





observing gravitational waves from space

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Outline



- Introduction to gravitational waves
- Gravitational wave sources in the millihertz regime
- LISA: a space-based gravitational wave observatory
- LISAPathfinder
- LISA status and organisation
- LISA scientific performances
- ► The French contribution to LISA:
 - Data Processing Center
 - Integration / performance control
- Conclusion and perspectives









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Some history ...

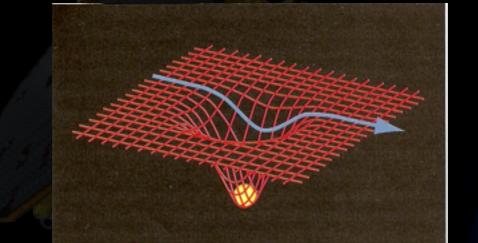
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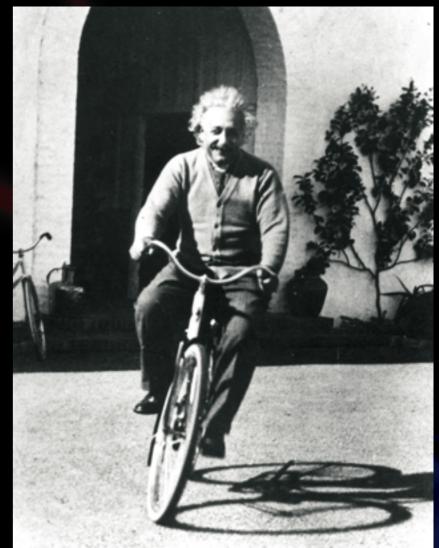
- Mass deforms geometry of space-time.
- Bodies are moving in a curve space.
- Gravitational information propagates at the speed of light.
- Dissipation of energy through deformation of space-time => gravitational waves

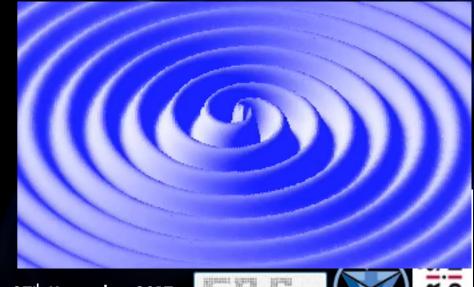
 $G_{\mu\nu} = \frac{8\pi G}{c^4} T_{\mu\nu}$

geometrical deformation

c⁴ distribution of energy









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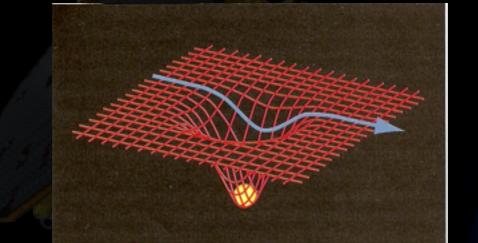
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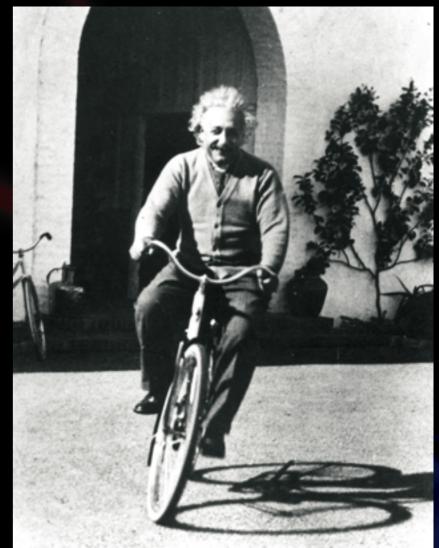
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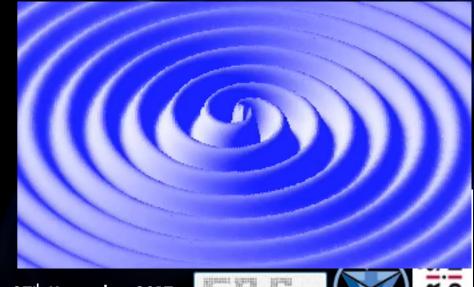
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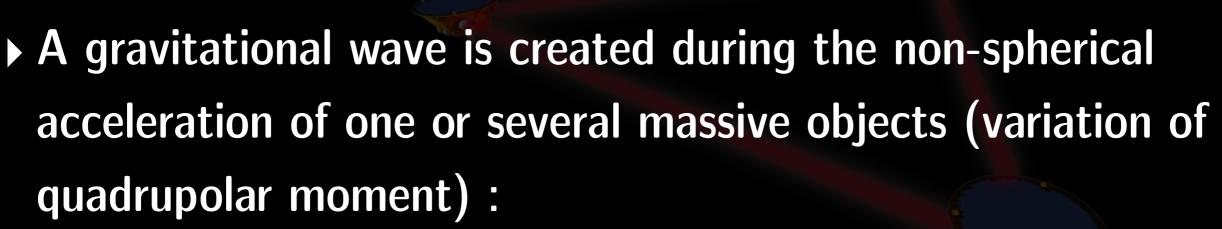




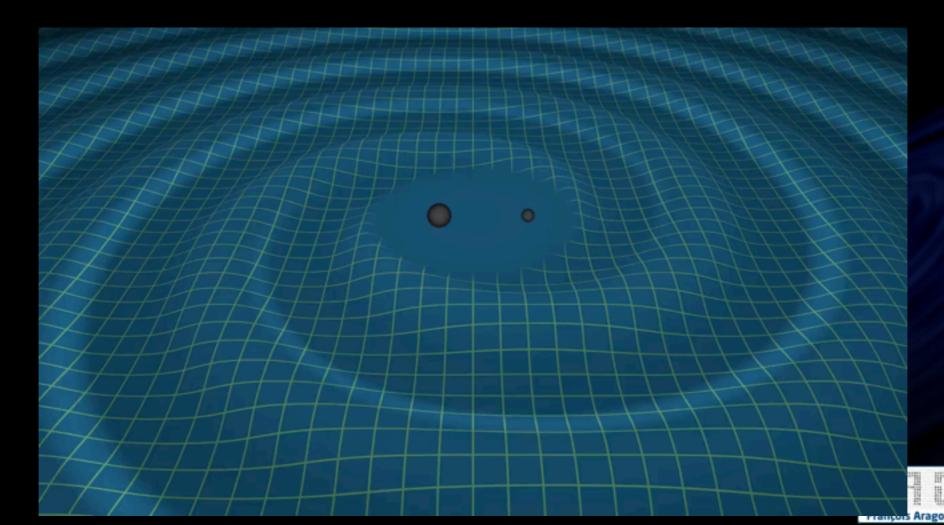








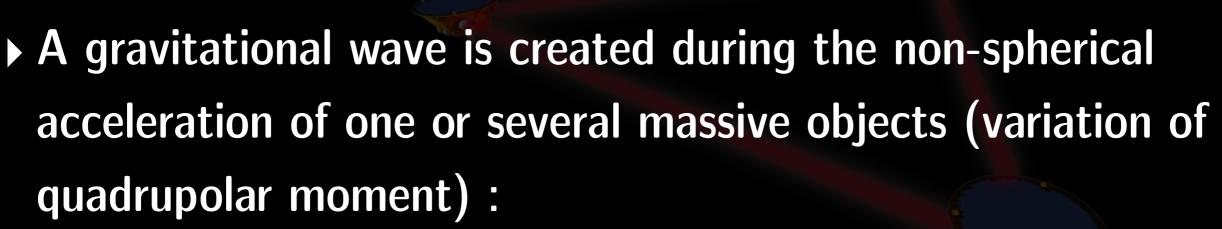
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- no emission: isolated, spherical body possibly in rotation



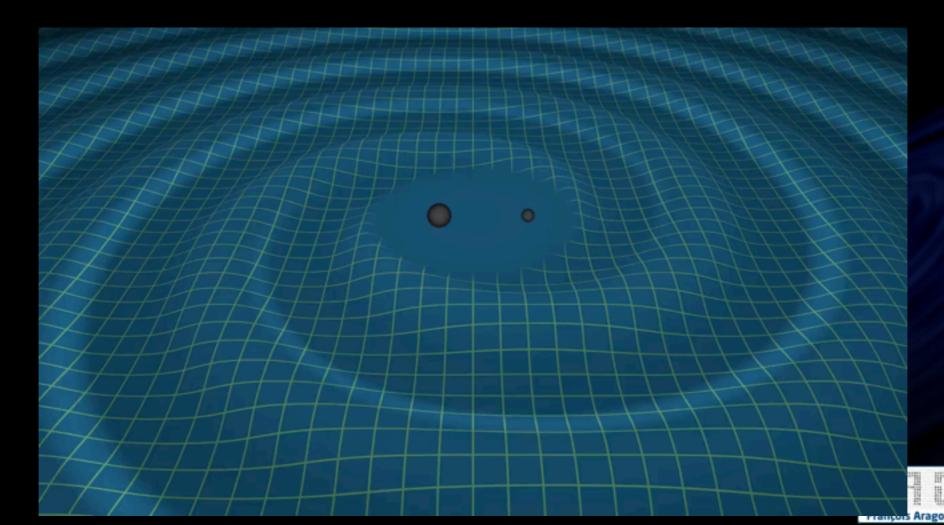








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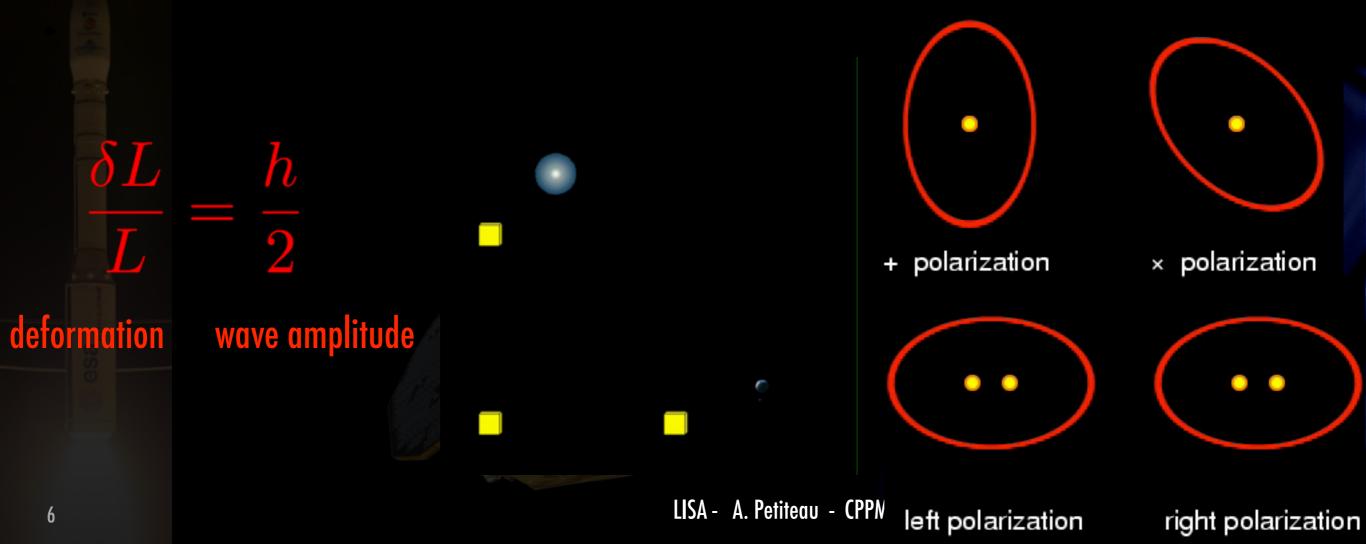




Effects of GWs



- Modification of distance between 2 objects:
 - Elastic deformation proportional to the distance between the 2 obj.,
 - Transverse deformation: perpendicular to the direction of propagation (different from ripples on water !),
 - Two components of polarisation : h_+ and h_\times

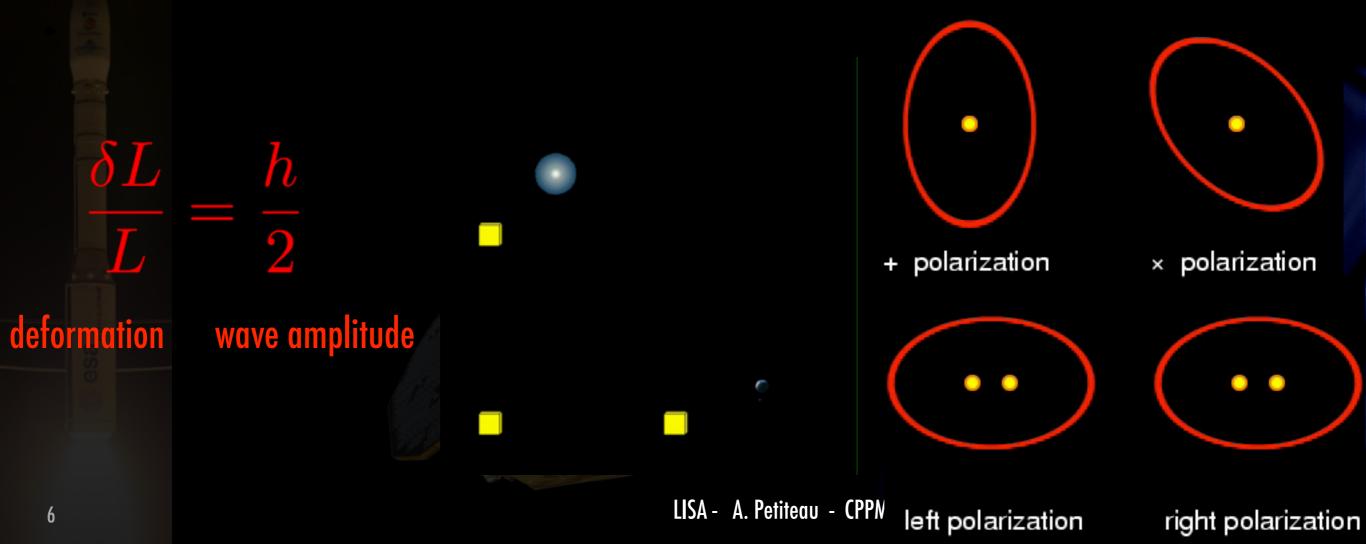




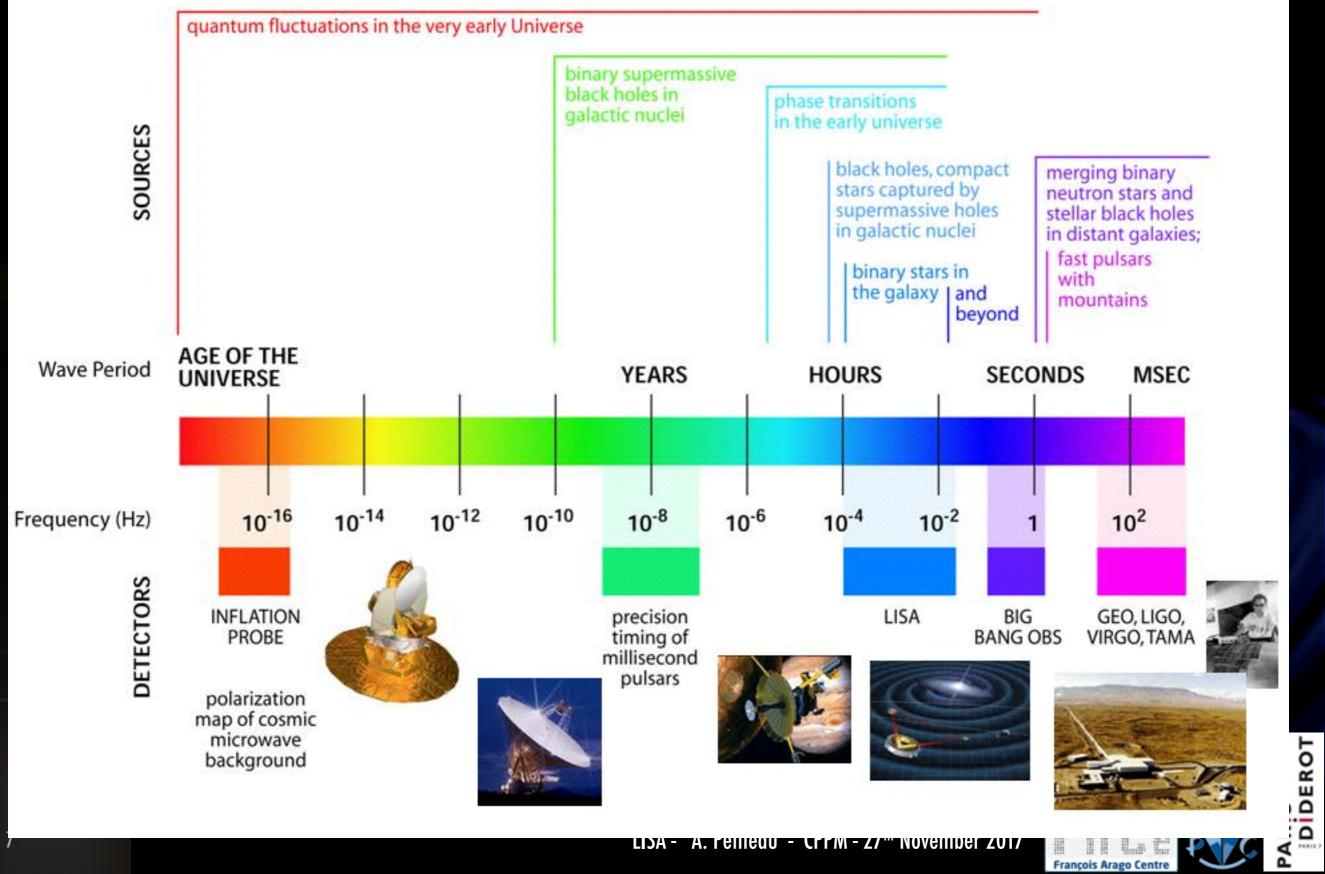
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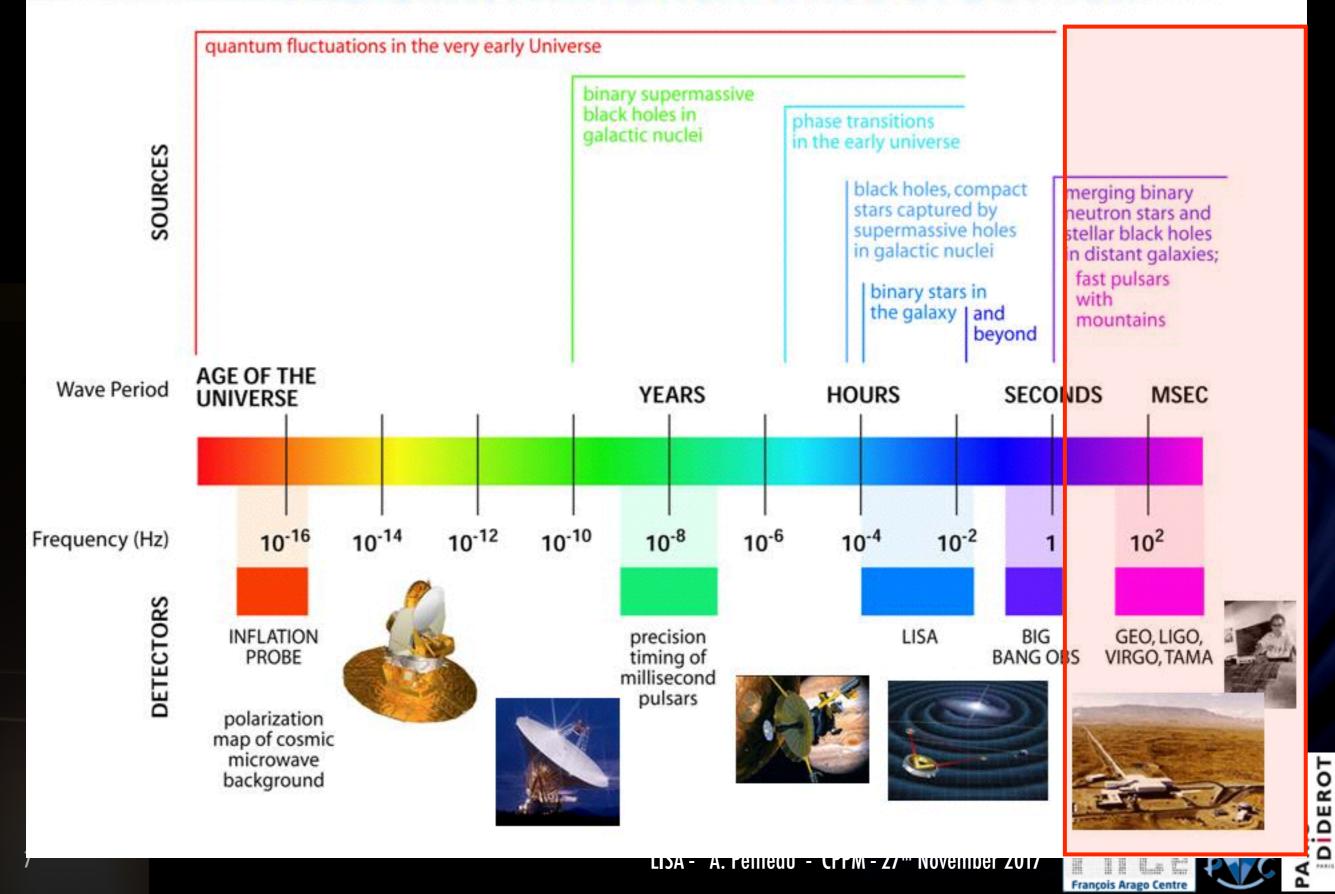


THE GRAVITATIONAL WAVE SPECTRUM

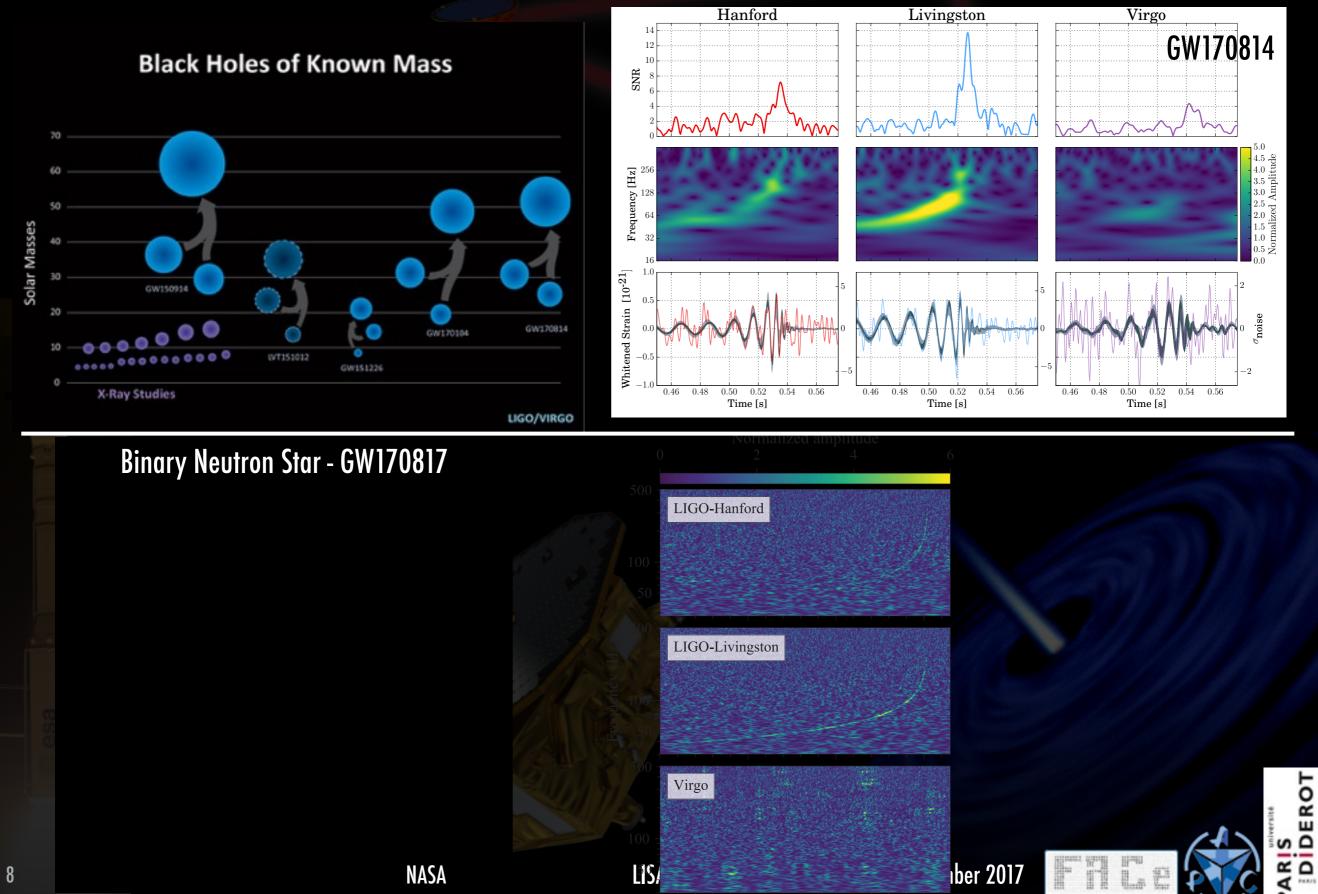


LISA - A. Feilieuu - CFFM - Z/ " November Zui/

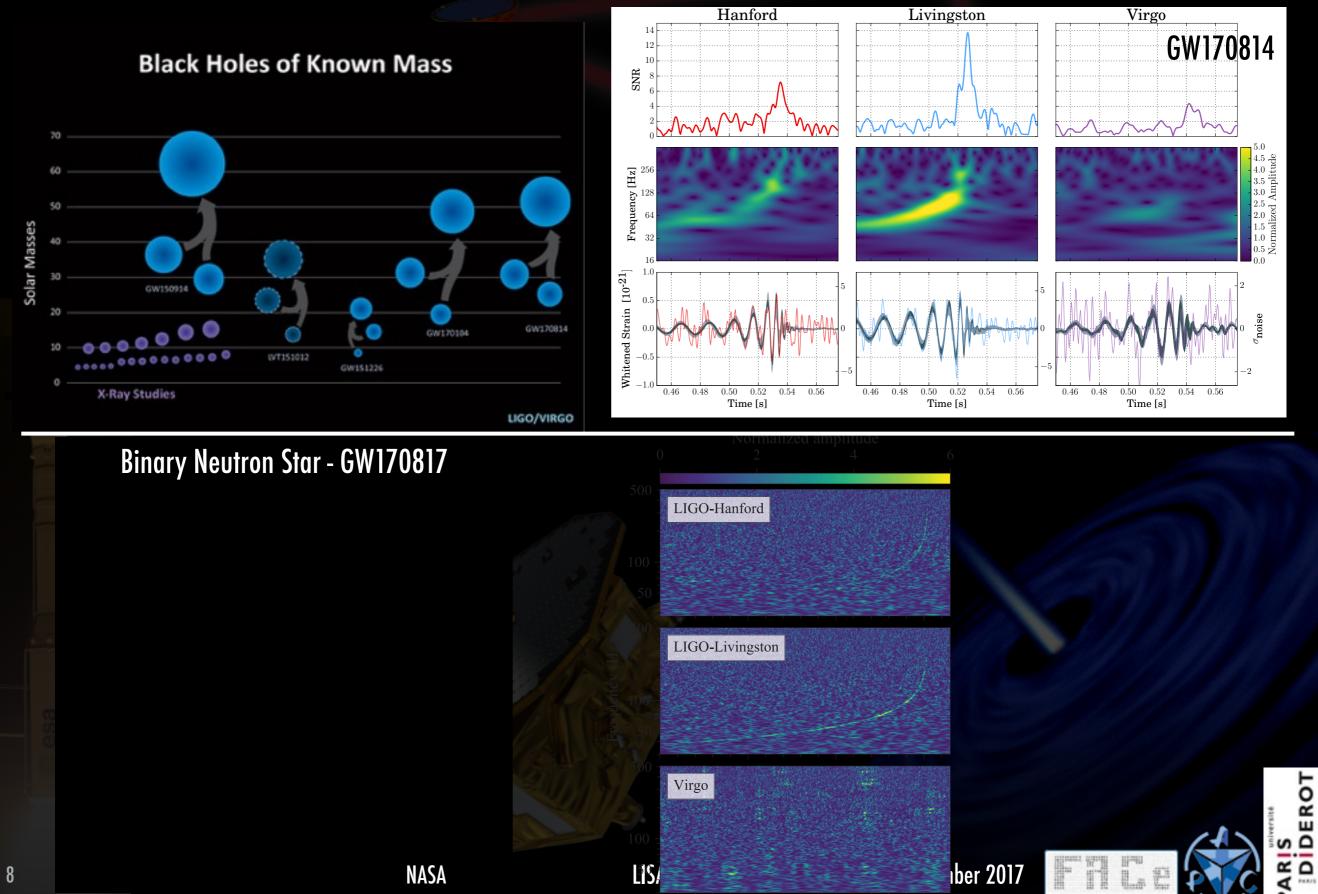
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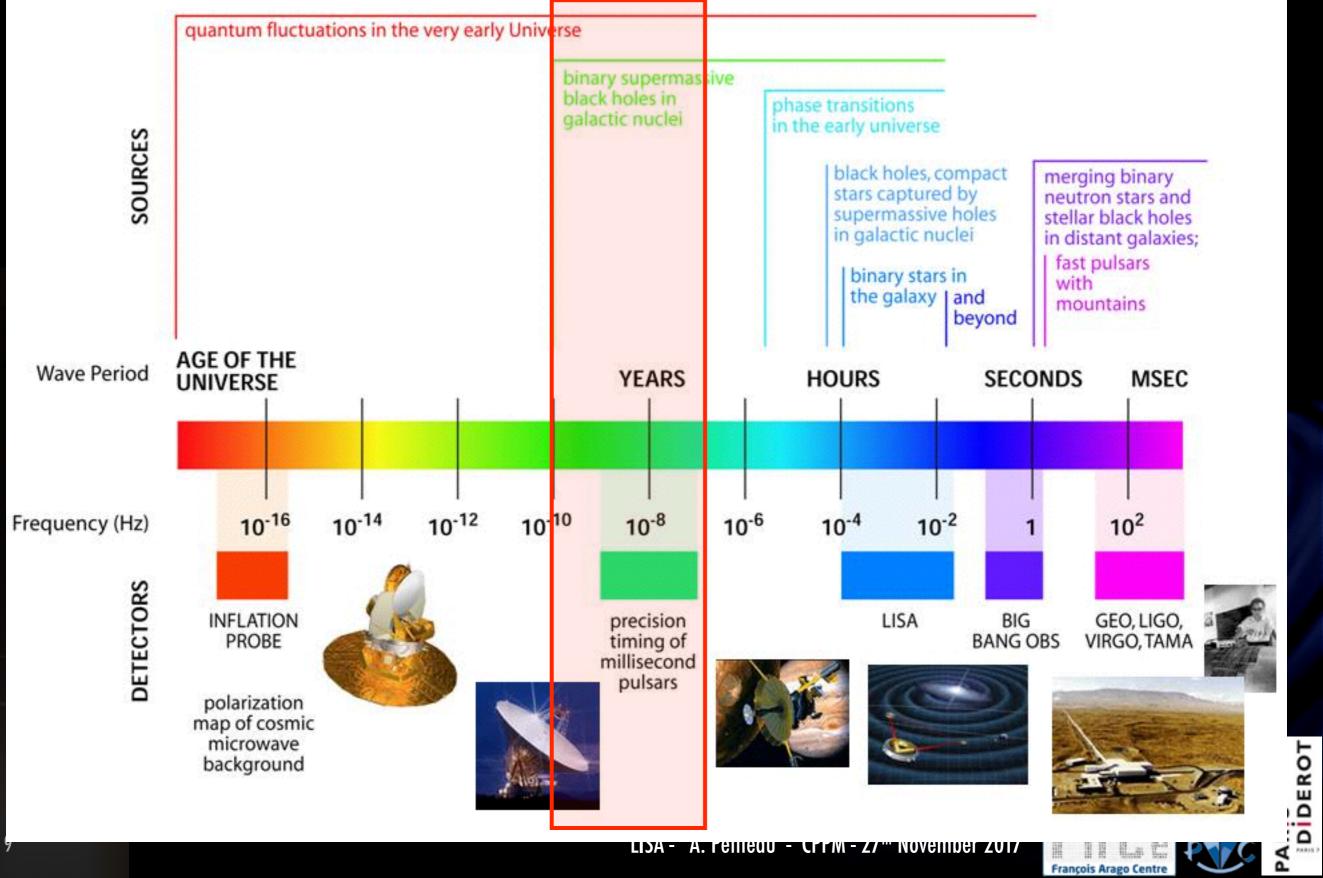
Ground-based obs.: GWs detected



