{split}$$



 $\nu_1 - \nu_1$



Side-band measurements on science interferometers

• Amplification of clock noises on sidebands by factor m_i





Phasemetre

- Input data from quadrant photodiode at 80 MHz
- Heterodyne interferometry
- Phase Locked Loop
- Output: phase and/or relative frequency fluctuation at 30 Hz

Drag Free Attitude Control System

- Use the output of the phasemeter: TM IFO & beam angles
- To control the spacecraft motion and Test-Mass motion on all degrees of freedom except the sensitive one





On-board computer

- Anti-aliasing filtering of the phase meter data then downsampling around few Hz for telemetry (2.5, 3.2, 4 or 5 Hz)
- Strong contraints on the filter; example attenuation > 240 dB
- Impact on the data: small delays added

$$\begin{split} s_{1} &= \mathcal{F}\Big\{\theta_{1}^{s}\Big[H_{1} + \mathbf{D}_{3}p_{2'} - p_{1} - \Big(\hat{\mathbf{n}}_{3}\cdot\mathbf{D}_{3}\frac{\mathbf{v}_{\Delta_{2'}}}{c} + \hat{\mathbf{n}}_{3'}\cdot\frac{\mathbf{v}_{\Delta_{1}}}{c}\Big) - a_{1}q_{1}\Big] + N_{1}^{s}\Big\}\\ s_{1'} &= \mathcal{F}\Big\{\theta_{1'}^{s}\Big[H_{1'} + \mathbf{D}_{2'}p_{3} - p_{1'} - \Big(\hat{\mathbf{n}}_{2'}\cdot\mathbf{D}_{2'}\frac{\mathbf{v}_{\Delta_{3}}}{c} + \hat{\mathbf{n}}_{2}\cdot\frac{\mathbf{v}_{\Delta_{1'}}}{c}\Big) - a_{1'}q_{1}\Big] + N_{1'}^{s}\Big\}\\ \epsilon_{1} &= \mathcal{F}\Big\{\theta_{1}^{\tau}\Big[p_{1'} - p_{1} + 2\hat{\mathbf{n}}_{3'}\cdot\Big(\frac{\mathbf{v}_{\Delta_{1}}}{c} - \frac{\mathbf{v}_{\delta_{1}}}{c}\Big) + \mu_{1'} - b_{1}q_{1}\Big] + N_{1}^{\epsilon}\Big\}\\ \epsilon_{1'} &= \mathcal{F}\Big\{\theta_{1'}^{\tau}\Big[p_{1} - p_{1'} + 2\hat{\mathbf{n}}_{2}\cdot\Big(\frac{\mathbf{v}_{\Delta_{1'}}}{c} - \frac{\mathbf{v}_{\delta_{1'}}}{c}\Big) + \mu_{1} - b_{1'}q_{1}\Big] + N_{1'}^{\epsilon}\Big\}\\ \tau_{1} &= \mathcal{F}\{\theta_{1}^{\tau}\Big[p_{1'} - p_{1} + \mu_{1'} - b_{1}q_{1}\Big] + N_{1}^{\tau}\}\\ \tau_{1'} &= \mathcal{F}\{\theta_{1'}^{\tau}\Big[p_{1} - p_{1'} + \mu_{1} - b_{1'}q_{1}\Big] + N_{1'}^{\tau}\Big\}\end{split}$$

Suppressing S/C jitter noise

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Suppressing S/C jitter noise

- TDI step 1: removing spacecraft jitter
 - Extract the spacecraft jitter by subtracting reference IFO from Test-Mass IFO
 - Then correcting the science IFO

$$egin{aligned} \xi_1 &= s_1 + heta_1^s heta_1^ au rac{\epsilon_1 - au_1}{2} + heta_1^s heta_{2'}^ au rac{\mathcal{D}_3(\epsilon_{2'} - au_{2'})}{2}\,, \ \xi_{1'} &= s_{1'} + heta_{1'}^s heta_{1'}^ au rac{\epsilon_{1'} - au_{1'}}{2} + heta_{1'}^s heta_3^ au rac{\mathcal{D}_{2'}(\epsilon_3 - au_3)}{2}\,, \end{aligned}$$

• Residual spacecraft jitter :

$$egin{aligned} &\xi_1^\Delta = heta_1^s (\mathcal{F} \mathbf{D}_3 - \mathcal{D}_3 \mathcal{F}) (\mathbf{\hat{n}}_3 \cdot \mathbf{v}_{\Delta_{2'}}) \,, \ &\xi_{1'}^\Delta = heta_{1'}^s (\mathcal{D}_{2'} \mathcal{F} - \mathcal{F} \mathbf{D}_{2'}) (\mathbf{\hat{n}}_{2'} \cdot \mathbf{v}_{\Delta_3}) \,. \end{aligned}$$

23

Suppressing laser noise

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Suppressing laser noise

• TDI step 2: removing half of the laser noise

• Combining reference IFOs and combining with the result of the previous TDI step

$$\eta_1 = heta_1^s \xi_1 + rac{ heta_{2'}^ au \mathcal{D}_3(au_{2'} + au_2)}{2} \,, \ \eta_{1'} = heta_{1'}^s \xi_{1'} + rac{ heta_{1'}^ au (au_{1'} + au_1)}{2} \,.$$

$$\begin{aligned} \tau_1 &= \mathcal{F}\{\theta_1^\tau[p_{1'} - p_1 + \mu_{1'} - b_1 q_1] + N_1^\tau\} \\ \tau_{1'} &= \mathcal{F}\{\theta_{1'}^\tau[p_1 - p_{1'} + \mu_1 - b_{1'} q_1] + N_{1'}^\tau\} \end{aligned}$$