Ground-based obs.: GWs detected



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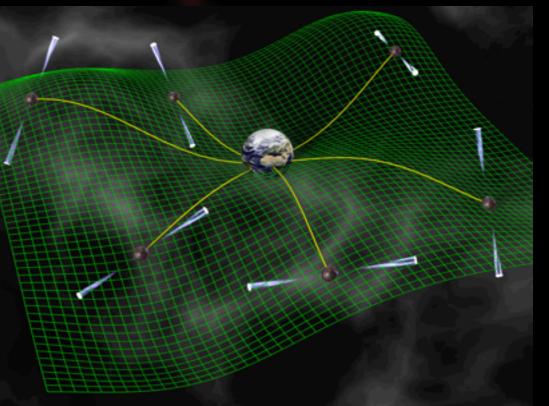


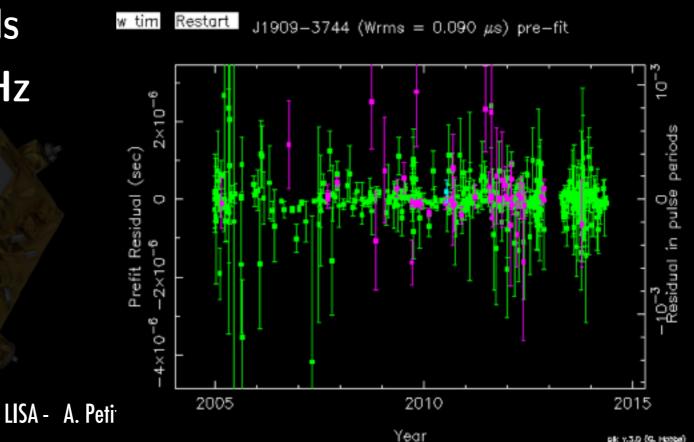


Pulsar Timing Array

- Pulsar: magnetized rotating neutron star emitting pulse as a lighthouse
- Millisecond pulsar = high precision clock
- Series of extremely regular pulses are perturbed by GWs passing between pulsar and Earth
- ▶ By timing an array of milliseconds pulsars we can detect GWs at nHz
 - SuperMassive BH binaries
 - **Cosmological backgrounds**







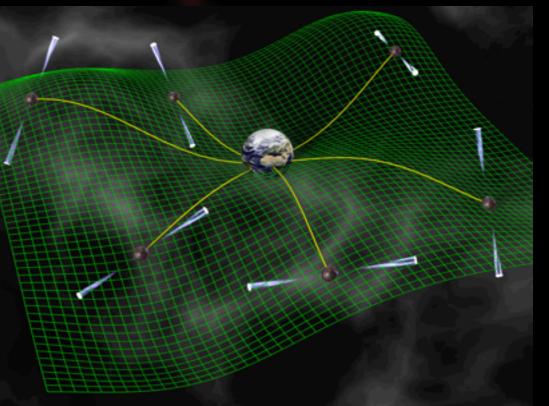
pik: Y.3.0 (G. H088)

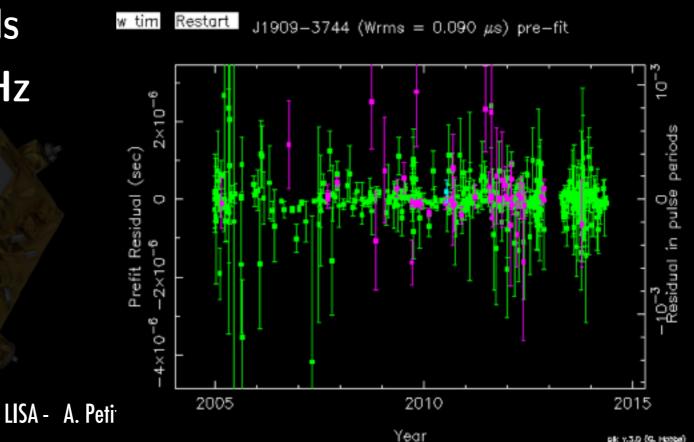


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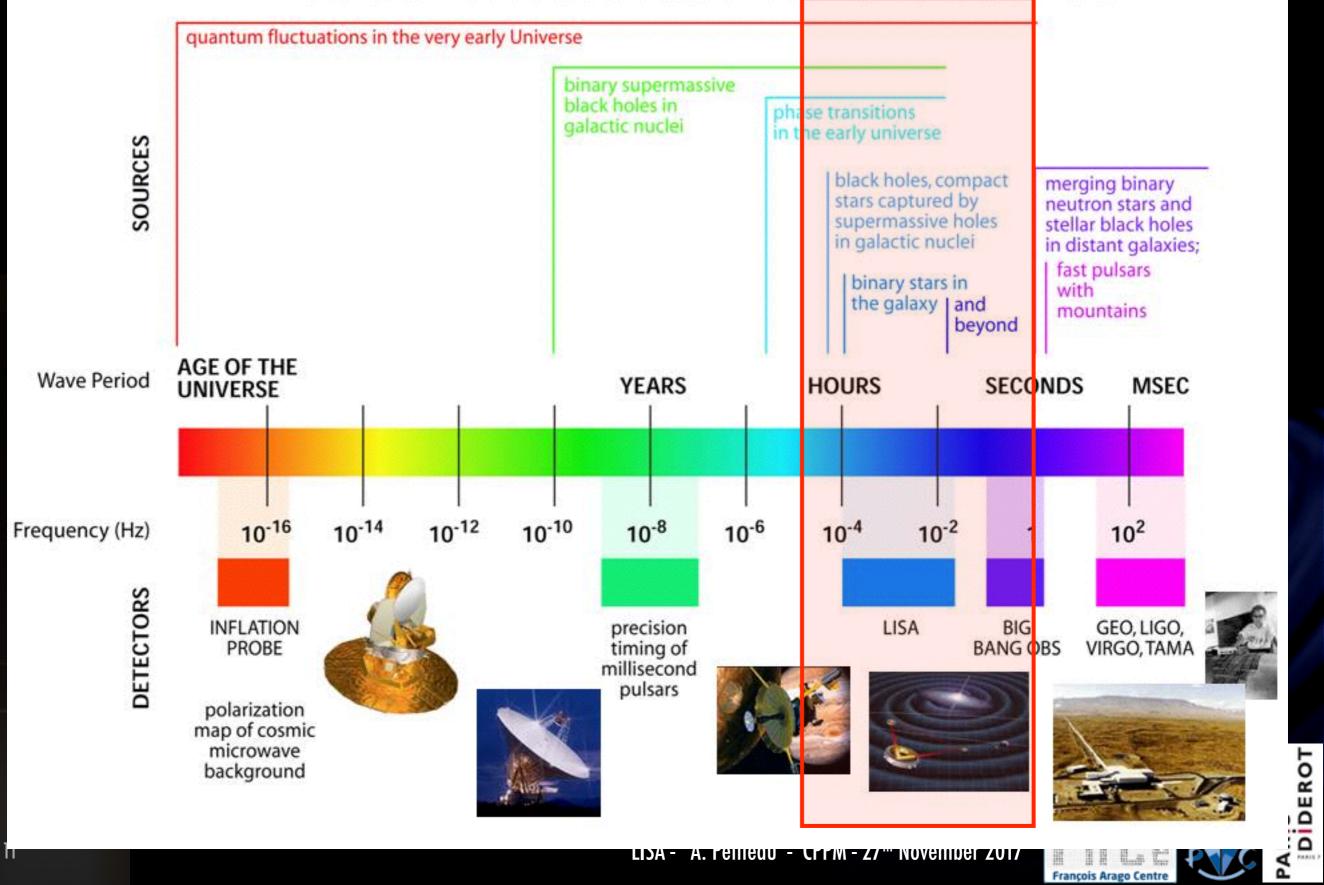






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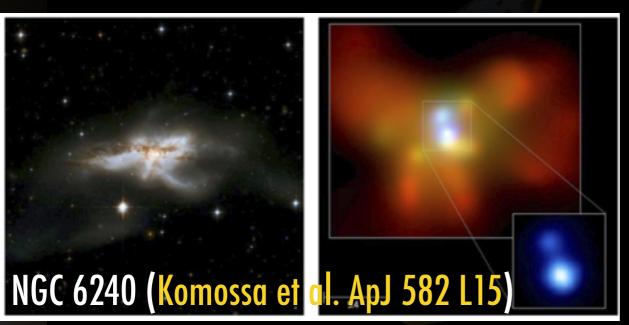


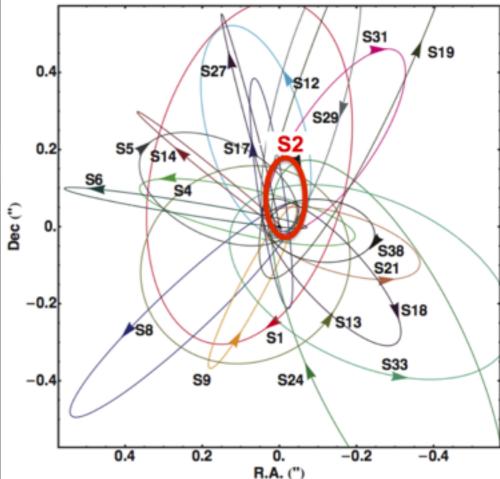
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Supermassive black hole binaries

- Observations of Sgr A*, a dark massive object of 4.5x10⁶ M_{Sun} at the centre of Milky Way.
- Supermassive Black Hole are indirectly observed in the centre of a large number of galaxies (Active Galactic Nuclei).
- Observations of galaxies mergers.
 - \rightarrow MBH binaries should exist.
- Observations of double AGN



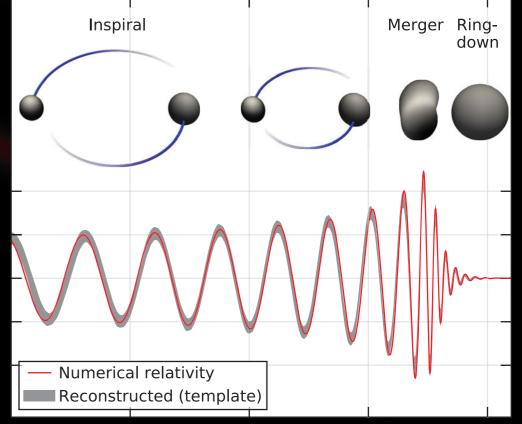






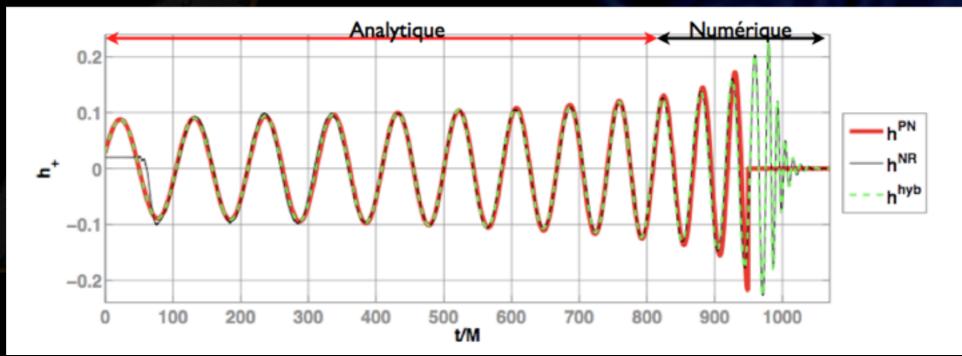
Supermassive black hole binaries

- GW emission: 3 phases:
 - Inspiral: Post-Newtonian,
 - Merger: Numerical relativity,
 - Ringdown: Oscillation of the resulting MBH.



No full waveform but several approximations exist :

- Phenomenological waveform,
- Effective One Body,

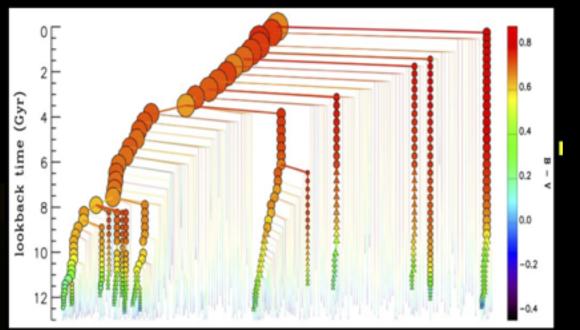




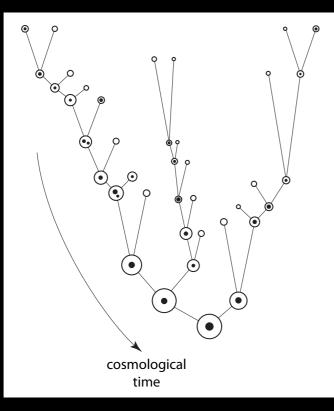
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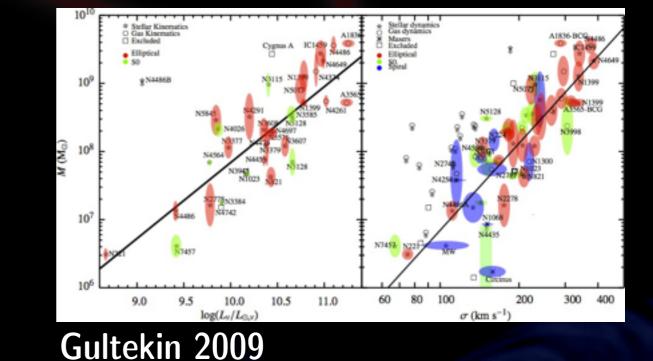
Galaxies merger tree (cosmological simulation)

"M - σ relation": the speed of stars in bulge is linked to the central MBH mass



From De Lucia et al 2006





Work from E. Barausse (IAP), A.
 Sesana (Univ. of Birmingham), M.
 Volonteri (IAP) et al.



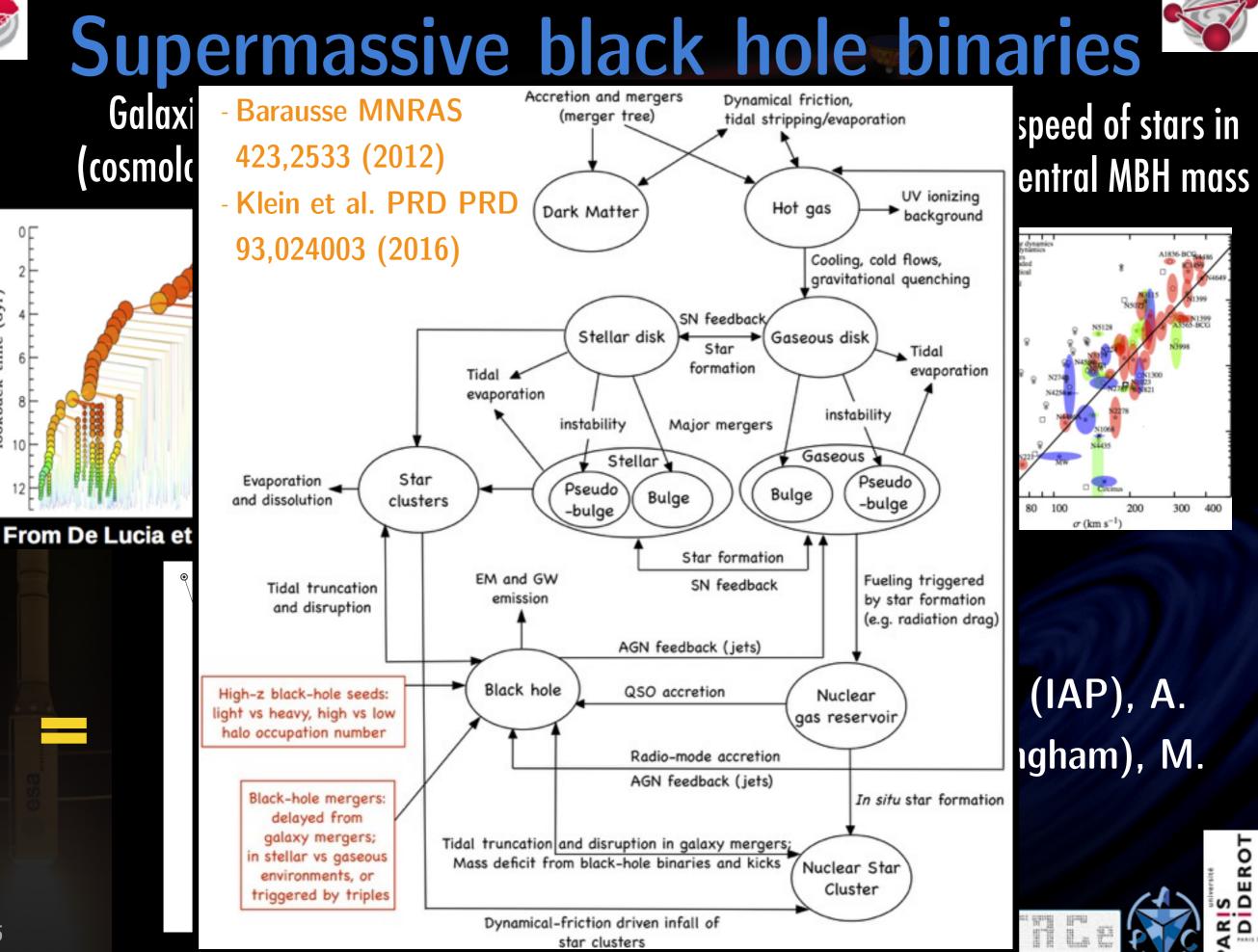




(Gyr)

time

ookback



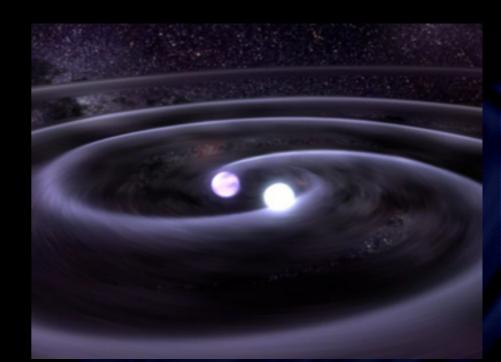
François Arago Cer





Compact solar mass binaries

- Large number of stars are in binary system.
- Evolution in white dwarf (WD) and neutron stars (NS).
 - => existence of WD-WD, NS-WD and NS-NS binaries
- Estimation for the Galaxy: 60 millions.
- Gravitational waves:
 - most part in the slow inspiral regime (quasi-monochromatic): GW at mHz
 - few are coalescing: GW event of few seconds at f > 10 Hz (LIGO/Virgo)



Several known system emitting around the mHz

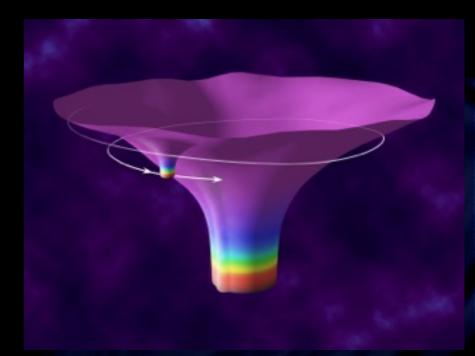




EMRIs

- Capture of a "small" object by massive black hole (10 – 10⁶ M_{Sun})
 - Mass ratio > 200
 - GW gives information on the geometry around the black hole.
 - Test General Relativity in stong field
 - Frequency : 0.1 mHz to 0.1 Hz
 - Large number of source could be observed by space-based interferometer





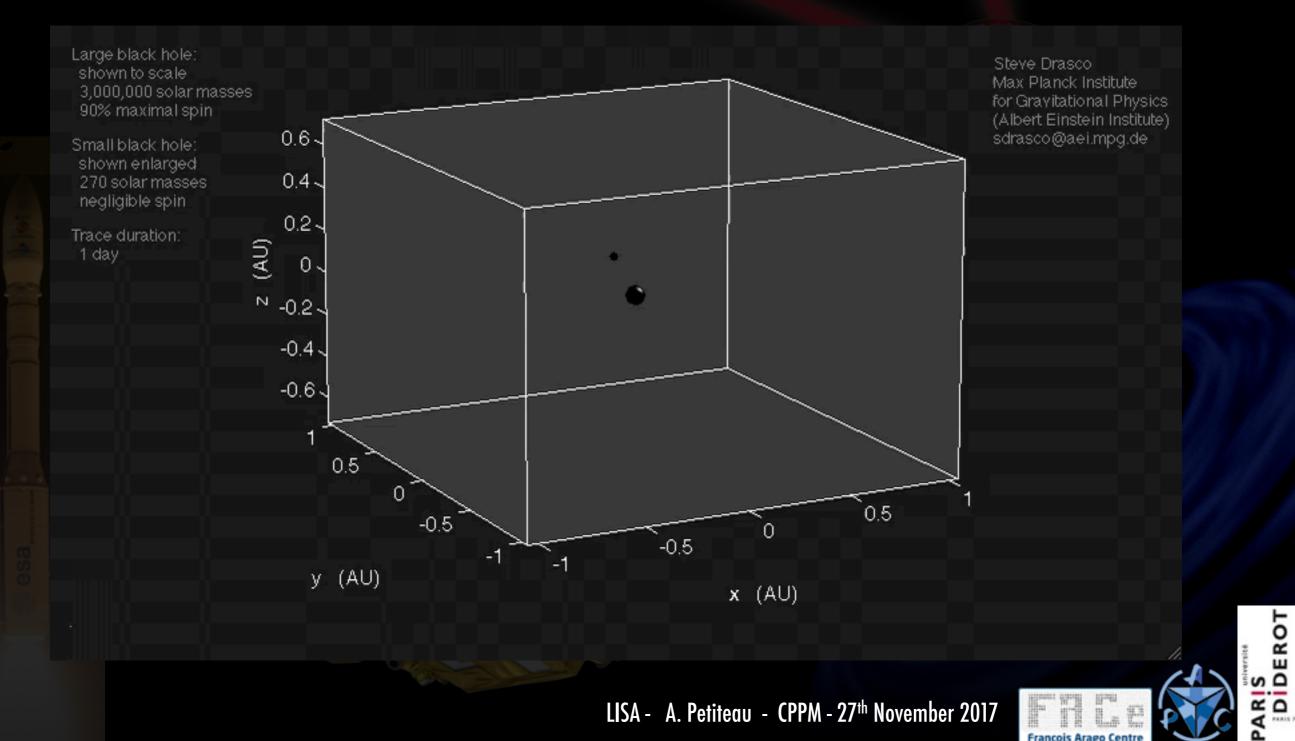




EMRIs



Extreme Mass Ratio Inspiral: small compact objects (10 M_{Sun}) orbiting around a SuperMassive Black Hole

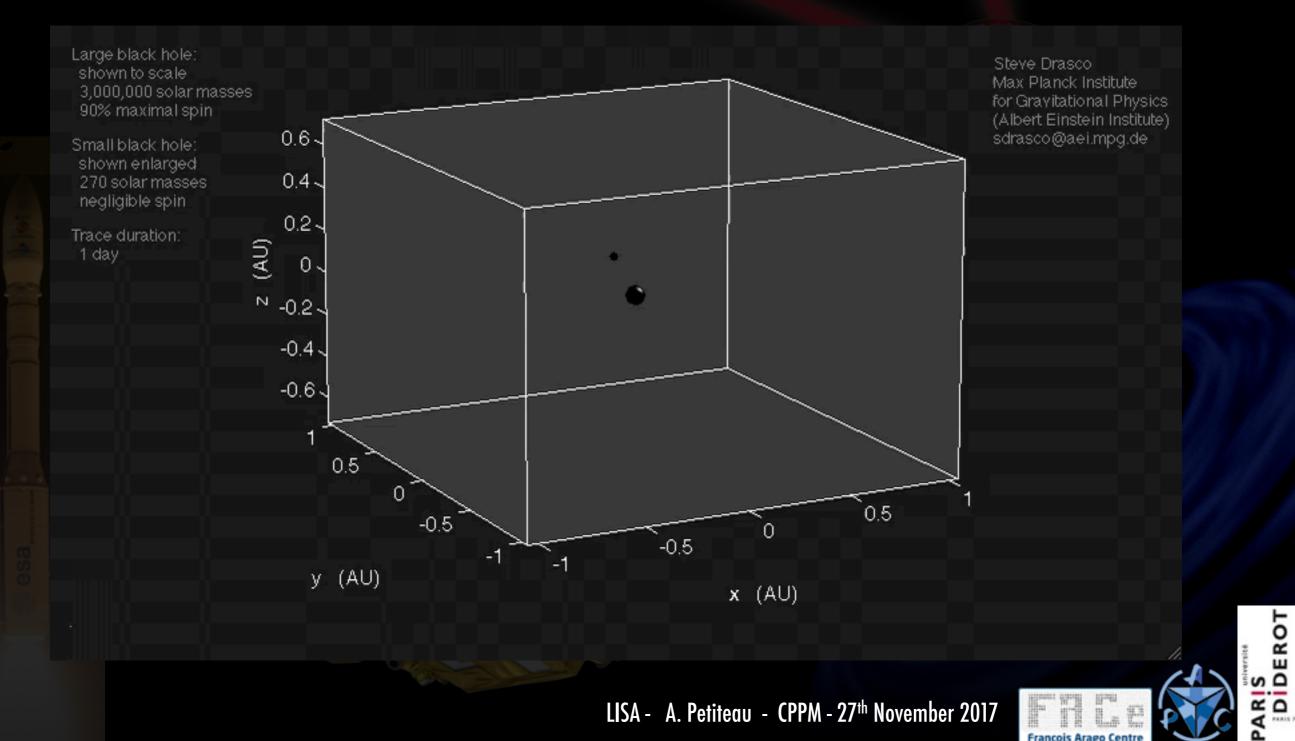




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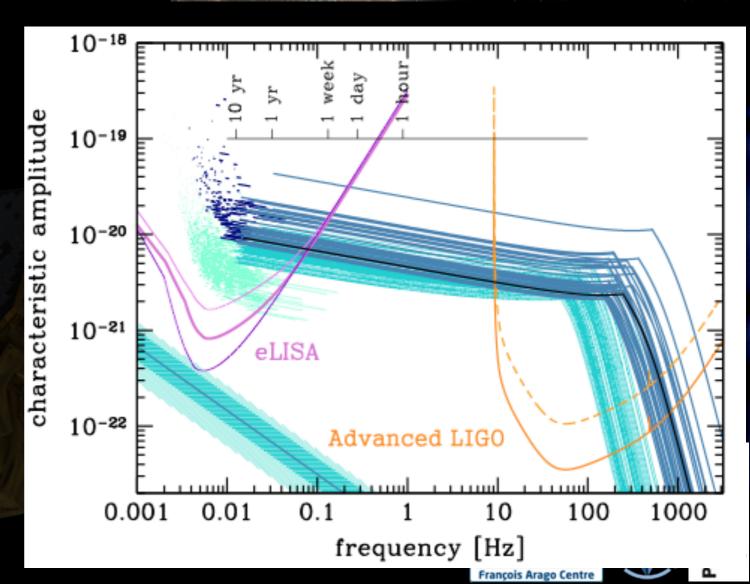




Black Hole Binaries

- LIGO/Virgo-type sources:
 binaries with 2 black
 holes of few tens solar
 masses.
- During most part of the inspiral time, emission in the mHz band
 multi-observatories
 GW astronomy
 A. Sesana, PRL 116, 231102 (2016)





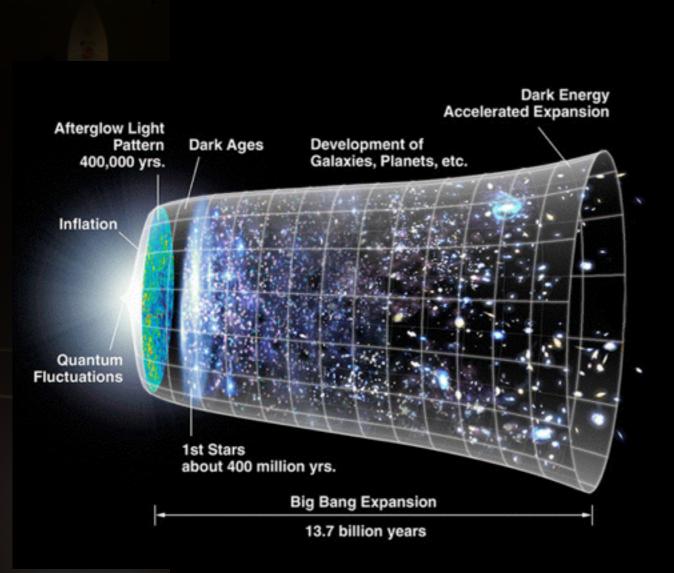


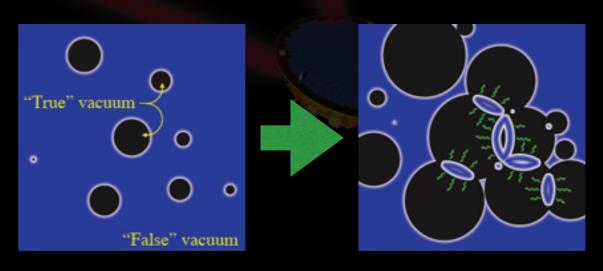


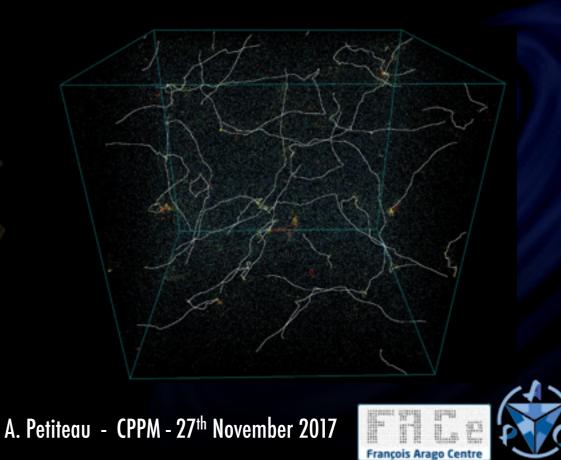
DIDEROT

Cosmological backgrounds

- Variety of cosmological sources for stochastic background :
 - First order phase transition in the very early Universe
 - Cosmic strings network











Unknown sources

► High potential of discovery in the mHz GW band ?









EROT

What can we learn ?

- The nature of gravity (testing the basis of general relativity)
- ► Fundamental nature of black hole: existence of horizon, ...
- Black holes as a source of energy,
- Nonlinear structure formation: seed, hierarchical assembly, accretion,
- Understanding the end of the life of massive stars,
- Dynamic of galactic nuclei,
- ▶ The very early Universe: Higgs TeV physics, topological defects, ...
- Constraining cosmological models,

=> Expand the new observational window on the Universe (with all the unexpected !): looking at dark side of the Universe !



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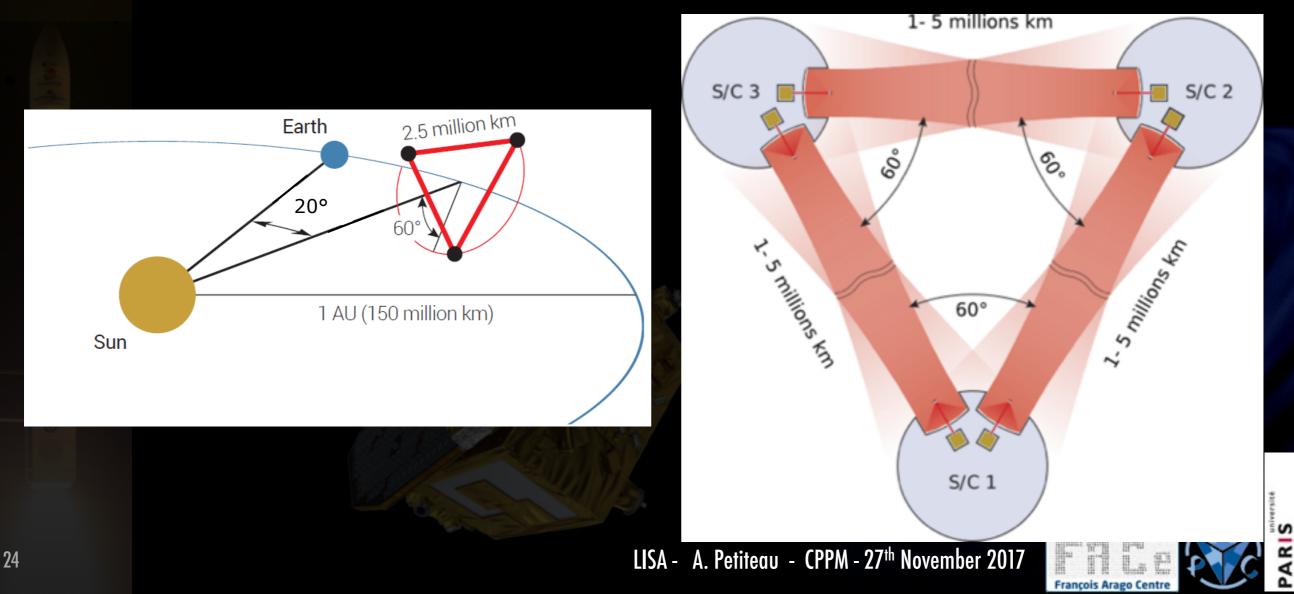






DIDEROT

- ► Laser Interferometer Space Antenna
- 3 spacecrafts on heliocentric orbits and distant from
 2.5 millions kilometers
- ► Goal: detect relative distance changes of 10⁻²¹: few picometers

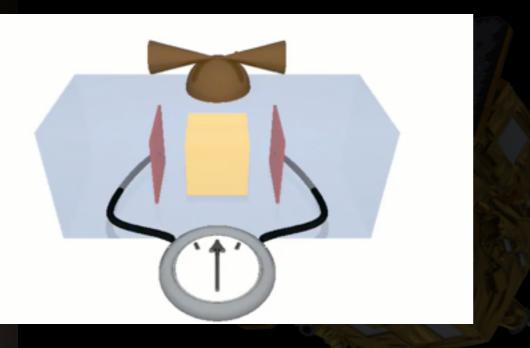


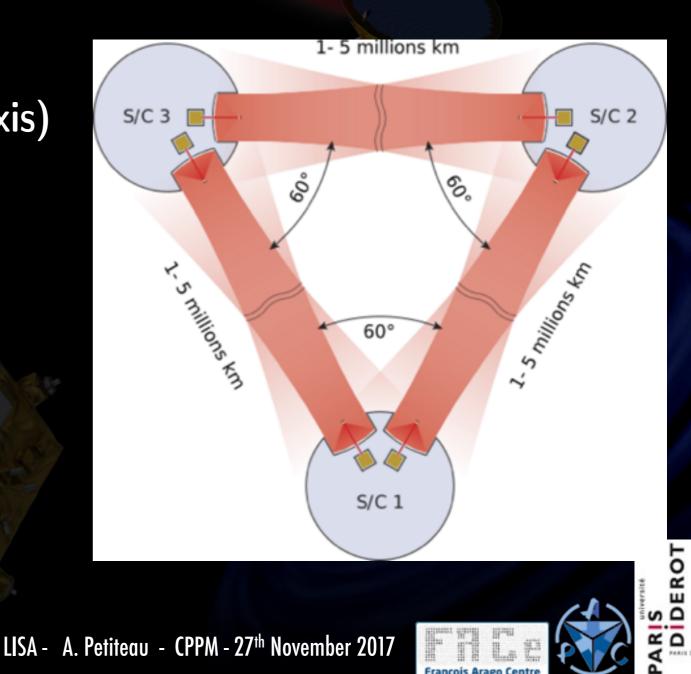






- Spacecraft (SC) should only be sensible to gravity:
 - the spacecraft protects test-masses (TMs) from external forces and always adjusts itself on it using micro-thrusters
 - Readout:
 - interferometric (sensitive axis)
 - capacitive sensing



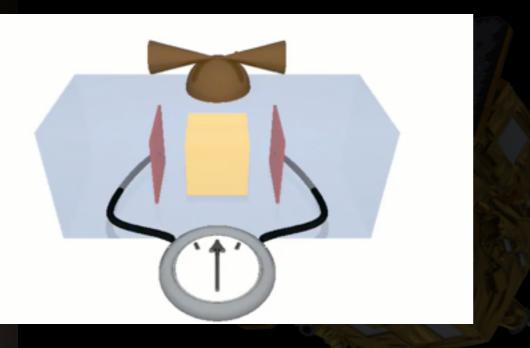


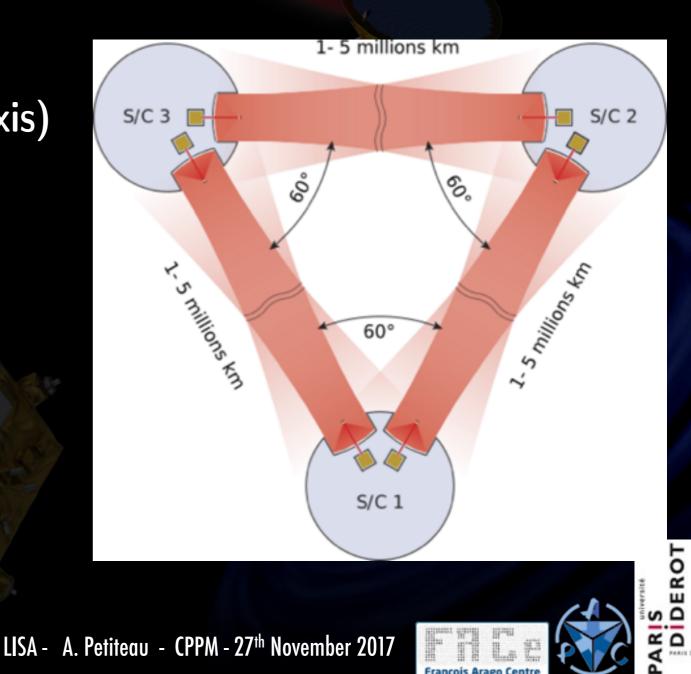






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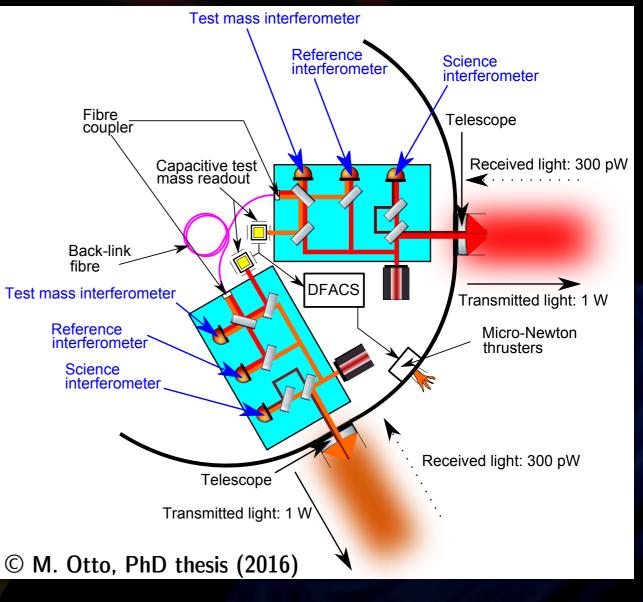


EROT

- Exchange of laser beam to form several interferometers
- Phasemeter measurements on each of the 6 Optical Benches:
 - Distant OB vs local OB
 - Test-mass vs OB
 - Reference using adjacent OB
 - Transmission using sidebands
 - Distance between spacecrafts

Noises sources:

- Laser noise : 10⁻¹³ (vs 10⁻²¹)
- Clock noise (3 clocks)
- Acceleration noise (see LPF)
- Read-out noises
- Optical path noises





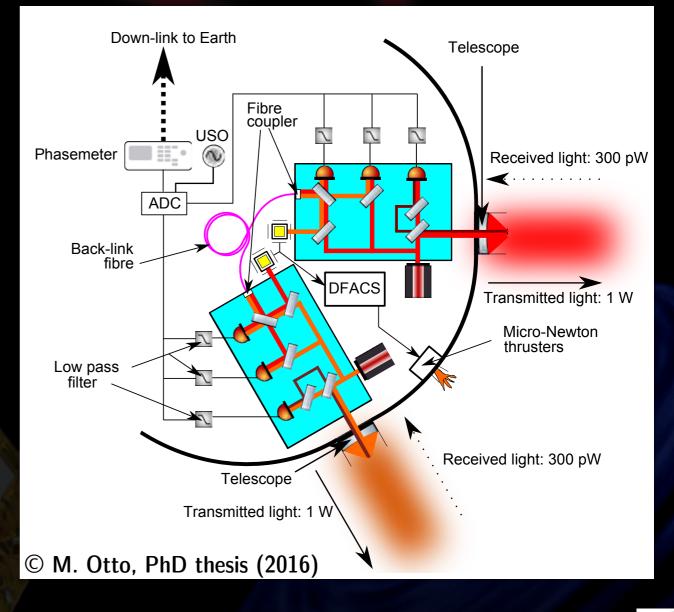


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LISA technology requirements

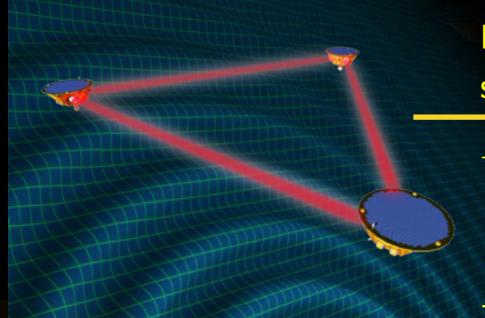


- Free flying test mass subject to very low parasitic forces:
 - Drag free control of spacecraft (non-contacting spacecraft)
 - Low noise microthruster to implement drag-free
 - Large gaps, heavy masses with caging mechanism
 - High stability electrical actuation on cross degrees of freedom
 - Non contacting discharging of test-masses
 - High thermo-mechanical stability of spacecraft
 - Gravitational field cancellation
 - Precision interferometric, local ranging of test-mass and spacecraft:
 - pm resolution ranging, sub-mrad alignments
 - High stability monolithic optical assemblies
 - Precision million km spacecraft to spacecraft precision ranging:
 - High stability telescopes
 - High accuracy phase-meter
 - High accuracy frequency stabilization





LISA data



Phasemeters (carrier, sidebands, distance)

+ Gravitational Reference Sensor **Auxiliary channels**





Corrections, calibrations

Resynchronisation (clocks)

Time-Delay Interferometry laser noise reduction

TDI data : 2 uncorrelated channels

GW data analysis

Catalog of GW sources with extracted waveforms



GW sources

- 10-100/yr SMBHBs
- 10-1000/yr EMRIs
- 60 millions Galactic binaries
- Large number of Black Hole binaries
- Cosmological backgrounds
- Unknown sources



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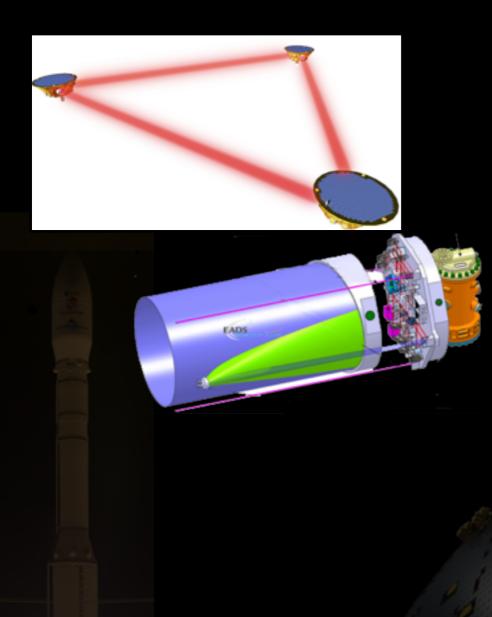


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Technological demonstrator for LISA



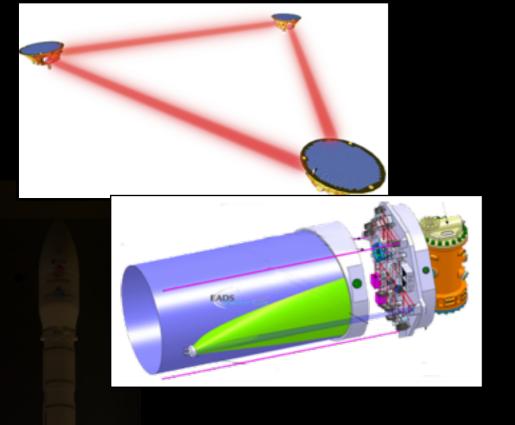
LISA :

- ► 3 spacecraft separated by millions of km
- Role of each spacecraft is to protect the fiducial test masses from external forces





Technological demonstrator for LISA



LISA :

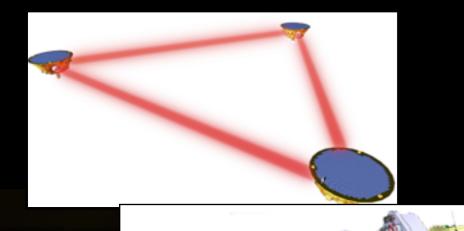
Locally measure distance from TM to SC using:

- Laser interferometry along sensitive axis (between SC)
- Capacitive sensing on orthogonal axes
- TM displacement measurements are used as input to DFACS which controls position and attitude of SC respect to the TM





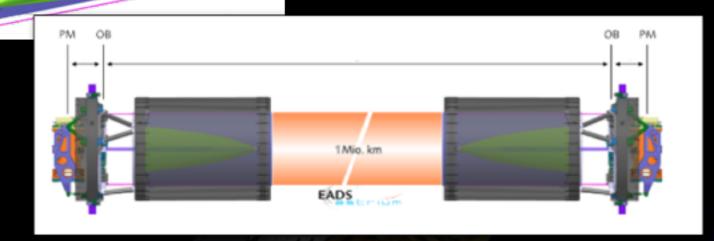
Technological demonstrator for LISA



LISA :

Measure distance along using laser interferometry

 $(TM1 \rightarrow SC1) + (SC1 \rightarrow SC2) + (SC2 \rightarrow TM2)$



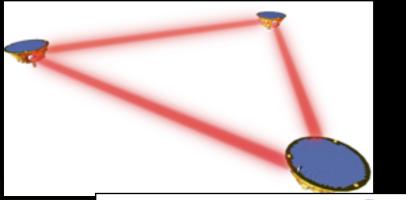


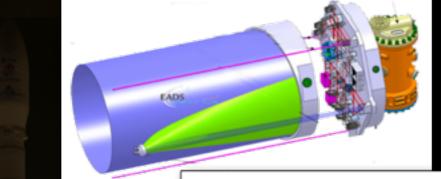


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LISAPathfinder

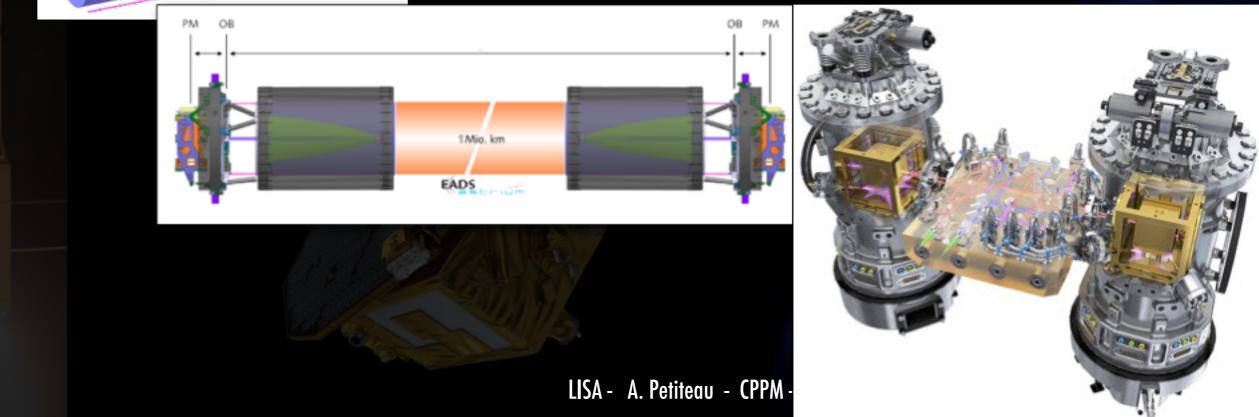
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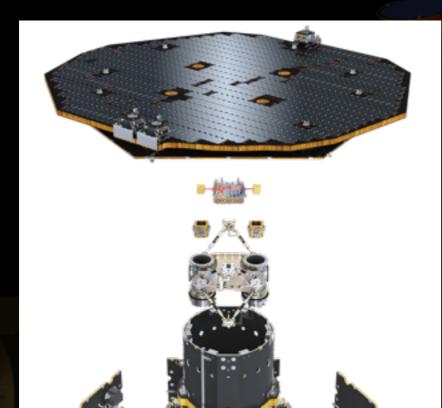
- 2 test masses / 2 inertial sensors
- Laser readout of TM1 \rightarrow SC and TM1 \rightarrow TM2
- Capacitive readout of all 6 d.o.f. of TM
- Drag-Free and Attitude Control System
- Micro-newton thrusters





LISAPathfinder timeline





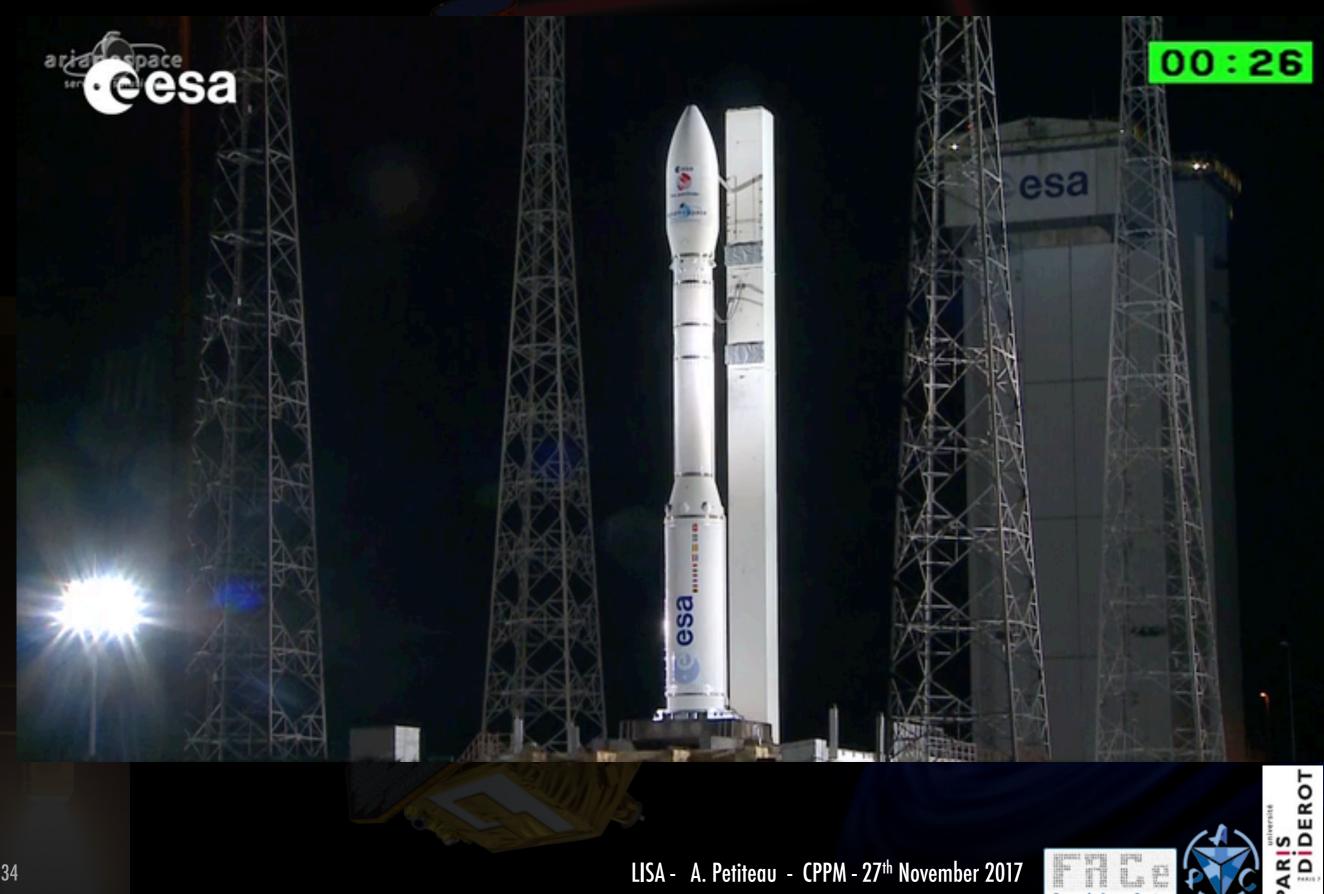






LISAPathfinder timeline







Francois Arago Centre

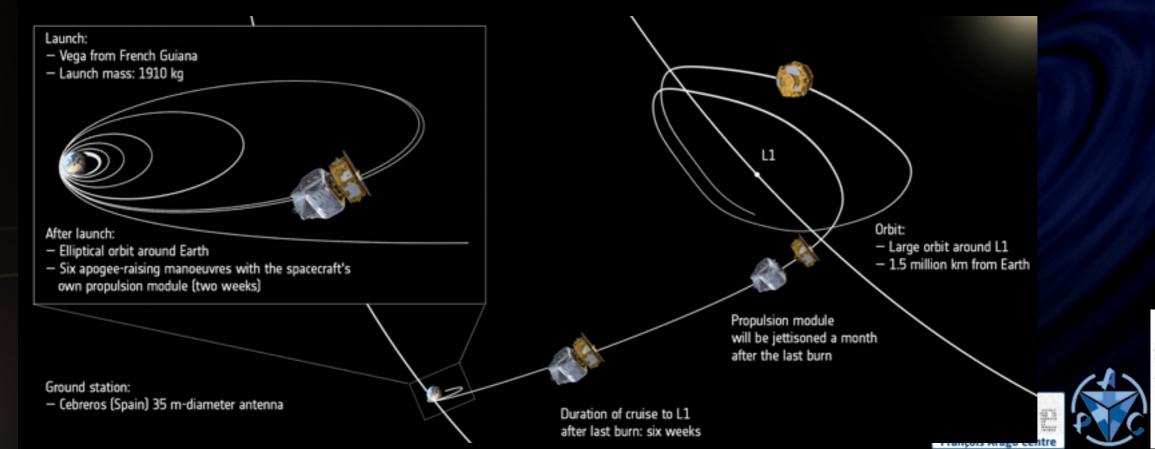




EROT

LISAPathfinder timeline

- ► 3/12/2015: Launch from Kourou
- ▶ 22/01/2016: arrived on final orbit & separation of propulsion module
- ▶ $17/12/2015 \rightarrow 01/03/2016$: commissioning
- ▶ $01/03/2016 \rightarrow 27/06/2016$: LTP operations (Europe)
- ▶ $27/06/2016 \rightarrow 11/2016$: DRS operations (US) + few LTP weeks
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- LISAPathfinder is testing :
 - Inertial sensor,
 - Drag-free and attitude control system
 - Interferometric measurement between 2 free-falling test-masses,
 - Micro-thrusters