=> Cancel all p_i

=> Situation equivalent of having one laser per spacecraft





Suppressing laser noise

- Reducing all laser noises
 - Several complex combinations
 - Can be seen as virtual interferometer



• TDI generation 1.5 takes into account the unequal arms

 $X_1 = \eta_{1'} + \mathcal{D}_{2'}\eta_3 + \mathcal{D}_{2'2}\eta_1 + \mathcal{D}_{2'23}\eta_{2'} - \eta_1 - \mathcal{D}_3\eta_{2'} - \mathcal{D}_{33'}\eta_{1'} - \mathcal{D}_{33'2'}\eta_3$

TDI generation 2 takes into account first order time evolution of arm length
 (3) (2)

$$egin{aligned} X_2 &= \eta_{1'} + \mathcal{D}_{2'}\eta_3 + \mathcal{D}_{2'2}\eta_1 - \mathcal{D}_{2'23}\eta_{2'} + \mathcal{D}_{2'233'}\eta_1 \ &+ \mathcal{D}_{2'233'3}\eta_{2'} + \mathcal{D}_{2'233'33'}\eta_{1'} + \mathcal{D}_{2'233'33'2'}\eta_3 \ &- \eta_1 - \mathcal{D}_3\eta_{2'} - \mathcal{D}_{33'}\eta_{1'} - \mathcal{D}_{33'2'}\eta_3 - \mathcal{D}_{33'2'2}\eta_1 \ &- \mathcal{D}_{33'2'22'}\eta_3 - \mathcal{D}_{33'2'22'2}\eta_1 - \mathcal{D}_{33'2'22'23}\eta_{2'} \,. \end{aligned}$$





Noises after TDI

 Assuming no clock noise and perfect ranging, laser noise and spacecraft jitter are reduced below other noises, i.e. acceleration and readout (here all IFO noises)





Flexing-filtering effect

• Coupling between filter and delay J.-B. Bayle et al., PRD (2019)

• => residual laser noise depend on the filter





Flexing-filtering effect

Analytic approximation (linear delay):

$$S_{X_2}(\omega) \approx 32S_p \omega^2 \sin^2(\omega L) \sin^2(2\omega L) \left(\dot{L_2}^2 + \dot{L_3}^2\right) K_F(\omega)$$

• with:

- L : travel time along arms
- S_p : pre-stabilised laser noise
- filter term:

$$K_{\mathcal{F}}(\omega) = 4f_s^{-2} \left| \sum_{k=1}^N k \alpha_{N+k} \sin\left(\frac{k\omega}{f_s}\right) \right|^2$$

- α_k : the 2 N + 1 filter coefficients of the FIR filter
- Filter output are

$$y_n = \sum_{k=0}^{2N} lpha_k x_{n-k}$$

$(k\omega)\Big|^2$

J.-B. Bayle et al., PRD (2019)

Suppressing clock noise

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Suppressing clock noise

Hartwig et Bayle, arXiv:2005.02430

• Use the sideband signal where the clock noise is amplified to build a calibration signal similar to TDI but clock noise only then subtract it from the standard TDI to form the corrected TDI.

$$\text{TDI}^{c} = \text{TDI} - \left(\sum_{i=1}^{3} -(a_{i} + b_{(i+1)'})R_{i} - (a_{i'} - b_{i'})R_{i'}\right) \text{ with } r_{1'} = \theta_{1'}^{s} \frac{s_{1} - s_{1}^{\text{sb}}}{\nu_{2'}^{m}},$$
$$r_{1'} = \theta_{1'}^{s} \frac{s_{1'} - s_{1'}^{\text{sb}}}{\nu_{3}^{m}},$$

R_i and R_i' combination of r_i and r_i' such that R_i = P_iq_i & R_{i'} = P_{i'}q_i with P_i and P_i' polynomial of delays.
▶ Noise level:

$$\begin{split} S_{X_1^c,\mathcal{F}}(\omega) &\approx 4\omega^2((A_1^2 + b_{1'}^2)\dot{L}_2^2 + a_1^2\dot{L}_3^2) \\ &\times \sin^2(\omega L)K_f(\omega)S_q(\omega) , \\ & \times \sin^2(\omega L)K_f(\omega)S_q(\omega) , \\ \end{split}$$

31



Suppressing modulation noise

 M_i are additional deviations present in the sidebands relative to that same pilot tone, e.g., due to noise added by the electro-optical modulators or the fibre amplifiers.

$$egin{aligned} s_1^{ ext{sb}} &= \mathcal{F}ig[heta_1^sig(
u_{2'}^m\mathcal{D}_3(q_2+M_{2'}) \ &-
u_1^m(q_1+M_1) \ &-(a_1+
u_{2'}^m-
u_1^m)q_1ig)+N_1^{ ext{sb}}ig]\,,\ s_{1'}^{ ext{sb}} &= \mathcal{F}ig[heta_{1'}^sig(
u_3^m\mathcal{D}_{2'}(q_3+M_3) \ &-
u_{1'}^m(q_1+M_{1'}) \ &-(a_{1'}+
u_3^m-
u_{1'}^mig)q_1ig)+N_{1'}^{ ext{sb}}ig]\,. \end{aligned}$$

The EOM imprinting the clock noise on side band are working at different frequency on primed and unprimed OB, for example M_i at 2.4 GHz and M_i['] at 2.401 GHz.