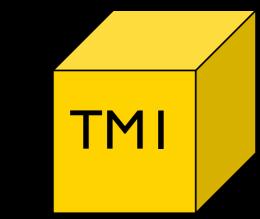
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X

by Joseph Martino







by Joseph Martino

TMI

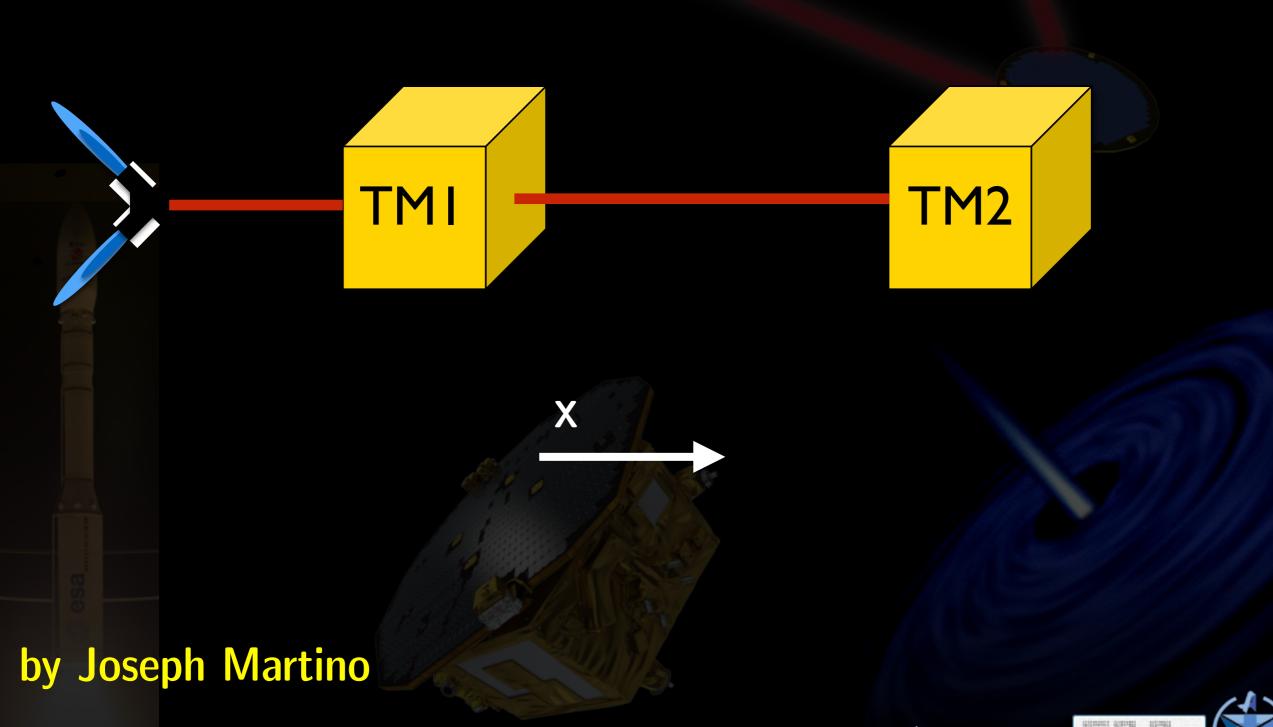
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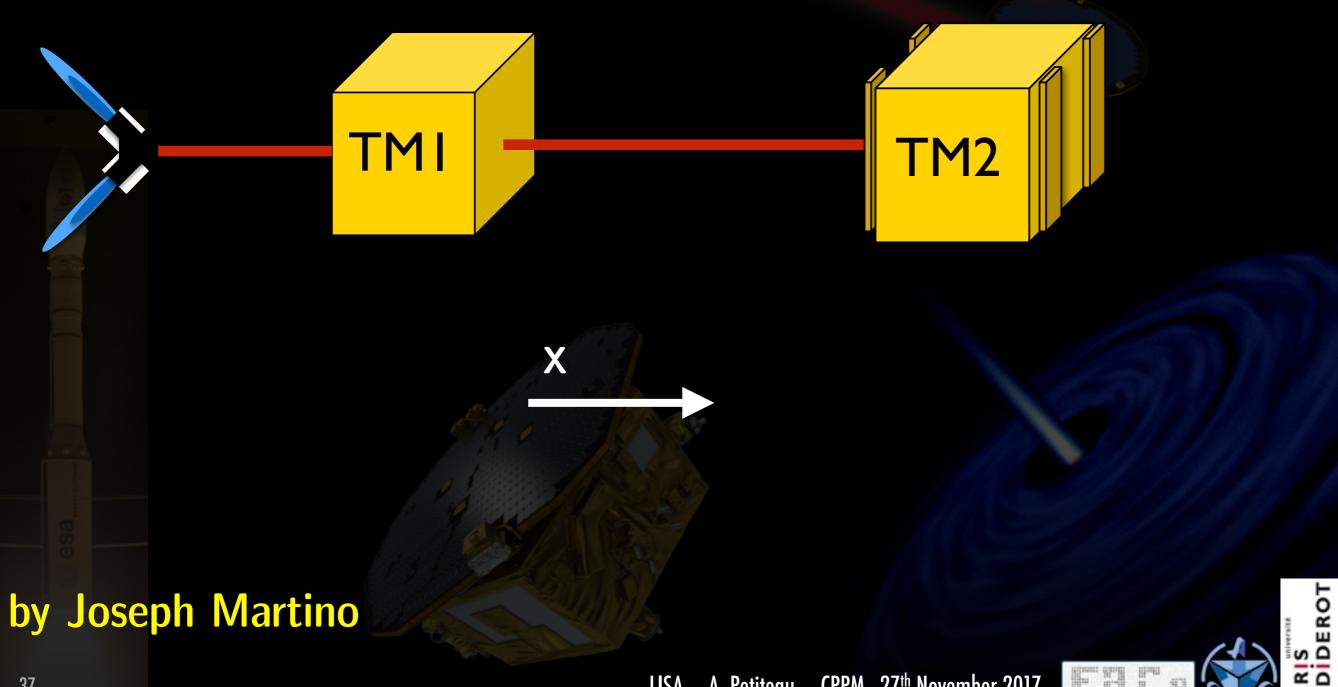
DIDEROT







Suspension (f<1mHz)

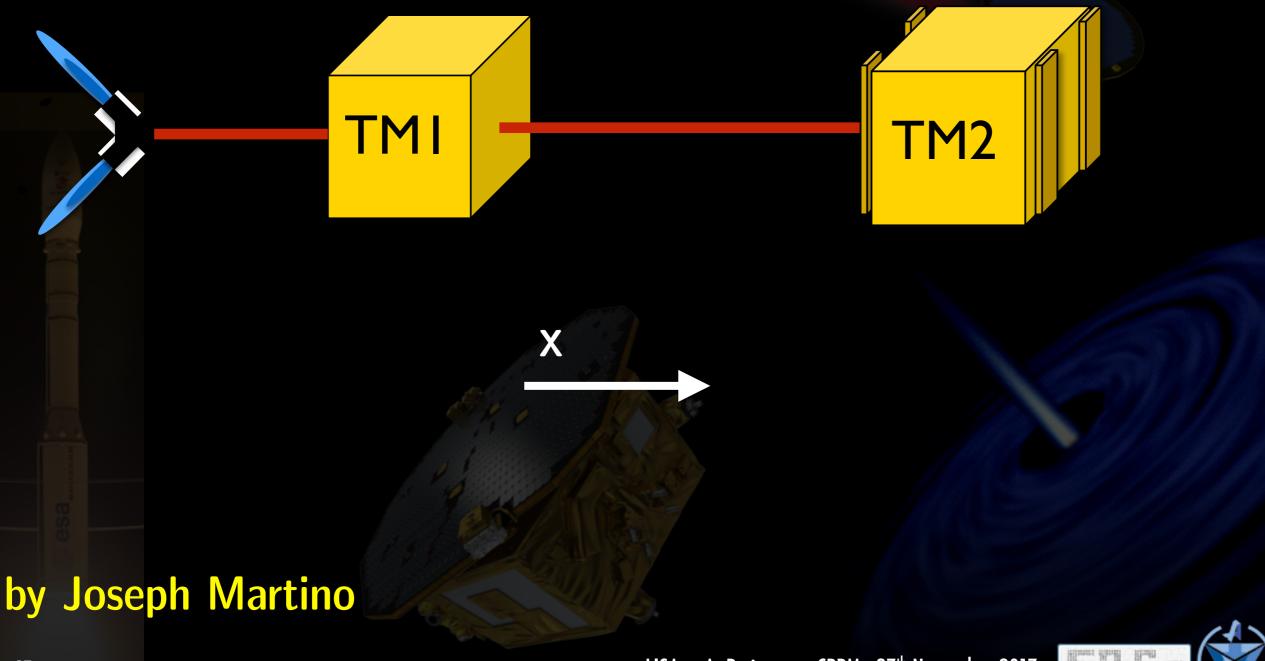




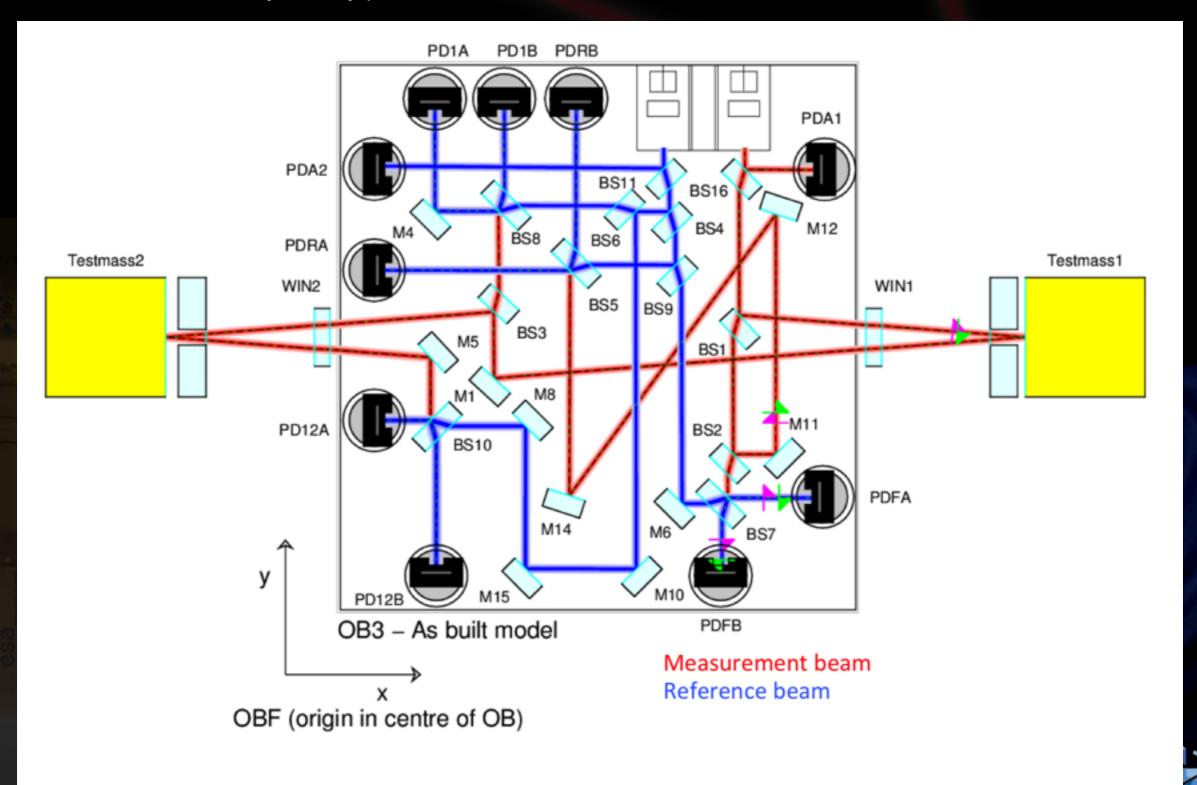
DIDEROT

The measurement - deltaG

$\label{eq:general} deltaG = d^2(o12)/dt^2 - Stiff * o12 - Gain * Fx2 \\ Suspension (f<1mHz) \\$



Optical bench deltaG = $d^2(o12)/dt^2$ - Stiff * o12 - Gain * Fx2



PARIS

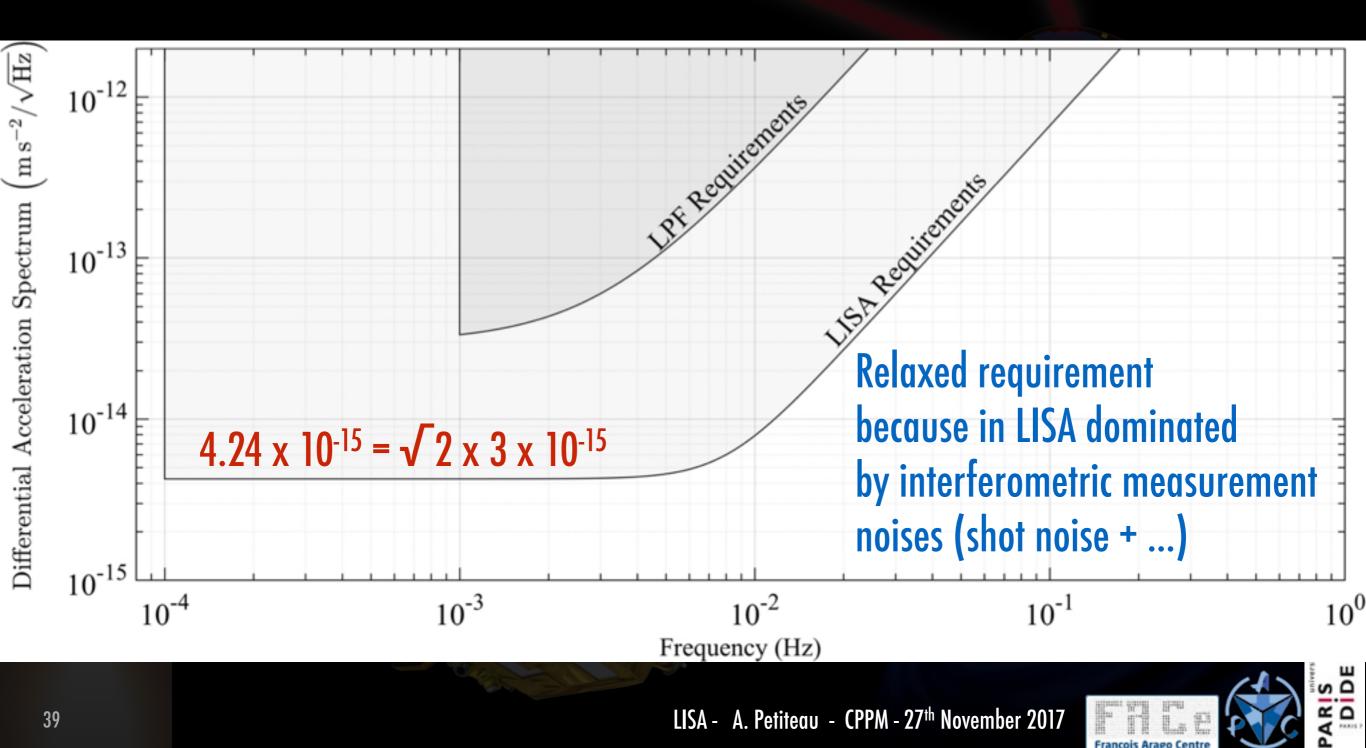
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Requirements: LPF vs LISA

Main LISAPathfinder (LPF) measurement : Δg : differential acceleration between the 2 test-masses



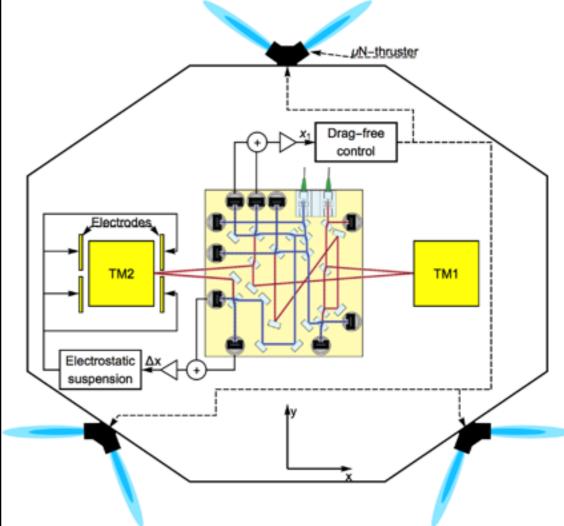


Requirements: LPF vs LISA



Why the LISAPathfinder requirements are restricted compare to LISA ones ?

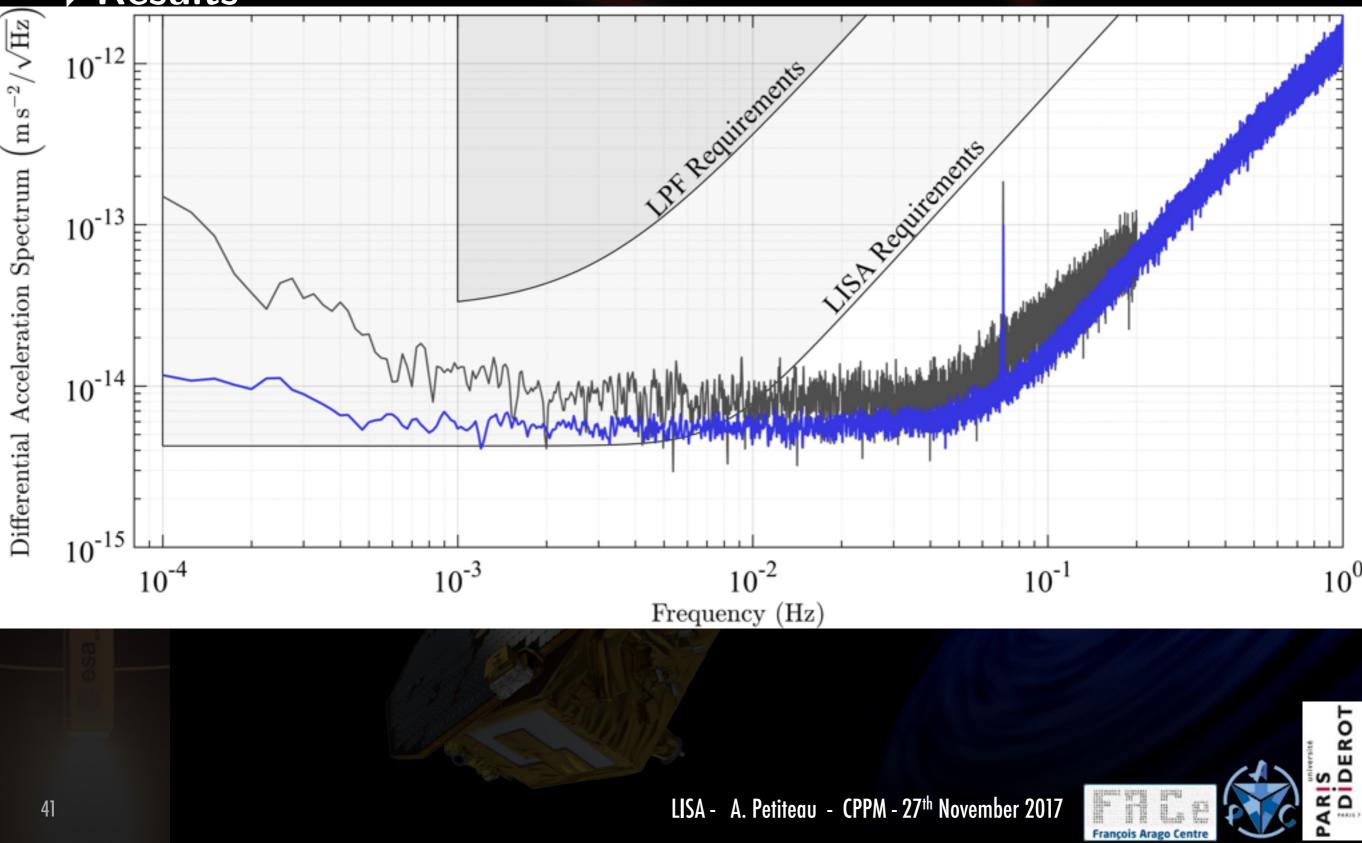
- We understand limitations with LISAPathfinder and correct for them in LISA
- Short arm limitation :
 - Gravitational field not perfectly flat
 => constant electrostatic actuation
 on test- mass 2
- f > 1 mHz : limit duration of industrial testing
- Industrial margin







M. Armano et al. PRL 116, 231101 (2016)



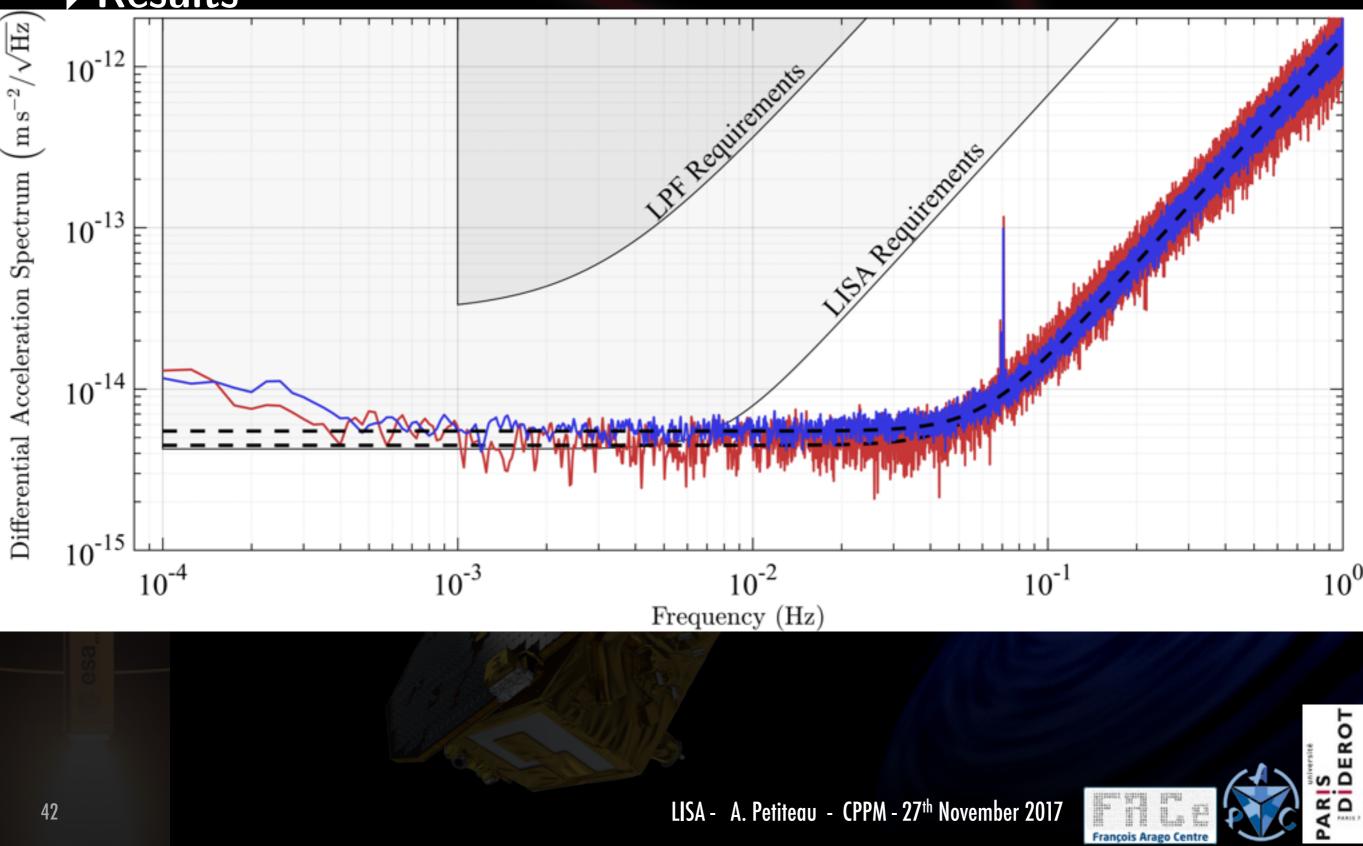
LISA - A. Petiteau - CPPM - 27th November 2017

François Arago Centre



Results

M. Armano et al. PRL 116, 231101 (2016)



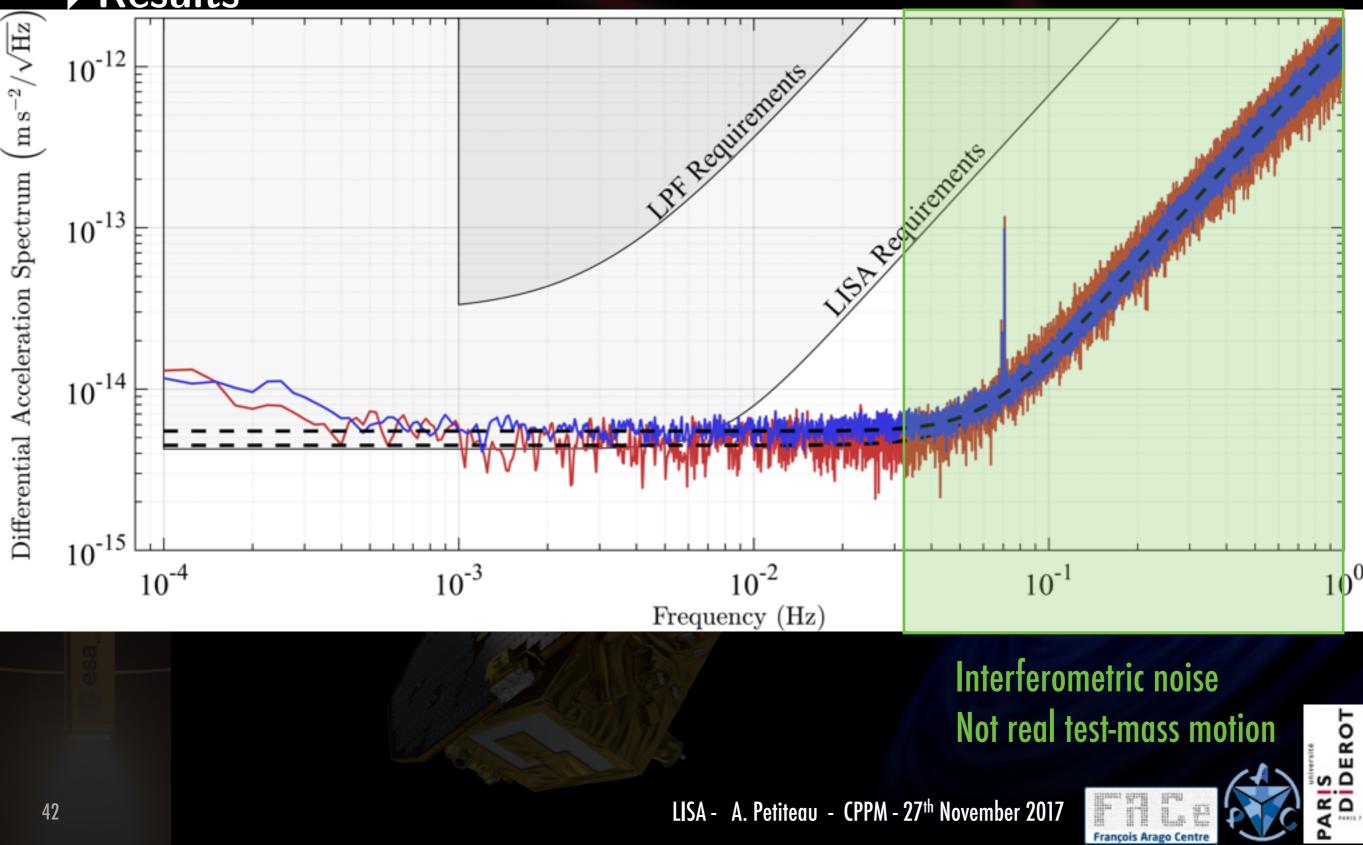
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Results

M. Armano et al. PRL 116, 231101 (2016)

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High frequency limit

Testmass2

PDA2

PDR/

PD12A



Testmass1

PDA1

WIN1

BS1

BS6

BS5

BS9

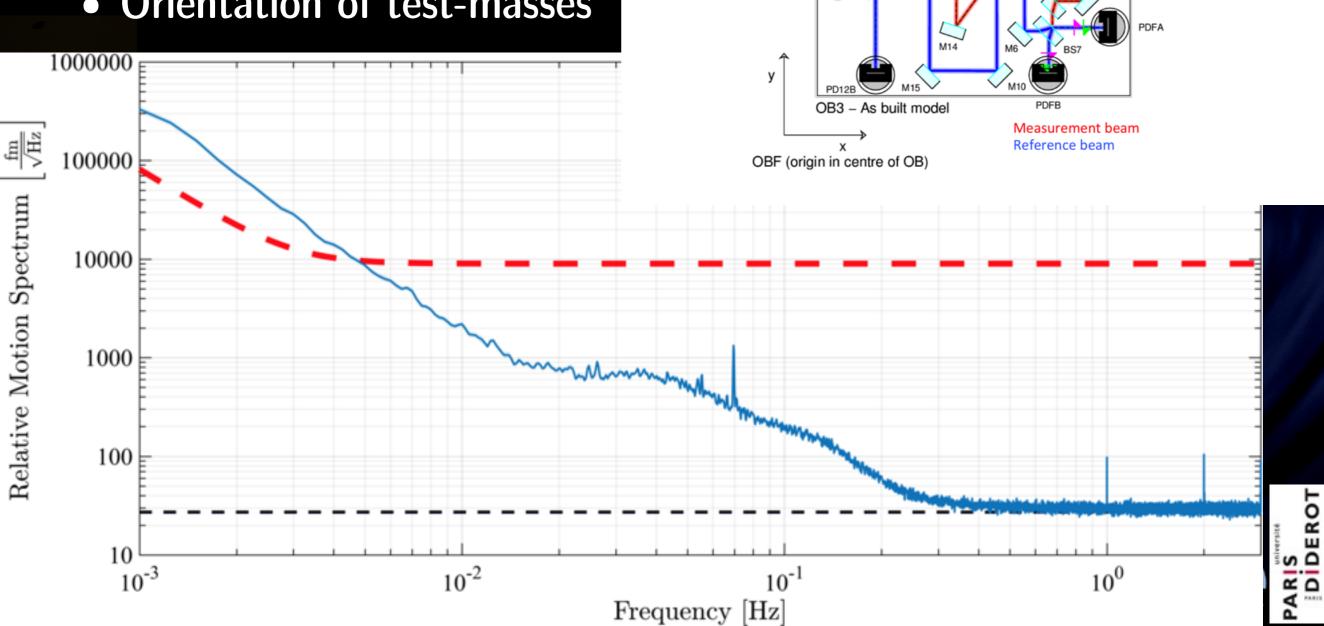
BS1

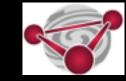
BS8

BS3



- Interferometric precision: **30** fm.Hz^{-1/2}
- Orientation of test-masses

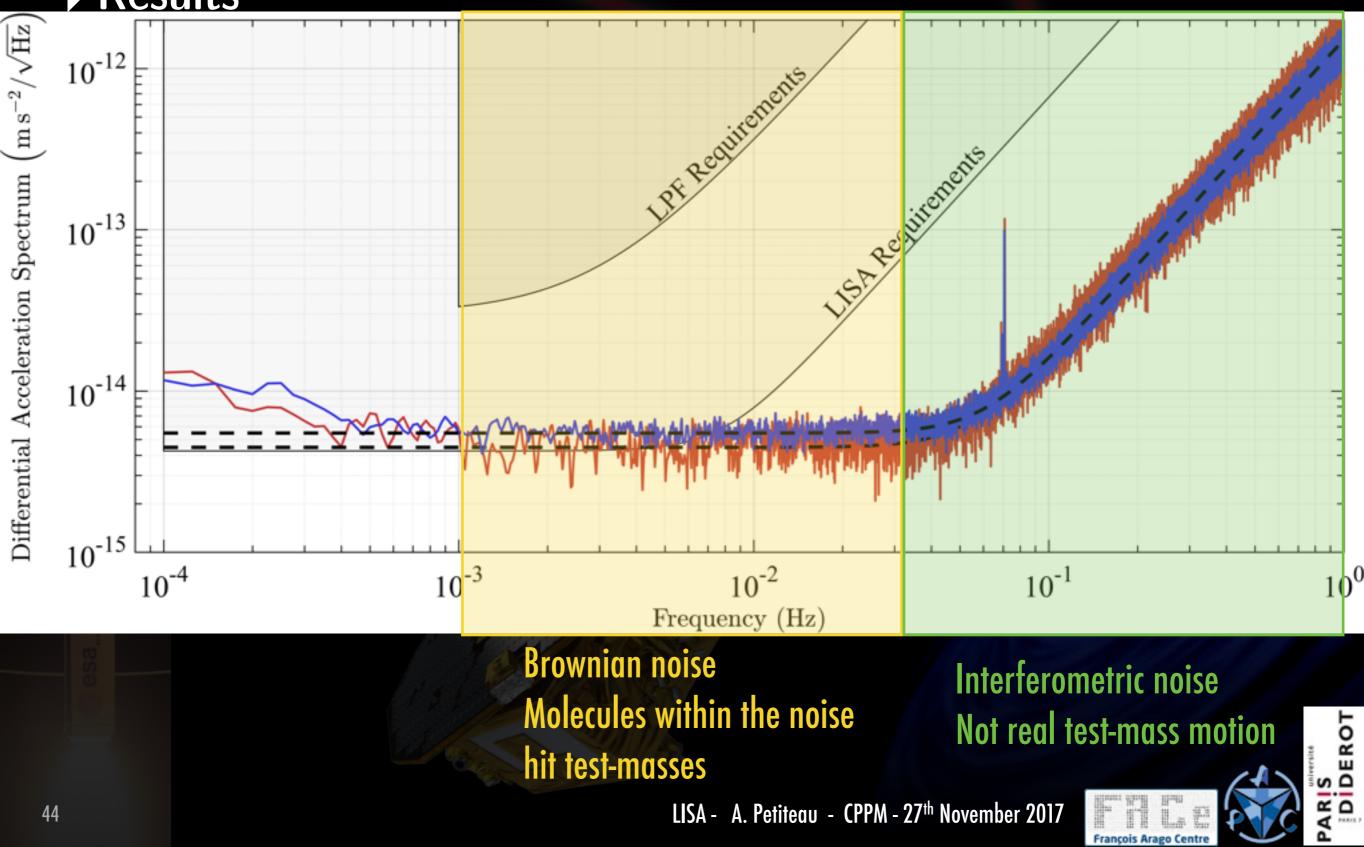






M. Armano et al. PRL 116, 231101 (2016)

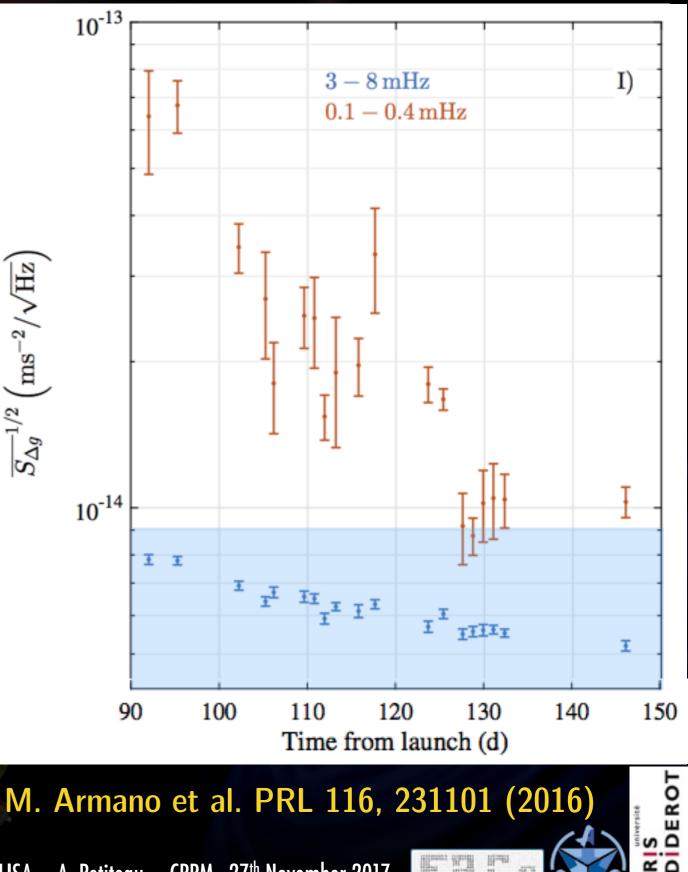
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Mid-frequency limit

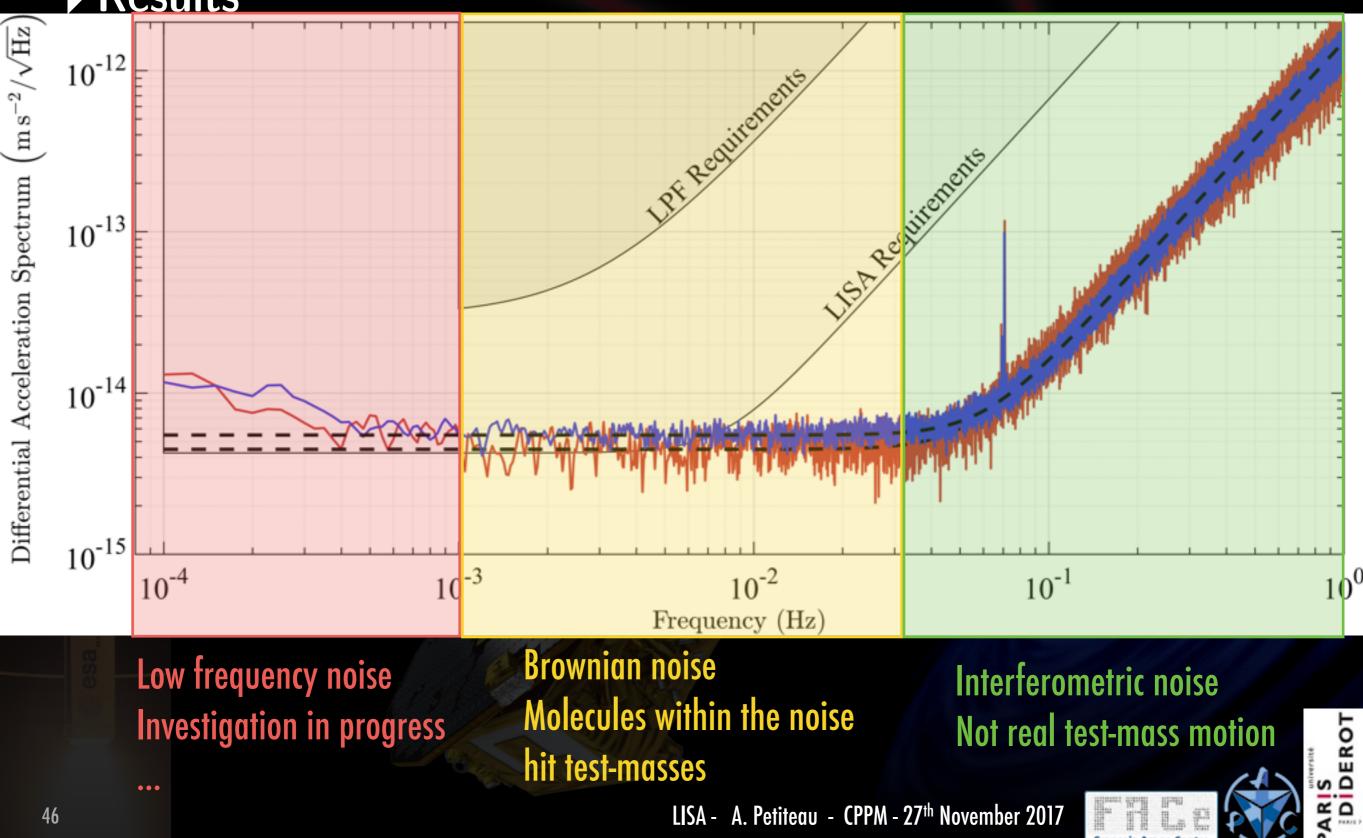
- Noise in 1–10 mHz: brownian noise due to residual pressure:
 - Molecules within the housing hitting the test-masses
 - Possible residual outgassing
- Evolution:
 - Pressure decreases with time => constant improvement
- For LISA:
 - Better evacuation system ... pump ?







M. Armano et al. PRL 116, 231101 (2016)



LISA - A. Petiteau - CPPM - 27th November 2017

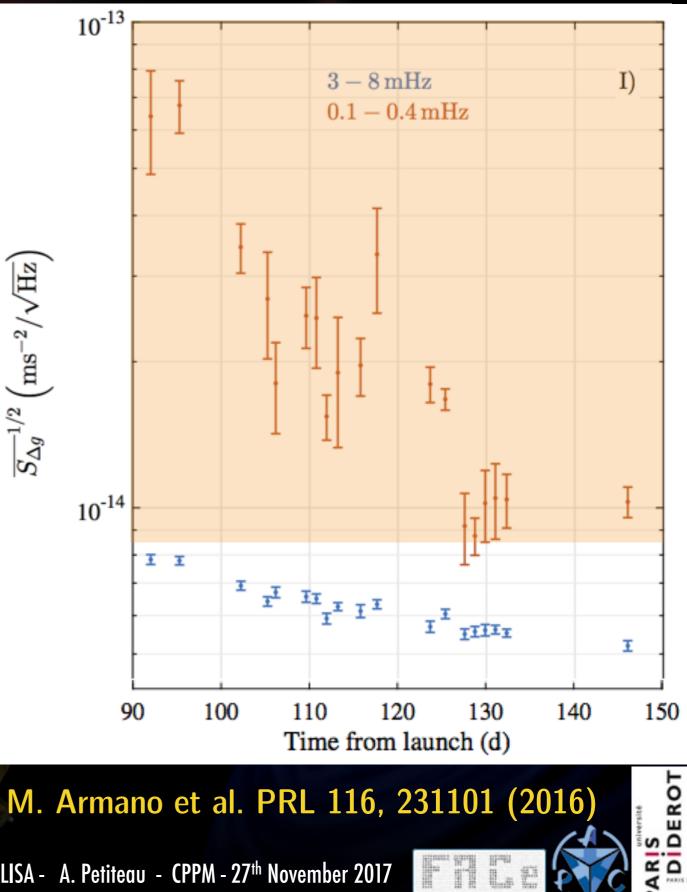
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Low-frequency limit

- Noise in 0.1 1 mHz:
- ► 50% understood: actuation noises
- Still 50% not completely explained:
 - 1/f slope
 - Temperature ? **Small glitches** ?
- Work in progress ...



$\Delta \vec{g}_{\text{tang}} = \vec{g}_{\text{tang},2} - \vec{g}_{\text{tang},1}$ $= (\vec{r}_2 - \vec{r}_1) \times \dot{\vec{\Omega}}$



2 gtan,2



by Joseph Martino

Angle Decorrelation - Euler Forces



Outline



- Introduction to gravitational waves
- Gravitational wave sources in the millihertz regime
- LISA: a space-based gravitational wave observatory
- LISAPathfinder
- **LISA** status and organization
- LISA scientific performances
- The French contribution to LISA:
 - Data Processing Center
 - Integration / performance control
- Conclusion and perspectives





History of LISA



- ▶ 1978: first study based on a rigid structure (NASA)
- ▶ 1980s: studies with 3 free-falling spacecrafts (US)
- ▶ 1993: proposal ESA/NASA: 4 spacecrafts
- ▶ 1996-2000: pre-phase A report
- ► 2000-2010: LISA and LISAPathfinder: ESA/NASA mission
- ▶ 2011: NASA stops => ESA continue: reduce mission
- ► 2012: selection of JUICE L1 ESA
- ▶ 2013: selection of ESA L3 : « The gravitational Universe »
- > 2015-2016: success of LISAPathfinder + detection GWs





LISA technology requirements

- Free flying test mass subject to very low parasitic forces:
 - ✓ Drag free control of spacecraft (non-contacting spacecraft)
 - ✓ Low noise microthruster to implement drag-free
 - ✓ Large gaps, heavy masses with caging mechanism
 - Validated with and a strait of the second strait of the second strait of the second se ✓ High stability electrical actuation on cross degrees of freedom
 - ✓ Non contacting discharging of test-masses
 - ✓ High thermo-mechanical stability of S/C
 - ✓ Gravitational field cancellation
 - Precision interferometric, local ranging of test-mass and spacecraft:

Stators

5

- ✓ pm resolution ranging, sub-mrad alignments
- ✓ High stability monolithic optical assemblies
- Precision million km spacecraft to spacecraft precision ranging:
 - ➡ High stability telescopes
 - High accuracy phase-meter and frequency distribution
 - ➡ High accuracy frequency stabilization (incl. TDI)



History of LISA



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Call for mission at ESA





The LISA Proposal

https://www.lisamission.org/ proposal/LISA.pdf

LISA Laser Interferometer Space Antenna

A proposal in response to the ESA call for L3 mission concepts

Lead Proposer Prof. Dr. Karsten Danzmann

2 Science performance

The science theme of The Gravitational Universe is addressed here in terms of Science Objectives (SOs) and (MRs) are expressed as linear spectral densities of the Science Investigations (SIs), and the Observational Re- sensitivity for a 2-arm configuration (TDI X). quirements (ORs) necessary to reach those objectives. etc. The majority of individual LISA sources will be biis the square root of this quantity, the linear spectral origin are also considered. density $\sqrt{S_b(f)}$, for a 2-arm configuration (TDI X). In

the following, any quoted SNRs for the Observational Requirements (ORs) are given in terms of the full 3arm configuration. The derived Mission Requirements

The sensitivity curve can be computed from the in-The ORs are in turn related to Mission Requirements dividual instrument noise contributions, with factors (MRs) for the noise performance, mission duration, that account for the noise transfer functions and the sky and polarisation averaged response to GWs. Requirenary systems covering a wide range of masses, mass ra-ments for a minimum SNR level, above which a source tios, and physical states. From here on, we use M to re- is detectable, translate into specific MRs for the obserfer to the total source frame mass of a particular system. vatory. Throughout this section, parameter estimation The GW strain signal, h(t), called the waveform, to- is done using a Fisher Information Matrix approach, gether with its frequency domain representation $\hat{h}(f)$, assuming a 4 year mission and 6 active links. For longencodes exquisite information about intrinsic param- lived systems, the calculations are done assuming a eters of the source (e.g., the mass and spin of the in- very high duty-cycle (> 95%). Requiring the capabilteracting bodies) and extrinsic parameters, such as inclination, luminosity distance and sky location. The curacy sets MRs that are generally more stringent than assessment of Observational Requirements (ORs) re- those for just detection. Signals are computed accordquires a calculation of the Signal-to-Noise-Ratio (SNR) ing to GR, redshifts using the cosmological model and and the parameter measurement accuracy. The SNR parameters inferred from the Planck satellite results, is approximately the square root of the frequency in- and for each class of sources, synthetic models driven tegral of the ratio of the signal squared, $\tilde{h}(f)^2$, to the by current astrophysical knowledge are used in order sky-averaged sensitivity of the observatory, expressed to describe their demography. Foregrounds from asas power spectral density Sh(f). Shown in Figure 2 trophysical sources, and backgrounds of cosmological

DIDEROT

Page 7

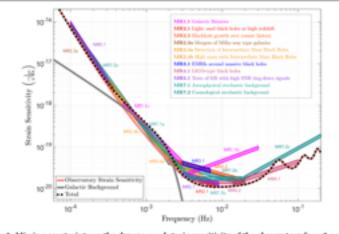


Figure 2: Mission constraints on the sky-averaged strain sensitivity of the observatory for a 2-arm configuration (TDI X), $\sqrt{S_b(f)}$, derived from the threshold systems of each observational requirement.

François Arago Centre

LISA - 2. SCIENCE PERFORMANCE

LISA - A. Petiteau





LISA science objectives

- SO1: Study the formation and evolution of compact binary stars in the Milky Way Galaxy.
- SO2: Trace the origin, growth and merger history of massive black holes across cosmic ages
- ► SO3: Probe the dynamics of dense nuclear clusters using EMRIs
- SO4: Understand the astrophysics of stellar origin black holes
- SO5: Explore the fundamental nature of gravity and black holes
- SO6: Probe the rate of expansion of the Universe
- ► SO8: Search for GW bursts and unforeseen sources





LISA concept in the proposal



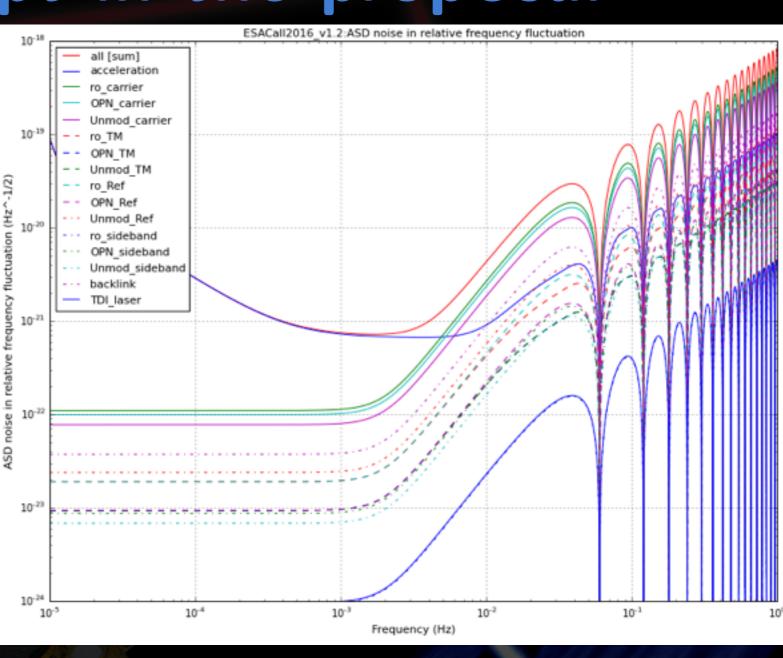
DEROT

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- ▶ 3 arms, 2.5 km
- Launch Ariane 6.4
- Propulsion:
 - micro-prop: cold gaz
 - prop. module
- Frequency band:

 $\begin{array}{ll} 100 \ \mu \mathrm{Hz} \leq f \leq 0.1 \, \mathrm{Hz} & \mathrm{req.} \\ 20 \ \mu \mathrm{Hz} \leq f \leq 1 \, \mathrm{Hz} & \mathrm{goal} \end{array}$

- Noise budget:
 - Acceleration => LISAPathfinder
 - Interferometric Measurement System