 $\mathcal{D}_3 \Delta M_2$

Suppressing modulation noise

Hartwig et Bayle, arXiv:2005.02430

To correct at the first order the modulation error, we form new quantities:

$$\Delta M_{1} = \theta_{1'}^{\tau} \left[\frac{\tau_{1}^{\text{sb}} - \tau_{1}}{2} + \frac{\tau_{1'}^{\text{sb}} - \tau_{1'}}{2} \right] \qquad r_{1} = \theta_{1}^{s} \frac{s_{1} - s_{1}^{s_{0}}}{\nu_{2'}^{m}}, \qquad r_{1}^{c} = r_{1}$$
$$r_{1'} = \theta_{1'}^{s} \frac{s_{1'} - s_{1'}^{\text{sb}}}{\nu_{3}^{m}}. \qquad r_{1'}^{c} = r_{1}$$

Then subtract a polynomial combination to the standard TDI to form the corrected TDI

$$TDI^{c} = TDI - \left(\sum_{i=1}^{3} b_{(i+1)'} P_{i} r_{i}^{c} - (a_{i} + b_{(i+1)'}) R_{i} - (a_{i'} - b_{i'}) R_{i'}\right)$$

Noise level:

$$S_{X_1^M}(\omega) \approx 4\sin^2(\omega L) \left| \tilde{f}(\omega) \right|^2 S_M(\omega)$$

 $\times \left[(a_1 - a_{1'})^2 + a_{2'}^2 + a_3^2 + 4b_{1'}(a_1 - a_{1'} + b_{1'}) \sin^2(\omega L) \right]$

$$S_{X_2^M}(\omega) \approx 4 \sin^2(2\omega L) S_{X_1^M}(\omega) ,$$

with
 $\left| \tilde{f}(\omega) \right|^2 = \left| \sum_{k=0}^{2N} \alpha_k \exp^{-jk\omega/f_s} \right|^2 J_{anv}$

_sb





Suppressing clock noise

Hartwig et Bayle, arXiv:2005.02430

Results: clock noise well suppressed but modulation residual

not negligible



TTL correction

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TTL correction

- TTL: coupling between rotation of beam with respect to the optical system and longitudinal phase measured by IFO
- See Hubert's slides
- Subtraction of correlation with pointing noise measured with
 DWS => done on ground after TDI
- Under development
- Preliminary noise budget existing (see dedicated TN by Ewan Fitzimons and performance TN)





- With 6 links, there is a large numbers of possible TDI combinations suppressing laser noise: generators
- ► X, Y, Z: Michelson equivalent

 $\bullet \alpha, \beta, \gamma$: Sagnac

$$\alpha_1 = \eta_1 - \eta_{1'} + \mathcal{D}_3 \eta_2 - \mathcal{D}_{2'1'} \eta_{2'} + \mathcal{D}_{31} \eta_3 - \mathcal{D}_{2'} \eta_{3'}$$

$$\begin{aligned} \alpha_2 &= (1 - \mathcal{D}_{2'1'3'})\eta_1 - (1 - \mathcal{D}_{312})\eta_{1'} + (1 - \mathcal{D}_{2'1'3'})\mathcal{D}_3\eta_2 \\ &- (1 - \mathcal{D}_{312})\mathcal{D}_{2'1'}\eta_{2'} + (1 - \mathcal{D}_{2'1'3'})\mathcal{D}_{31}\eta_3 - (1 - \mathcal{D}_{312})\mathcal{D}_{2'}\eta_{3'} \end{aligned}$$

 $\boldsymbol{\xi}$: fully symmetric Sagnac: senses the constellation rotation,

& combines all interferometric signals with exactly one delay

$$\zeta_1 = \mathcal{D}_1 \eta_1 + \mathcal{D}_2 \eta_2 + \mathcal{D}_3 \eta_3 - \mathcal{D}_{1'} \eta_{1'} - \mathcal{D}_{2'} \eta_{2'} - \mathcal{D}_{3'} \eta_{3'}$$

$$\begin{split} \zeta_2 = & (\mathcal{D}_{11'} - \mathcal{D}_{2'3'1'})\eta_1 + (\mathcal{D}_{1'2'} - \mathcal{D}_{322'})\eta_2 \\ &+ (\mathcal{D}_{13} - \mathcal{D}_{2'3'3})\eta_3 - (\mathcal{D}_{1'1} - \mathcal{D}_{321})\eta_{1'} \\ &- (\mathcal{D}_{1'2'} - \mathcal{D}_{322'})\eta_{2'} - (\mathcal{D}_{13} - \mathcal{D}_{2'3'3})\eta_{3'} \end{split}$$
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Specific generators when 2 links are missing



• TDI 1.5:

$$\begin{split} P_1 &= -(\mathcal{D}_2 - \mathcal{D}_{3'1'})\eta_2 + (\mathcal{D}_2 - \mathcal{D}_{11'2})\eta_{2'} - (\mathcal{D}_{3'} - \mathcal{D}_{11'3'})\eta_3 + (\mathcal{D}_{3'} - \mathcal{D}_{21})\eta_{3'} \,, \\ E_1 &= -(1 - \mathcal{D}_{11'})\eta_1 + (1 - \mathcal{D}_{11'})\eta_{1'} - (\mathcal{D}_3 - \mathcal{D}_{2'1'})\eta_2 + (\mathcal{D}_{2'} - \mathcal{D}_{31})\eta_{3'} \,, \\ U_1 &= -(\mathcal{D}_{3'} - \mathcal{D}_{11'3'})\eta_{1'} + (1 - \mathcal{D}_{3'2'1'})\eta_2 - (1 - \mathcal{D}_{11'})\eta_{2'} + (\mathcal{D}_1 - \mathcal{D}_{3'2'})\eta_{3'} \,. \end{split}$$

• TDI 2.0 in Hartwig et Bayle, arXiv:2005.02430





Noise uncorrelated generators:

• Form a basis of TDI generators by diagonalising the noise correlation matrix

(S_{XX}	S_{XY}	S_{XZ}	
	S_{XY}	S_{YY}	S_{YZ}	
	S_{XZ}	S_{YZ}	S_{ZZ}	

• If all the noise of the same type are at the same level, identical PSD, $S_{XX} = S_{YY} = S_{ZZ}$, and CSD, $S_{XY} = S_{XZ} = S_{YZ}$

$\int S$	S_{XX}	S_{XY}	S_{XY})
5	S_{XY}	S_{XX}	S_{XY}	
15	S_{XY}	S_{XY}	S_{XX}	J

• Diagonalisation => generator A, E and T.