0.4 mHz

2 mHz

LISA - A. Petiteau - CPPM - 27th November 2017

 $S_{\rm IFO}^{1/2} \le 10 \cdot 10^{-12} \frac{\rm m}{10^{-12}}$

 $S_a^{1/2} \le 3 \cdot 10^{-15} \frac{\text{m s}^{-2}}{(1-1)^{-15}}$

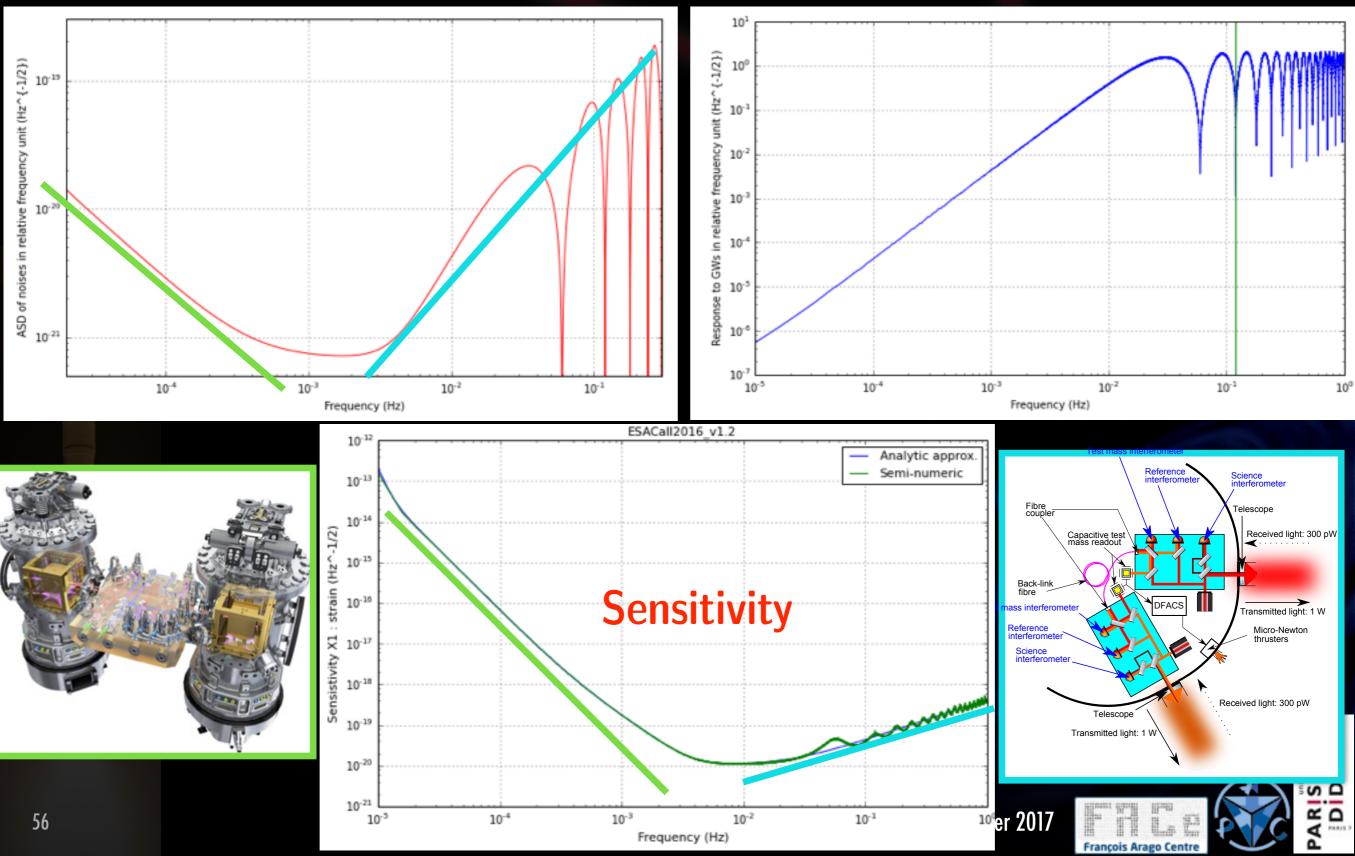


Sensitivity



Noises

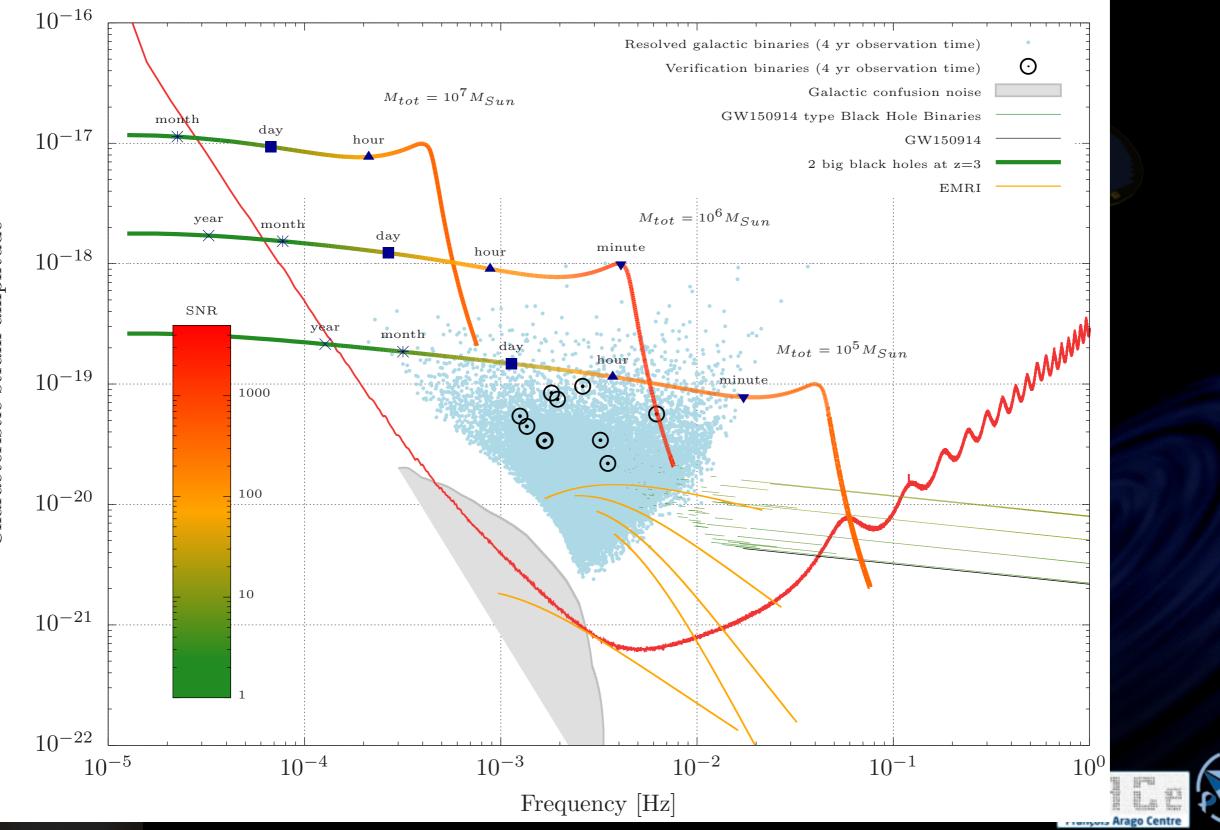
Response of the detector to GWs





PARIS DIDEROT

GW sources

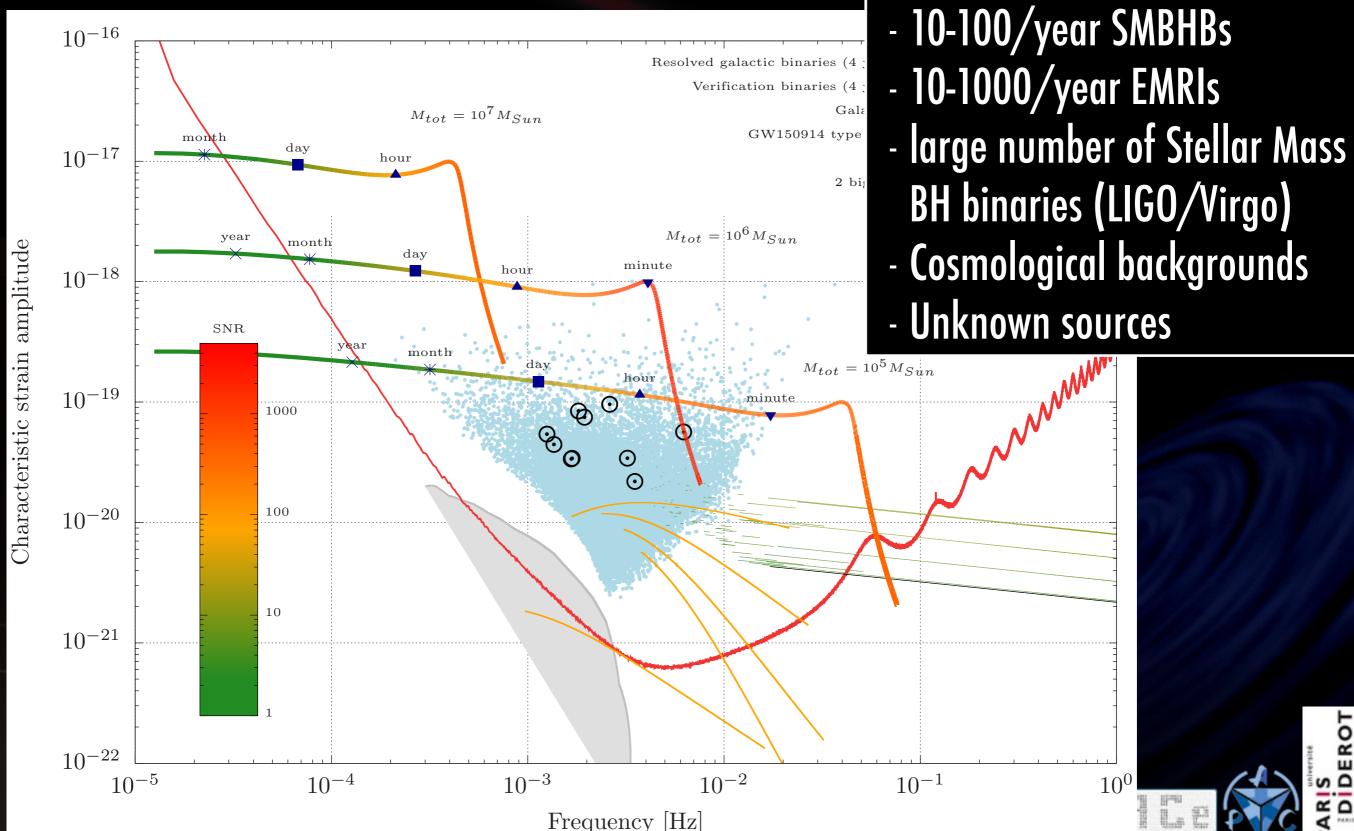


GW sources



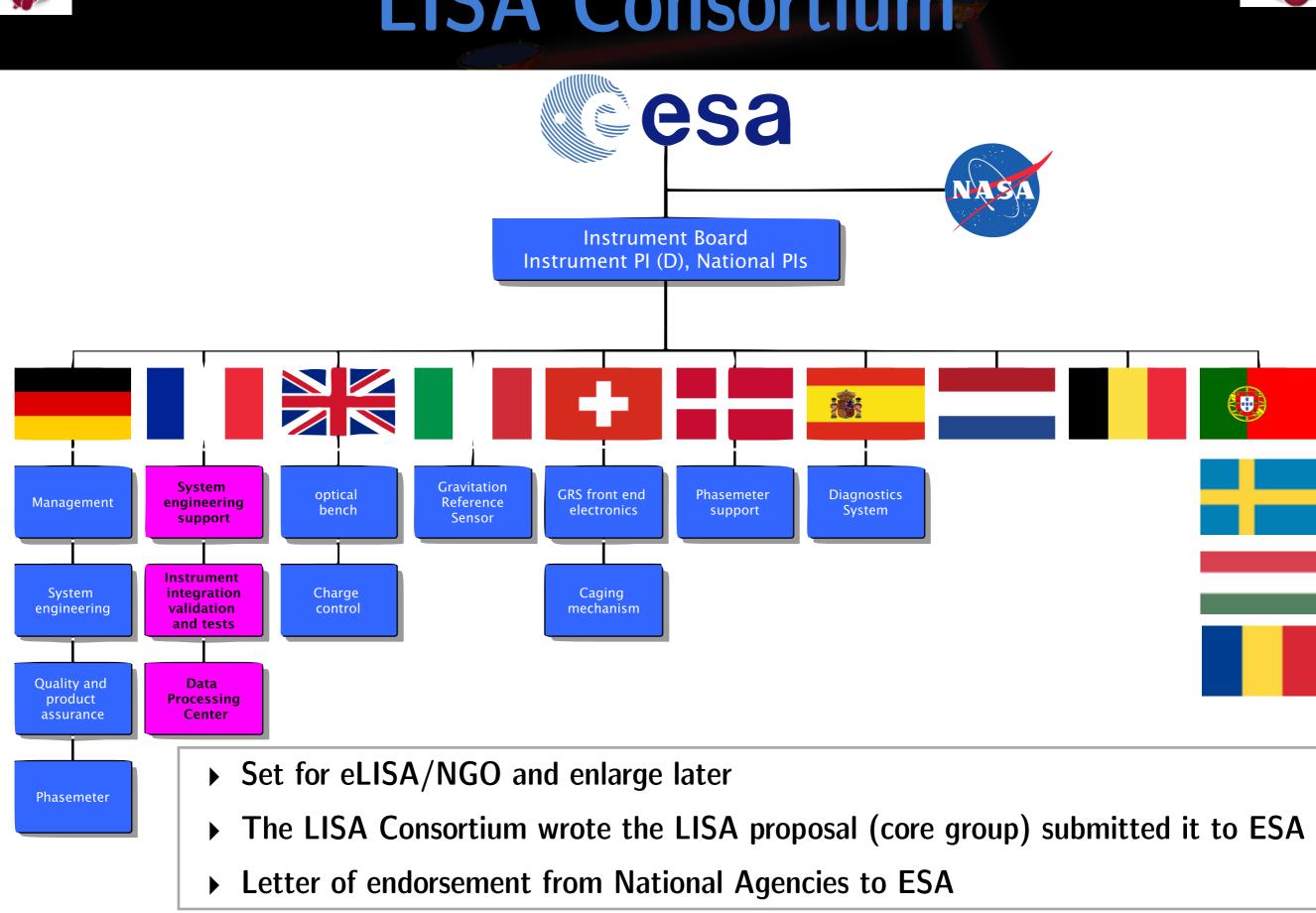
DIDEROT

- 6 x10⁷ galactic binaries





LISA Consortium



the m in district, part François Arago Centre

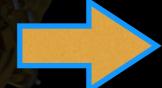


LISA at ESA



DEROT

- > 25/10/2016 : Call for mission
- > 13/01/2017 : submission of «LISA proposal» (LISA consortium)
- ▶ 8/3/2017 : Phase 0 mission (CDF 8/3/17 → 5/5/17)
- > 20/06/2017 : LISA mission approved by SPC
- ▶ 8/3/2017 : Phase 0 payload (CDF June → November 2017)
- ► 2018 \rightarrow 2020 : competitive phase A : 2 companies compete
- ▶ $2020 \rightarrow 2022$: B1: start industrial implementation
- ► 2022-2024 : mission adoption
- During about 8.5 years : construction
- ► 2030-2034 : launch Ariane 6.4
- ▶ 1.5 years for transfert
- > 4 years of nominal mission









ESA Phase 0 mission

- ▶ 13 Concurrent Design Facility from March to May 2017
- Conducted by ESA with few members of the consortium
- Drivers: thermal stability/range, mechanical stability, mass, power, data rate, volume, integration, …
- Several studied options:
 - Propulsion: chemical (CP) / electrical (EP & EP+)
 - Micro-propulsion: cold-gas (CP & EP)/ electrical (EP+)
 - Communication,
 - Shape,
 - Launch strategies, orbits,





ESA Phase 0 mission



EP+

4.4

170.6

175

117

20

1522

0

EP

190.2

80.7

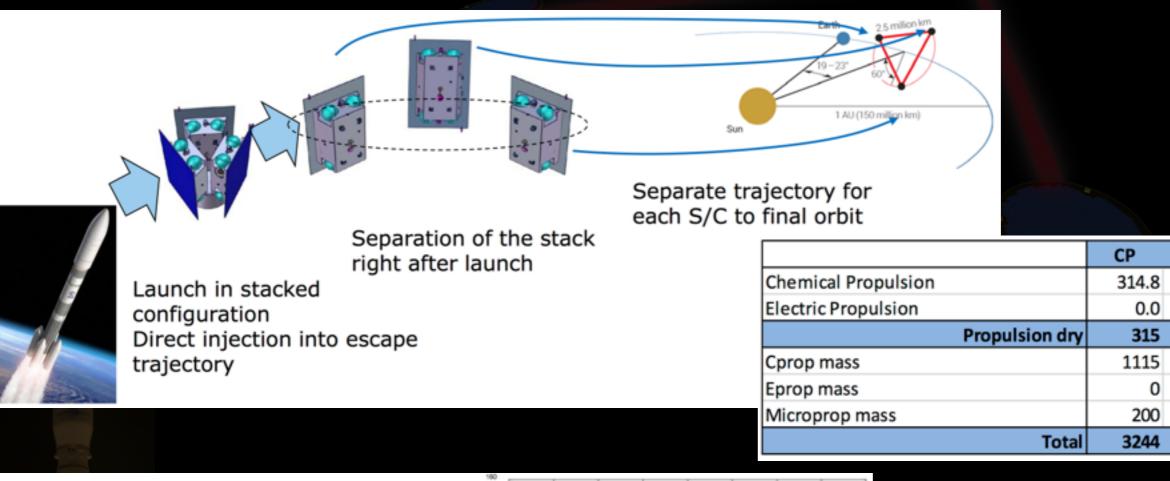
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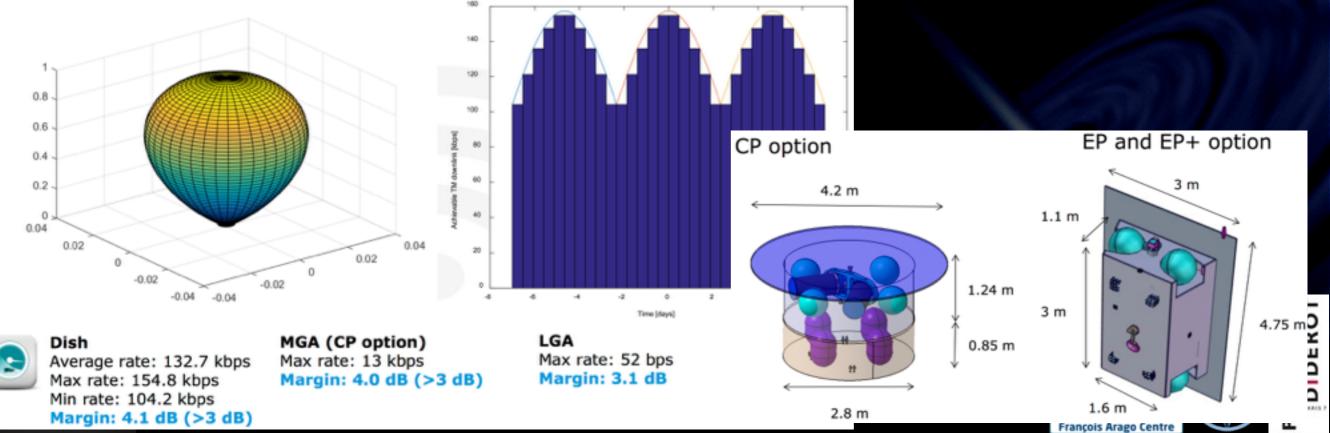
148

240

1881

0









- From June to November
- Conducted by Payload Coordination Team with ESA
- Support of ESA CDF
 - => Write the Payload Definition Document:
 - System requirements
 - Architecture
 - Budgets
 - Commissioning
 - Communications
 - Control
 - Critical items
 - Data
 - Electrical
 - Environment

- Subsystems:
 - Laser
 - Diagnostics
 - Gravitational Reference Sensor
 - Mechanisms
 - Optical Bench
 - Telescope
 - Constellation Acquisition Sensor
 - PhaseMeter

DEROT





ESA next steps

- 1. Payload Definition Document (PDD)
- 2. ESA appointed the Science Study Team: responsible for the Science Requirement Document (SRD):
- 3. SRD + PDD => input to the Mission Requirement Document (MRD)
- 4. MRD is used to defined the ITT: invitation to industries All these steps need to be done by december 2017
- 5. March 2018: start of phase A

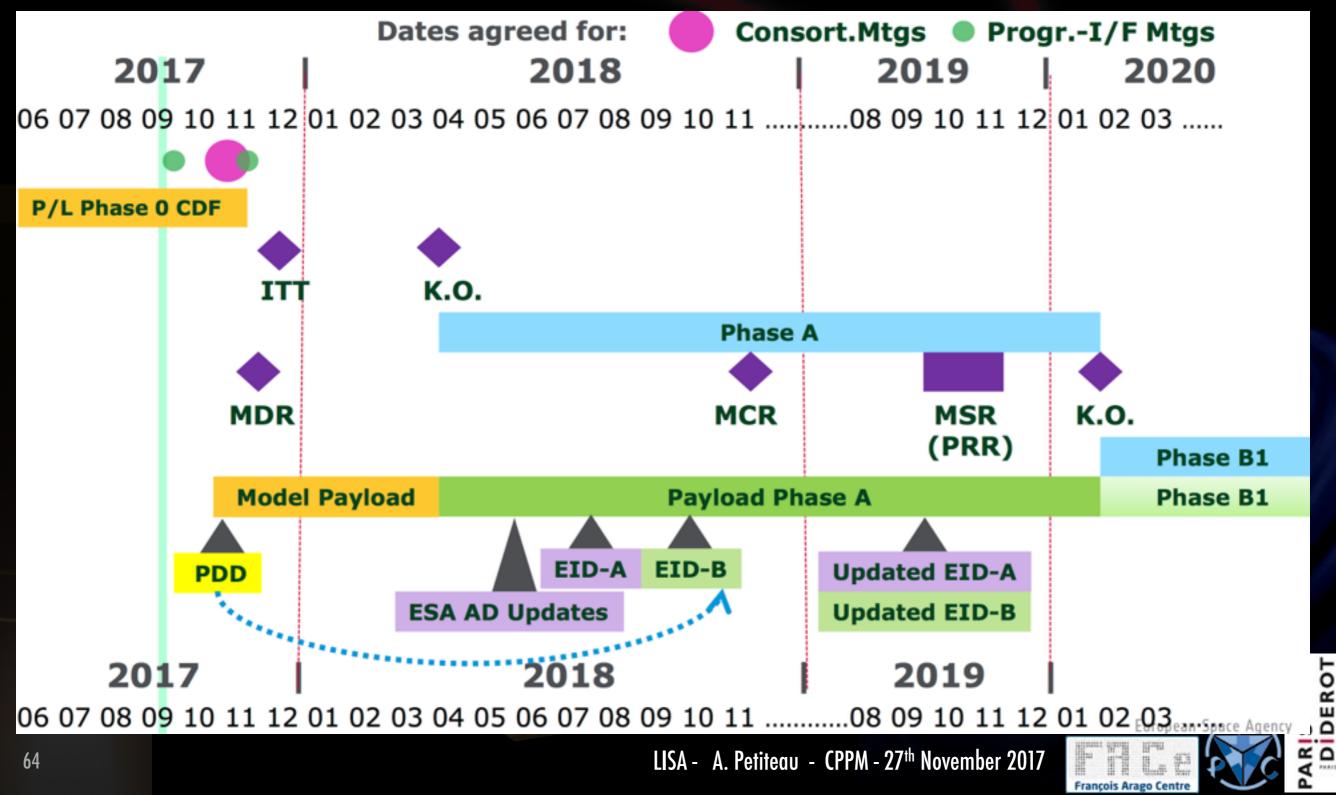




ESA Next steps

Mission Definition/Consolidation/Selection/Adoption Review,

Payload Definition Doc, Experiment Interface Doc.





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ESA Team(s)

- Project Study Scientist: Paul McNamara
- Project Study Manager: Martin Gelher
- ► CDF: Diego Escorial Olmos + ESA experts
- Science Study Team:
 - LISA Europe:
 - K. Danzmann (Germany)
 - M. Colpi (Italy)
 - P. Jetzer (Switzerland)
 - M. Hewitson (Germany)
 - G. Nelemans (Neederland)
 - A. Petiteau (France)
 - C. Sopuerta (Spain)
 - H. Ward (UK)

- External:
 - N. Tanvir
 - J. Hjorth
- LISA US:
 - K. Holley-Bockelmann
 - D. Shoemaker
- Observers:
 - O. Jennrich (ESA)
 - I. Thorpe (NASA)
 - R. Sambruna (NASA)

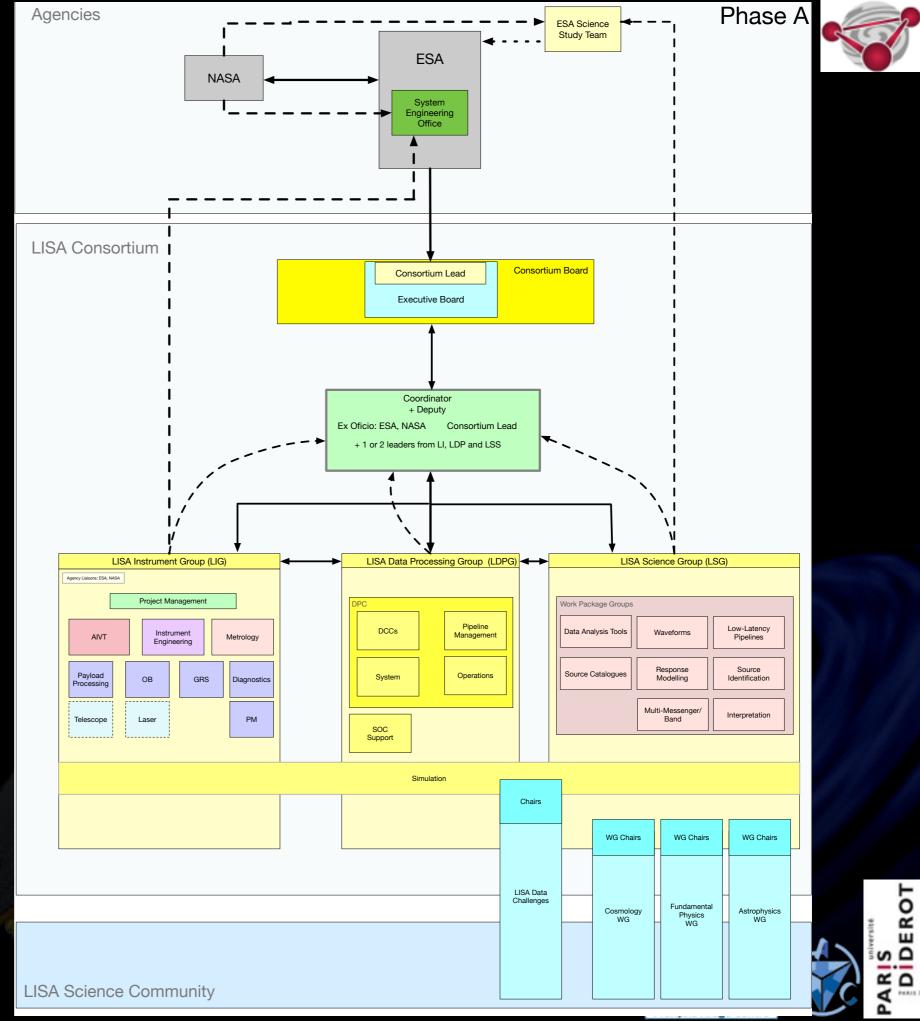




DEROT



Consortium organisation



66



Outline



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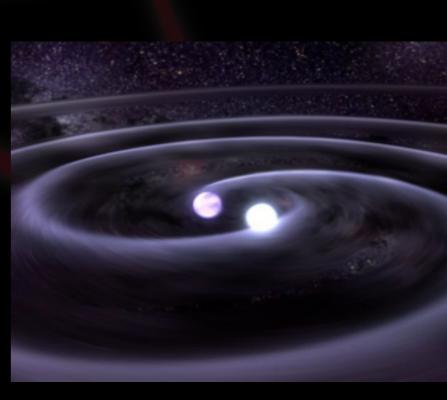




Galactic binaries

- Gravitational wave:
 - quasi monochromatic
- Duration: permanent
- Signal to noise ratio:
 - detected sources: 7 1000
 - confusion noise from non-detected sources
- Event rate:
 - 25 000 detected sources
 - more than 10 guarantied sources (verification binaries)

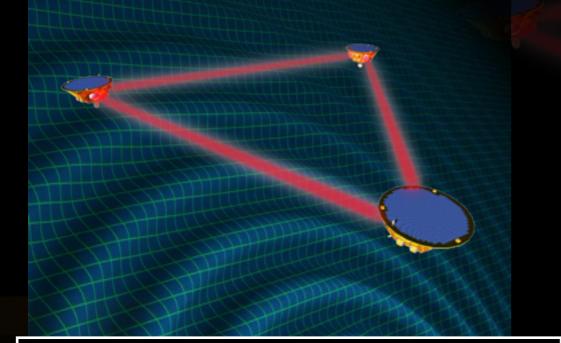




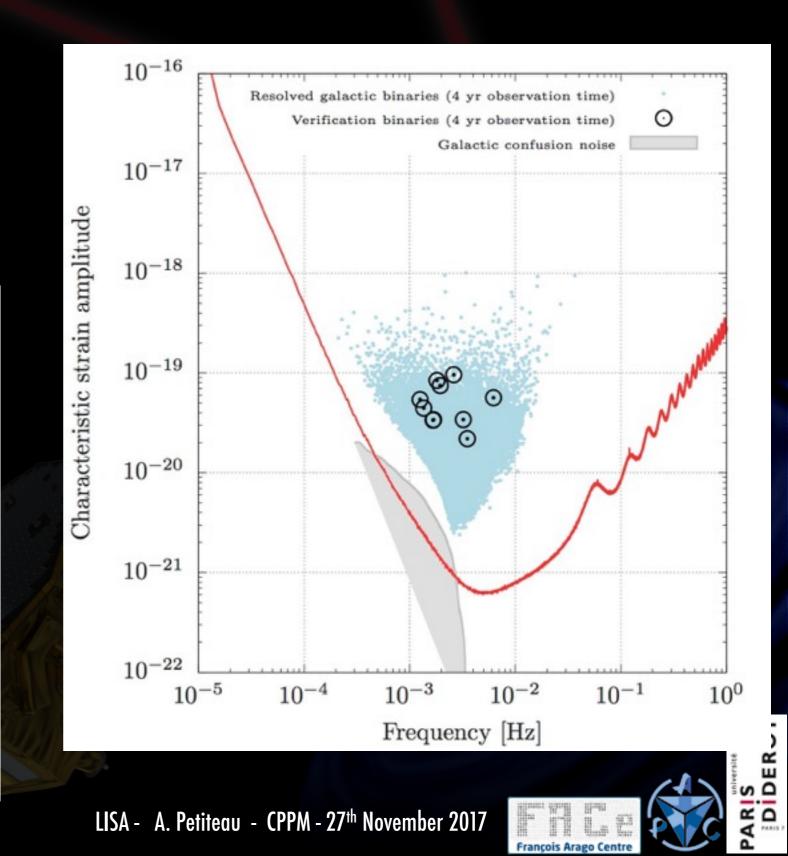


Galactic binaries





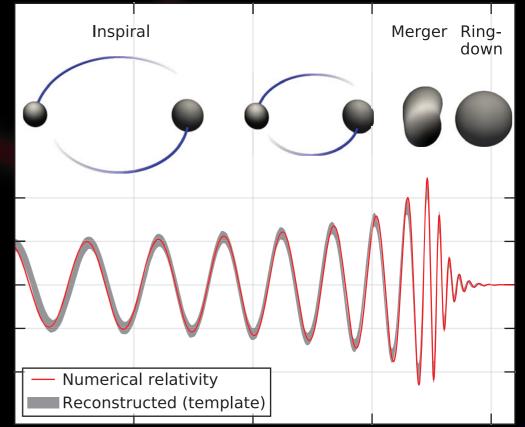
GW sources - 6 x10⁷ galactic binaries



Super Massive Black Hole Binaries

• Gravitational wave:

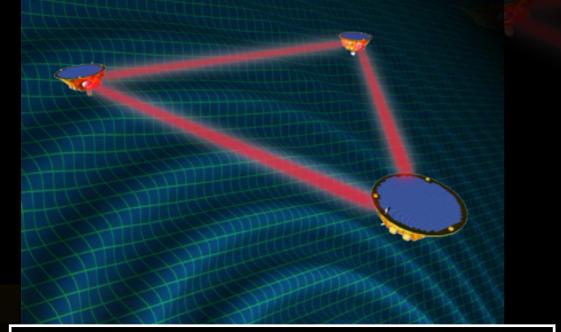
- Inspiral: Post-Newtonian,
- Merger: Numerical relativity,
- Ringdown: Oscillation of the resulting MBH.



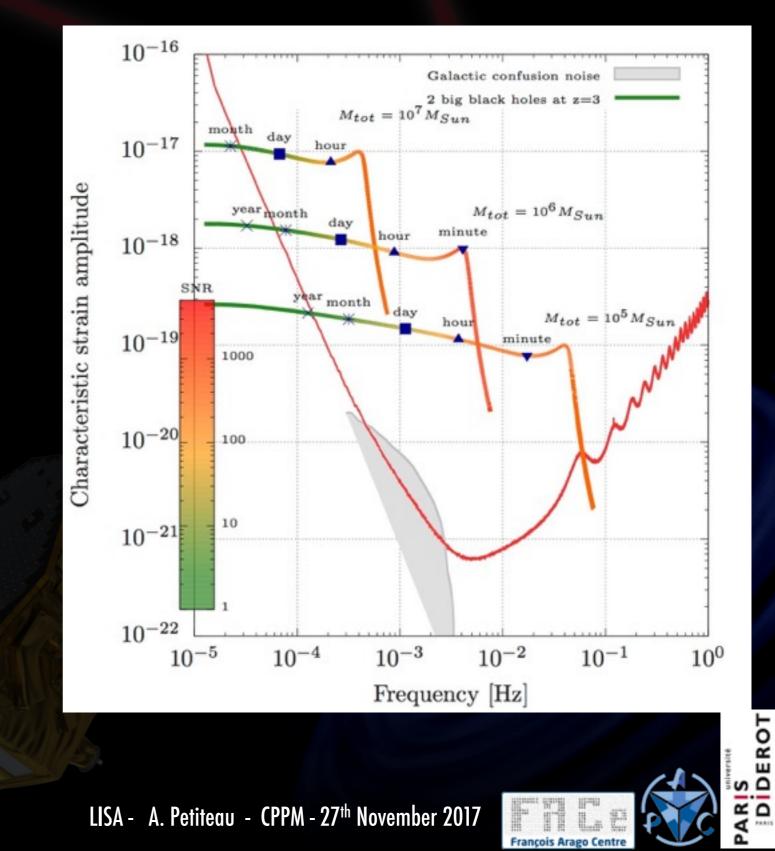
Duration: between few hours and several months Signal to noise ratio: until few thousands Event rate: 10-100/year







OG sources - 6 x10⁷ galactic binaries - 10-100/year SMBHBs





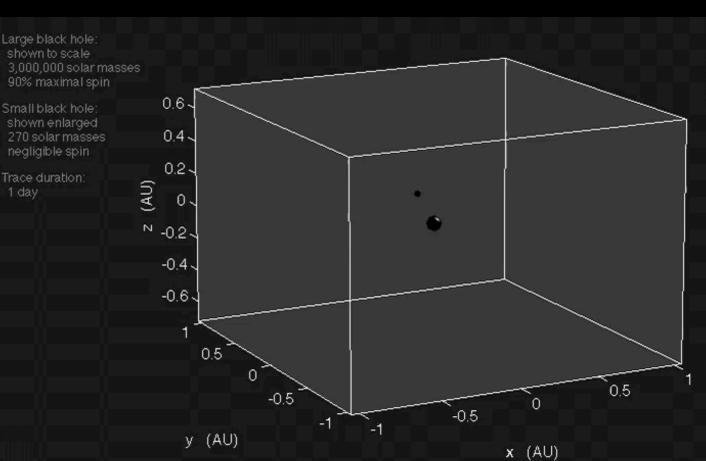




- **Gravitational wave:**
 - very complex waveform
 - No precise simulation at the moment
- Duration: about 1 year
- Signal to Noise Ratio: from tens to few hundreds

1 day

Event rate: from few events per year to few hundreds



Steve Drasco Max Planck Institute for Gravitational Physics (Albert Einstein Institute) sdrasco@aei.mpd.de



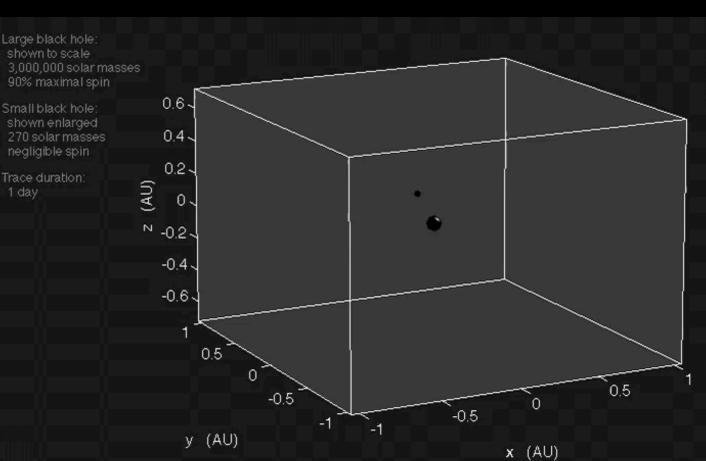




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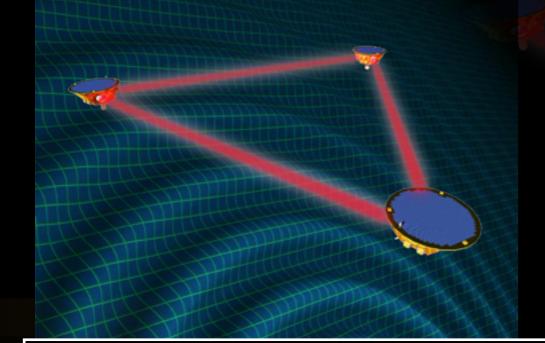