- Several possibilities for A, E, T
- Standard one, convention in the Consortium, used in LDC

$$E = \frac{X - 2Y + Z}{\sqrt{6}}, \ A = \frac{Z - X}{\sqrt{2}}, \ T = \frac{X + Y + Z}{\sqrt{3}}$$

- Noise level:
 - $S_A = S_E = S_{XX} S_{XY}$
 - $S_T = S_{XX} + 2 S_{XY}$
- ► Response to GW:
 - T very weak
 - A, E => main generators used for science





Noises & TDI

Noise budget to be done at the TDI level



Input of science analysis :

- Data analysis
- Figures of Merits



Noises & TDI





Sensitivity





Response of the detector to GWs



TDI transfert function for noises

- Study the propagation of acceleration and noises through TDI
- Others noises neglected or assumed to be cancelled by TDI
- 4 interferometric measurements in spacecraft 1 (simplified version)

$$\begin{cases} s_{1}^{c} = \theta_{1}^{2'} N_{1} \\ \tau_{1} = 0 \\ \epsilon_{1} = 2\theta_{1}^{1'} \delta_{1} \\ s_{1}^{sb} = \theta_{1}^{2'} N_{1} \end{cases} \begin{cases} s_{1'}^{c} = \theta_{1}^{3} N_{1'} \\ \tau_{1'} = 0 \\ \epsilon_{1'} = -2\theta_{1}^{1} \delta_{1'} \\ s_{1'}^{sb} = -2\theta_{1'}^{1} \delta_{1'} \\ s_{1'}^{sb} = -2\theta_{1'}^{1} \delta_{1'} \end{cases}$$





Suppressing spacecraft jitter noise

$$\begin{split} \xi_1 &= s_1^c - \theta_1^{2'} \theta_1^{1'} \frac{\epsilon_1 - \tau_1}{2} - \theta_1^{2'} \theta_{2'}^2 \frac{D_3 \epsilon_{2'}(t) - D_3 \tau_{2'}}{2} \\ &= \theta_1^{2'} \left(N_1 - \delta_1 + D_3 \delta_{2'} \right) \\ \xi_{1'} &= s_{1'}^c - \theta_{1'}^3 \theta_{1'}^1 \frac{\epsilon_{1'}(t) - \tau_{1'}(t)}{2} - \theta_{1'}^3 \theta_{3'}^{3'} \frac{D_{2'} \epsilon_3 - D_{2'} \tau_3}{2} \\ &= \theta_{1'}^3 \left(N_{1'} + \delta_{1'} - D_{2'} \delta_3 \right) \end{split}$$

Suppressing half of the laser noise:

$$\begin{split} \eta_1 &= \theta_1^{2'} \xi_1 + \frac{\theta_{2'}^2 D_3 \tau_{2'} - \theta_2^{2'} D_3 \tau_2}{2} \\ &= N_1 - \delta_1 + D_3 \delta_{2'} \\ \eta_{1'} &= \theta_{1'}^3 \xi_{1'} - \frac{\theta_{1'}^1 \tau_{1'} - \theta_1^{1'} \tau_1}{2} \\ &= N_{1'} + \delta_{1'} - D_{2'} \delta_3 \end{split}$$



TDI transfert function for noises

► TDI X 1.5 (time domain)

$X_{1.5}$	=	$\eta_{1'} + D_{2'}\eta_3 + D_{2'}D_2\eta_1 + D_{2'}D_2D_3\eta_{2'} - \eta_1 - D_3\eta_{2'} - D_3D_{3'}\eta_{1'} - D_3D_{3'}D_{2'}\eta_3$
	=	$N_{1'}+\delta_{1'}-D_{2'}\delta_3$
		$+D_{2'}N_3 - D_{2'}\delta_3 + D_{2'}D_2\delta_{1'}$
		$+D_{2'}D_2N_1 - D_{2'}D_2\delta_1 + D_{2'}D_2D_3\delta_{2'}$
		$+D_{2'}D_2D_3N_{2'}+D_{2'}D_2D_3\delta_{2'}-D_{2'}D_2D_3D_{3'}\delta_1$
		$-N_1+\delta_1-D_3\delta_{2'}$
		$-D_3N_{2'} - D_3\delta_{2'} + D_3D_{3'}\delta_1$
		$-D_3 D_{3'} N_{1'} - D_3 D_{3'} \delta_{1'} + D_3 D_{3'} D_{2'} \delta_3$
		$-D_3 D_{3'} D_{2'} N_3 + D_3 D_{3'} D_{2'} \delta_3 - D_3 D_{3'} D_{2'} D_2 \delta_{1'}$
$X_{1.5}$	=	$(1 - D_3 D_{3'})N_{1'} - (1 - D_{2'} D_2)N_1 + (D_{2'} - D_3 D_{3'} D_{2'})N_3 - (D_3 - D_{2'} D_2 D_3)N_{2'}$
		$+(1+D_3D_{3'}-D_{2'}D_2-D_{2'}D_2D_3D_{3'})\delta_1+(1-D_3D_{3'}+D_{2'}D_2-D_3D_{3'}D_{2'}D_2)\delta_{1'}$
		$-2(D_3 - D_{2'}D_2D_3)\delta_{2'} - 2(D_{2'} - D_3D_{3'}D_{2'})\delta_3$



TDI transfert function for noises

Computation of the PSD:

- OMS terms $PSD [(1 - D_{2'}D_2)N_1] = \left\langle \left(1 - e^{-i\omega(L_{2'} + L_2)}\right) \left(1 - e^{i\omega(L_{2'} + L_2)}\right) \tilde{N}_1 \tilde{N}_1^* \right\rangle$ $= \left\langle \left(e^{-i\omega\frac{L_{2'} + L_2}{2}} - e^{i\omega\frac{L_{2'} + L_2}{2}}\right) \tilde{N}_1 \tilde{N}_1^* \right\rangle$ $= 4 \sin^2 \left(\omega \frac{L_{2'} + L_2}{2}\right) S_{OMS_i}$
- Acceleration terms

$$PSD\left[(1 + D_{3}D_{3'} - D_{2'}D_{2} - D_{2'}D_{2}D_{3}D_{3'})\delta_{1}\right] = \left\langle \left(1 + e^{-i\omega(L_{3} + L_{3'})} - e^{-i\omega(L_{2'} + L_{2})} - e^{-i\omega(L_{2'} + L_{2} + L_{3} + L_{3'})}\right)(...)^{*} \tilde{\delta}_{1}\tilde{\delta}_{1}^{*}\right\rangle$$

$$= \left\langle \left(\left(1 + e^{-i\omega(L_{3} + L_{3'})}\right)\left(1 - e^{-i\omega(L_{2'} + L_{2})}\right)\right)(...)^{*} \tilde{\delta}_{1}\tilde{\delta}_{1}^{*}\right\rangle$$

$$= \left\langle \left(e^{-i\omega\frac{L_{3} + L_{3'}}{2}} + e^{i\omega\frac{L_{3} + L_{3'}}{2}}\right)^{2} \left(-e^{-i\omega\frac{L_{2'} + L_{2}}{2}} + e^{i\omega\frac{L_{2'} + L_{2}}{2}}\right)^{2} \delta_{1}\delta_{1}^{*}\right\rangle$$

$$= 16\cos^{2}\left(\omega\frac{L_{3} + L_{3'}}{2}\right)\sin^{2}\left(\omega\frac{L_{2'} + L_{2}}{2}\right)S_{acc_{i}}$$
(97)



Computation of PSD:

$$\begin{split} PSD_{X_{1.5}} &= 4\sin^2\left(\omega\frac{L_2+L_{2'}}{2}\right) \left(S_{OMS_1}+S_{OMS_{2'}}+4S_{acc_{2'}}+4\cos^2\left(\omega\frac{L_3+L_{3'}}{2}\right)S_{acc_1}\right) \\ &+4\sin^2\left(\omega\frac{L_{3'}+L_3}{2}\right) \left(S_{OMS_{1'}}+S_{OMS_3}+4S_{acc_3}+4\cos^2\left(\omega\frac{L_2+L_{2'}}{2}\right)S_{acc_{1'}}\right) \end{split}$$

Approx. : All armlength equal

$PSD_{X_{1.5}} = 4\sin^2(\omega L) \left(S_{OMS_1} + S_{OMS_{2'}} + S_{OMS_{1'}} + S_{OMS_3} + 4 \left(S_{acc_{2'}} + S_{acc_3} + \cos^2(\omega L) \left(S_{acc_1} + S_{acc_{1'}} \right) \right) \right)$

Approx. : All noises of the same type have the same PSD

 $PSD_{X_{1.5}} = 16 \sin^2(\omega L) (S_{OMS} + (3 + \cos(2\omega L)) S_{acc})$


TDI transfert function for noises

► TDI X 1.5:

$PSD_{X_{1.5}} = 16 \sin^2(\omega L) (S_{OMS} + (3 + \cos(2\omega L)) S_{acc})$





TDI transfert function for noises

► TDI X 2.0:

$PSD_{X_{2.0}} = 64 \sin^2(\omega L) \sin^2(2\omega L) (S_{OMS} + (3 + \cos(2\omega L)) S_{acc})$





Model:

• multiple beams. Example of MOSA 1:

$$\begin{split} b_{sc,2'\to1} &= \mathbf{D}_3 \left[p_{2'} + N_{TX,2'}^{\mathrm{op}} - k_{2'} \hat{\mathbf{n}}_3. \frac{[K \ \vec{\delta}_{2'} + \vec{\Delta}_{2'}]}{1 + K} + \frac{K}{1 + K} N_{2'}^{sens} \right] \\ &+ ds_3 + gw_3 - k_{2'} \hat{\mathbf{n}}_{3'}. \frac{[K \ \vec{\delta}_1 + \vec{\Delta}_1]}{1 + K} + \frac{K}{1 + K} N_1^{sens} + N_{RX,1}^{\mathrm{op}} \right] \\ b_{TM,1'\to1} &= p_{1'} + \mu_{1'\to1} + N_{TM,1}^{\mathrm{op}} \\ b_{ref,1'\to1} &= p_1 + \mu_{1'\to1} + N_{ref,1}^{\mathrm{op}} \\ b_{sc,1\to1} &= p_1 + N_{loc\to sc,1}^{\mathrm{op}} \\ b_{TM,1\to1} &= p_1 + 2k_1 \hat{\mathbf{n}}_{3'}. \frac{(\vec{\Delta}_1 - \vec{\delta}_1)}{1 + K} + N_{loc\to TM,1}^{\mathrm{op}} \\ b_{ref,1\to1} &= p_1 + N_{loc\to ref,1}^{\mathrm{op}} \end{split}$$