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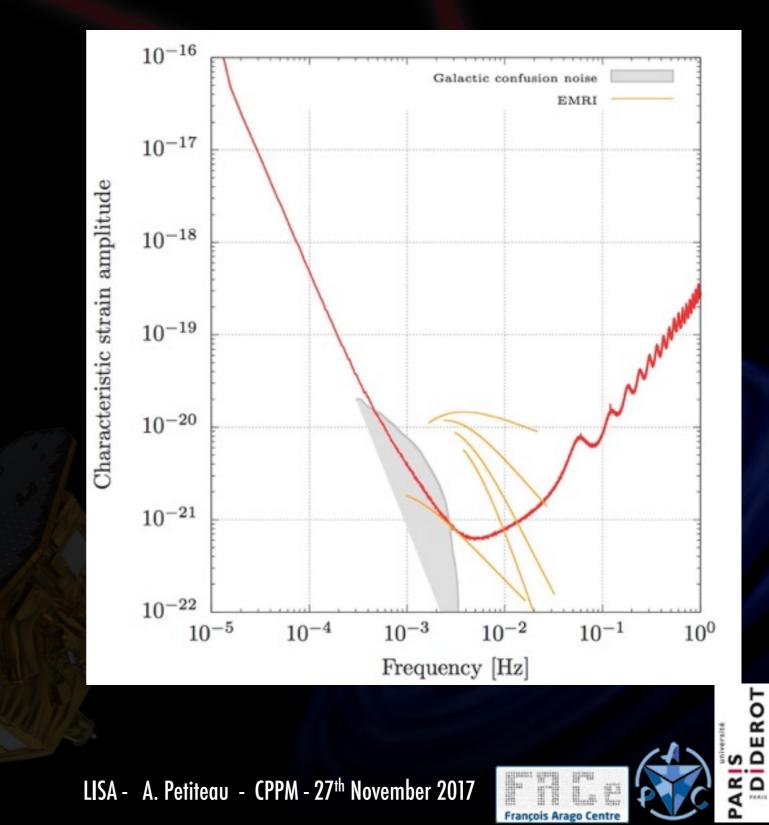








OG sources - 6 x10⁷ galactic binariess - 10-100/year SMBHBs - 10-1000/years EMRIs





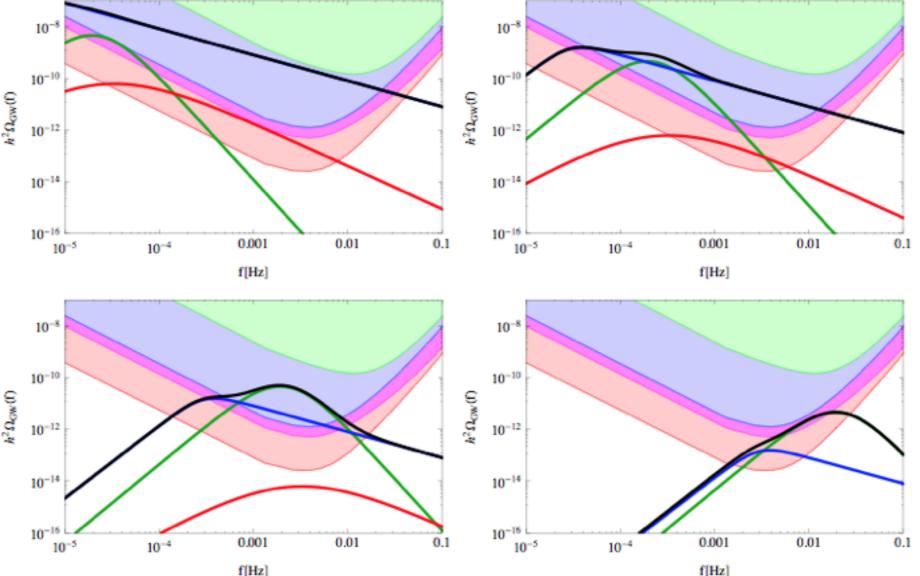


Cosmological backgrounds

- ► Work in progress for LPF-LISA ...
- But studies done in the context of eLISA already showed:
 - Ex: first order phase transition in the very early Universe Caprini et al.
 JCAP 04, 001

 (2016)

 Cosmic strings
 - Cosmic strings network



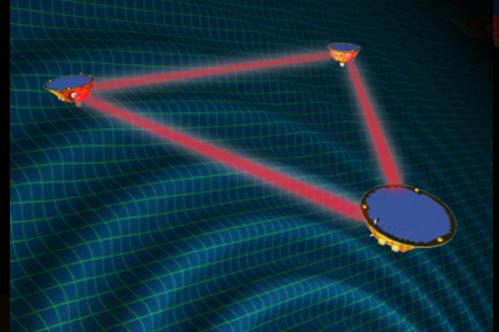


Others sources

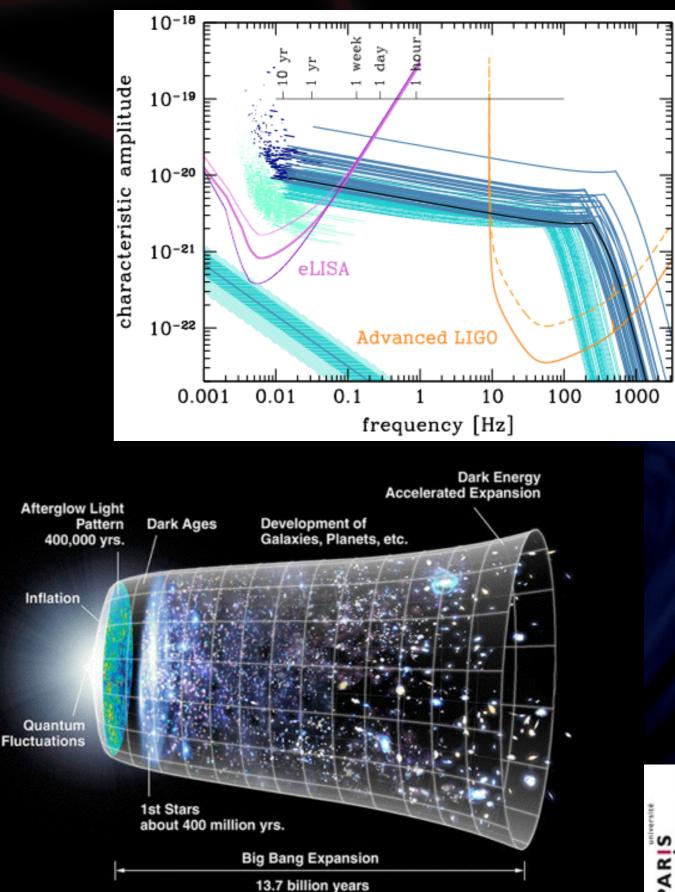


DEROT

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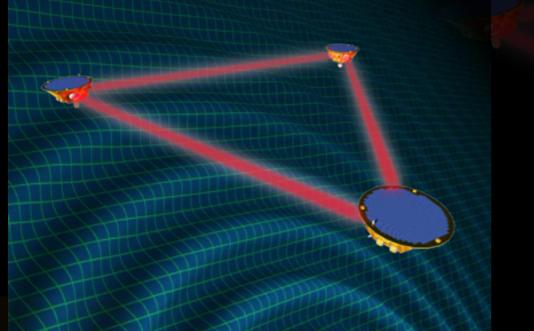
GW sources - 6 x10⁷ galactic binaries - 10-100/year SMBHBs - 10-1000/year EMRIs - large number of Stellar Origin BH binaries (LIGO/Virgo) - Cosmological backgrounds - Unknown sources



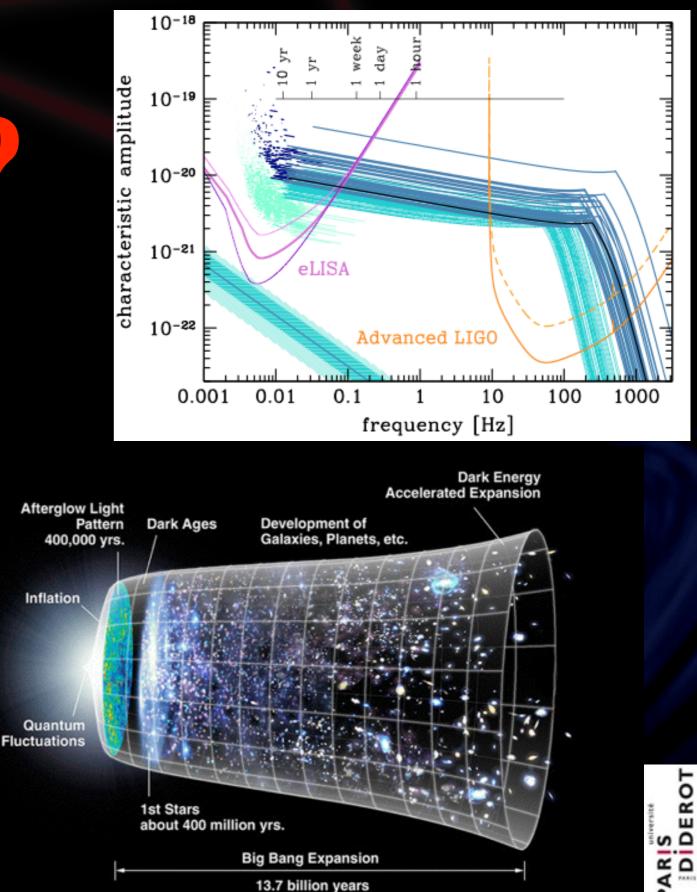


Others sources





GW sources - 6 x10⁷ galactic binaries - 10-100/year SMBHBs - 10-1000/year EMRIs - large number of Stellar Origin BH binaries (LIGO/Virgo) - Cosmological backgrounds - Unknown sources





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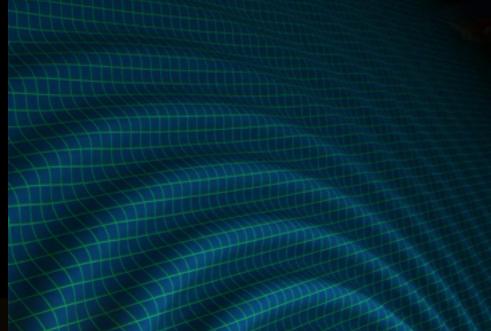








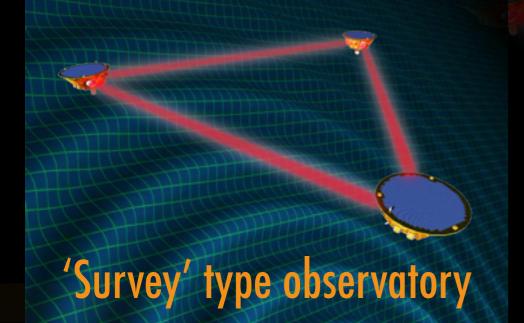


















Phasemeters (carrier, sidebands, distance)

+ Gravitational Refe-rence Sensor
+ Auxiliary channels

'Survey' type observatory

Gravitational wave sources emitting between 0.02mHz and 1 Hz



DEROT





Phasemeters (carrier, sidebands, distance)

+ Gravitational Refe -rence Sensor
 + Auxiliary channels

'Survey' type observatory





a m	
(2)	
1.5	
100	

'Survey

	Λ		
	Source		



	Source		Measurement	Channel Count	Sample Rate [Hz]		Rate [bits/s]
			Payload				
			Inter-S/C IFO	2	3,0	64	384,0
Dhacanat			Test Mass IFO	2	3,0	64	384,0
Phasemet	IFO Longitudinal		Test mass y IFO	0	3,0	64	0,0
			Reference IFO	2	3,0	64	384,0
sidebands	, (Clock Sidebands error point	4	3,0	64	768,0
					3,0	32	96,0
	Freg reference		feedback	2	3,0	32	192,0
+ Gravita	tic		clock sidebands monitoring (local pilot tone beat)	1	3,0	32	96,0
	IFO Angular		SC η,φ	4	0,0	32	384,0
-rence	Se		ΤΜ η,φ	4	0,0	32	384,0
TCHEC A			TM θ (from y IFO)	0	- / -	32	0,0
	Ancillary		Time Semaphores	4	0,0	64	768,0
Auxilia			PRDS metrology	4	3,0	32	384,0
	Optical Monitoring		Optical Truss	0	0,0	32	0,0
type obcervatory			TM x,y,z	0	0,0	32 32	0,0 192,0
' type observatory	DFACS / GRS Cap. Sens.		TM X,y,2 TM θ,η,φ	6	.,.	32	192,0 192,0
			breathing errorpoint	0	-,-	32	0,0
			breathing actuator	2	.,.	32	0,0 64,0
			TM applied torques	12	-	24	288,0
	DFACS		TM applied forces	12	y -	24	288,0
			SC applied torques	3	,	24	72,0
			SC applied forces	3	-	24	72,0
		Themometers	EH	16	0,1	32	51
			OB	20	0,1	32	64
			Telescope	10	-,.	32	32
			interface	10		32	32
ional wave sources	Science Diagnostics	Magnetometers	ТМ	12		32	38
		radiation monitor		1			30
F		FIOS output powers (Inloop and Out of Loop)		6	3,0	32	576
between 0.02mHz		pressure sensor		0	0,1	32	0
		body mic	CGAS tanks	0	3,0	32	0
		1	breathing mechanism	0	3,0	32	0

Gravitati emitting and 1 Hz

	i i i i i i i i i i i i i i i i i i i	7	5,0	52	504,0
	Outlined Trans				0,0
	•				0,0
DFACS / GRS Cap. Sens.					192,0
					192,0
					0,0
	-				64,0
					288,0
					288,0
					72,0
					72,0
		16			51
Themometers					64
		10			32
					32
-	ТМ		0,1	32	38
		1			30
(Inloop and Out of		6	3,0	32	576
		2	0.4		
	CCAC to all a				0
body mic					0
RIN monitoring	2 lasers, 2 frequencies, 2				0
RIN monitoring	quadratures	8	3,0	32	768
			0,0		0
			0,0		0
					1000
					7984
	Platform				
					4000
					4000
	Totals				
					11984
Raw Rate per SC Packetisation Overhead [10%]					1198
					13182
					39546
	Magnetometers radiation monitor FIOS output powers	Optical TrussTM x,y,zTM θ,η,φbreathing errorpointbreathing actuatorTM applied torquesTM applied torquesSC applied torquesSC applied forcesSC applied forcesSC applied forcesSC applied torquesThemometersThemometersTMradiation monitorFIOS output powers (Inloop and Out of Loop)pressure sensorbody micCGAS tanks breathing mechanism 2 lasers, 2 frequencies, 2 quadraturesRIN monitoringPlatform	Optical Truss0TM x,y,z6TM θ,η,φ6breathing errorpoint0breathing actuator2TM applied torques12TM applied torques3SC applied torques3SC applied forces3SC applied forces3SC applied forces10ThemometersEHOB20Telescope10interface10MagnetometersTMFIOS output powers6(Inloop and Out of Loop)0body micCGAS tanks breathing mechanism 2 lasers, 2 frequencies, 2 quadratures8RIN monitoringPlatform	Optical Truss 0 3,0 TM x,y,z 6 1,0 TM θ,η,φ 6 1,0 breathing errorpoint 0 1,0 breathing actuator 2 1,0 TM applied torques 12 1,0 TM applied torques 12 1,0 SC applied torques 3 1,0 SC applied torques 3 1,0 SC applied forces 3 1,0 B 20 0,1 Themometers TM 12 0,1 Telescope 10 0,1 1 FIOS output powers TM 12 0,1 (Intoop and Out of Loop) CGAS tanks 0 3,0 2 lasers, 2 frequencies, 2 8 3,0 0,0 0,0	Optical Truss 0 3,0 32 TM x,yz 6 1,0 32 TM 9,n,\$ 6 1,0 32 breathing errorpoint 0 1,0 32 truss 12 1,0 32 trusplied torques 12 1,0 24 SC applied torques 3 1,0 32 Themometers TM 12 0,1 32 Interface 10 0,1 32 30





Phasemeters (carrier, sidebands, distance)

+ Gravitational Refe -rence Sensor
 + Auxiliary channels

'Survey' type observatory







Phasemeters (carrier, sidebands, distance)

+ Gravitational Refe--rence Sensor Auxiliary channels



Calibrations corrections

Resynchronisation (clock)

Time-Delay Interferometry reduction of laser noise

2 data channels TDI non-correlated





Phasemeters (carrier, sidebands, distance)

+ Gravitational Refe--rence Sensor Auxiliary channels



Calibrations corrections

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2 data channels TDI non-correlated

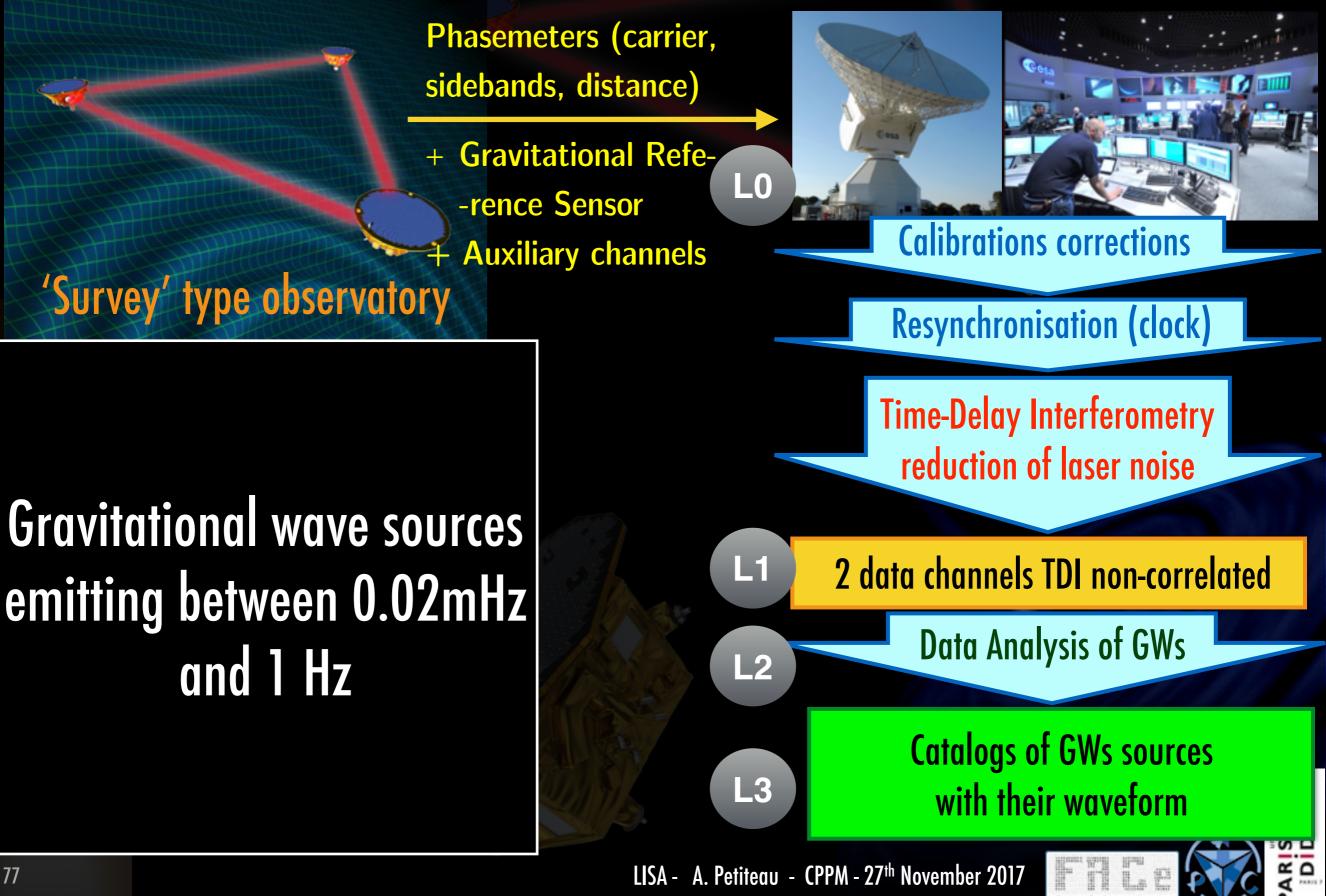
Data Analysis of GWs

Catalogs of GWs sources with their waveform





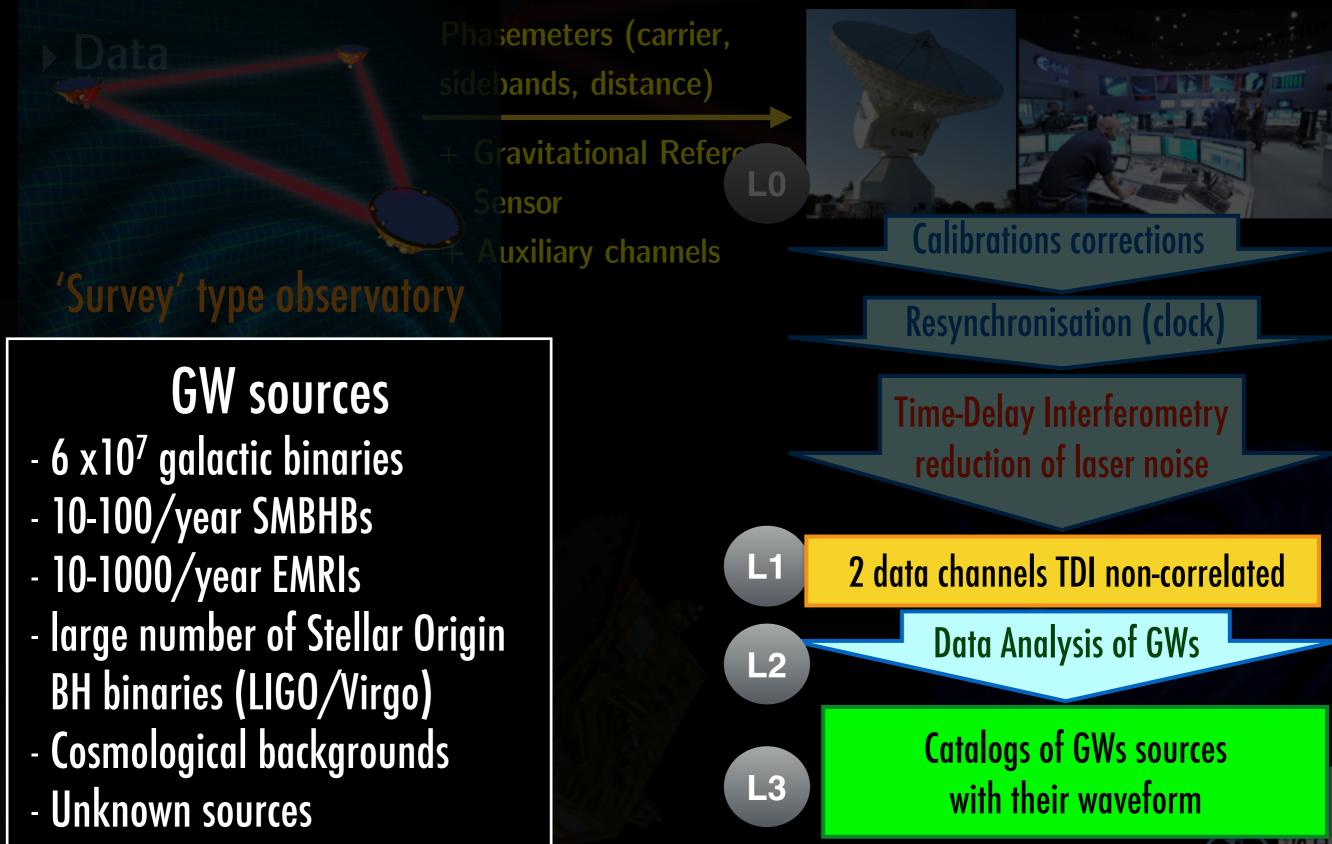






LISA DPC





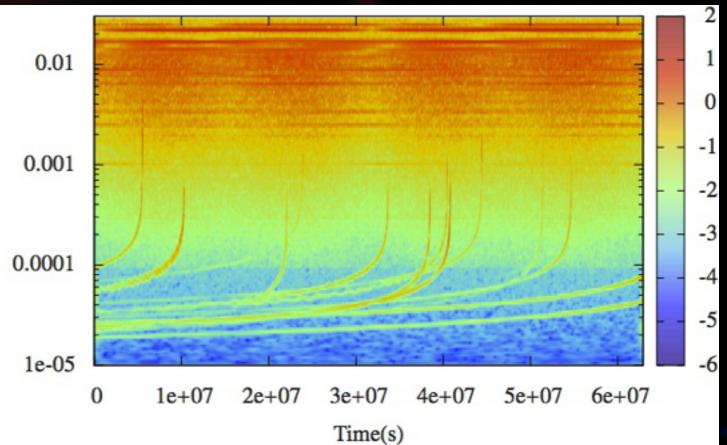
François Arago Centre

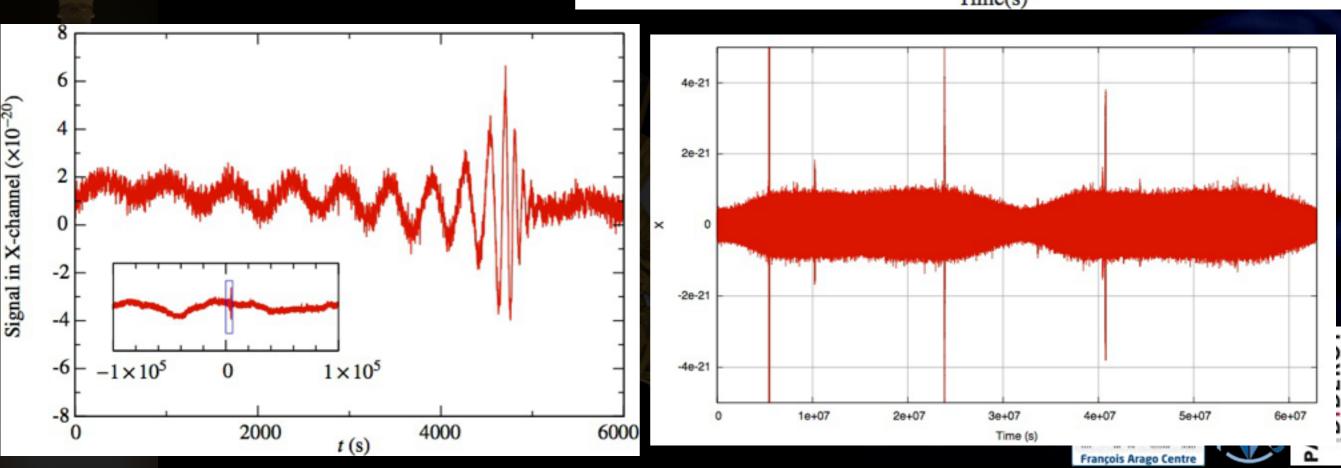


GWs in LISA data

Frequency (Hz)

- Example of simulated data (LISACode):
 - about 100 SMBHs,
 - Galactic binaries

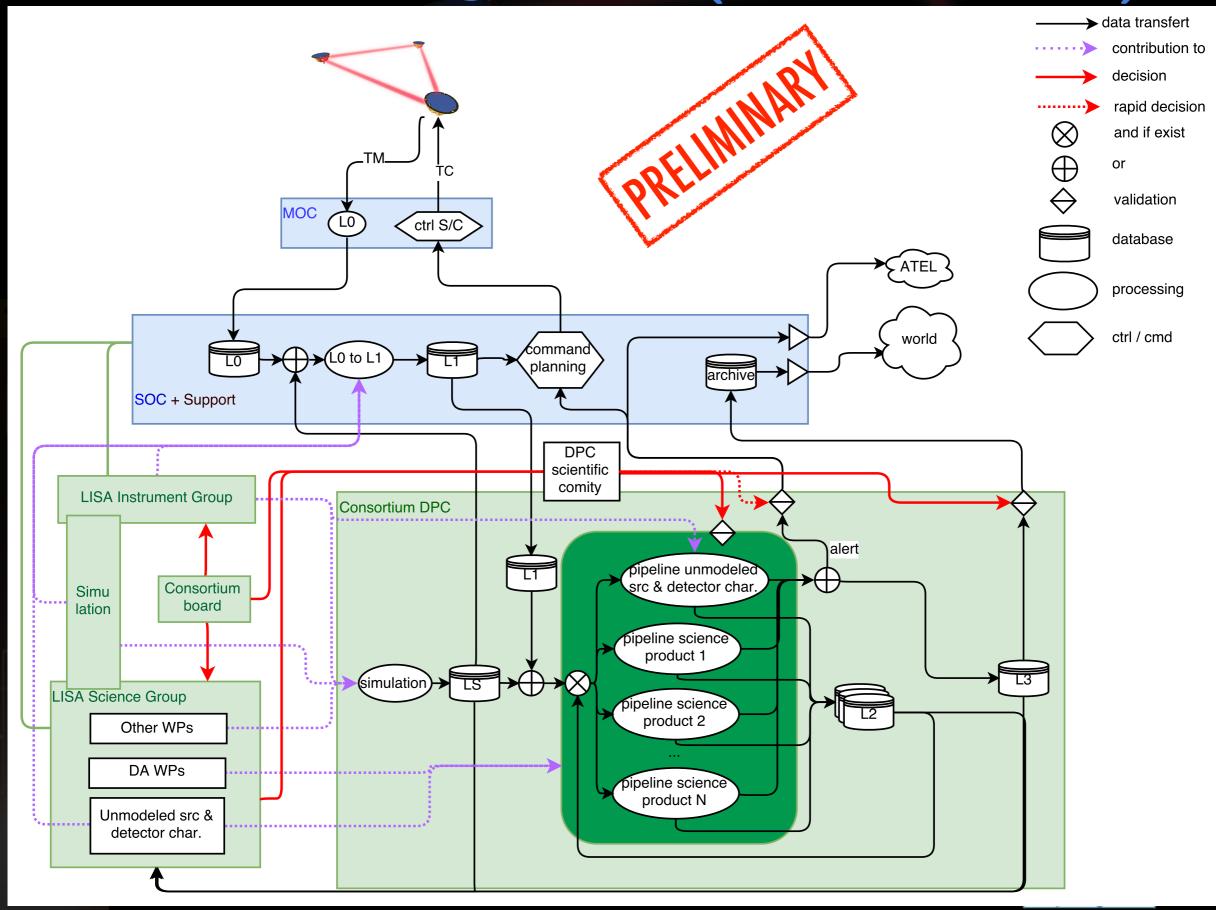




Ground Segment (Consortium)



DIDEROT



80

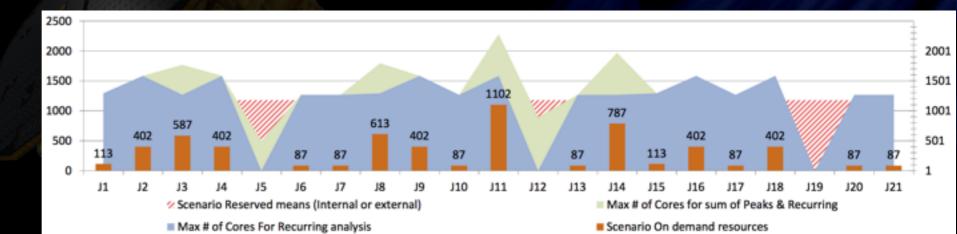


DPC CNES Phase 0



- ▶ In 2013-2014, CNES did a phase-0 with APC & CapGemini
- Results of this Phase-0 :
 - Doable within a reasonable budget (~ 22 millions euros)
 - Developments & pipelines: First analysis of this kind + potential unknown sources \rightarrow Keep flexibility + continuous evolution
 - Infrastructure : fluctuations of the computational charge : permanent sources + transient sources + continuous evolution of codes (full reprocessing phase)

mixed infrastructure based on regular cluster + cloud to absorb variation of needs with time





Current vision of the DPC

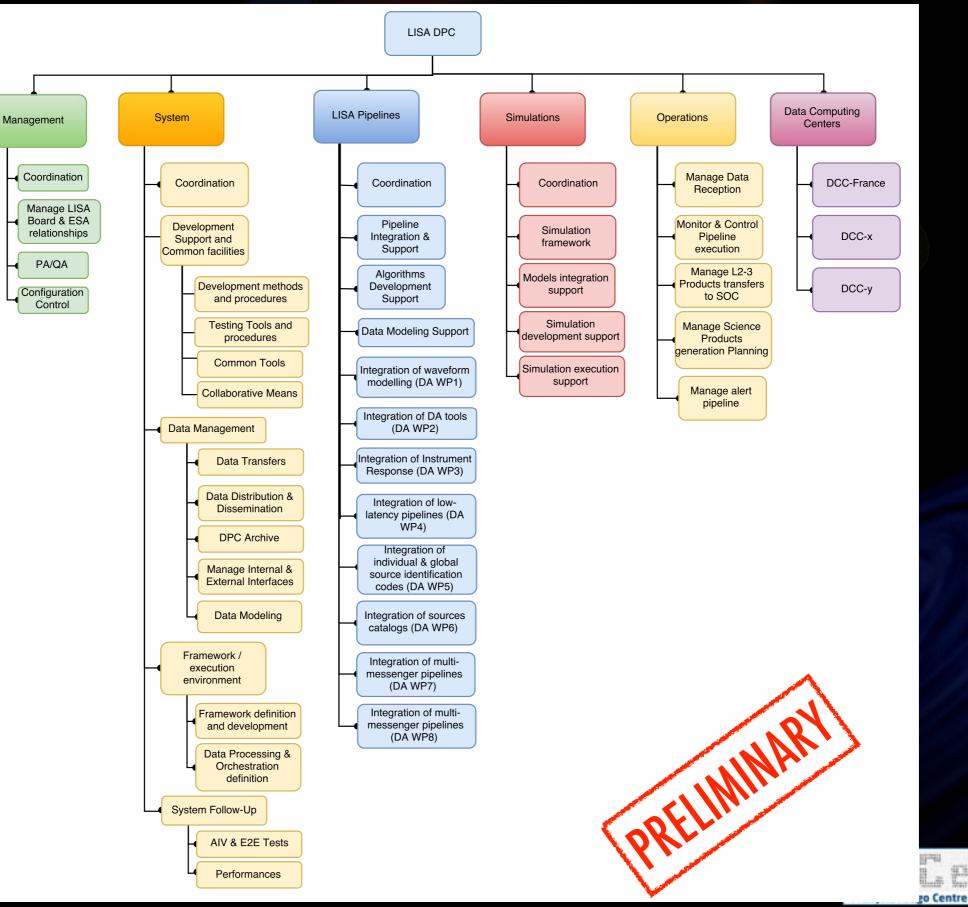


- DPC: unique entity responsable for the data processing (driving, integration of software block, ...)
- DPC in charge of delivering L2 & L3 products + what's necessary to reproduce/refine the analysis (i.e. input data + software + its running environment + some CPU to run it).
- Data Computing Centres (DCC): hardware, computer rooms (computing and storage) taking part to the data processing activities.
- ► The DPC software « suite » can run on any DCC.
 - Software: codes (DA & Simu.) + services (LDAP, wiki, database) + OS.
- First solutions:
 - Separation of hardware and software: ligth virtualization, ...
 - Collaborative development: continuous integration, ...





DPC Organisation



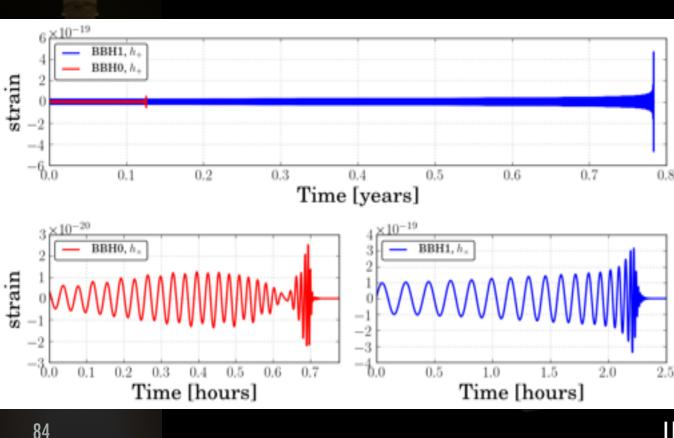


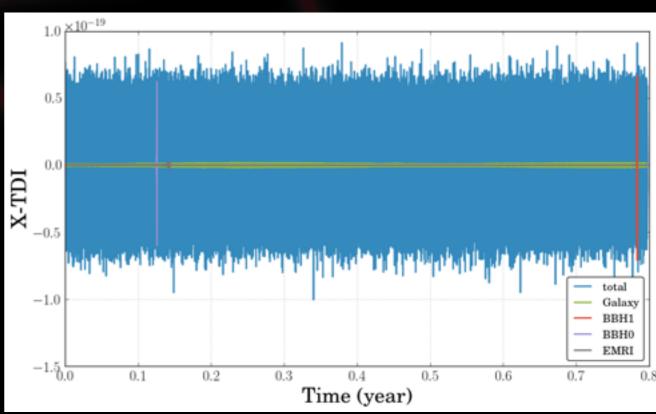


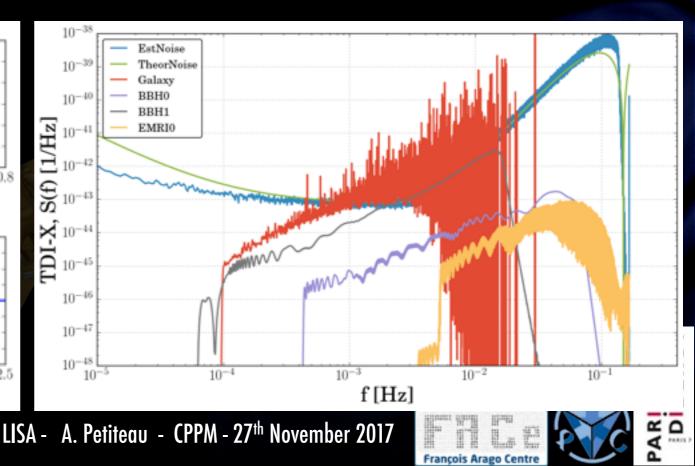


LISA Data Challenges

- ▶ Mock LDC: 2005→2011
- ▶ 2017: start of the LDC
- Develop data analysis
- Design the pipelines of the mission
- Example of the potential data for LDC1











Proto-DPC: basics

- Development environment: in prod. <u>https://elisadpc.in2p3.fr/home</u>
 - Goals
 - Ease the collaborative work: reason why it's already started
 - Guarantee reproducibility of a rapidly evolving & composite DA pipeline
 - Keep control of performance, precision, readability, etc
 - Use existing standard tool
 - Control version system
 - Continuous integration (like in Euclid, LSST)
 - Docker image
 - Done:
 - Simple install of open & standard tools (Jenkins, SonarQube, gitlab CI)
 - Worked on moving from 'simple' to 'automatic' using Docker
 - More projects, more users to come.







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Proto-DPC: basics



- Data basis & data model: in R&D
 - Motivations
 - Data sharing among people and computing centers
 - Mainly processed, temporary or intermediate data: need meta data management to use them
 - A lot of information: a web 2.0 (intuitive) interface is mandatory (search engine, DB request, tree view to show data dependancies, etc)
 - Context
 - Not very big LISA data volume
 - But still implies some specific developments even if using standard data format. One has to define LISA data model first



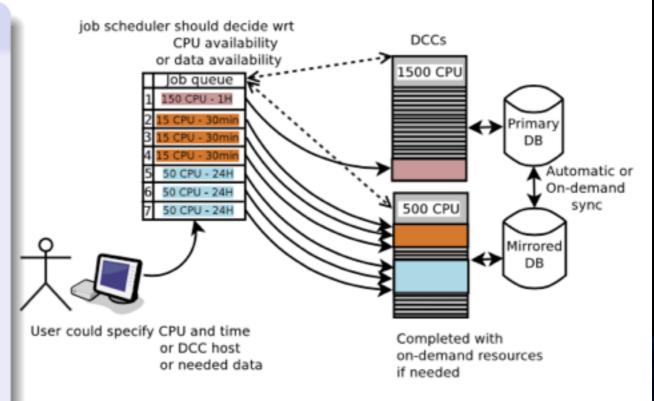


Proto-DPC: basics

Execution environment: in R&D

Objectives: a composite computer center

- Pooling of CPU resources with a single scheduler for all DCCs
 - the user-friendly way to go
 a dynamic CPU pool to adapt the resources to the actual needs (the economic way)
 transfering data if needed
- Assumptions
 - it's easy to plug new hardware
 - it's easy to transfer data



same principles than grid computing with a shorter learning phase.

R&D activities

- Docker ochestrator R&T study performed by CNES
- APC involved in the French cloud network
- Doing some actual testing of cloud platform and containers orchestration (singularity).





Outline



- Introduction to gravitational waves
- Gravitational wave sources in the millihertz regime
- LISA : a space-based gravitational wave observatory
- LISAPathfinder
- LISA scientific performances
- The French contribution to LISA:
 - Data Processing Center
 - Integration / performance control
- Conclusion and perspectives



Integration & performance model

- In LISA, the "instrument" is the satellites' constellation !
 - Highly integrated spacecraft
 - Strong interactions between subsystems (payload & platform)
- A (very) precise knowledge of the noise sources and detector response is required.
- The Consortium must have the hands on a complete and precise performance model:
 - End-to-end simulator (development just started ...)
 - Validation and performance tests designs
 - Tests and checkout benches development
 - Integration and qualification activities









Consortium is responsible for delivering integrated/tested/ validated MOSA



Additional elements:

- MOSA support structure,
- Phase Measurement Subsystem (PMS),
- Laser Assembly (LA),
- Diagnostic subsystem (temperature sensors & heaters)





AIV/T



Consortium is responsible for delivering integrated/tested/ validated MOSA



Additional elements:

- MOSA support structure,
- Phase Measurement Subsystem (PMS),
- Laser Assembly (LA),
- Diagnostic subsystem (temperature sensors & heaters)



ESA/NASA





Models philosophy

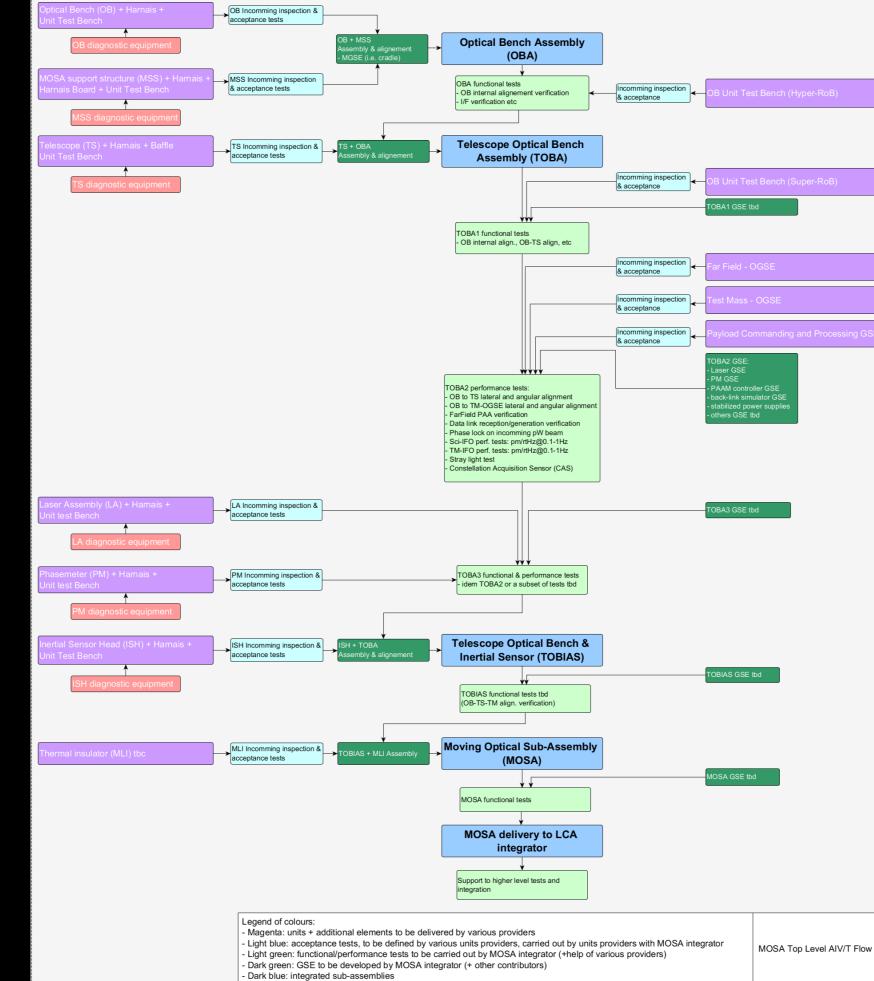
- ► MOSA Elegant BreadBoard (EBB) [TBC]
 - Demonstrates development criticality
- MOSA Structural and Thermal Model (STM)
 - Validate thermal and mechanical models;
 - AIVT flows with dummy for telescope, GRS, OB
- MOSA (Engineering) Qualification Model (E)QM
 - Representative for MOSA design concept & AIVT process
- ► 6 Flight Models (FM)
- Spare





Top Level MOSA AIV/T flow











- ► Far-Field OGSE => simulate laser beam from distant S/C
- ► Test Mass OGSE => mimic the movements of the TM
- Super/Hyper RoB => readout of heterodyne photodetectors
- Phase Reference Distribution System OGSE
- Payload Commanding and Processing GSE
- Dedicated GSE: stable laser, phasemeter, test bench for stray light characterization, real time command/control, ...
- MOSA MGSE => mechanical load of MOSA + FF-OGSE
- Climatic chamber, cleanliness (class 100)



EROT







- ► LISA will observe GWs between 10⁻⁵ and 1 Hz:
 - Large number of sources: compact objects binaries with large range of masses, stochastic backgrounds, ...
 - Huge scientific potential: physic, astrophysics, cosmology, ...
- LISAPathfinder: success
 - Performances > 7 times better than the requirements
- LISAPathfinder + detections of Ground-based observatories
 => Green light for LISA: large extension of the new window opened with LIGO/Virgo
 - => speed-up of the ESA planning:
 - Already done: call for mission, selection, phase 0
 - Next: phase A starting in April 2018

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EROT

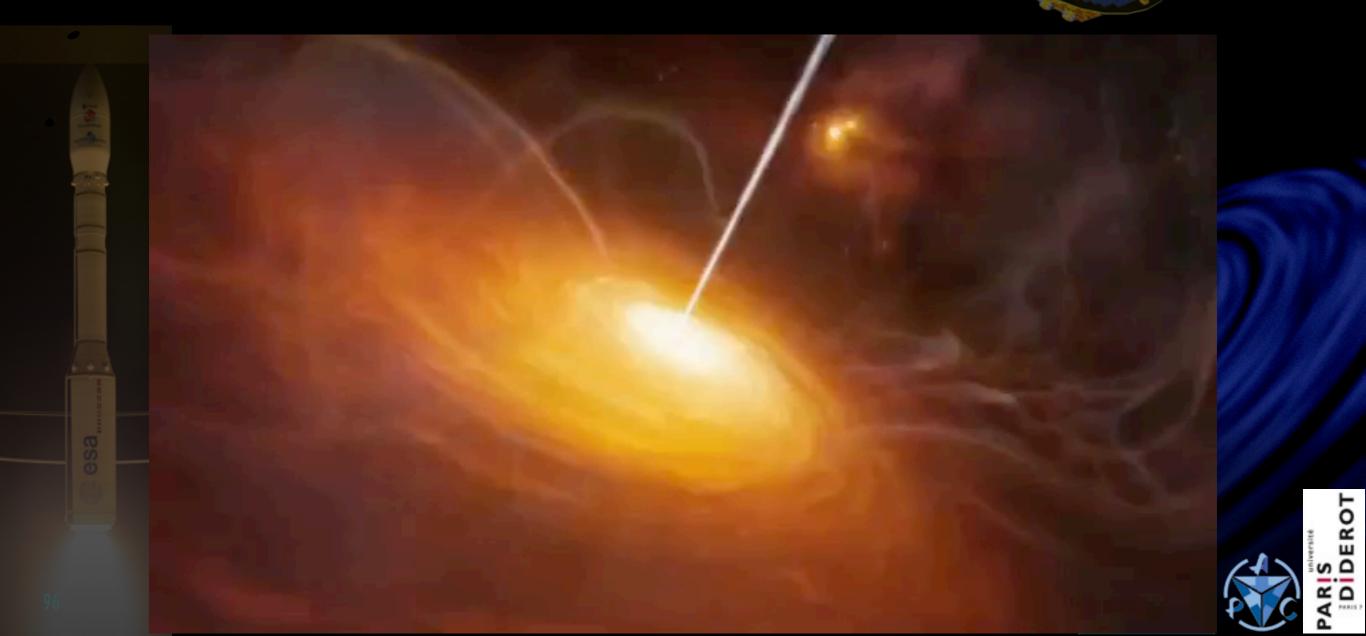
- LISA Consortium re-organised in a stronger and suitable structure for phase A https://elisadpc.in2p3.fr/lisafrance
- French contributions organized via LISAFrance
 - Data Processing Center: flexible and distributed, proto-DPC in dev.
 - => LISA Data Challenge; contact proto-DPC team; contribute to one of the activities organize by the LISA Data Processing Group
 - AIV/T of MOSA and performance control: number of GSE, models, etc to develop => join the phase A activities leaded by CNES
 - Science => join a working group; contribute to work packages (more than > 60 WPs for Science/DA in definition)
- LISA project started: a lot to do, a lot of possible contribution!







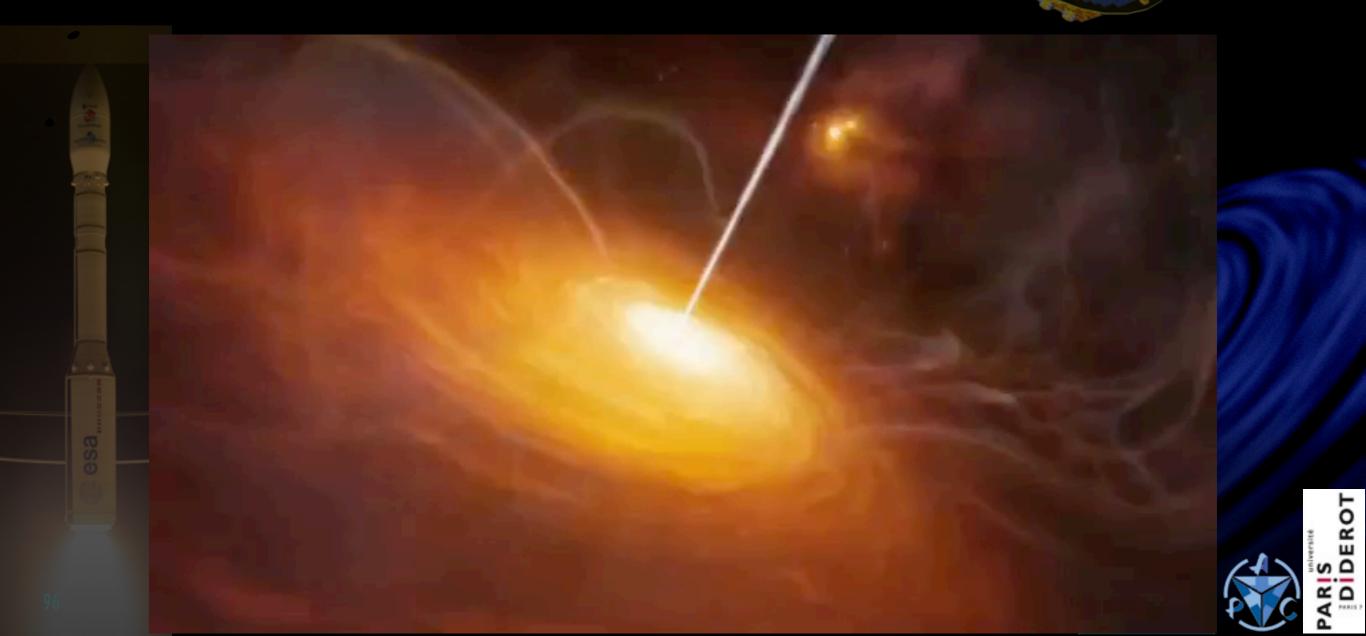
Thank you !







Thank you !





LISA Data Processing Group



- Responsibilities (for phase A ...):
- Develop and coordinate the unique DPC for LISA;
- Prepare and execute the pipelines to produce L2 and L3 products as well as other defined scientific products and deliver them to the SOC;
- Manage the interfaces between pipeline design and pipeline implementation;
- Aid in the management and execution of large-scale simulations and provide structures for data management;
- Coordinate the definition and development of LISA DCC;
- Aid in the definition of Consortium support to ESA's ground-segment;
- Coordinate the definition and implementation of the data analysis frameworks and operations environment;
- Coordinate the development and management of data analysis pipelines;







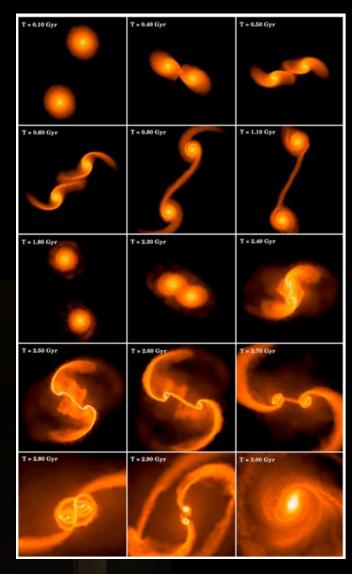
- ► The development of the DPC includes:
 - definition and maintenance of the pipeline and data analysis development environment;
 - design and implementation of the pipeline and analysis operations environment;
 - design and implementation of data storage facilities and databases;
 - implementation and operation of consortium IT services;
 - management and implementation of pipelines for simulation and data analysis.





MBH binaries: Formation &





Colpi & Dotti (2009) Review, astro-ph 0906.4339 Talk F. Combes * Talk J. de Freitas-Pacheco

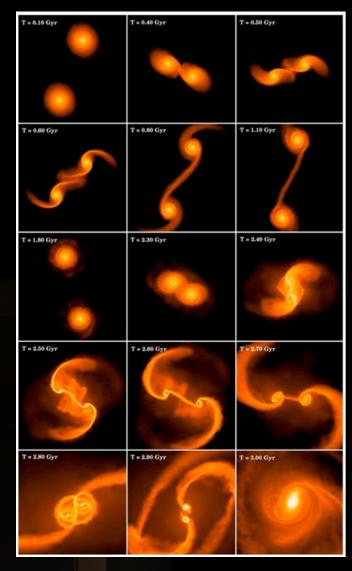
Galaxies mergers	Binary formation	Close binary	Merger			
$HE = \begin{bmatrix} 1 & 0 & 0 \\ 0 & 1 & 1 \\ 0 & 0 & 0 \\ 1 & 1 & 0 \\ 0 & 1 & 1 \end{bmatrix} = \begin{bmatrix} 1 & 4 & at & z = 0 \\ 0 & 1 & 1 & 0 \\ 1 & 2 & 3 & 4 & 5 & 6 \\ 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 2 & 3 & 4 & 5 & 6 & 7 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & 1 & 0 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 0 & 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & at & z = 3 \\ 0 & 1 & 1 & 1 & 0 & at & z = 3 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0 & 0 \\ 1 & 1 & 1 & 0 & 0$						
100 kpc → 100 pc	100 pc → sub-parsec	sub-pc→few M (au)				
\sim few Gyr	~ few Myr		\sim few hours			
 Dynamical friction Stellar formation Tidal shocks Gas dynamics Callegari & al. (2009) ApJ 696 L89 	 ★ Gas-dynamical friction ★ Circularization ★ Orbital angular momentum can flip ★ 3 bodies interaction ★ Dotti & al. (2009) MNRAS 396-1640 	due to <u>gravitational</u> <u>wave</u> emission	 ★ <u>GW</u> "burst", ★ Recoil velocities of remnant BH. 			
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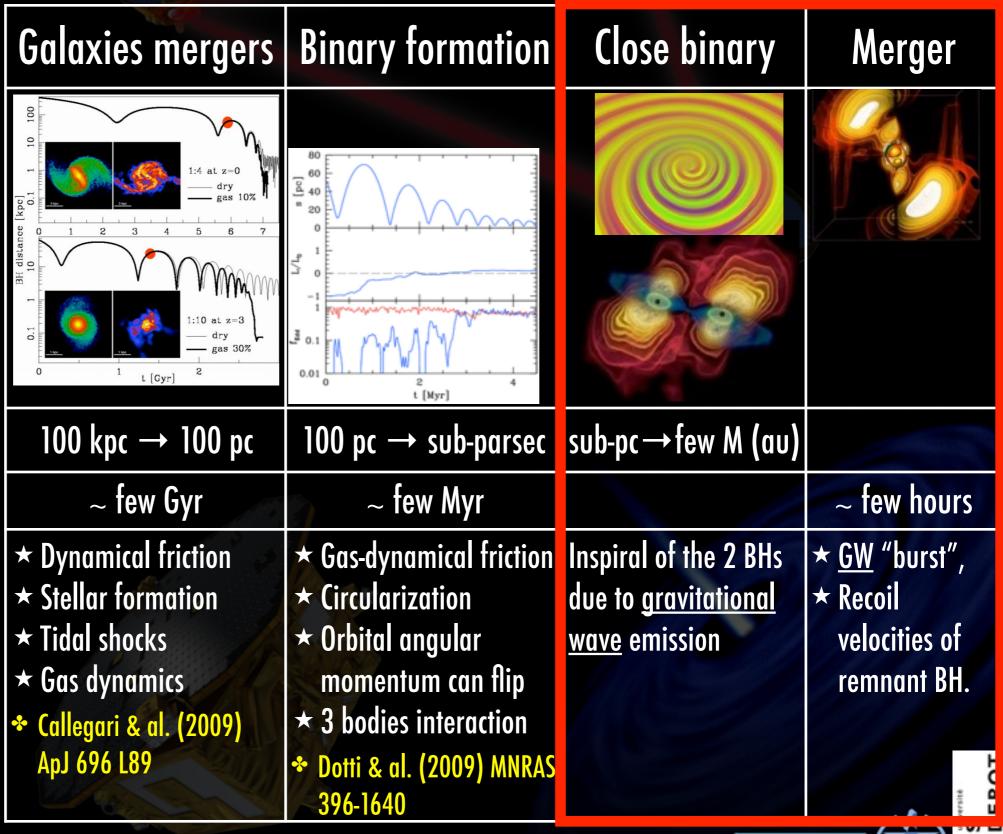


MBH binaries: Formation &





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observing gravitational waves from space

Antoine Petiteau

(APC – Université Paris-Diderot) Co-PI LISA, PI of LISAPathfinder for France, Member of LISA Board, Executive Board and Science Study Team

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