Model:

• Measurements. Example of MOSA 1:

$$\begin{split} s_{1}^{c} &= \mathcal{F}\left[\theta_{1}^{2'}\left(b_{sc,2'\to1}-b_{sc,1\to1}\right)+N_{sc,1}^{ro}\right] \\ \epsilon_{1} &= \mathcal{F}\left[\theta_{1}^{1'}\left(b_{TM,1'\to1}-b_{TM,1\to1}\right)+N_{\epsilon,1}^{ro}\right] \\ \tau_{1} &= \mathcal{F}\left[\theta_{1}^{1'}\left(b_{ref,1'\to1}-b_{ref,1\to1}\right)+N_{\tau,1}^{ro}\right] \\ s_{1}^{sb} &= \mathcal{F}\left[\theta_{1}^{2'}\left(b_{2'\to1}+\mathbf{D}_{3}[m_{3}q_{3}]-b_{1\to1}-m_{1}q_{1}\right)+N_{s^{sb},1}^{ro}\right] \end{split}$$



- Doing the same procedure as the one described before, we get transfer function TDI X2.0 for all noises sources
- To be used in performance model (see Joseph's talk)
- Approximations:
 - There is no coupling between S/C and DFACS (two independent bodies): $\mathsf{K}=\mathsf{0}$
 - All armlengths of the constellation are equal.
 - All noises of the same type have the same PSD, i.e

 $PSD\{p_{12}\} = PSD\{p_{21}\} = PSD\{p_{13}\} = PSD\{p_{31}\} = PSD\{p_{32}\} = PSD\{p_{23}\} = S_p$

• The laser nominal frequencies are almost constant and equal, so are the laser wavelengths.



for unsuppressed noise

Noise category	Noise symbol	Transfer function in PSD
Test-mass acceleration noise \checkmark	δ	$4C(\omega)\left(3+\cos\left(2\omega\frac{L}{c}\right)\right)$
Backlink fiber noise 🗸	μ	$2C(\omega)\sin^2(\omegarac{L}{c})$
Science and reference IFO noise \checkmark	$N_{sci}^{ro}, N_{RX}^{op}, N_{TX}^{op},$	$4C(\omega)$
	$N_{ref}^{op}, N_{loc ightarrow sci}^{op}, N_{loc ightarrow ref}^{op}$	
Test-mass IFO noise \checkmark	$N_{tm}^{op}, N_{loc \rightarrow tm}^{op}, N_{tm}^{ro}$	$C(\omega)\left(3+\cos\left(2\omega\frac{L}{c}\right)\right)$
Telescope noise	N_{tel}^{op}	$4C(\omega)\left(3+\cos\left(2\omega\frac{L}{c}\right)\right)$

with
$$C(\omega) = 16 \sin^2\left(\omega \frac{L}{c}\right) \sin^2\left(2\omega \frac{L}{c}\right)$$



for suppressed noise

Noise category	Noise symbol	Transfer function in PSD
Residual laser frequency noise \checkmark	p	$4C(\omega)\omega^2\left(\left(\frac{\dot{L}}{c}\right)^2 K_{\mathcal{F}}(\omega) + \left(\frac{\bar{\mu}}{c}\right)^2\right)$
Residual S/C translational jitter noise	Δ	$4C(\omega) \; \omega^2 \left(rac{\dot{L}}{c} ight)^2 K_{\mathcal{F}}(\omega)$
Residual clock jitter noise	q	$2C(\omega) \ \omega^2 \left(rac{\dot{L}}{c} ight)^2 K_{\mathcal{F}}(\omega)$
		$+C(\omega) \delta \nu_{BN}^2 \left(6 + 12 \sin^2\left(\omega \frac{L}{c}\right)\right)$
Residual modulation errors	m	$C(\omega) \left \tilde{f}(\omega) \right ^2 \left(6 + 12 \sin^2 \left(\omega \frac{L}{c} \right) \right)$
TTL		$4C(\omega)$

- \dot{L}_i , the time derivative of the armlength L_i (in m.s⁻¹);
- $\omega = 2\pi f;$
- $\bar{\mu}$, the average ranging bias (in m);
- ν_{BN} , the beatnote frequency (Hz);
- $\delta \nu_{BN}$, the fractional error in measured value of beatnote frequency
- $K_{\mathcal{F}}$ the delay-filter factor (in s²),

$$K_{\mathcal{F}}(\omega) = 4f_s^{-2} \left| \sum_{k=1}^N k \alpha_{N+k} \sin\left(\frac{k\omega}{f_s}\right) \right|^2$$

• $\left|\tilde{f}(\omega)\right|^2$, the filter response (dimensionless):

$$\left|\tilde{f}(\omega)\right|^2 = \left|\sum_{k=0}^{2N} \alpha_k \exp^{-jk\omega/f_s}\right|^2$$
(24)

- f_s , the sampling frequency (in Hz),
- α_k , the coefficients of the Finite Impulse Response filter (dimensionless),
- 2N + 1 the number of coefficients of the filter (dimensionless),

We can write the filter output y_n as a function of the past input samples x_{n+k} and 2N+1 coefficients α_k

$$y_n = \sum_{k=0}^{2N} \alpha_k x_{n-k} \tag{25}$$

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TDI transfert function for noises

• Correlations:

- If a noise source appears in several terms when doing combining measurements at TDI level correlation appears while it was not present in the measurement.
- Examples:
 - thermal noise presents on acceleration noise and OMS noise
 - telescope noise for emitted beam and received beam:

 $N_{TX,i}^{op}(t) = \alpha N_{RX,i}^{op}(t)$

$$\begin{split} PSD\left[XX_{1.5}^{\text{corr tels op}}\right] &= 16\left[1+\alpha^2+2\alpha\cos^2(\omega L)\right]\sin^2(\omega L)S_{N_{RX}^{op}}\\ PSD\left[YY_{1.5}^{\text{corr tels op}}\right] &= PSD\left[ZZ_{1.5}^{\text{corr tels op}}\right] = PSD\left[XX_{1.5}^{\text{corr tels op}}\right] \end{split}$$



GW signal



The single link response (the laser light emitted by "s"ender to the "r"eceiver) to GW is given as:

$$y_{rs}^{GW} = \frac{\Phi_{rs}(t - \hat{k}.\vec{R}_s - L_{rs}) - \Phi_{rs}(t - \hat{k}.\vec{R}_r)}{2(1 - \hat{k}.\hat{n}_{rs})}$$

$$\Phi_{rs} = \hat{n}_{rs} h_{ij} \hat{n}_{rs}$$

where:

- k is direction of GW propagation,
- R_{r,s} vector position of a sender/receiver

 h_{ij}^{SSB}

- n_{rs} unit vector of the link
- Source:
 - ecliptic latitude β and longitude λ
 - polarisation ψ
- ► Apply TDI on the y_{rs} directly

 $egin{array}{rcl} \epsilon^+_{ij}&=&(\hat{u}\otimes\hat{u}-\hat{v}\otimes\hat{v})_{ij}\ \epsilon^ imes_{ij}&=&(\hat{u}\otimes\hat{v}+\hat{v}\otimes\hat{u})_{ij} \end{array}$

$$egin{aligned} \hat{u} &= -\hat{e}_{\phi} \sim rac{\partial \hat{k}}{\partial \lambda} \ \hat{v} &= -\hat{e}_{ heta} \sim rac{\partial \hat{k}}{\partial eta} \end{aligned}$$

$$\hat{k} = -\hat{e}_r = -\{\cos\beta\cos\lambda,\cos\beta\sin\lambda,\sin\beta\}$$

 $= (h_+ \cos 2\psi - h_\times \sin 2\psi)\epsilon_{ii}^+ + (h_+ \sin 2\psi + h_\times \cos 2\psi)\epsilon_{ii}^+$



GW signal & TDI

- TDI on the y_{rs} : signal for a particular source (h_{ij}, β , λ , ψ)
- General TDI transfer function to GW: average over polarisation and sky. Complex to compute:
 - Semi-analytical computation

$$< R_L(f) >= \left(4\sin\left(\frac{2\pi fL}{c}\right)\right)^2 \left(\frac{L}{c}\right)^2 (2\pi f)^2 R_{\Sigma}(f,L)$$

with R_{Σ} : average response in phase for a standard Michelson. Semi-analytical computation with numerical integral (see Larson et al. 2000, PRD 62(6):062001)

- Numerical simulation using white noise stochastic background emitted from many sources regularly distributed over the sky
- Analytical computation in long wavelength approximation

$$R_{LW} = \frac{3}{20} (\omega L)^2 \sin^2(\omega L)$$

.021



GW signal & TDI





- In reality the laser are not 6 free running lasers but there is one main laser on which one the other are locked with offsets
- The time evolution of offset: frequency planning



Different locking schemes (TN from Gerhard Heinzel)

• Example

$$s_{2}^{\times} = os_{3' \rightarrow 2}$$

$$\tau_{3}^{\times} = os_{3' \rightarrow 3}$$

$$s_{1'}^{\times} = os_{3 \rightarrow 1'}$$

$$\tau_{1}^{\times} = os_{1' \rightarrow 1}$$

$$s_{2'}^{\times} = os_{1 \rightarrow 2'}$$



$$\begin{array}{rcl} \overbrace{s_{2'}}^{\swarrow} &=& os_{3' \rightarrow 2} \\ \tau_{3}^{\bigstar} &=& os_{3' \rightarrow 3} \\ \overbrace{s_{1'}}^{\bigstar} &=& os_{3 \rightarrow 1'} \\ \tau_{1}^{\bigstar} &=& os_{1' \rightarrow 1} \\ \overbrace{s_{2'}}^{\bigstar} &=& os_{1 \rightarrow 2'} \end{array}$$

Provide the following formulation for the 5 locked lasers:

$$\begin{array}{lll} p_{2} & = & \theta_{2}^{3'}(N_{s,2}^{ro} - os_{3' \rightarrow 2}) + b_{sc,3' \rightarrow 2} - N_{loc \rightarrow sc,2}^{op} \\ p_{3} & = & \theta_{3}^{3'}(N_{\tau,3}^{ro} - os_{3' \rightarrow 3}) + b_{ref,3' \rightarrow 3} - N_{loc \rightarrow ref,3}^{op} \\ p_{1'} & = & \theta_{1'}^{3}(N_{s,1'}^{ro} - os_{3 \rightarrow 1'}) + b_{sc,3 \rightarrow 1'} - N_{loc \rightarrow sc,1'}^{op} \\ p_{1} & = & \theta_{1}^{1'}(N_{\tau,1}^{ro} - os_{1' \rightarrow 1}) + b_{ref,1' \rightarrow 1} - N_{loc \rightarrow ref,1}^{op} \\ p_{2'} & = & \theta_{2'}^{1}(N_{s,2'}^{ro} - os_{1 \rightarrow 2'}) + b_{sc,1 \rightarrow 2'} - N_{loc \rightarrow sc,2'}^{op}. \end{array}$$



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- For unsuppressed noises (readout noise, optical path noise, acceleration noise, ...) => same PSD as for free running lasers
- For suppressed noise:
 - depend on the locking configuration
 - not the same for X, Y & Z
 - Work in progress via analytical formulation and simulation (locking implemented in last version of LISANode)



INREP: Analytical formulations

- TDI transfer functions without & with phase locking, PSD & CSD, TDI1.5 & TDI2.0:
 - Unsuppressed noises => done
 - Suppressed noises:
 - laser frequency noise => partially (flexing-filtering, armlength uncertainties bias & stochastic, interpolation)
 - Translational spacecraft jitter noise => done
 - Clock jitter noise => done
 - Clock correction residuals => partially (modulation, sideband readout noise)
 - TTL => partially



INREP: Analytical formulations

- TDI transfer functions without & with phase locking, PSD & CSD, TDI1.5 & TDI2.0:
 - Correlated noises:
 - Telescope => done
 - Backlink => done
 - correlation need to be identified first => on demand
- Cross-checking with LISANode:
 - 70% done for PSD
 - To do for CSD





Key ingredients of TDI

- Measurements
- Arm length knowledge
- Synchronisation of data
- ► No aliasing => onboard filtering
- Interpolation
- Mitigate the impact of artefacts, i.e. gaps and glitches



Armlength



- Knowledge of the absolute armlength is crucial
- Several combined technics:
 - Direct measurement: pseudo-random code imprinted on laser beam => precision at the meter level (TBC)
 - Kalman filter:
 - combine a dynamic model of the whole LISA constellation (orbits, PSD of clock jitters) and phasemeter raw measurements
 - => precision at cm level
 - **TDI-Ranging**: minimisation of noises in post-processing (data analysis) with armlength treated as parameters



INREP: Clock synchronisation

rance





Anti-aliasing filtering

- On-board sampling rate at few tens Hz (10Hz, 12.8Hz, 16Hz ?)
- ► Telemetry sampling rate at few Hz (2.5Hz, 3.2Hz, 4Hz, 5Hz ?)
 - => on-board filtering
- To avoid any aliasing residual at TDI level, strong reduction requested
 => typically > 240 dB
- Linear response => FIR
- Coupling with delay used in TDI (flexing -filtering)
- Computing cost





Interpolation

- Discrete input timeseries=> we need to have a very precise interpolation in order to get measurement between samples
- Which method ?

70

 we are using Lagrange interpolation with 32 points but other methods could be tested



Importance of the sampling



 $D_i x = D_i x(t) = x \left(t - \frac{L_i}{c} \right)$



Artefacts

- We are expecting artefacts in LISA data:
 - Gaps

• Glitches

- => Interaction of TDI with these artefacts have to be studied
- ► For example it could increase the gap size





Data analysis & TDI

- The input of GW search pipelines are TDI data:
 - X, Y, Z
 - A, E, T : assuming all noise level are the same, noise matrix is diagonal => likelihood computation is just a sum
- Most of the methods are using matched filtering => TDI has to be included in the template:
 - Computation of template have to be very fast
 - approximation in the TDI implementation
 - very fast TDI ?





TDI infinity

- Main idea: likelihood directly in terms of the basic phase measurements with marginalization over the laser phase noises in the limit of infinite laser-noise variance
- ► 2 studies:
 - Vallisneri et al. 2020 arXiv:2008.12343
 - Baghi et al. 2020 arXiv:2010.07224





New conventions

- Ongoing uniformisation of the convention in the LISA Consortium
- Spacecraft, indexed from 1
 to 3 clockwise looking down
 the z-axis, onto the solar
 panels
- MOSAs, indexed with two numbers ij



Spacecraft MOSA is mounted on spacecraft MOSA receives light from





New conventions

Measurements

$$\begin{cases} s_{12}^{c} = \theta_{12}^{21} N_{12} \\ \tau_{12} = 0 \\ \epsilon_{12} = 2\theta_{12}^{13} \delta_{12} \\ s_{12}^{sb} = \theta_{12}^{21} N_{12} \end{cases} \begin{cases} s_{13}^{c} = \theta_{13}^{31} N_{13} \\ \tau_{13} = 0 \\ \epsilon_{13} = -2\theta_{13}^{12} \delta_{13} \\ s_{13}^{sb} = \theta_{13}^{31} N_{13} \end{cases}$$

Advantage: natural chaining of indices when concatenating delays

 $X_{1.5} = \eta_{13} + D_{13}\eta_{31} + D_{13}D_{31}\eta_{12} + D_{13}D_{31}D_{12}\eta_{21} - \eta_{12} - D_{12}\eta_{21} - D_{12}D_{21}\eta_{13} - D_{12}D_{21}D_{13}\eta_{31}$

 $\begin{aligned} X_{2.0} &= \eta_{13} + D_{13}\eta_{31} + D_{13}D_{31}\eta_{12} + D_{13}D_{31}D_{12}\eta_{21} + D_{13}D_{31}D_{12}D_{21}\eta_{12} \\ &+ D_{13}D_{31}D_{12}D_{21}D_{12}\eta_{21} + D_{13}D_{31}D_{12}D_{21}D_{12}D_{21}\eta_{13} + D_{13}D_{31}D_{12}D_{21}D_{12}D_{21}D_{13}\eta_{31} \end{aligned}$

